

## Topic #13

### 16.31 Feedback Control

#### State-Space Systems

- **Full-state Feedback Control**
- How do we change the poles of the state-space system?
- Or, even if we can change the pole locations.
- Where do we change the pole locations to?
- How well does this approach work?

# Full-state Feedback Controller

- Assume that the single-input system dynamics are given by

$$\begin{aligned}\dot{x} &= Ax + Bu \\ y &= Cx\end{aligned}$$

so that  $D = 0$ .

- The multi-actuator case is quite a bit more complicated as we would have many extra degrees of freedom.

- Recall that the system poles are given by the eigenvalues of  $A$ .
  - Want to use the input  $u(t)$  to modify the eigenvalues of  $A$  to change the system dynamics.
- Assume a full-state feedback of the form:

$$u = r - Kx$$

where  $r$  is some **reference input** and the **gain**  $K$  is  $\mathcal{R}^{1 \times n}$

- If  $r = 0$ , we call this controller a **regulator**

- Find the closed-loop dynamics:

$$\begin{aligned}\dot{x} &= Ax + B(r - Kx) \\ &= (A - BK)x + Br \\ &= A_{cl}x + Br \\ y &= Cx\end{aligned}$$

- **Objective:** Pick  $K$  so that  $A_{cl}$  has the desired properties, *e.g.*,
  - $A$  unstable, want  $A_{cl}$  stable
  - Put 2 poles at  $-2 \pm 2j$
- Note that there are  $n$  parameters in  $K$  and  $n$  eigenvalues in  $A$ , so it looks promising, but what can we achieve?
- **Example #1:** Consider:

$$\dot{x} = \begin{bmatrix} 1 & 1 \\ 1 & 2 \end{bmatrix} x + \begin{bmatrix} 1 \\ 0 \end{bmatrix} u$$

– Then

$$\det(sI - A) = (s - 1)(s - 2) - 1 = s^2 - 3s + 1 = 0$$

so the system is unstable.

– Define  $u = - \begin{bmatrix} k_1 & k_2 \end{bmatrix} x = -Kx$ , then

$$A_{cl} = A - BK = \begin{bmatrix} 1 & 1 \\ 1 & 2 \end{bmatrix} - \begin{bmatrix} 1 \\ 0 \end{bmatrix} \begin{bmatrix} k_1 & k_2 \end{bmatrix} = \begin{bmatrix} 1 - k_1 & 1 - k_2 \\ 1 & 2 \end{bmatrix}$$

– So then we have that

$$\det(sI - A_{cl}) = s^2 + (k_1 - 3)s + (1 - 2k_1 + k_2) = 0$$

– Thus, by choosing  $k_1$  and  $k_2$ , we can put  $\lambda_i(A_{cl})$  anywhere in the complex plane (assuming complex conjugate pairs of poles).

- To put the poles at  $s = -5, -6$ , compare the *desired characteristic equation*

$$(s + 5)(s + 6) = s^2 + 11s + 30 = 0$$

with the closed-loop one

$$s^2 + (k_1 - 3)s + (1 - 2k_1 + k_2) = 0$$

to conclude that

$$\left. \begin{array}{l} k_1 - 3 = 11 \\ 1 - 2k_1 + k_2 = 30 \end{array} \right\} \begin{array}{l} k_1 = 14 \\ k_2 = 57 \end{array}$$

so that  $K = [ 14 \ 57 ]$ , which is called **Pole Placement**.

- Of course, it is not always this easy, as the issue of **controllability** must be addressed.
- **Example #2:** Consider this system:

$$\dot{x} = \begin{bmatrix} 1 & 1 \\ 0 & 2 \end{bmatrix} x + \begin{bmatrix} 1 \\ 0 \end{bmatrix} u$$

with the same control approach

$$A_{cl} = A - BK = \begin{bmatrix} 1 & 1 \\ 0 & 2 \end{bmatrix} - \begin{bmatrix} 1 \\ 0 \end{bmatrix} [ k_1 \ k_2 ] = \begin{bmatrix} 1 - k_1 & 1 - k_2 \\ 0 & 2 \end{bmatrix}$$

so that

$$\det(sI - A_{cl}) = (s - 1 + k_1)(s - 2) = 0$$

So the feedback control can modify the pole at  $s = 1$ , but it cannot move the pole at  $s = 2$ .

- **The system cannot be stabilized with full-state feedback control.**

- What is the reason for this problem?
  - It is associated with loss of controllability of the  $e^{2t}$  mode.
- Consider the basic controllability test:

$$\mathcal{M}_c = [ B \mid AB ] = \left[ \begin{array}{c|c} \begin{bmatrix} 1 \\ 0 \end{bmatrix} & \begin{bmatrix} 1 & 1 \\ 0 & 2 \end{bmatrix} \end{array} \begin{bmatrix} 1 \\ 0 \end{bmatrix} \right]$$

So that  $\text{rank } \mathcal{M}_c = 1 < 2$ .

- Consider the **modal test** to develop a little more insight:

$$A = \begin{bmatrix} 1 & 1 \\ 0 & 2 \end{bmatrix}, \text{ decompose as } AV = V\Lambda \Rightarrow \Lambda = V^{-1}AV$$

where

$$\Lambda = \begin{bmatrix} 1 & 0 \\ 0 & 2 \end{bmatrix} \quad V = \begin{bmatrix} 1 & 1 \\ 0 & 1 \end{bmatrix} \quad V^{-1} = \begin{bmatrix} 1 & -1 \\ 0 & 1 \end{bmatrix}$$

Convert

$$\dot{x} = Ax + Bu \xrightarrow{z=V^{-1}x} \dot{z} = \Lambda z + V^{-1}Bu$$

where  $z = [ z_1 \ z_2 ]^T$ . But:

$$V^{-1}B = \begin{bmatrix} 1 & -1 \\ 0 & 1 \end{bmatrix} \begin{bmatrix} 1 \\ 0 \end{bmatrix} = \begin{bmatrix} 1 \\ \mathbf{0} \end{bmatrix}$$

so that the dynamics in modal form are:

$$\dot{z} = \begin{bmatrix} 1 & 0 \\ 0 & 2 \end{bmatrix} z + \begin{bmatrix} 1 \\ \mathbf{0} \end{bmatrix} u$$

- With this zero in the modal  $B$ -matrix, can easily see that the mode associated with the  $z_2$  state is **uncontrollable**.
  - **Must assume that the pair  $(A, B)$  are controllable.**

## Ackermann's Formula

- The previous outlined a design procedure and showed how to do it by hand for second-order systems.
  - Extends to higher order (controllable) systems, but tedious.
- **Ackermann's Formula** gives us a method of doing this entire design process in one easy step.

$$K = [0 \ \dots \ 0 \ 1] \mathcal{M}_c^{-1} \Phi_d(A)$$

where

- $\mathcal{M}_c = [B \ AB \ \dots \ A^{n-1}B]$
  - $\Phi_d(s)$  is the characteristic equation for the closed-loop poles, which we then evaluate for  $s = A$ .
  - It is explicit that the **system must be controllable** because we are inverting the controllability matrix.
- Revisit **example # 1**:  $\Phi_d(s) = s^2 + 11s + 30$

$$\mathcal{M}_c = [B \ | \ AB] = \left[ \begin{array}{c|c} \begin{bmatrix} 1 \\ 0 \end{bmatrix} & \begin{bmatrix} 1 & 1 \\ 1 & 2 \end{bmatrix} \end{array} \begin{bmatrix} 1 \\ 0 \end{bmatrix} \right] = \begin{bmatrix} 1 & 1 \\ 0 & 1 \end{bmatrix}$$

So

$$\begin{aligned} K &= [0 \ 1] \begin{bmatrix} 1 & 1 \\ 0 & 1 \end{bmatrix}^{-1} \left( \begin{bmatrix} 1 & 1 \\ 1 & 2 \end{bmatrix}^2 + 11 \begin{bmatrix} 1 & 1 \\ 1 & 2 \end{bmatrix} + 30I \right) \\ &= [0 \ 1] \left( \begin{bmatrix} 43 & 14 \\ 14 & 57 \end{bmatrix} \right) = [14 \ 57] \end{aligned}$$

- Automated in Matlab: `place.m` & `acker.m` (see `polyvalm.m` too)

- Where did this formula come from?
- For simplicity, consider a third-order system (case #2), but this extends to any order.

$$A = \begin{bmatrix} -a_1 & -a_2 & -a_3 \\ 1 & 0 & 0 \\ 0 & 1 & 0 \end{bmatrix} \quad B = \begin{bmatrix} 1 \\ 0 \\ 0 \end{bmatrix} \quad C = [ b_1 \quad b_2 \quad b_3 ]$$

- See key benefit of using **control canonical** state-space model
- This form is useful because the characteristic equation for the system is obvious  $\Rightarrow \det(sI - A) = s^3 + a_1s^2 + a_2s + a_3 = 0$

- Can show that

$$\begin{aligned} A_{cl} = A - BK &= \begin{bmatrix} -a_1 & -a_2 & -a_3 \\ 1 & 0 & 0 \\ 0 & 1 & 0 \end{bmatrix} - \begin{bmatrix} 1 \\ 0 \\ 0 \end{bmatrix} [ k_1 \quad k_2 \quad k_3 ] \\ &= \begin{bmatrix} -a_1 - k_1 & -a_2 - k_2 & -a_3 - k_3 \\ 1 & 0 & 0 \\ 0 & 1 & 0 \end{bmatrix} \end{aligned}$$

so that the characteristic equation for the system is still obvious:

$$\begin{aligned} \Phi_{cl}(s) &= \det(sI - A_{cl}) \\ &= s^3 + (a_1 + k_1)s^2 + (a_2 + k_2)s + (a_3 + k_3) = 0 \end{aligned}$$

- We then compare this with the desired characteristic equation developed from the desired closed-loop pole locations:

$$\Phi_d(s) = s^3 + (\alpha_1)s^2 + (\alpha_2)s + (\alpha_3) = 0$$

to get that

$$\left. \begin{array}{l} a_1 + k_1 = \alpha_1 \\ \vdots \\ a_n + k_n = \alpha_n \end{array} \right\} \begin{array}{l} k_1 = \alpha_1 - a_1 \\ \vdots \\ k_n = \alpha_n - a_n \end{array}$$

- To get the specifics of the Ackermann formula, we then:
  - Take an arbitrary  $A, B$  and transform it to the control canonical form ( $x \rightsquigarrow z = T^{-1}x$ )
  - Solve for the gains  $\hat{K}$  using the formulas above for the state  $z$  ( $u = \hat{K}z$ )
  - Then switch back to gains needed for the state  $x$ , so that

$$K = \hat{K}T^{-1}$$

$$(u = \hat{K}z = Kx)$$

- Pole placement is a very powerful tool and we will be using it for most of this course.