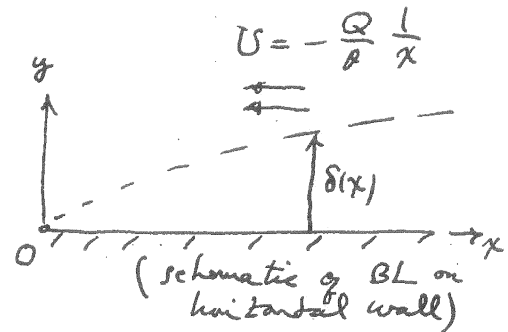
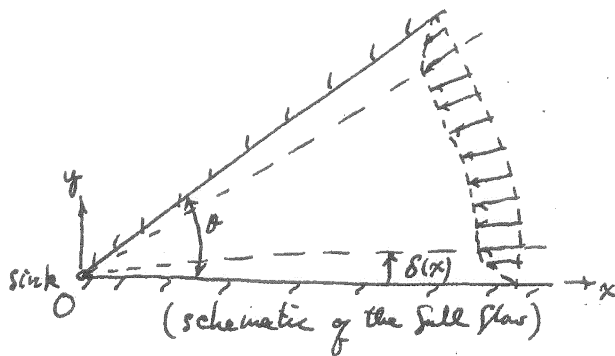


Problem Set 3 SolutionSteady boundary layer along a wall towards a corner sink\* Boundary-layer equations

$$\frac{\partial u}{\partial x} + \frac{\partial v}{\partial y} = 0 \quad (\text{continuity})$$

$$u \frac{\partial u}{\partial x} + v \frac{\partial u}{\partial y} = U \frac{dU}{dx} + \nu \frac{\partial^2 u}{\partial y^2} \quad (x\text{-mom.})$$

$$\text{Boundary conditions: } \begin{cases} u = v = 0 & (y=0) \\ u \rightarrow U = -\frac{Q}{L} \frac{1}{x} & (y/\delta \rightarrow \infty) \end{cases}$$

In terms of streamfunction  $\Psi(x, y)$ ,

$$u = \frac{\partial \Psi}{\partial y}, \quad v = -\frac{\partial \Psi}{\partial x}$$

continuity is automatically satisfied, and

$$x\text{-mom} \Rightarrow \frac{\partial \Psi}{\partial y} \frac{\partial^2 \Psi}{\partial x \partial y} - \frac{\partial \Psi}{\partial x} \frac{\partial^2 \Psi}{\partial y^2} = -\frac{Q^2}{L^2 x^3} + \nu \frac{\partial^3 \Psi}{\partial y^3}$$

$$\text{and BC's} \Rightarrow \begin{cases} \Psi = \frac{\partial \Psi}{\partial y} = 0 & (y=0) \\ \frac{\partial \Psi}{\partial y} \rightarrow -\frac{Q}{L} \frac{1}{x} & (y/\delta \rightarrow \infty) \end{cases}$$

\* Similarity solution

$$x = \lambda^\alpha x', \quad y = \lambda^\beta y', \quad \psi = \lambda^\gamma \psi'$$

$$\text{Gov. eqn} \Rightarrow \lambda^{2\gamma - 2\beta - \alpha} \sim \lambda^{-3\alpha} \sim \lambda^{\gamma - 3\beta}$$

$$\therefore -3\alpha = -2\beta + \gamma \Rightarrow \beta = \alpha + \frac{1}{3}\gamma$$

$$-3\alpha = -\alpha + 2\gamma - 2\beta \Rightarrow 2\alpha = 2\beta - 2\gamma$$

$$\text{Hence } 2\alpha = 2\alpha + \frac{2}{3}\gamma - 2\gamma \Rightarrow \underline{\underline{\gamma = 0, \alpha = \beta}}$$

$$\text{Also } \psi_y \rightarrow -\frac{Q}{\theta} \frac{1}{x} \Rightarrow \lambda^{\gamma - \beta} \sim \lambda^{-\alpha} \text{ automatically met}$$

$$\text{Thus, by } \underline{\underline{\psi = f(y/x), \quad \eta = y/x}}$$

$$\frac{\partial}{\partial x} = \frac{\partial}{\partial \eta} \frac{\partial \eta}{\partial x} = -\frac{\eta}{x} \frac{d}{d\eta}, \quad \frac{\partial}{\partial y} = \frac{\partial}{\partial \eta} \frac{\partial \eta}{\partial y} = \frac{1}{x} \frac{d}{d\eta}$$

After some algebra, we find that  $f$  satisfies

$$-\frac{df}{d\eta} \left( \eta \frac{df}{d\eta} \right)' + \eta \frac{df}{d\eta} \frac{d^2 f}{d\eta^2} = -\frac{Q^2}{\theta^2} + \nu \frac{d^3 f}{d\eta^3}$$

$$\Rightarrow \underline{\underline{\nu \frac{d^3 f}{d\eta^3} = \frac{Q^2}{\theta^2} - \left( \frac{df}{d\eta} \right)^2}}$$

with BC

$$\begin{cases} df/d\eta = 0 & \text{at } \eta = 0 \\ df/d\eta \rightarrow -\frac{Q}{\theta} & \text{as } \eta \rightarrow \infty (\eta \rightarrow \infty) \\ (d^2 f/d\eta^2 \rightarrow 0 & \text{as } \eta \rightarrow \infty) + (\text{too much to ask}) \end{cases}$$

Write 
$$\underline{\underline{\frac{df}{d\eta} = -\frac{Q}{\theta} F(\eta)}}$$

Then, 
$$\underline{\underline{-\frac{\nu\theta}{Q} \frac{d^2F}{d\eta^2} = 1 - F^2}}$$

with BC 
$$\begin{cases} F(0) = 0 & \eta = 0 \\ F \rightarrow 1 & \eta \rightarrow \infty \end{cases}$$

Integrate once: 
$$-\frac{2\nu\theta}{2Q} F'^2 = F - \frac{1}{3}F^3 + C$$

As  $\eta \rightarrow \infty$   $F' \rightarrow 0 \Rightarrow C = -\frac{2}{3}$

Hence 
$$-\frac{\nu\theta}{2Q} F'^2 = F - \frac{1}{3}F^3 - \frac{2}{3} = -\frac{1}{3}(F-1)^2(F+2)$$

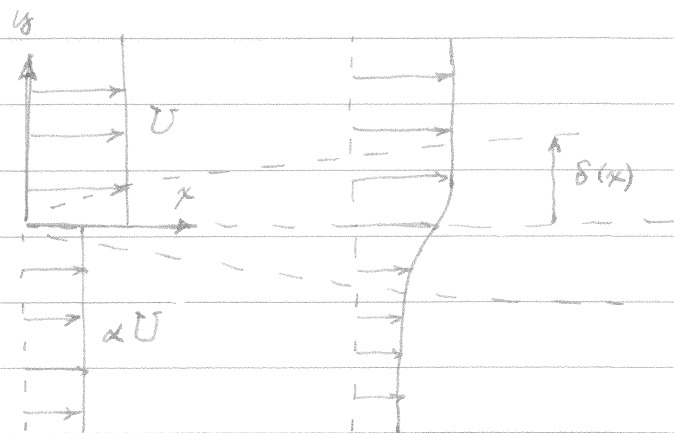
For  $0 \leq F \leq 1$ , RHS is always  $< 0 \Rightarrow \underline{\underline{Q > 0}}$

So, 
$$\frac{\nu\theta}{2Q} \left(\frac{dF}{d\eta}\right)^2 = \frac{1}{3}(F-1)^2(F+2)$$

$$u = \psi_y = \frac{1}{x} \frac{df}{d\eta} = -\frac{Q}{\theta x} F(\eta) \Rightarrow \underline{\underline{\frac{u}{U(x)} = F(\eta)}}$$

$$\underline{\underline{F(\eta) = 3 \tanh^2 \left[ \frac{y}{x} \sqrt{\frac{Q}{2\nu\theta}} + \tanh^{-1} \sqrt{\frac{2}{3}} \right] - 2}}$$

Problem Set 3, Problem 2; solution



This problem is a generalization of the Blasius flat-plate boundary layer. Since the 'outer' potential flow is uniform,  $U \frac{\partial U}{\partial x} = 0$ , and there is no pressure gradient in the boundary layer equation:

$$u \frac{\partial u}{\partial x} + v \frac{\partial u}{\partial y} = \nu \frac{\partial^2 u}{\partial y^2}$$

Introducing streamfunction  $\psi(x,y)$ :  $u = \frac{\partial \psi}{\partial y}$ ,  $v = -\frac{\partial \psi}{\partial x}$

$$\Rightarrow \frac{\partial \psi}{\partial y} \frac{\partial^2 \psi}{\partial x \partial y} - \frac{\partial^2 \psi}{\partial y^2} \frac{\partial \psi}{\partial x} = \nu \frac{\partial^3 \psi}{\partial x^3}$$

with BC (i)  $\psi = 0$  ( $y=0$ ) (dividing streamline is flat)

(ii)  $\psi \rightarrow Uy$  ( $y/\delta \rightarrow \infty$ )

(iii)  $\psi \rightarrow \alpha Uy$  ( $y/\delta \rightarrow -\infty$ )

This problem can be solved by the same similarity variables as in the Blasius boundary layer:

$$\psi = \sqrt{\nu U x} f(\eta), \quad \eta = \frac{y}{\sqrt{\nu x/U}} \quad (\delta \sim \sqrt{\nu x/U})$$

$$\Rightarrow \quad \underline{ff'' + 2f''' = 0} \quad (\text{Blasius equation})$$

$$\underline{\text{BC}} \quad \begin{cases} f = 0 & (\eta = 0) \\ f' \rightarrow 1 & (\eta \rightarrow \infty) \\ f' \rightarrow \alpha & (\eta \rightarrow -\infty) \end{cases}$$

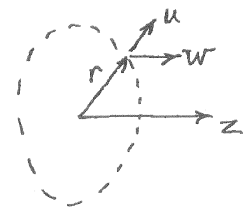
There is no closed-form solution of this boundary-value problem [See R.C. Lock, Quarterly Journal of Mechanics and Applied Mathematics, 4, 42-63, 1951; he shows how to obtain numerical solution and also considers the more general problem where the two streams have different densities and viscosities; eg air over water]

Problem Set No. 3; Problem 3 Solution

Jet from a point source

\* Boundary-layer equations (axisymmetric flow)  $(r, z) \leftrightarrow (u, w)$

(continuity)  $\frac{\partial}{\partial r}(ru) + \frac{\partial}{\partial z}(rw) = 0$   
 (Momentum)  $u \frac{\partial w}{\partial r} + w \frac{\partial w}{\partial z} = \frac{\nu}{r} \frac{\partial}{\partial r} \left( r \frac{\partial w}{\partial r} \right)$



\* Momentum flow rate at the origin

$$M = 2\pi \int_0^{\infty} r dr \rho w^2 = 2\pi \rho \int_0^{\infty} r w^2 dr$$

\* Boundary conditions are  $\begin{cases} u = 0, \frac{\partial w}{\partial r} = 0 & \text{at } r = 0 \\ w \rightarrow 0 & \text{as } r \rightarrow \infty \end{cases}$

Also, from the momentum equation above,

$$\int_0^{\infty} 2\pi r dr \left\{ u \frac{\partial w}{\partial r} + w \frac{\partial w}{\partial z} \right\} = \int_0^{\infty} 2\pi r dr \left\{ \frac{\nu}{r} \frac{\partial}{\partial r} \left( r \frac{\partial w}{\partial r} \right) \right\}$$

$$\Rightarrow \underbrace{\int_0^{\infty} 2\pi r dr u \frac{\partial w}{\partial r}}_{\downarrow} + \int_0^{\infty} 2\pi r dr w \frac{\partial w}{\partial z} = 2\pi r w \frac{\partial w}{\partial r} \Big|_0^{\infty} = 0$$

$$2\pi r u w \Big|_0^{\infty} - 2\pi \int_0^{\infty} dr w \frac{\partial}{\partial r} (ru) - \frac{\partial}{\partial z} (rw)$$

$$\therefore 2 \int_0^{\infty} 2\pi r dr \frac{1}{2} \frac{\partial}{\partial z} (w^2) = 0 \Rightarrow \int_0^{\infty} r dr w^2 = \text{const at any } z$$

$$\Rightarrow M = 2\pi \rho \int_0^{\infty} r w^2 dr = \text{const at any } z$$

\* Introduce stream function  $\Psi(r, z)$  s.t.

$$\underline{u = \frac{1}{r} \Psi_z, \quad w = -\frac{1}{r} \Psi_r ;}$$

Then, continuity  $\frac{\partial}{\partial r}(ru) + \frac{\partial}{\partial z}(rw) = 0$  is automatically satisfied

Momentum equation  $\Rightarrow$

$$-\frac{\nu}{r} \frac{\partial \Psi}{\partial z} \frac{\partial}{\partial r} \left( \frac{1}{r} \Psi_r \right) + \frac{1}{r} \Psi_r \frac{\partial}{\partial z} \left( \frac{1}{r} \Psi_r \right) = \frac{\nu}{r} \frac{\partial}{\partial r} \left( r \frac{\partial}{\partial r} \left( \frac{1}{r} \Psi_r \right) \right)$$

Now, try the transformation

$$\Psi \rightarrow \lambda^\alpha \Psi', \quad r \rightarrow \lambda^\beta r', \quad z = \lambda^\gamma z'$$

Then, invariance of momentum eqn  $\Rightarrow$

$$\lambda^{2\alpha - 3\beta - \gamma} \sim \lambda^{\alpha - 3\beta} \Rightarrow \alpha = \gamma$$

$$\text{Also, from } \int_0^\infty r dr w^2 = \text{const} \Rightarrow \lambda^{2\beta + 2\alpha - 4\beta} \sim 1$$

$$\Rightarrow \alpha = \beta$$

$$\therefore \underline{\alpha = \beta = \gamma}$$

So, the appropriate similarity variable is of the form

$$\underline{\eta = \frac{1}{B} \frac{r}{z}} \quad \text{and} \quad \underline{\Psi = A z f(\eta)}$$

\* Substitute into the momentum equation (algebra)

$$\eta = \frac{1}{B} \frac{r}{z}, \quad \Psi = Az f(\eta)$$

$$\frac{\partial}{\partial r} = \frac{\partial}{\partial \eta} \frac{\partial \eta}{\partial r} = \frac{1}{Bz} \frac{\partial}{\partial \eta}, \quad \frac{\partial}{\partial z} = \frac{\partial}{\partial z} + \frac{\partial}{\partial \eta} \frac{\partial \eta}{\partial z} = \frac{\partial}{\partial z} - \frac{r}{Bz^2} \frac{\partial}{\partial \eta} = \frac{\partial}{\partial z} - \frac{\eta}{z} \frac{\partial}{\partial \eta}$$

$$w = \frac{1}{r} \frac{\partial \Psi}{\partial r} = - \frac{Az}{r} \frac{1}{Bz} f'_{\eta} \frac{1}{Bz\eta} = \underline{\underline{\frac{-A}{B^2 \eta z} f'_{\eta} = W}}$$

$$\frac{1}{r} \frac{\partial \Psi}{\partial z} = \frac{1}{Bz\eta} \left\{ Af - \frac{\eta}{z} Af' f_{\eta} \right\} \Rightarrow \underline{\underline{u = \frac{A}{Bz\eta} (f - \eta f')}}$$

So, momentum equation  $\Rightarrow$

$$- A(f - \eta f') \frac{1}{Bz} \frac{\partial}{\partial \eta} \left( \frac{A}{B^2 \eta z} f' \right) + \frac{1}{Bz} \frac{\partial}{\partial \eta} (Az f) \left( \frac{\partial}{\partial z} - \frac{\eta}{z} \frac{\partial}{\partial \eta} \right) \left( \frac{A}{B^2 \eta z} f' \right) = \nu \frac{1}{Bz} \frac{\partial}{\partial \eta} \left\{ Bz\eta \frac{1}{Bz} \frac{\partial}{\partial \eta} \left( \frac{A}{B^2 \eta z} f' \right) \right\}$$

Re-arranging, after some cancellation,

$$A \left\{ f \left( \frac{f'}{\eta} \right)' + \frac{f'^2}{\eta} \right\} = \nu \frac{\partial}{\partial \eta} \left\{ \eta \frac{\partial}{\partial \eta} \left( \frac{f'}{\eta} \right) \right\}$$

Choose  $A = \nu, B = 1$

Then,  $\left( \frac{f f'}{\eta} \right)' = \left( \eta \left( \frac{f'}{\eta} \right)' \right)'$

\* B.C.  $u = 0$  at  $r = 0 \Rightarrow$

$$\frac{1}{\eta} (f - \eta f') \rightarrow 0 \text{ as } \eta \rightarrow 0$$

$$\text{ie, } \frac{f}{\eta} - f' \rightarrow 0 \text{ as } \eta \rightarrow 0$$

Also  $w = \frac{\nu}{\eta z} f'_{\eta}$  finite as  $r \rightarrow 0$   $f'(0) \rightarrow 0$  and hence  $f(0) = 0$



\* Solve ode for  $f$

Integrating once,  $\frac{ff'}{\eta} = \eta \left(\frac{f'}{\eta}\right)' + \text{const} = f'' - \frac{1}{\eta}f' + \text{const}$

BC  $\Rightarrow \frac{f}{\eta}, f' \rightarrow 0$  as  $\eta \rightarrow 0 \Rightarrow \text{const} = 0$

$\therefore ff' = \eta f'' - f'$  or  $\eta f'' + f' = ff' + 2f'$   
 $\Rightarrow (\eta f')' = \left(\frac{f^2}{2} + 2f\right)'$

Integrating again,  $\eta f' = \frac{f^2}{2} + 2f = \frac{f}{2}(f+4)$

$\Rightarrow \frac{d\eta}{2\eta} = \frac{df}{f(f+4)} \Rightarrow \ln \frac{f}{f+4} = 2 \ln \eta + \ln C$

$\Rightarrow f(\eta) = \frac{4C\eta^2}{1-C\eta^2}$

\* Suppose  $M = 2\pi\rho \int_0^\infty w^2 r dr = 2\pi\rho \int_0^\infty \left(\frac{1}{r}\Psi_r\right)^2 dr$   
 $= 2\pi\rho \int_0^\infty \left(\frac{\nu}{\eta z} f'\right)^2 \eta z^2 d\eta = 2\pi\rho \nu^2 \int_0^\infty \frac{f'^2}{\eta} d\eta$   
 $= 2\pi\rho \nu^2 \int_0^\infty \frac{1}{\eta} \left\{ \frac{d}{d\eta} \left( \frac{4C\eta^2}{1-C\eta^2} \right) \right\}^2 d\eta = -2\pi\rho \frac{\nu^2}{3} 32C = M$

This determines  $C = -\frac{3M}{64\pi\rho\nu^2} < 0$

$\Rightarrow f(\eta) = \frac{-4K\eta^2}{1+K\eta^2} \quad (K \equiv -C)$

\* Velocity components

$$\left. \begin{aligned} u &= \frac{1}{r} \psi_z = \frac{\nu}{z\eta} (f - \eta f') \\ w &= -\frac{1}{r} \psi_r = -\frac{\nu}{\eta z} f' \end{aligned} \right\}$$

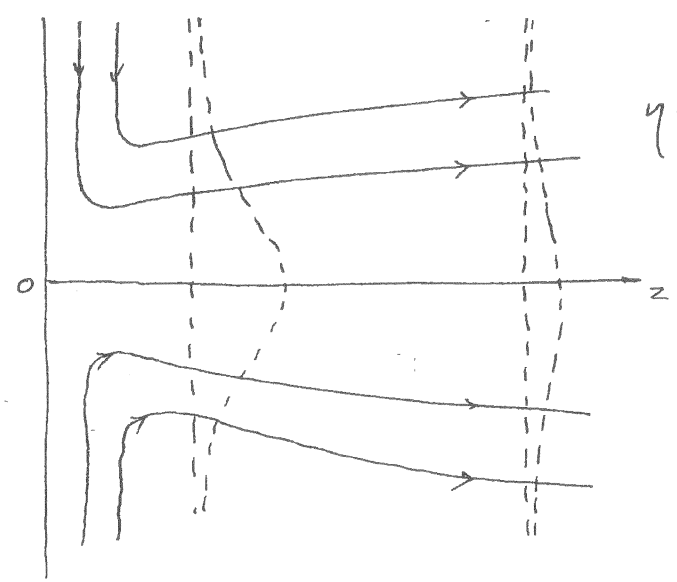
Near the axis ( $r \rightarrow 0, \eta \rightarrow 0$ )  $f \rightarrow -4K\eta^2$

$$\therefore \begin{cases} u \sim \frac{\nu}{z} (-4K + 8K)\eta = 4K\nu \frac{r}{z^2} > 0 \\ w \sim \frac{\nu}{z} 8K > 0 \quad (\text{ie, } w \gg u) \end{cases}$$

Far from the axis ( $r \rightarrow \infty, \eta \rightarrow \infty, z$  fixed)

$$f = \frac{-4K\eta^2}{1 + K\eta^2} \sim -4 \left( 1 - \frac{1}{K\eta^2} + \dots \right)$$

$$\therefore \begin{cases} u \sim -\frac{4\nu}{r} < 0 \Rightarrow \text{flow is entrained from the jet edges} \\ w \sim \frac{\nu}{z} \left( \frac{8}{K\eta^4} \right) = \frac{8\nu}{K} \frac{z^3}{r^4} \quad (\text{ie, } w \ll |u|) \end{cases}$$



$\eta = \frac{r}{z} \Rightarrow$  radius of jet grows like  $z$

\* How good is the boundary-layer approximation?

Viscous term omitted:  $\nu \frac{\partial^2 w}{\partial z^2} = -\nu \frac{\partial^2}{\partial z^2} \left( \frac{1}{r} \frac{\partial \psi}{\partial r} \right) =$   
 $= -\nu \frac{A}{B^2} \left( \frac{\partial}{\partial z} - \frac{\eta}{z} \frac{\partial}{\partial \eta} \right)^2 \left( \frac{f'}{\eta z} \right) \equiv \nu F(\eta) \frac{1}{z^3}$

Viscous term included:  $\nu \frac{1}{r} \frac{\partial}{\partial r} \left( r \frac{\partial w}{\partial r} \right) =$   
 $= \nu \frac{1}{B \eta z} \frac{1}{B z} \frac{\partial}{\partial \eta} \left( B \eta z \frac{1}{B z} \frac{\partial}{\partial \eta} \left( -\frac{A}{B \eta^2 z} f' \right) \right) \equiv \nu G(\eta) \frac{1}{z^3}$

These two terms are comparable  $O\left(\frac{1}{z^3}\right)$  for  $\eta = O(1), z \gg 1$

Note that the Reynolds number  $Re = \frac{U \delta}{\nu} \equiv \frac{W_{max} \cdot \delta}{\nu}$

$W_{max} \sim \frac{\nu}{z} \delta K, \quad \delta = O(z) \Rightarrow \underline{\underline{Re = O(1)}}$

[ In 2-D jet:  $U \sim \frac{1}{x^{1/3}}, \delta \sim x^{2/3}$   
 $\Rightarrow Re = \frac{U \delta}{\nu} \sim x^{1/3} \gg 1$  for  $x$  large  
 i.e., boundary-layer approximation becomes better as  $x$  increases ]

\* Even if we include  $\nu \frac{\partial^2 w}{\partial z^2}$  term, solution would not be exact because pressure-gradient term  $-\partial p / \partial z$  has been omitted as well.