

**UNIFIED HANDOUT**  
**MATERIALS AND STRUCTURES - #M-8**  
Fall, 2008

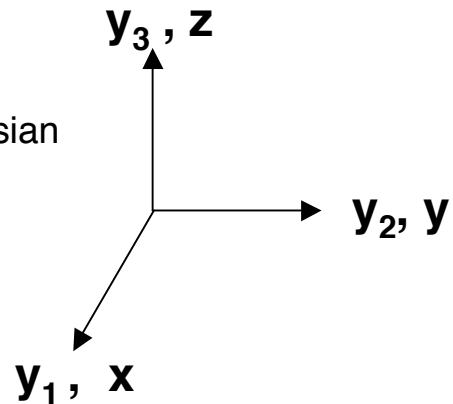
**Review of Stress, Strain and Elasticity**

**NOTATION REVIEW** (e.g., for strain)

<u>Engineering</u>	=	<u>Contracted</u>	=	<u>Engineering “Tensor”</u>	=	<u>Tensor</u>
$\epsilon_x$	=	$\epsilon_1$	=	$\epsilon_{xx}$	=	$\epsilon_{11}$
$\epsilon_y$	=	$\epsilon_2$	=	$\epsilon_{yy}$	=	$\epsilon_{22}$
$\epsilon_z$	=	$\epsilon_3$	=	$\epsilon_{zz}$	=	$\epsilon_{33}$
$\gamma_{yz}$	=	$\epsilon_4$	=	$2 \epsilon_{yz}$	=	$2 \epsilon_{23}$
$\gamma_{xz}$	=	$\epsilon_5$	=	$2 \epsilon_{xz}$	=	$2 \epsilon_{13}$
$\gamma_{xy}$	=	$\epsilon_6$	=	$2 \epsilon_{xy}$	=	$2 \epsilon_{12}$

**EQUATIONS OF ELASTICITY**

Right-handed rectangular Cartesian coordinate system



15 equations/15 unknowns

1. Equilibrium (3)

$$\left. \begin{aligned} \frac{\partial \sigma_{11}}{\partial y_1} + \frac{\partial \sigma_{21}}{\partial y_2} + \frac{\partial \sigma_{31}}{\partial y_3} + f_1 &= 0 \\ \frac{\partial \sigma_{12}}{\partial y_1} + \frac{\partial \sigma_{22}}{\partial y_2} + \frac{\partial \sigma_{32}}{\partial y_3} + f_2 &= 0 \\ \frac{\partial \sigma_{13}}{\partial y_1} + \frac{\partial \sigma_{23}}{\partial y_2} + \frac{\partial \sigma_{33}}{\partial y_3} + f_3 &= 0 \end{aligned} \right\} \frac{\partial \sigma_{mn}}{\partial y_m} + f_n = 0$$

## 2. Strain-Displacement (6)

$$\left. \begin{aligned} \varepsilon_{11} &= \frac{\partial u_1}{\partial y_1} & \varepsilon_{21} = \varepsilon_{12} &= \frac{1}{2} \left( \frac{\partial u_1}{\partial y_2} + \frac{\partial u_2}{\partial y_1} \right) \\ \varepsilon_{22} &= \frac{\partial u_2}{\partial y_2} & \varepsilon_{31} = \varepsilon_{13} &= \frac{1}{2} \left( \frac{\partial u_1}{\partial y_3} + \frac{\partial u_3}{\partial y_1} \right) \\ \varepsilon_{33} &= \frac{\partial u_3}{\partial y_3} & \varepsilon_{32} = \varepsilon_{23} &= \frac{1}{2} \left( \frac{\partial u_2}{\partial y_3} + \frac{\partial u_3}{\partial y_2} \right) \end{aligned} \right\} \varepsilon_{mn} = \frac{1}{2} \left( \frac{\partial u_m}{\partial y_n} + \frac{\partial u_n}{\partial y_m} \right)$$

## 3. Stress-Strain (6)

$$\sigma_{mn} = E_{mnpq} \varepsilon_{pq}$$

Generalized Hooke's Law:

• Anisotropic:

$$\begin{Bmatrix} \sigma_{11} \\ \sigma_{22} \\ \sigma_{33} \\ \sigma_{23} \\ \sigma_{13} \\ \sigma_{12} \end{Bmatrix} = \begin{bmatrix} E_{1111} & E_{1122} & E_{1133} & 2E_{1123} & 2E_{1113} & 2E_{1112} \\ E_{1122} & E_{2222} & E_{2233} & 2E_{2223} & 2E_{2213} & 2E_{2212} \\ E_{1133} & E_{2233} & E_{3333} & 2E_{3323} & 2E_{3313} & 2E_{3312} \\ E_{1123} & E_{2223} & E_{3323} & 2E_{2323} & 2E_{1323} & 2E_{1223} \\ E_{1113} & E_{2213} & E_{3313} & 2E_{1323} & 2E_{1313} & 2E_{1213} \\ E_{1112} & E_{2212} & E_{3312} & 2E_{1223} & 2E_{1213} & 2E_{1212} \end{bmatrix} \begin{Bmatrix} \varepsilon_{11} \\ \varepsilon_{22} \\ \varepsilon_{33} \\ \varepsilon_{23} \\ \varepsilon_{13} \\ \varepsilon_{12} \end{Bmatrix}$$

• Orthotropic:

$$\begin{Bmatrix} \sigma_{11} \\ \sigma_{22} \\ \sigma_{33} \\ \sigma_{23} \\ \sigma_{13} \\ \sigma_{12} \end{Bmatrix} = \begin{bmatrix} E_{1111} & E_{1122} & E_{1133} & 0 & 0 & 0 \\ E_{1122} & E_{2222} & E_{2233} & 0 & 0 & 0 \\ E_{1133} & E_{2233} & E_{3333} & 0 & 0 & 0 \\ 0 & 0 & 0 & 2E_{2323} & 0 & 0 \\ 0 & 0 & 0 & 0 & 2E_{1313} & 0 \\ 0 & 0 & 0 & 0 & 0 & 2E_{1212} \end{bmatrix} \begin{Bmatrix} \varepsilon_{11} \\ \varepsilon_{22} \\ \varepsilon_{33} \\ \varepsilon_{23} \\ \varepsilon_{13} \\ \varepsilon_{12} \end{Bmatrix}$$

Compliance Form:  $\varepsilon_{mn} = S_{mnpq} \sigma_{pq}$

where:  $\tilde{E}^{-1} = \tilde{S}$

## DEFINITION OF ENGINEERING CONSTANTS

1. Longitudinal (Young's) (Extensional) Moduli:

$$E_{mm} = \frac{\sigma_{mm}}{\varepsilon_{mm}} \quad \text{due to } \sigma_{mm} \text{ applied } \underline{\text{only}} \quad (\text{no summation on } m)$$

2. Poisson's Ratios:

$$\nu_{nm} = -\frac{\varepsilon_{mm}}{\varepsilon_{nn}} \quad \text{due to } \sigma_{nn} \text{ applied } \underline{\text{only}} \quad (\text{for } n \neq m)$$

$$\text{Reciprocity: } \nu_{nm} E_m = \nu_{mn} E_n \quad (\text{no sum}) \\ (m \neq n)$$

3. Shear Moduli:

$$G_{mn} = \frac{\sigma_{mn}}{2\varepsilon_{mn}} \quad \text{due to } \sigma_{mn} \text{ applied } \underline{\text{only}}$$

## "ENGINEERING" STRESS-STRAIN EQUATIONS (using contracted notation)

• Orthotropic form

In terms of ENGINEERING CONSTANTS (using *contracted notation*):

$$\begin{Bmatrix} \varepsilon_1 \\ \varepsilon_2 \\ \varepsilon_3 \\ \varepsilon_4 \\ \varepsilon_5 \\ \varepsilon_6 \end{Bmatrix} = \begin{bmatrix} \frac{1}{E_1} & -\frac{\nu_{12}}{E_1} & -\frac{\nu_{13}}{E_1} & 0 & 0 & 0 \\ -\frac{\nu_{21}}{E_2} & \frac{1}{E_2} & -\frac{\nu_{23}}{E_2} & 0 & 0 & 0 \\ -\frac{\nu_{31}}{E_3} & -\frac{\nu_{32}}{E_3} & \frac{1}{E_3} & 0 & 0 & 0 \\ 0 & 0 & 0 & \frac{1}{G_4} & 0 & 0 \\ 0 & 0 & 0 & 0 & \frac{1}{G_5} & 0 \\ 0 & 0 & 0 & 0 & 0 & \frac{1}{G_6} \end{bmatrix} \begin{Bmatrix} \sigma_1 \\ \sigma_2 \\ \sigma_3 \\ \sigma_4 \\ \sigma_5 \\ \sigma_6 \end{Bmatrix}$$

• Isotropic form

$$\begin{Bmatrix} \varepsilon_1 \\ \varepsilon_2 \\ \varepsilon_3 \\ \varepsilon_4 \\ \varepsilon_5 \\ \varepsilon_6 \end{Bmatrix} = \begin{bmatrix} 1/E & -\nu/E & -\nu/E & 0 & 0 & 0 \\ -\nu/E & 1/E & -\nu/E & 0 & 0 & 0 \\ -\nu/E & -\nu/E & 1/E & 0 & 0 & 0 \\ 0 & 0 & 0 & 1/G & 0 & 0 \\ 0 & 0 & 0 & 0 & 1/G & 0 \\ 0 & 0 & 0 & 0 & 0 & 1/G \end{bmatrix} \begin{Bmatrix} \sigma_1 \\ \sigma_2 \\ \sigma_3 \\ \sigma_4 \\ \sigma_5 \\ \sigma_6 \end{Bmatrix}$$

with:  $G = \frac{E}{2(1 + \nu)}$

**TRANSFORMATIONS**

$$\tilde{\sigma}_{mn} = l_{\tilde{m}p} l_{\tilde{n}q} \sigma_{pq}$$

$$\tilde{\varepsilon}_{mn} = l_{\tilde{m}p} l_{\tilde{n}q} \varepsilon_{pq}$$

$$\tilde{x}_m = l_{\tilde{m}p} x_p$$

$$\tilde{u}_m = l_{\tilde{m}p} u_p$$

$$\tilde{E}_{mnpq} = l_{\tilde{m}r} l_{\tilde{n}s} l_{\tilde{p}t} l_{\tilde{q}u} E_{rstu}$$

where:  $l_{\tilde{m}n} = \cosine \text{ of angle from } \tilde{y}_m \text{ to } y_n$