

## G. Design Equations and Procedure for Beam-Columns (Braced Frame)

There is no *standard* set of design steps but the following procedure may be suggested.

### Step 1: Design Load

Moments should be computed at both the top and bottom of the column.  $M_{ntx}$  and  $M_{nty}$  are the maximum design moments in the  $x$ - and  $y$ -axis of the member.

$$\begin{aligned} P_u &= 1.2P_D + 1.6P_L \\ M_{ntx} &= 1.2M_{Dx} + 1.6M_{Lx} \\ M_{nty} &= 1.2M_{Dy} + 1.6M_{Ly} \end{aligned}$$

### Step 2: Initial Member Selection. (Equivalent Axial Load Method)

Beam-column design is a trial and error process in which a trial section is checked for compliance with the AISC interaction equations (H1-1a) and (H1-1b). Initial guess of the member is made by using AISC Table 3-2 and the Column Tables. AISC/LRFD Specification (H1-a) can be rewritten, by multiplying each term by  $\phi P_n$ , as

$$P_u + \frac{8\phi P_n}{9\phi_b M_{nx}} M_{ux} + \frac{8\phi P_n}{9\phi_b M_{ny}} M_{uy} \leq \phi P_n$$

or at the limit state,

$$P_u + \frac{8\phi P_n}{9\phi_b M_{nx}} M_{ux} + \frac{8\phi P_n}{9\phi_b M_{ny}} M_{uy} = \phi P_n$$

Multiplication of the third term by  $M_{nx}/M_{nx}$  and letting

$$m = \frac{8\phi P_n}{9\phi_b M_{nx}} \quad \text{and} \quad u = \frac{M_{nx}}{M_{ny}}$$

the equivalent load ( $P_{ueq}$ ) is obtained

$$P_u + mM_{ux} + muM_{uy} = \phi P_n = P_{ueq}$$

where the values  $m$  (bending factor) are found in the AISC Table 3-2 and  $u$  are obtained by guessing from the Column Tables.

**Step 3: Check member.**

- (a) **Column Effect:** Calculate the axial strength  $= \phi_c P_n$ . It is useful to compute the slenderness parameter  $\lambda_c$  for both the  $x$ - and  $y$ -axis for steps (d) and (e):

$$\lambda_{cx} = \frac{K_x L_x}{r_x} \sqrt{\frac{F_y}{\pi^2 E}}$$

$$\lambda_{cy} = \frac{K_y L_y}{r_y} \sqrt{\frac{F_y}{\pi^2 E}}$$

- (b) **Beam Effect** ( $x$ -direction): Calculate the bending design strength  $= \phi_b M_{nx}$  for the  $x$ -axis. Check both LB and LTB.
- (c) **Beam Effect** ( $y$ -direction): Calculate the bending design strength  $= \phi_b M_{ny}$  for the  $y$ -axis. This analysis is similar to step (b) except that  $y$ -axis properties ( $S_y$  and  $Z_y$ ) are used. Consider only LB in the flange since there will be no LTB in the  $y$ -axis.
- (d) **Moment Magnification** ( $x$ -axis direction): Calculate  $C_{mx}$  for the  $x$ -axis moments using:

$$C_{mx} = 0.6 - 0.4 \left( \frac{M_1}{M_2} \right)_x \quad (\text{H1-4})$$

Here,  $M_1$  and  $M_2$  are the end moments with the condition  $|M_1| \leq |M_2|$  and the sign of the value  $M_1/M_2$  is:

$$\begin{aligned} (M_1/M_2)_x > 0 & \quad \text{for reverse curvature} \\ (M_1/M_2)_x \leq 0 & \quad \text{for single curvature} \end{aligned}$$

Calculate  $B_{1x}$  for the  $x$ -axis using the formula:

$$B_{1x} = \frac{C_{mx}}{1 - P_u/P_{e1x}} \geq 1 \quad (\text{H1-3})$$

The Euler buckling load,  $P_{e1x}$ , is calculated using the  $x$ -axis properties regardless of which axis is weaker:

$$P_{e1x} = \frac{\pi^2 E A_g}{K_x L_x} = \frac{F_y A_g}{\lambda_{cx}^2}$$

- (e) **Moment Magnification** ( $y$ -axis direction): Repeat step (d) for the  $y$ -axis using the formulas:

$$C_{my} = 0.6 - 0.4 \left( \frac{M_1}{M_2} \right)_y \quad (\text{H1-4})$$

$$B_{1y} = \frac{C_{my}}{1 - P_u/P_{e1y}} \geq 1 \quad (\text{H1-3})$$

The Euler buckling load,  $P_{e1y}$ , is calculated using the  $y$ -axis properties regardless of which axis is weaker:

$$P_{e1y} = \frac{\pi^2 E A_g}{K_y L_y} = \frac{F_y A_g}{\lambda_{cy}^2}$$

- (f) **Interaction:**

If  $P_u/\phi_c P_n \geq 0.2$  then

$$\frac{P_u}{\phi_c P_n} + \frac{8}{9} \left( \frac{M_{ux}}{\phi_b M_{nx}} + \frac{M_{uy}}{\phi_b M_{ny}} \right) = \text{interaction ratio} \leq 1 \quad (\text{H1-1a})$$

If  $P_u/\phi_c P_n < 0.2$  then

$$\frac{P_u}{2\phi_c P_n} + \left( \frac{M_{ux}}{\phi_b M_{nx}} + \frac{M_{uy}}{\phi_b M_{ny}} \right) = \text{interaction ratio} \leq 1 \quad (\text{H1-1a})$$

- (g) **Redesign:** If the interaction ratio falls in the range between 0.95 and 1.0, then no redesign may be necessary. Otherwise, it is necessary to check a new section using the general formula:

$$\text{New Weight} = \text{Old Weight} \times \frac{\text{Load}}{\text{Capacity}} = \text{Old Weight} \times \text{Interaction}$$