Early Flight Control System Overview

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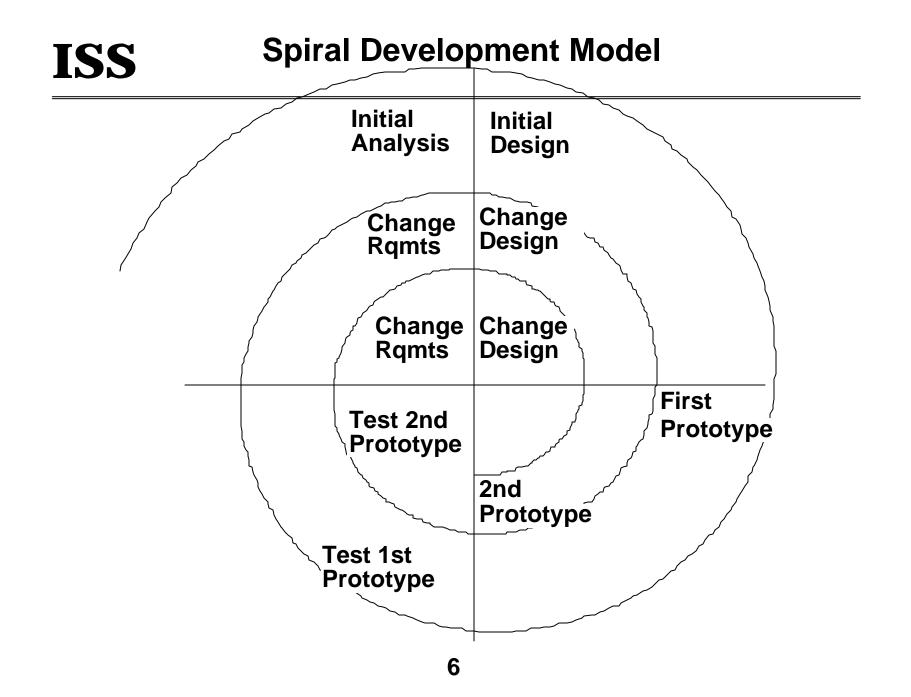
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- EFCS Task Statement
- Development Approach
- Architecture
- Test Environment
- Relevance to Redesign

- The SSFPO, in Feb. 1992, asked Draper to develop an Early Flight Control System (EFCS) as a feasibility demonstration of flight critical SDP-level functions essential for controlling the Space Station Freedom for Mission Builds 2-4.
 - Develop and demonstrate a system that could be used to provide schedule relief.
 - Implement simplified (as compared to the "mainline") versions of the essential systems (DMS, GN&C, EPS, C&T, etc.).
 - Replace the truss avionics with an MDM-based system.
 - Follow mainline truss avionics external interface specifications (to the SSCC, the Shuttle, and all lower level MDMs and firmware controllers).
 - Use rapid prototyping.

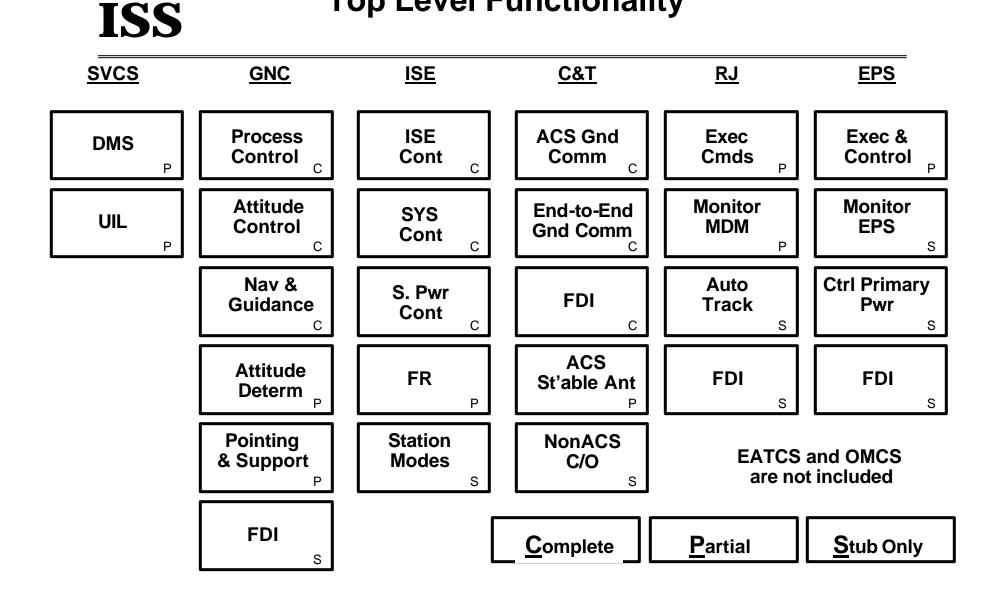
- Utilized a small multi-disciplined System Engineering Team.
- Designed an integrated system architecture allowing adding and modifying capabilities as required.
- Includes SSF system requirements for unmanned operations.
- Used the Rapid-prototyping life cycle (Progressive Refinement):
 - Risk reduction methodology for system development.
 - Early evaluation of system designs.
 - Early identification of performance issues.
 - Minimal early documentation.

- Used Mainline Requirements documentation as starting point for requirements.
- Added GPS, for time, position, velocity and attitude.
- Simplified where appropriate.
 - Eliminated functions to support payloads or manned operation.
- Generated requirements for each subsystem.

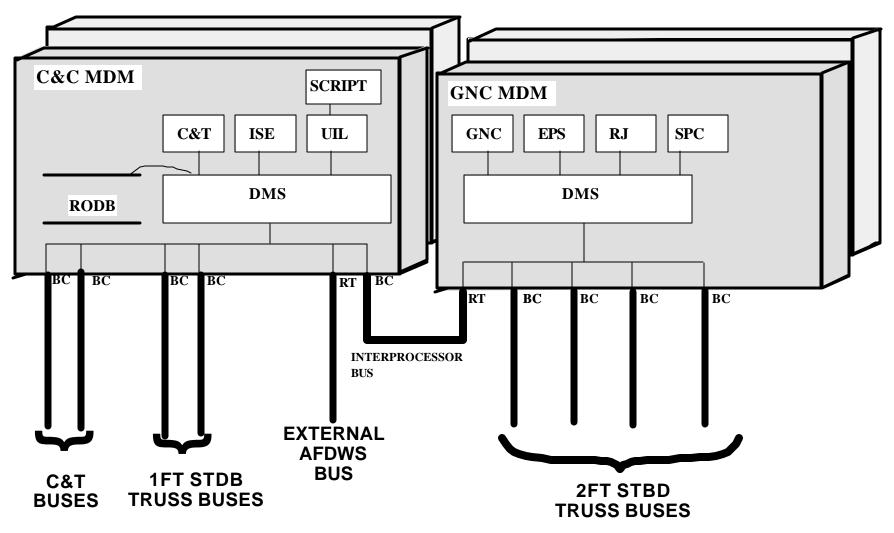


- Development process is "progressive refinement".
 - Four demonstrations were scheduled:
 - First demonstration (Nov. 92) showed basic capability for each system, integrated into a Station-level simulation.
 - Second demonstration (May 93) refined and added further capability to each system.
 - Third demonstration was scheduled to refine the overall system, add further capability, show that Draper on-board software meets external interfaces, and show that GCS meets FSW interfaces.
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- Redesign changed priorities; redesigned Data Management System interface.
- Fourth demonstration was scheduled to show that the integrated system was essentially complete.

Top Level Functionality



ISS EFCS Functional Block Diagram



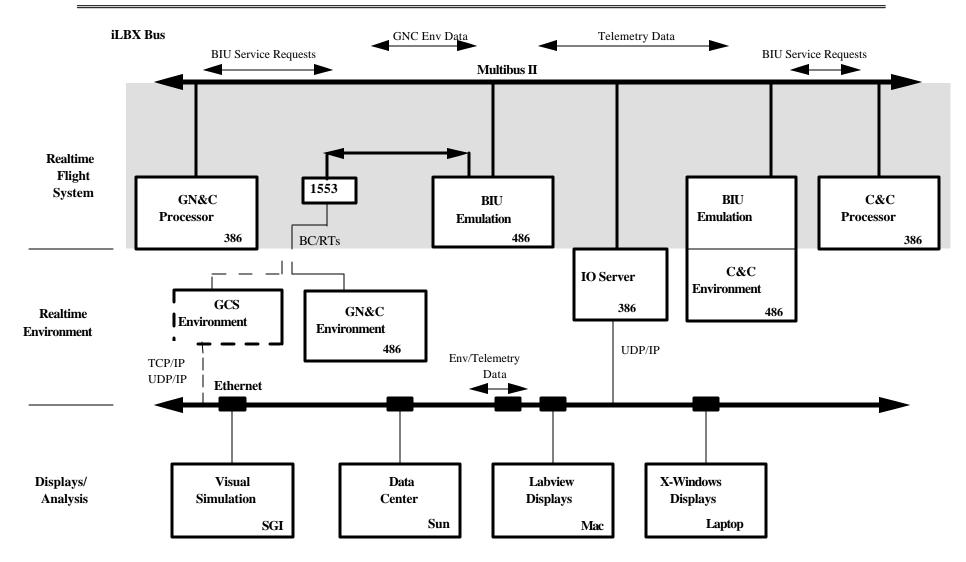
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- Development utilizes the following process:
 - Integrated system development and testing is performed on a non-realtime host based configuration.
 - This integrated system is then moved to the Realtime Testbed.
- This two-phased process permits:
 - System development, system integration and integrated testing is performed without complication of realtime operations.
 - Realtime-specific modifications to integrated system are made as required when the integrated system is ported to the realtime testbed.

ISS Host-Based Configuration

- Initial integration is performed, non-real-time, on host computer.
- Host and real-time testbed are running identical software except for machine-dependent routines.
- All systems and all environment modules, are linked together into one Ada program (real-time environment uses multiple Ada programs).
 - Application interfaces remain the same.
- Unique within Space Station Program.
- Benefits
 - Instrumenting software for debugging does not affect timing.
 - It is possible to stop a simulation, look at data, and then continue.
 - Many simulations can run at the same time.

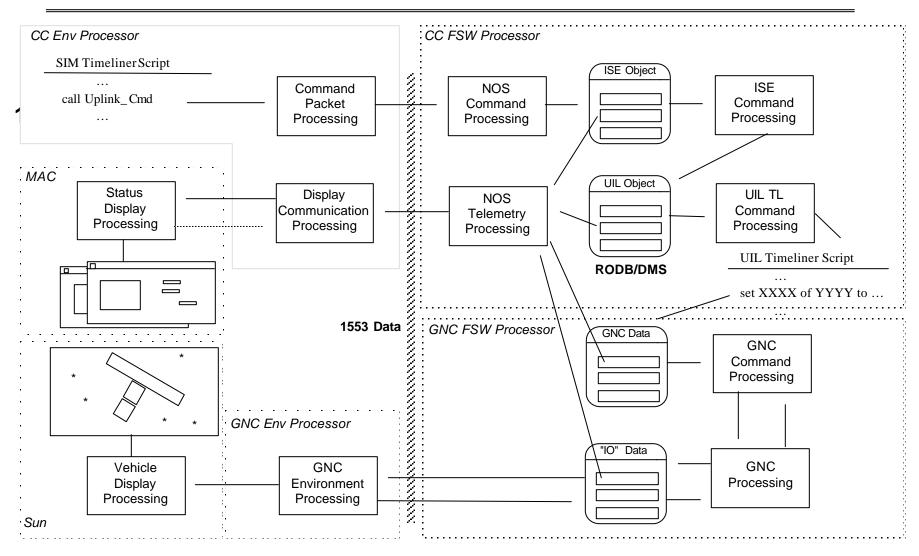
ISS EFCS Test Bed Configuration



- Currently running flight software on the SDP emulation.
 - 80386DX (about twice as fast as MDM 80386SX).
 - Multibus II backplane bus.
 - No EEPROM; all RAM.
- Real-time environment models, along with a model of a Bus Interface Unit, run on 80486.
- Ethernet card is used by the Environment processors to send simulation data to "outside world".
- Data Center collects and logs data, sends data to displays or analysis programs.

Demo Data Flow

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- For the Control System software, the following roles need to be partitioned among the available personnel:
 - Overall leader
 - » Responsible for creating the Software Development Plan, maintaining the schedule, creating status reports, etc.
 - Requirements Analyst
 - Responsible for writing the Software Requirements Specification (SRS)
 - Control algorithm developer
 - » Responsible for the design of the control systems
 - Generates at least the Top-Level Design documentation for the Control system
 - Software architect
 - » Responsible for the high-level software design
 - Creates at least the Top-Level Design documentation laying out the structure of the software

- Control software coder
 - » Writes the Control software
- Design documenter
 - » Writes the Detailed Design document
- Test Lead
 - » Writes the Software Test Plan
- Test SW algorithm developer
- Test SW coder
- Version Control person
 - » Responsible for dealing with the version control system
- Integration lead
 - » The problem solver. Responsible for integrating the Control software with the other software in the ISS, and getting it to work
- Display developer
 - » Takes telemetry data and displays it

Guidelines

- Expect requirements changes
 - Trying to stay ahead of the main developers means NASA or the contractors might change something
- The customer wants demonstrations. Part of the job is making sure the demonstrations are professional
 - Look good
 - Provide enough information to show the system working well
- All the software was developed quickly. There is no guarantee that problems are all due to new software