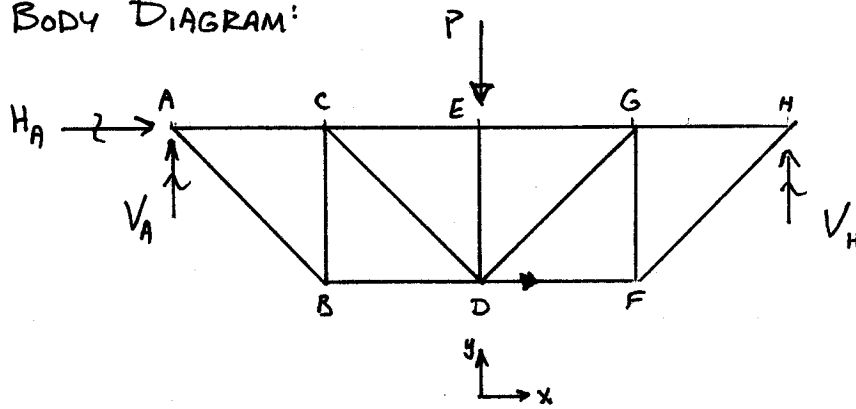


MX 8

FREE BODY DIAGRAM:



APPLY EQUILIBRIUM TO FIND REACTIONS

$$\rightarrow \sum F_x = 0$$

$$H_A = 0 \text{ N}$$

$$\uparrow \sum F_y = 0$$

$$V_A + V_H - P = 0$$

$$+ (\sum M_A = 0$$

$$V_H(4L) - P(2L) = 0$$

$$\Rightarrow \begin{cases} V_H = \frac{P}{2} \\ V_A = \frac{P}{2} \end{cases}$$

TO DETERMINE THE DEFLECTION OF D, WE NEED TO EMPLOY COMPATIBILITY + CONSTITUTIVE LAWS.

OUR CONSTITUTIVE LAW FOR BAR DEFORMATION IS:

$$\delta_{ij} = \frac{F_{ij} L_{ij}}{AE}$$

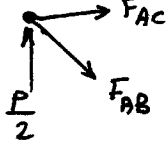
SO WE'LL NEED TO SOLVE FOR THE BAR FORCES IN ORDER TO DETERMINE THEIR EXTENSIONS, AND HENCE THE TRUSS DEFLECTION.

BECAUSE OF SYMMETRY, I ONLY NEED TO FIND HALF OF THE BAR FORCES. ALL OF THE PAIRS MIRRORED IN THE D-E AXIS WILL HAVE THE SAME BAR FORCE:

$$\begin{aligned} F_{AC} &= F_{CH} & F_{BD} &= F_{DC} \\ F_{AB} &= F_{FH} & F_{CD} &= F_{DG} \\ F_{BC} &= F_{FC} & F_{CE} &= F_{EG} \end{aligned}$$

SOLVE FOR INDEPENDENT BAR FORCES:

MOJ @ A:



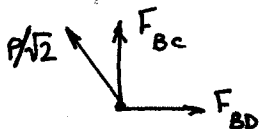
$$\sum F_y = \frac{P}{2} - F_{AB} \cos 45 = 0$$

$$F_{AB} = \frac{P}{\sqrt{2}}$$

$$\sum F_x = F_{AC} + F_{AB} \sin 45 = 0$$

$$F_{AC} = -\frac{P}{2}$$

MOJ @ B:



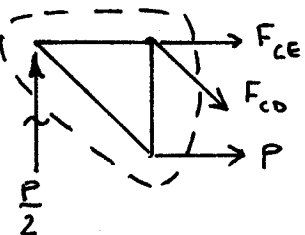
$$\sum F_y = 0 = F_{BC} + (P/\sqrt{2}) \cos 45$$

$$F_{BC} = -P/2$$

$$\sum F_x = 0 = F_{BD} - (P/\sqrt{2}) \sin 45$$

$$F_{BD} = P/2$$

MOS:



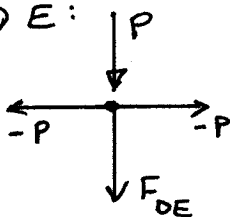
$$\sum F_y = 0 = \frac{P}{2} - F_{CD} \cos 45$$

$$F_{CD} = \frac{P}{\sqrt{2}}$$

$$\sum M_D = 0 = -F_{CE} \cdot k - \frac{P}{2} \cdot (2k) = 0$$

$$F_{CE} = -P$$

MOJ @ E:



$$\sum F_y = 0$$

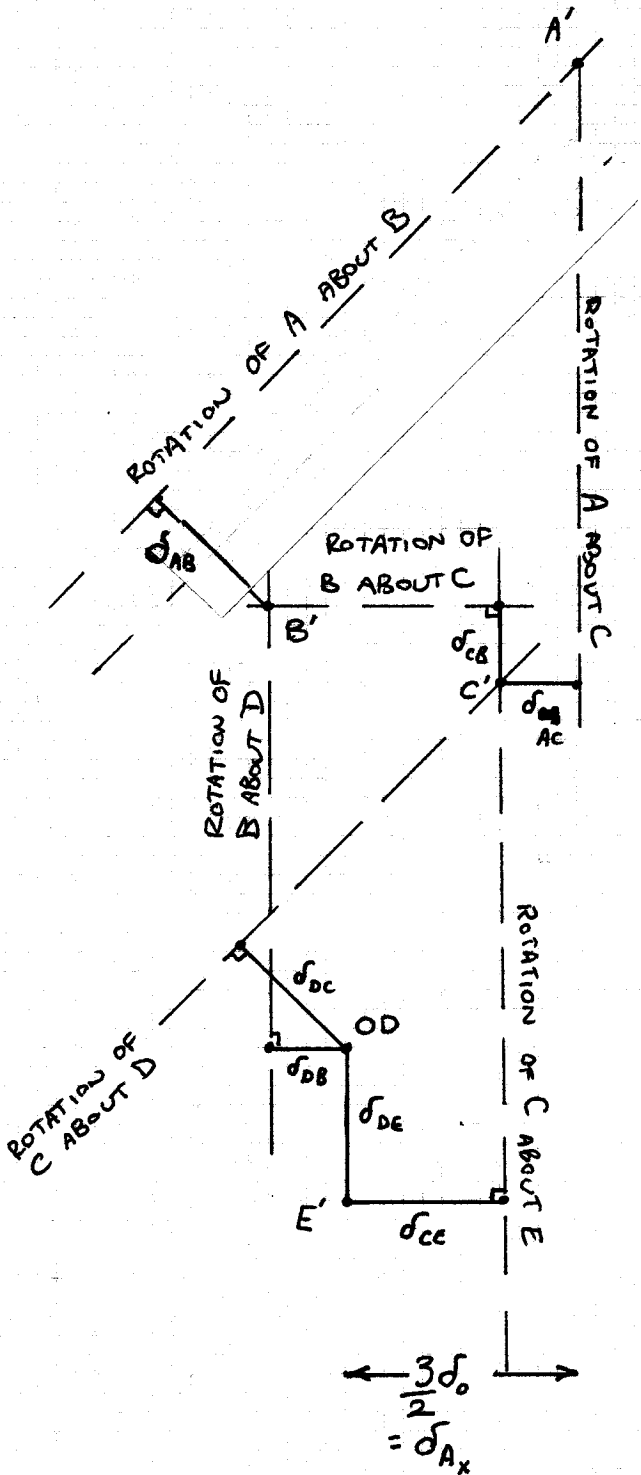
$$-P - F_{DE} = 0$$

$$F_{DE} = -P$$

BAR	FORCE $\left(\frac{F_{ij}}{P}\right)$	LENGTH $\left(\frac{L_{ij}}{L}\right)$	DEFORMATION $\frac{\delta_{ij}}{PL/AE}$
AB	$+\frac{1}{\sqrt{2}}$	$\sqrt{2}$	$+1$
AC	$-\frac{1}{2}$	1	$-\frac{1}{2}$
CB	$-\frac{1}{2}$	1	$-\frac{1}{2}$
CE	-1	1	-1
CD	$+\frac{1}{\sqrt{2}}$	$\sqrt{2}$	$+1$
BD	$+\frac{1}{2}$	1	$+\frac{1}{2}$
ED	-1	1	-1
EG	-1	1	-1
DG	$+\frac{1}{\sqrt{2}}$	$\sqrt{2}$	$+1$
DF	$+\frac{1}{2}$	1	$+\frac{1}{2}$
GF	$-\frac{1}{2}$	1	$-\frac{1}{2}$
GH	$-\frac{1}{2}$	1	$-\frac{1}{2}$
FH	$+\frac{1}{\sqrt{2}}$	$\sqrt{2}$	$+1$

NOW WE CAN GO AHEAD AND PLOT OUR TRUSS DEFLECTION DIAGRAM.

$$\delta_o = \frac{PL}{AE}$$



IF MY HINGE POINT A' ENDS UP DISPLACED FROM MY ORIGIN BY δ_{Ax} AND δ_{Ay} , THEN MY ORIGIN OD IS DISPLACED FROM A' BY $-\delta_{Ax}$ AND $-\delta_{Ay}$.

$$\frac{13\delta_o}{2} = \delta_{Ay}$$

IF I NOW CONSIDER THE FIXED FRAME OA , WHERE A AND A' ARE THE SAME, I CAN FIND THE DEFLECTION OF D' IN THE FIXED FRAME, WHICH IS JUST ITS DISPLACEMENT FROM A' , NAMELY $-\delta_{Ax} \hat{i} - \delta_{Ay} \hat{j}$.

THE JOINT D WILL TRANSLATE DOWN BY $\frac{13}{2} \frac{PL}{AE}$, AND LEFT BY $\frac{3}{2} \frac{PL}{AE}$

ESTIMATE OF TRUSS DEFLECTIONS

BARS IN EXPERIMENTAL TRUSS MADE OF STEEL

- HOLLOW WITH 22 mm OUTER DIAMETER AND

1.5 mm WALL THICKNESS (IGNORE END FITTINGS)

$$A \approx 2\pi r t \approx 10.5 \times 10^{-3} \times 2 \times \pi \times 1.5 \times 10^{-3} \\ \approx 100 \text{ mm}^2$$

$$L = 0.5 \text{ m}$$

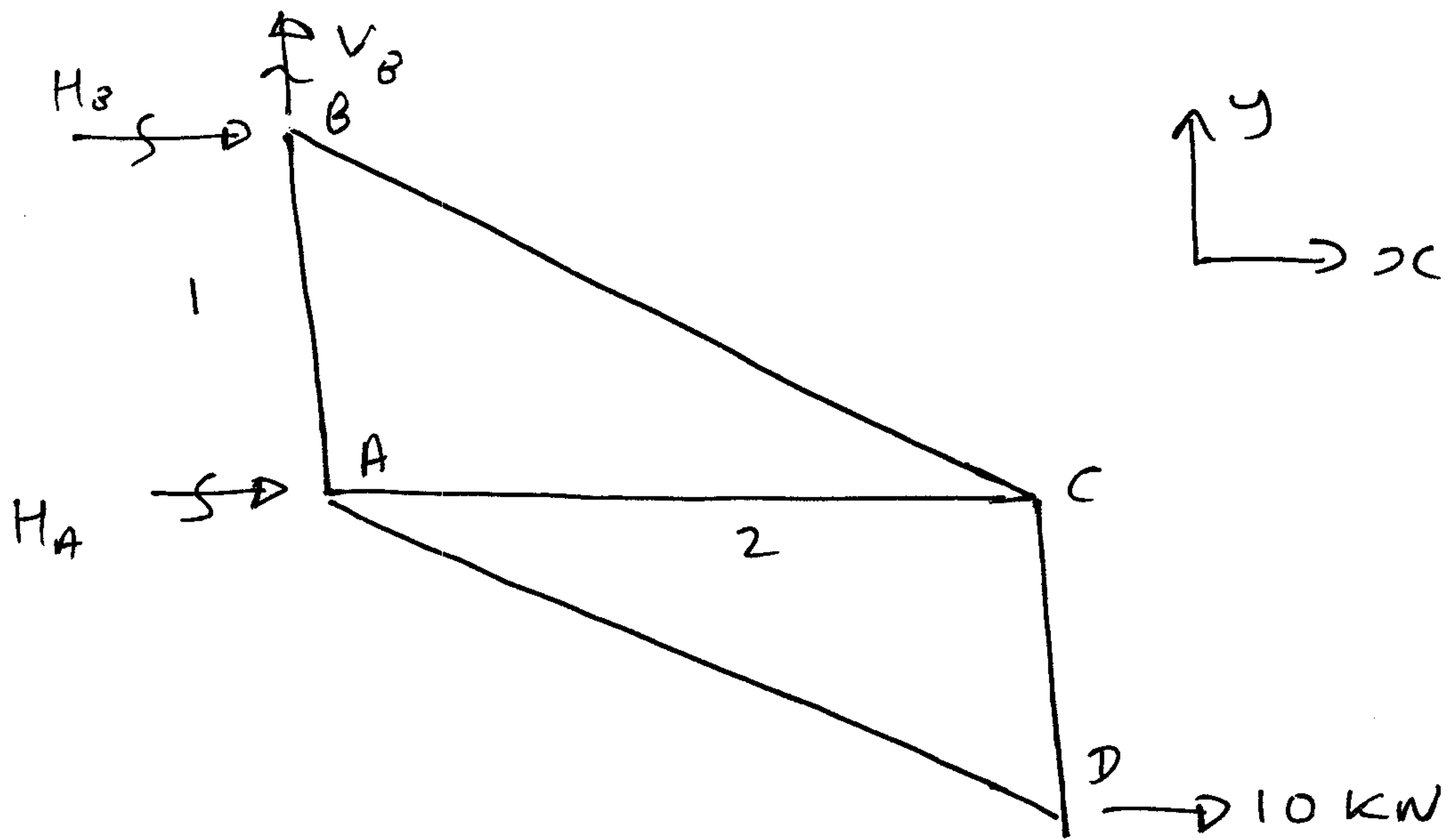
$$E = 210 \text{ GPa}$$

CENTER POINT DEFLECTION

$$\frac{\delta_D}{P} = \left(\frac{0.5}{100 \times 10^{-6} \times 210 \times 10^9} \right) \left(\frac{-13\hat{j}}{2} - \frac{3\hat{i}}{2} \right)$$

$$\boxed{\frac{\delta_D}{P} = -\frac{3.1 \times 10^{-7}}{2} \hat{j} - \frac{7.1 \times 10^{-8}}{2} \hat{i} \quad \text{N}^{-1}}$$

M9



$$\sum \vec{F}_x = 0: 10 + H_B + H_A = 0 \quad (1)$$

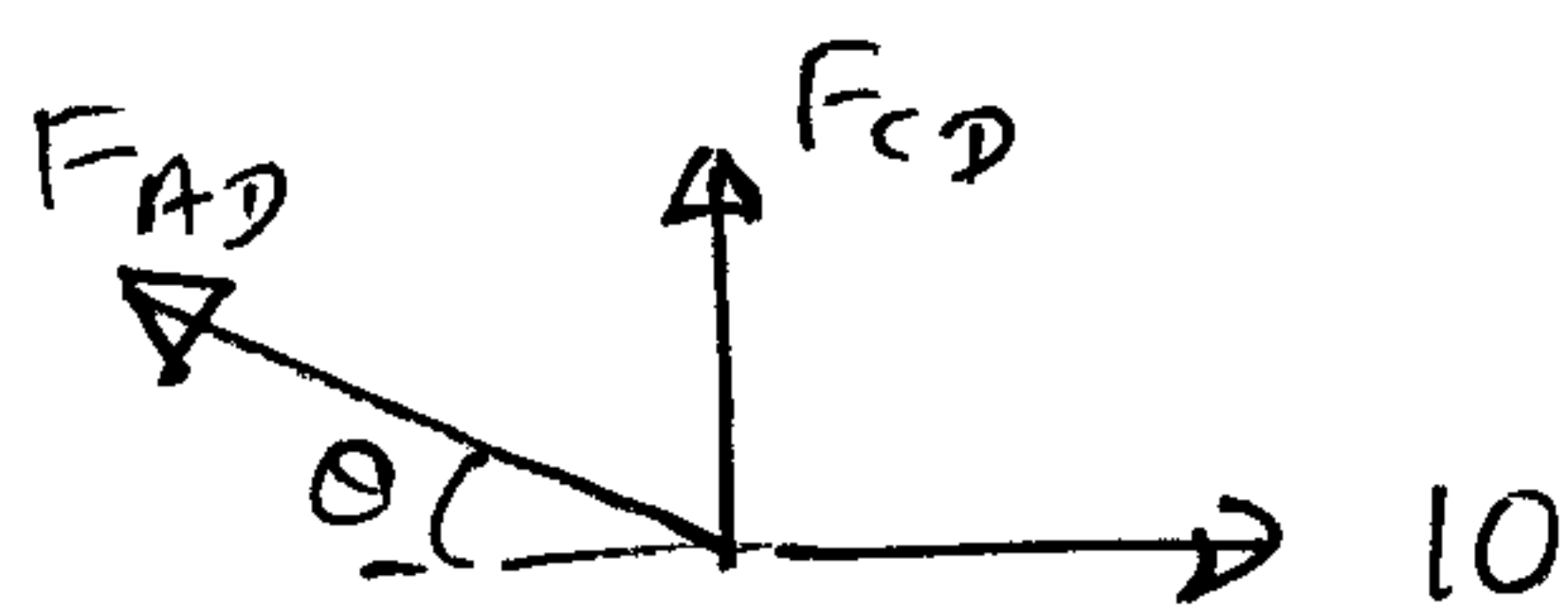
$$\sum \vec{F}_y \uparrow = 0: V_B + 0 = 0 \Rightarrow V_B = 0 \quad (2)$$

$$\sum (M_B) = 0: H_A \cdot 1 + 10 \cdot 2 = 0 \Rightarrow H_A = -20 \text{ kN} \quad (3) \Leftarrow$$

Substitute in (1) $H_B = +10 \text{ kN} \Leftarrow$

Bar Forces

Moment @ D



$$\cos \theta = \frac{2}{\sqrt{5}}$$

$$\sin \theta = \frac{1}{\sqrt{5}}$$

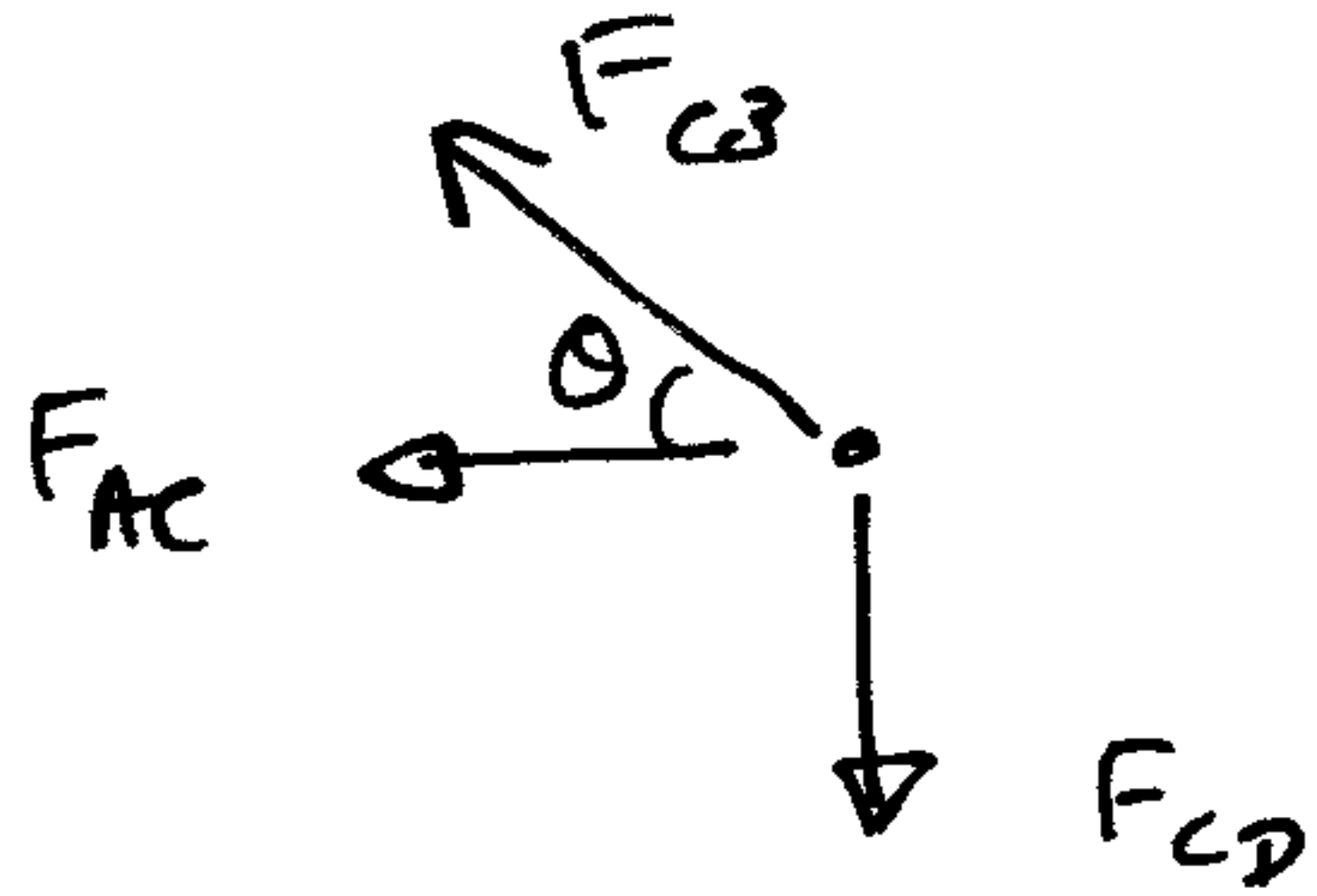
$$\sum \vec{F}_x = 0 \quad -F_{AD} \cos \theta + 10 = 0 \Rightarrow F_{AD} = \frac{10 \cdot \sqrt{5}}{2} = +11.2 \text{ kN}$$

$$\sum \vec{F}_y \uparrow = 0 \quad F_{AD} \sin \theta + F_{CD} = 0$$

$$F_{CD} = -F_{AD} \sin \theta = -\frac{10 \cdot \sqrt{5}}{2} \cdot \frac{1}{\sqrt{5}} = -5 \text{ kN} \Leftarrow$$

MOS @ C.

Σ



$$\sin \theta = \frac{1}{\sqrt{5}}$$

$$\cos \theta = \frac{2}{\sqrt{5}}$$

$$\Sigma F_y \uparrow = 0$$

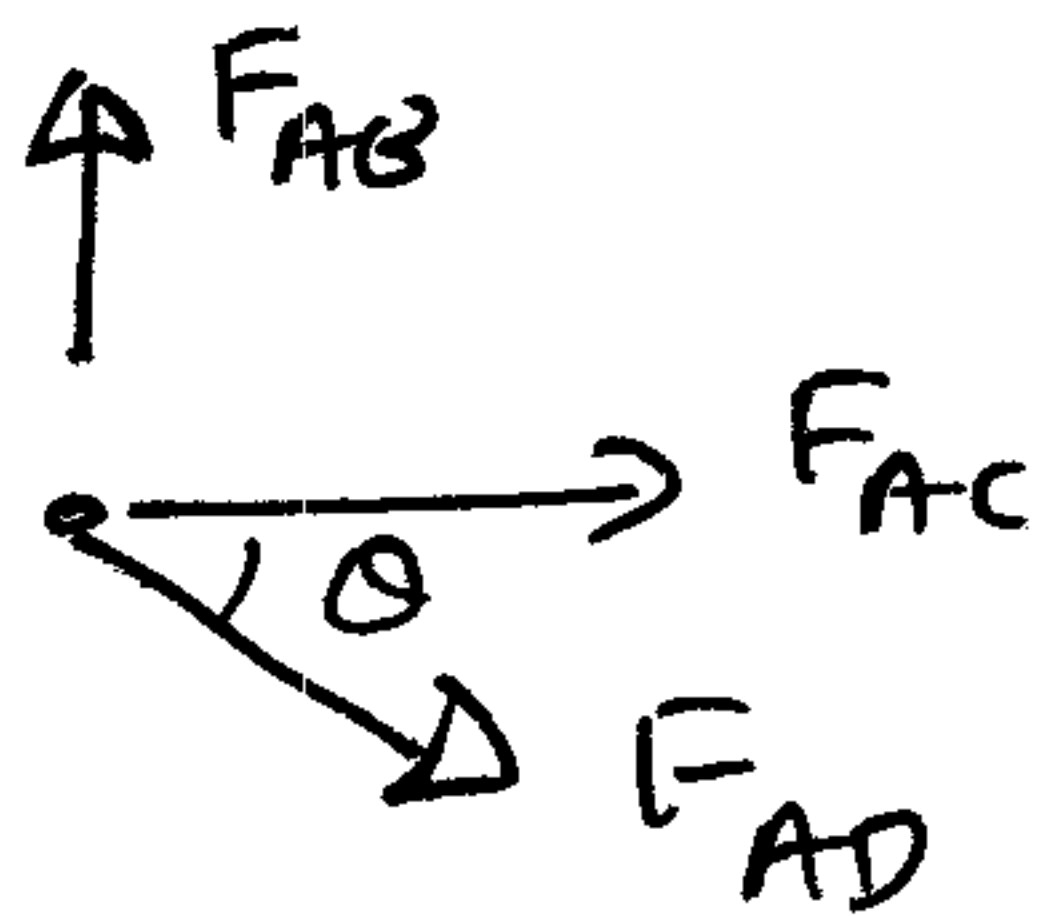
$$F_{CB} \sin \theta - F_{CD} = 0$$

$$F_{CB} = + \frac{F_{CD}}{\sin \theta} = \underline{-5\sqrt{5} \text{ KN} \Leftarrow}$$

$$\Sigma \vec{F}_x = 0 : -F_{AC} - F_{CD} \cos \theta = 0$$

$$F_{AC} = -F_{CD} \cos \theta = +5\sqrt{5} \cdot \frac{2}{\sqrt{5}} = +10 \text{ KN} \Leftarrow$$

MOS @ A



$$\cos \theta = \frac{2}{\sqrt{5}}$$

$$\sin \theta = \frac{1}{\sqrt{5}}$$

$$\Sigma F_y \uparrow = 0$$

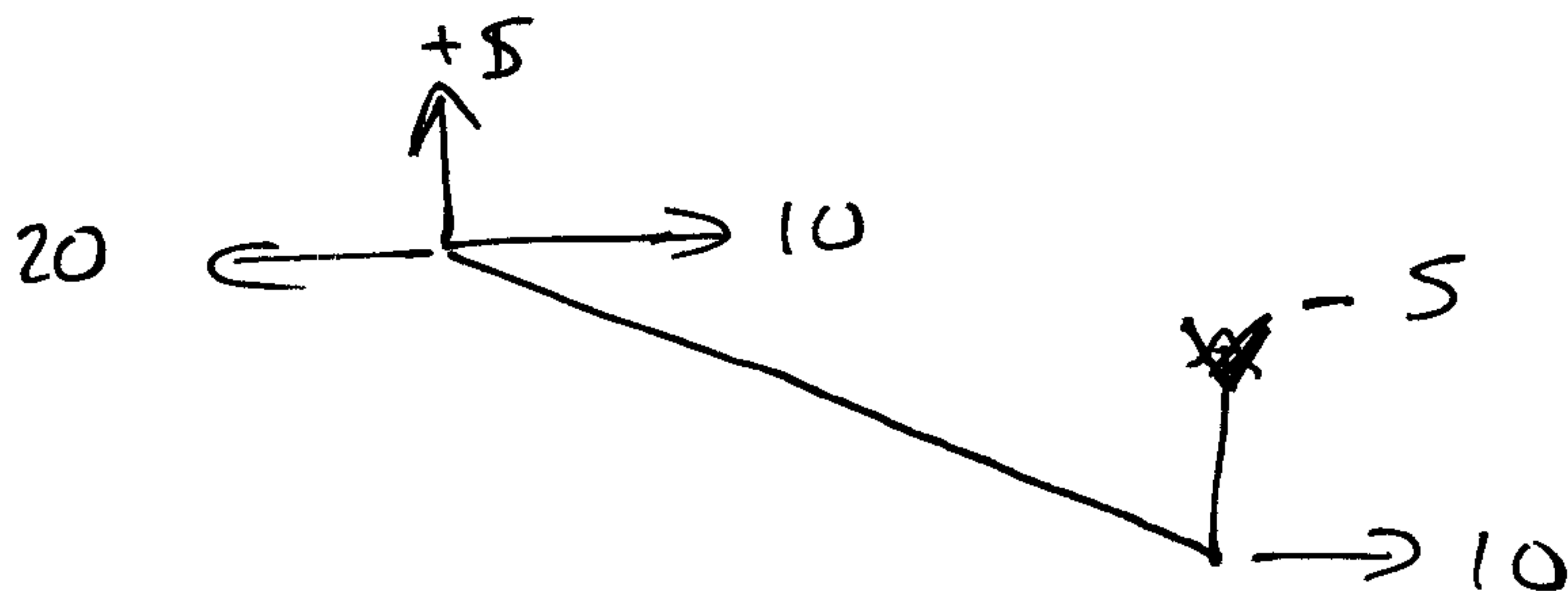
$$F_{AB} - F_{AD} \sin \theta = 0$$

$$\Sigma \vec{F}_x = 0$$

$$F_{AB} = F_{AD} \sin \theta$$

$$= 10 + 5\sqrt{5} \cdot \frac{1}{\sqrt{5}} = +5 \text{ KN} \Leftarrow$$

check MOS



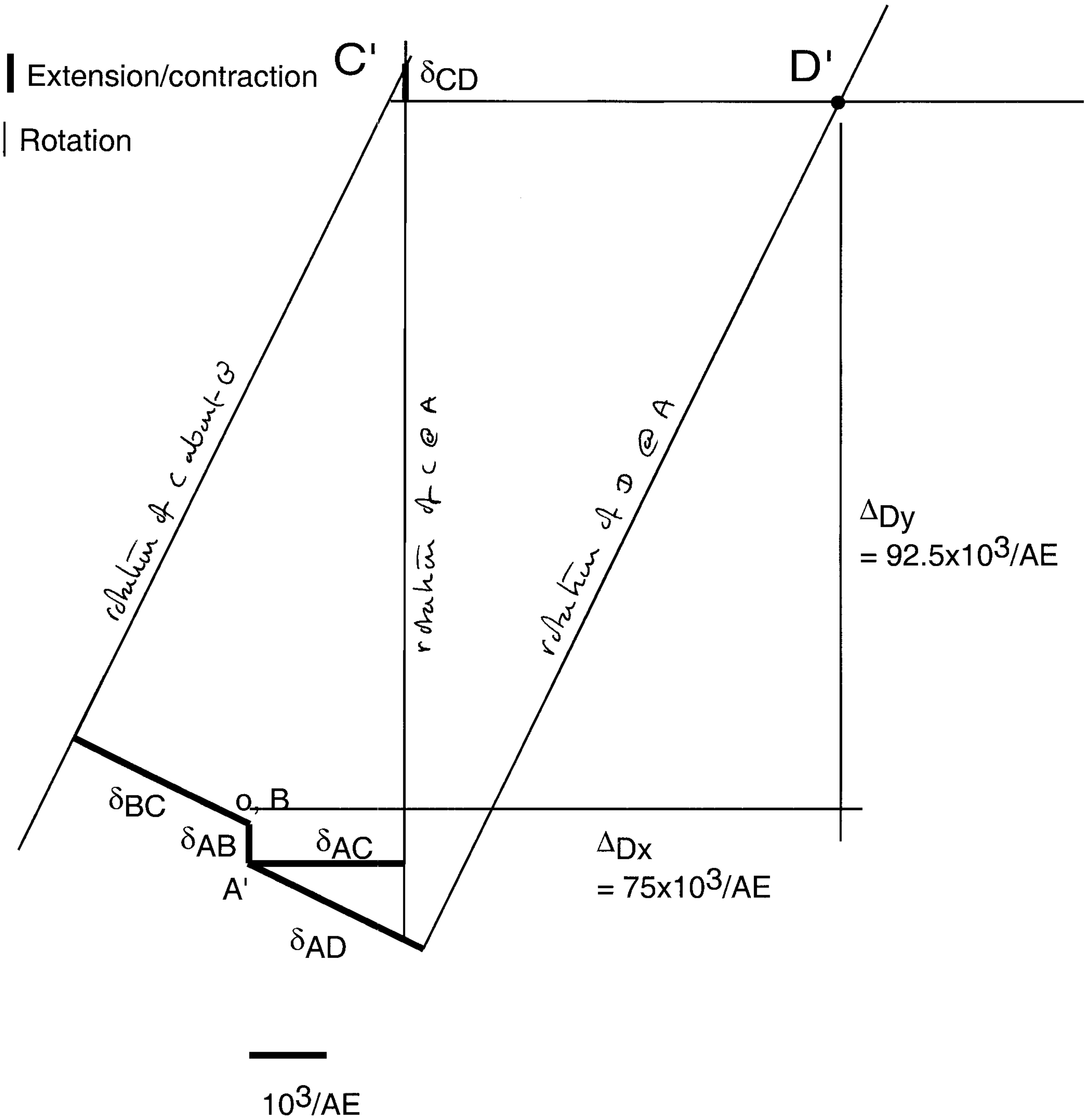
OK!

Bar	Force (kN)	F/P	Length	Length/l	$\delta / PL/AE$	$\frac{\delta}{AE} \times 10^3$
AB	+5	$+\frac{1}{2}$	1	1	$+\frac{1}{2}$	+5
AC	+10	+1	2	2	+2	+20
BC	$-5\sqrt{5}$	$-\sqrt{5}/2$	$\sqrt{5}$	$\sqrt{5}$	$-\frac{5}{2}$	-25
AD	$+5\sqrt{5}$	$+\sqrt{5}/2$	$\sqrt{5}$	$\sqrt{5}$	$+\frac{5}{2}$	+25
CD	-5	$-\frac{1}{2}$	1	1	$-\frac{1}{2}$	-5

Draw displacement diagram (see attached)

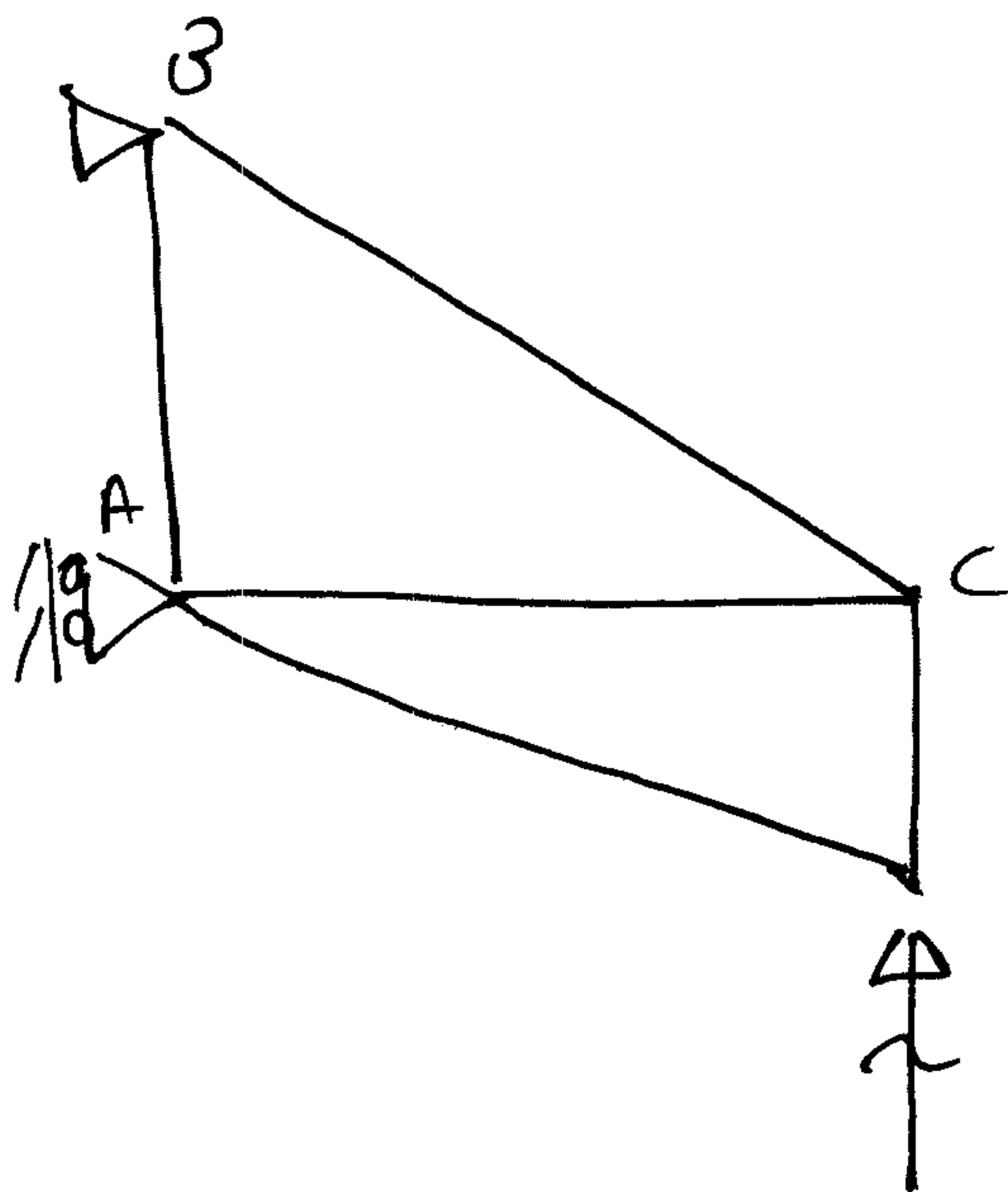
$$\text{Vertical deflection} = \frac{92.5 \times 10^3}{AE} = 2.64 \times 10^{-3} \text{ m} \\ = 2.64 \text{ mm} \leftarrow$$

$$\text{Horizontal deflection} = \frac{75 \times 10^3}{AE} = 2.14 \times 10^{-3} \text{ m} \\ = 2.14 \text{ mm} \leftarrow$$

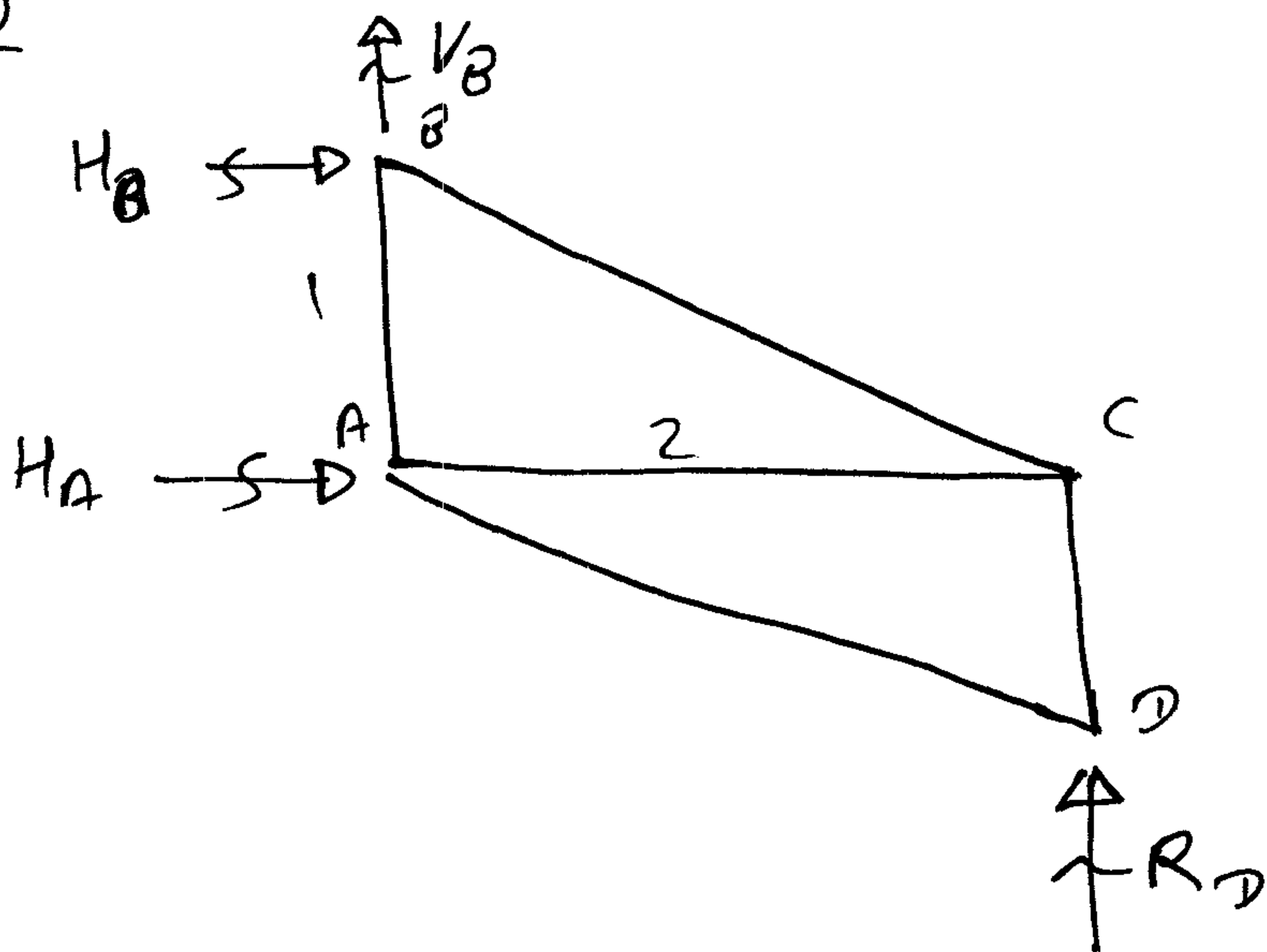


M10

Use superposition (or set up as a set of unknowns)



FBD



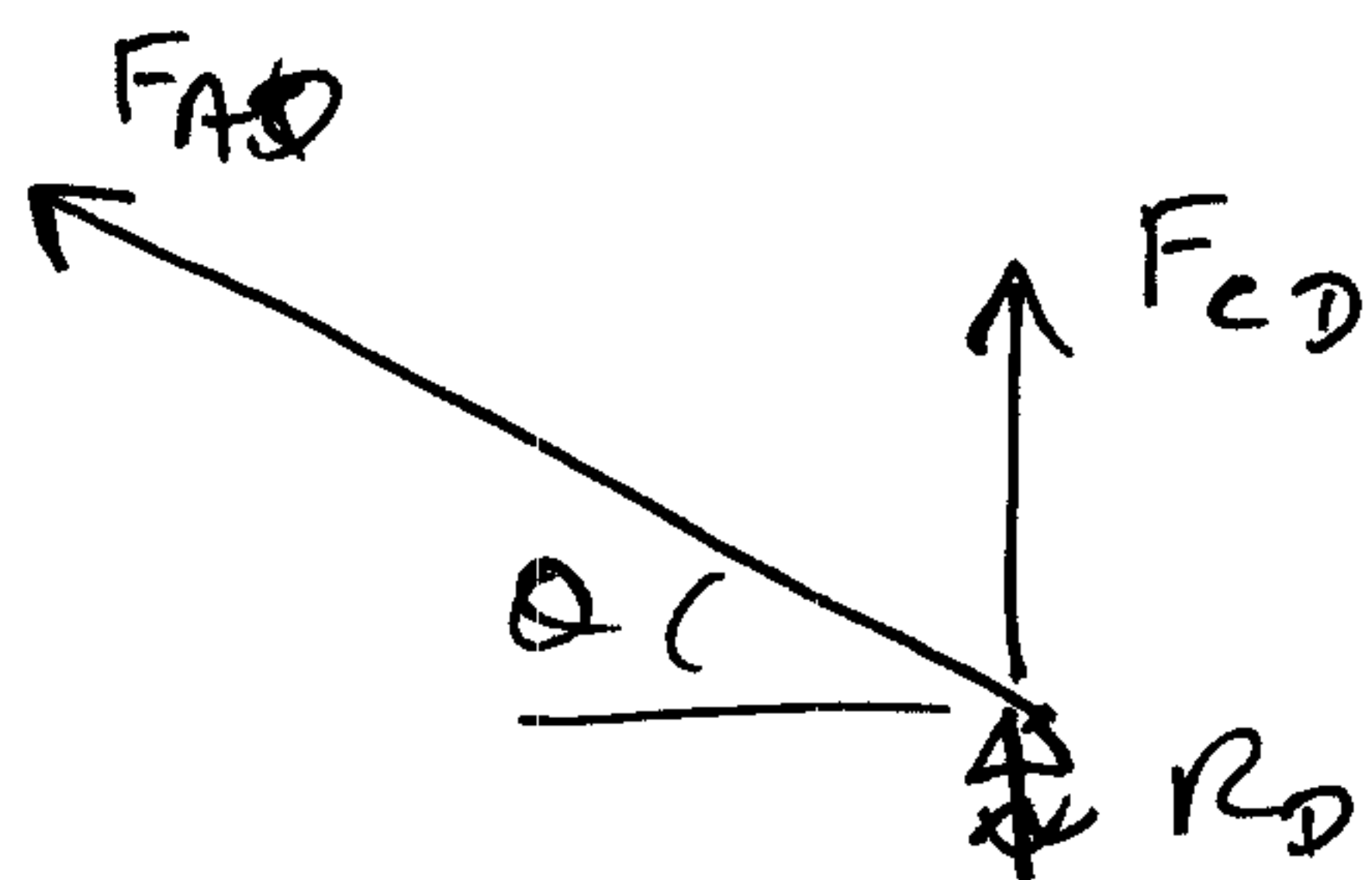
$$\sum F_y \uparrow = 0: V_B + R_D = 0 \quad (1) \quad V_B = -R_D$$

$$\sum \vec{F}_x = 0: H_A + H_B = 0 \quad (2)$$

$$\sum \vec{M}_A = 0: H_A \cdot 1 + R_D \cdot 2 = 0 \quad H_A = -2R_D$$

$$H_B = +2R_D.$$

Bar forces: Method of joints



$$\cos \theta = \frac{2}{\sqrt{5}}$$

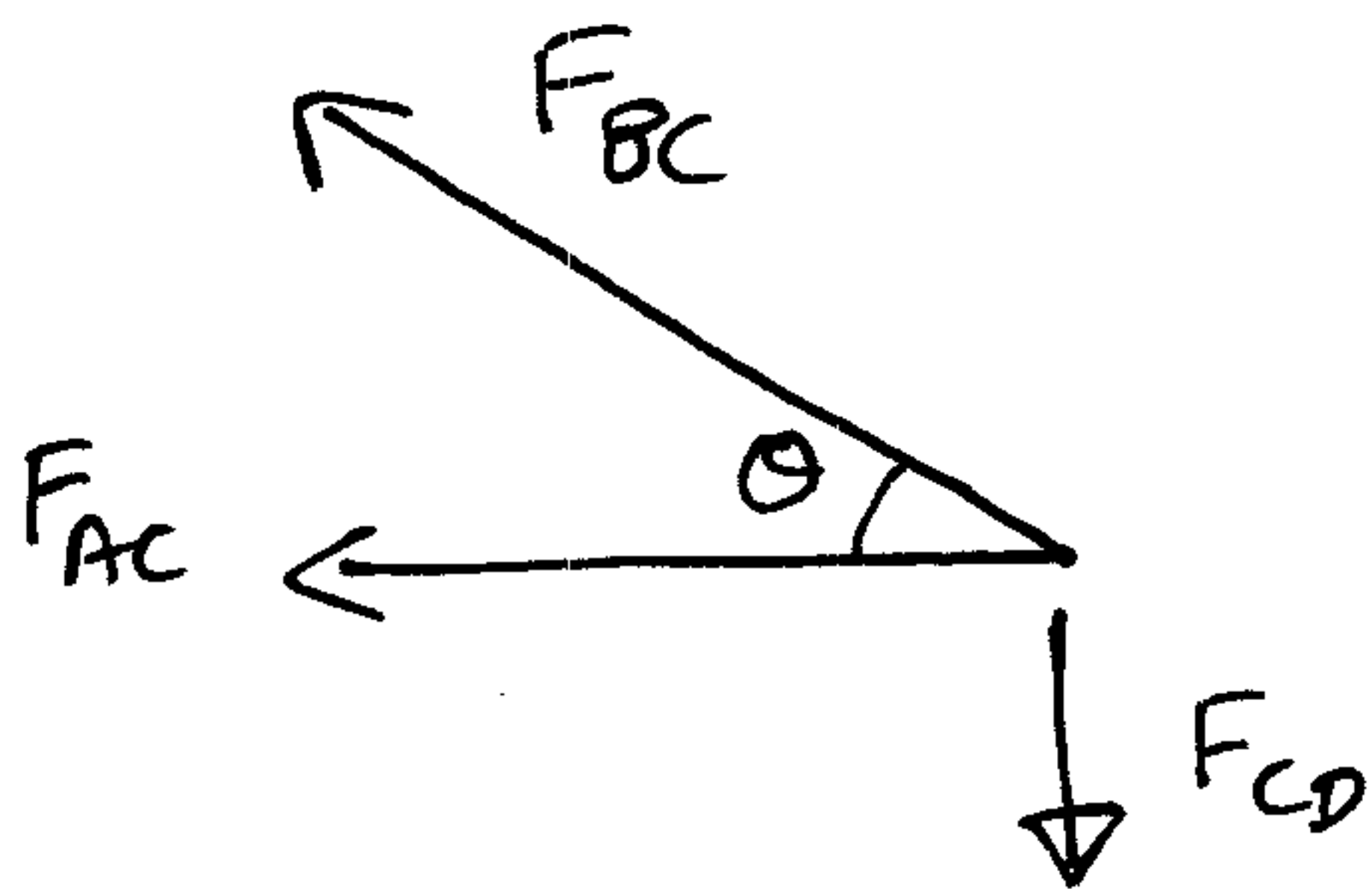
$$\sin \theta = \frac{1}{\sqrt{5}}$$

$$\sum \vec{F}_x = 0: F_{AD} \cos \theta = 0 \Rightarrow F_{AD} = 0$$

$$\sum F_y \uparrow = 0 \quad F_{AD} \sin \theta + F_{CD} + R_D = 0$$

$$F_{CD} = -R_D$$

2

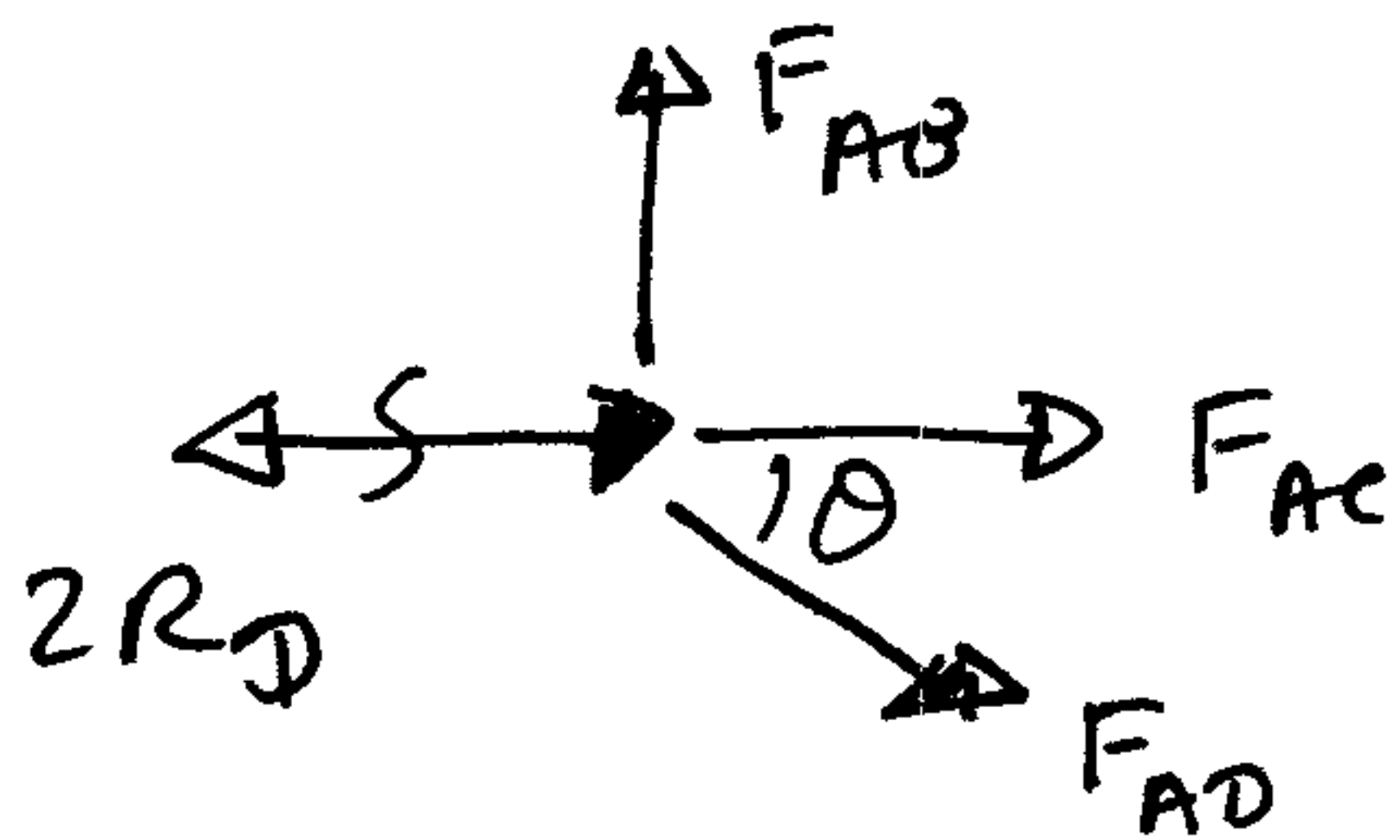


$$\sum F_y \uparrow = 0: F_{BC} \sin \theta - F_{CD} = 0$$

$$F_{BC} = \underline{\underline{-R_D \sqrt{5}}}$$

$$\sum \vec{F}_x = 0: -F_{AC} - F_{BC} \cos \theta = 0$$

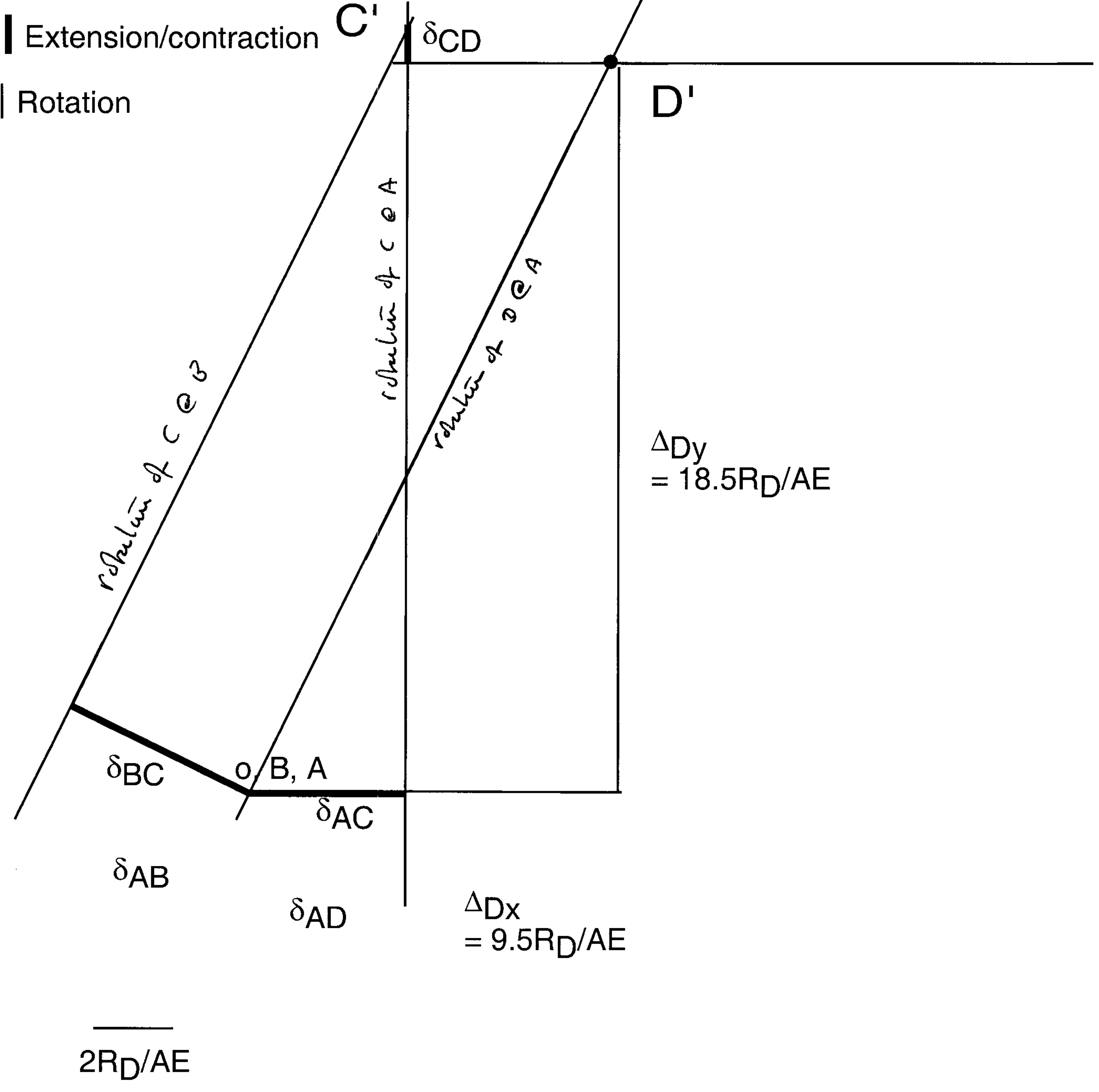
$$F_{AC} = R_D \sqrt{5} \cdot \frac{2}{\sqrt{5}} = 2R_D \leftarrow$$



$$\sum F_y \uparrow = 0 \quad F_{AB} - F_{AD} \sin \theta = 0$$

$$F_{AB} - 0 = 0$$

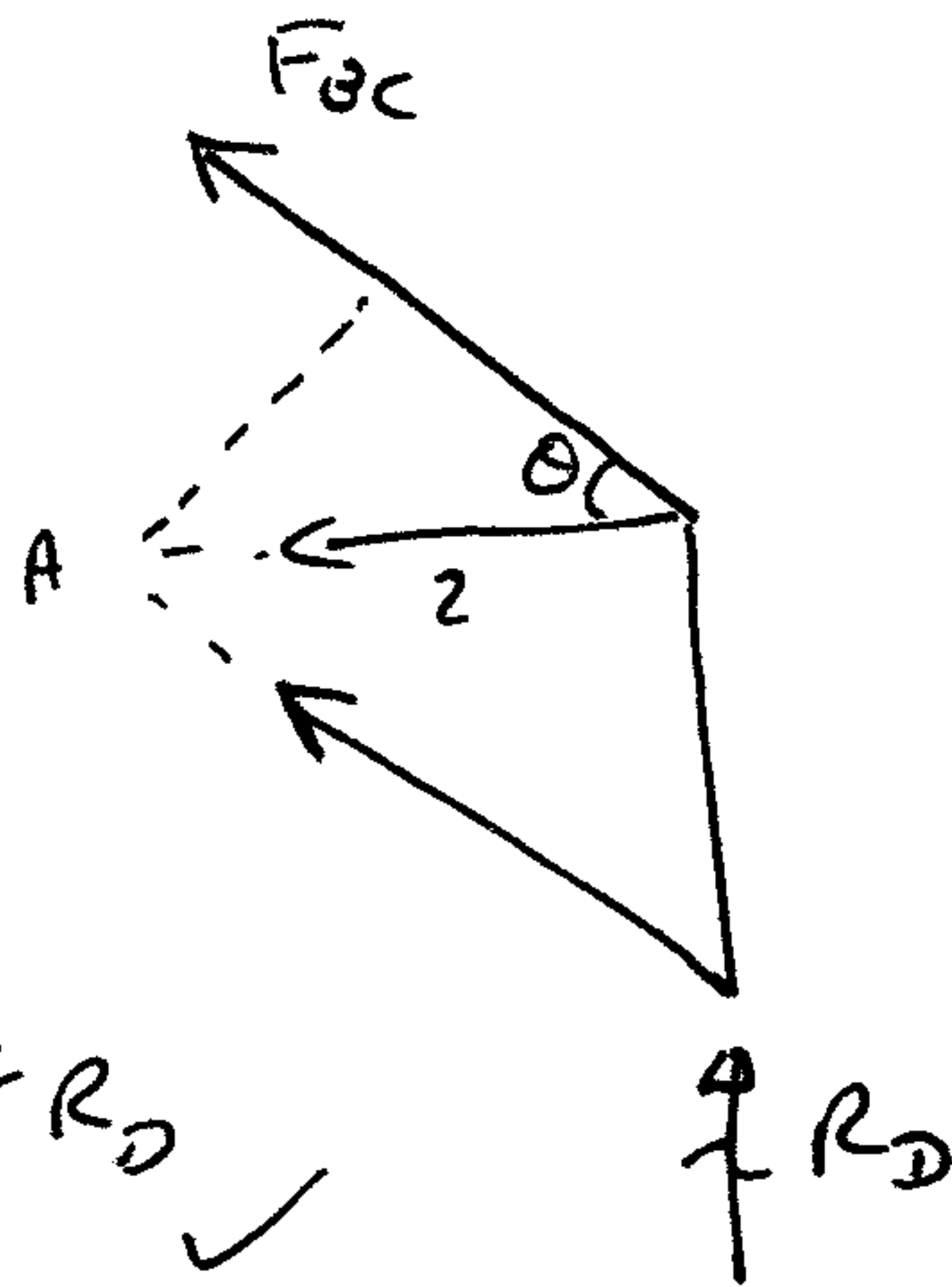
$$F_{AB} = 0$$



check MoS

$$\sin \theta = \frac{1}{\sqrt{5}}$$

$$\cos \theta = \frac{2}{\sqrt{5}}$$



$$\sum (M_A) = 0$$

$$R_D \cdot 2 + F_{BC} \cdot 2 \sin \theta = 0$$

$$F_{BC} = -\frac{2 R_D \sqrt{5}}{2} = -\sqrt{5} R_D \quad \checkmark$$

check

Bar	Length	Force (R_D)	$\delta / R_D / AE$
AB	1	0	0
AC	2	2	4
BC	$\sqrt{5}$	$-\sqrt{5}$	-5
AD	$\sqrt{5}$	0	0
CD	1	-1	-1

Draw displacement diagram

D displaces upward $\Delta D_y = \frac{18.5 R_D}{AE}$

Since D is on a roller $\delta^{(M9)} + \delta^{(M10)} = 0$

$$\frac{92.5 \times 10^3}{AE} + \frac{18.5 R_D}{AE} = 0 \quad R_D = -\frac{92.5 \times 10^3}{18.5} = -5 \text{ kN} \in$$

Horizontal deflection:

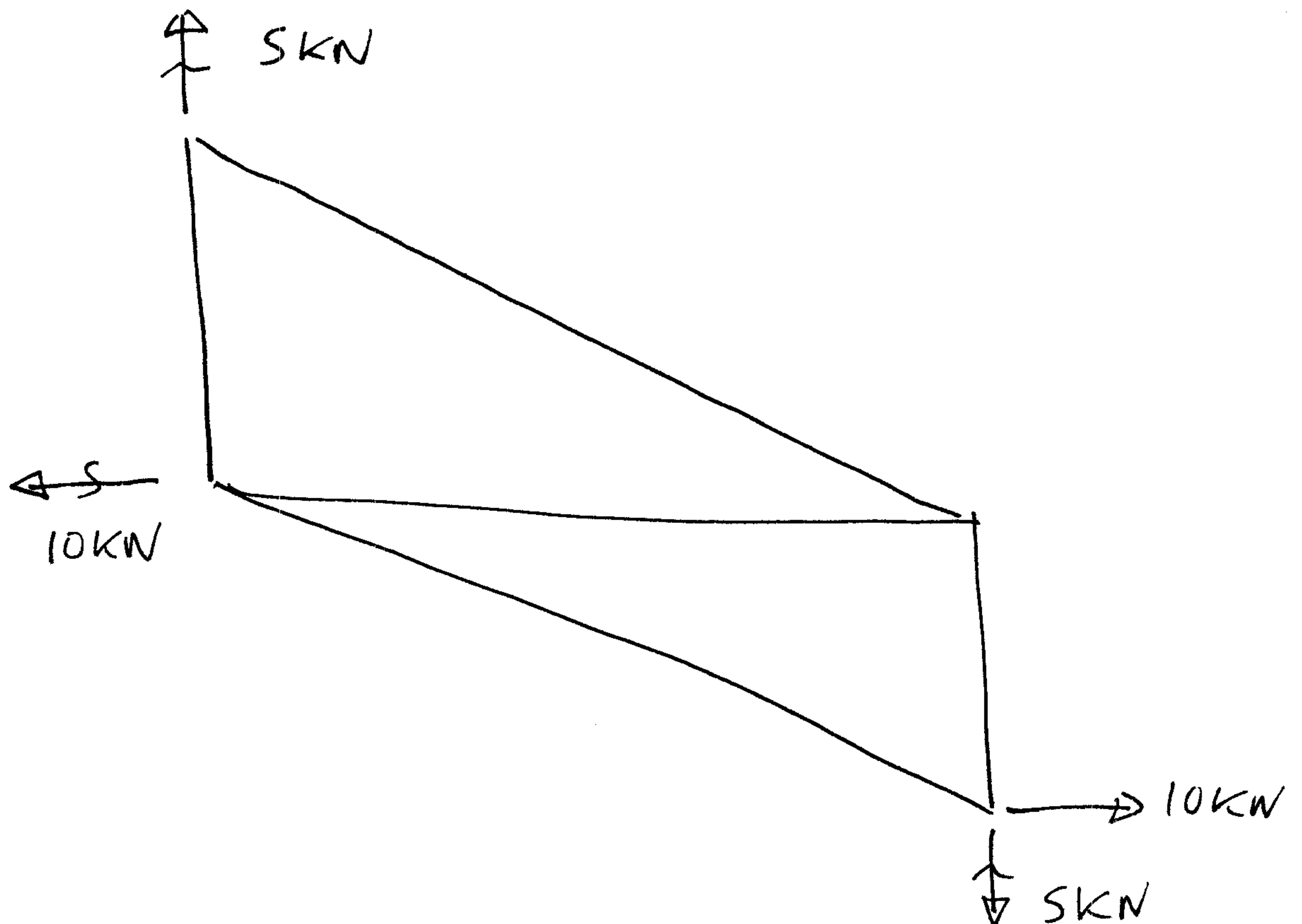
$$\Delta D_x^{M9} + \Delta D_x^{M10} = \frac{75 \times 10^3}{AE} + \frac{(9.5 \times -5) \times 10^3}{AE} = 786 \times 10^{-6} \\ = 0.79 \text{ mm} \Leftarrow$$

Reactions

$$H_B = H_B^{M9} + H_B^{M10} = +10 + 2(-5) = 0 \Leftarrow$$

$$H_A = H_A^{M9} + H_A^{M10} = -20 + 2(-2(-5)) = -10 \text{ kN} \Leftarrow$$

$$V_B = V_B^{M9} + V_B^{M10} = 0 - (-5) = +5 \text{ kN} \Leftarrow$$



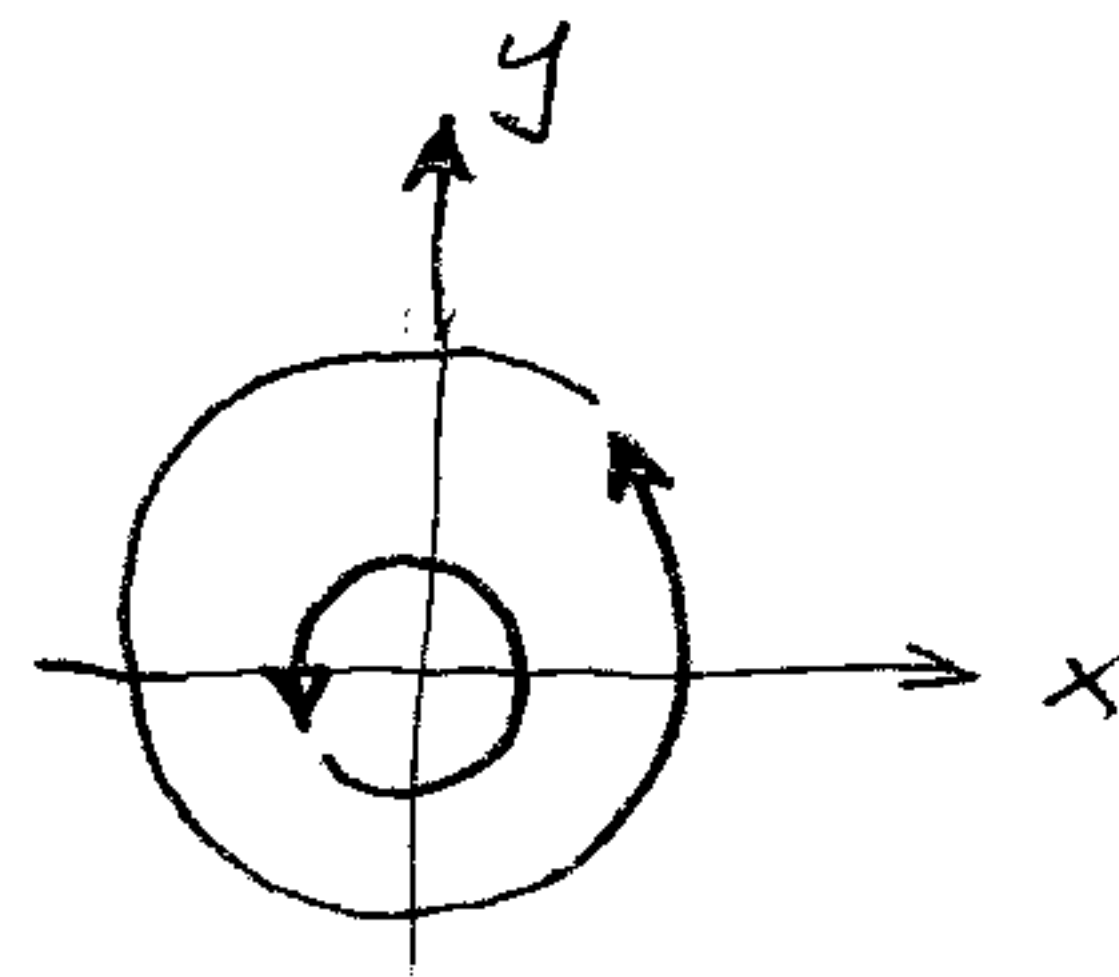
$$\frac{dy}{dx} = \frac{v}{u} = -\frac{x}{y}$$

$$y \, dy = -x \, dx$$

$$\frac{1}{2} y^2 = -\frac{1}{2} x^2 + C$$

$$x^2 + y^2 = 2C$$

circles of radius $\sqrt{2C}$



For steady flow, with $\rho = \text{const}$, must have $\nabla \cdot \vec{U} = \frac{\partial u}{\partial x} + \frac{\partial v}{\partial y} = 0$

$$u = \frac{-y}{x^2 + y^2}$$

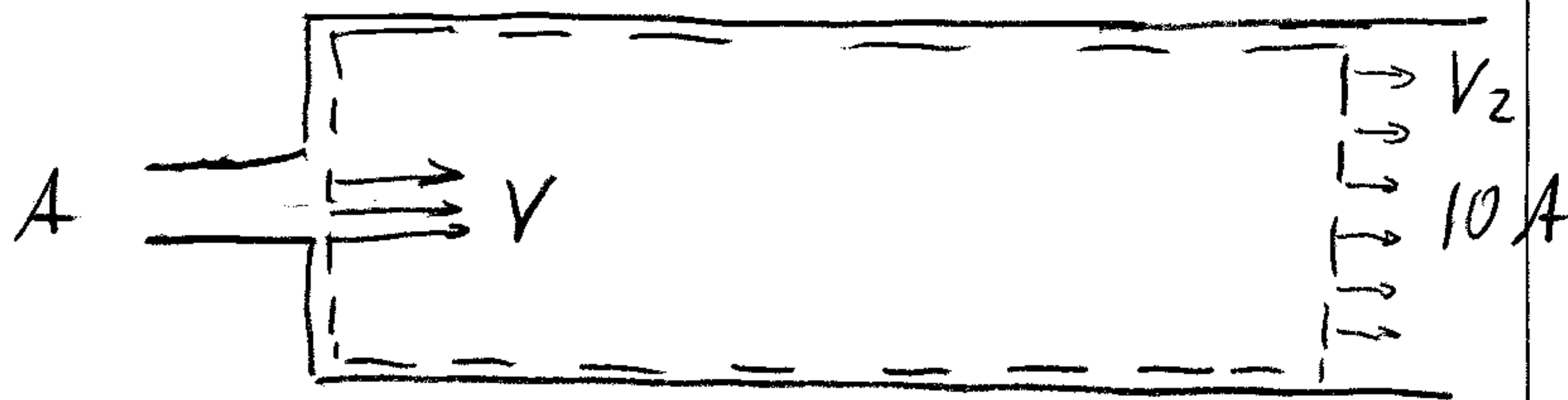
$$\frac{\partial u}{\partial x} = \frac{y \cdot 2x}{(x^2 + y^2)^2}$$

$$v = \frac{x}{x^2 + y^2}$$

$$\frac{\partial v}{\partial y} = \frac{-x \cdot 2y}{(x^2 + y^2)^2}$$

$$\left. \begin{array}{l} \frac{\partial u}{\partial x} \\ \frac{\partial v}{\partial y} \end{array} \right\} = 0 \quad \checkmark$$

Control Volume:



a) mass conservation

$$\oint \rho \vec{V} \cdot \hat{n} dA = -\rho V A + \rho V_2 (10A) = 0 \rightarrow \boxed{V_2 = \frac{1}{10} V}$$

b) momentum conservation

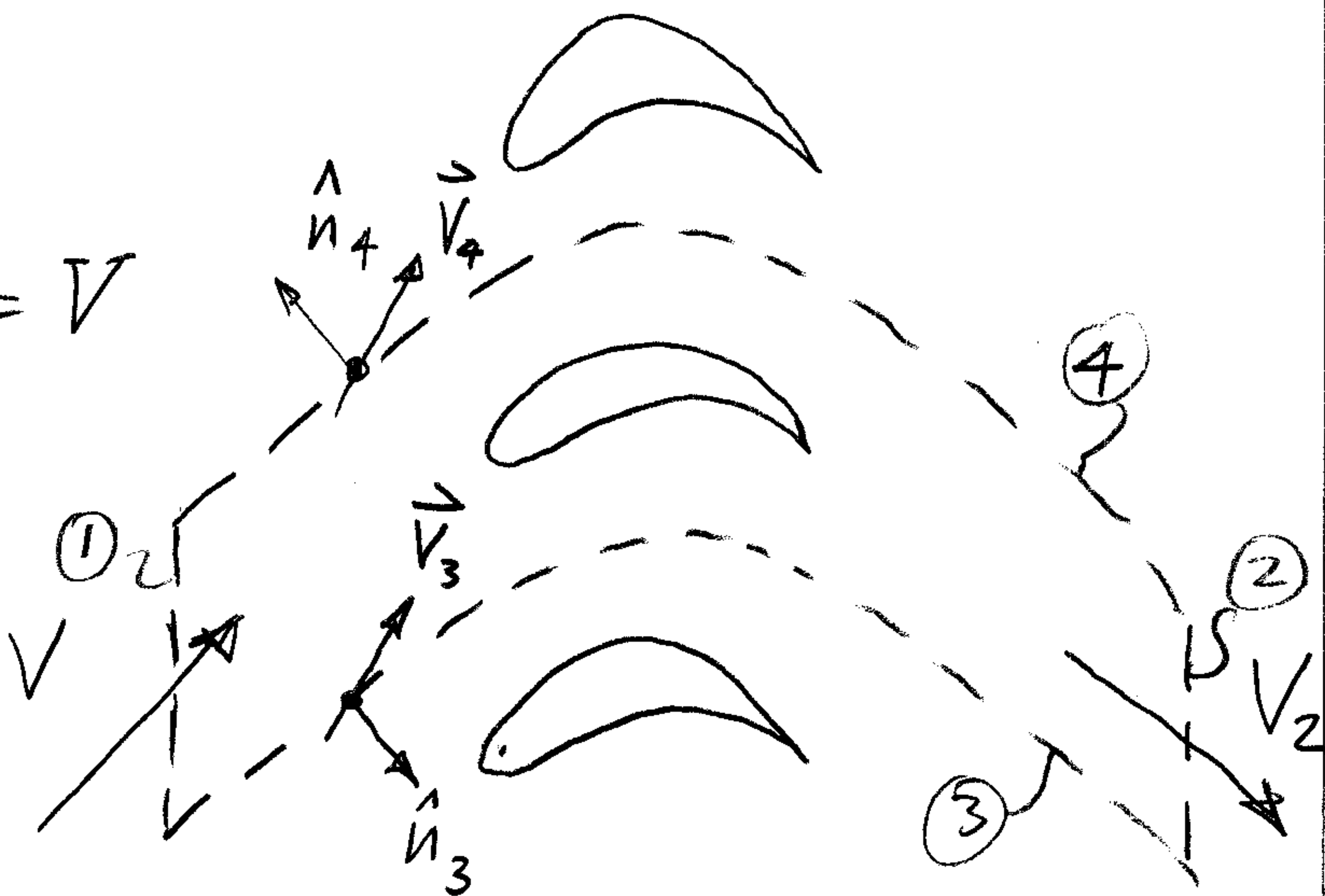
$$0 = \oint \left[p \hat{n} + \rho (\vec{V} \cdot \hat{n}) \vec{V} \right] dA = -p_1 \cdot 10A - \rho V^2 A + p_2 \cdot 10A + \rho V_2^2 10A$$

$$0 = (p_2 - p_1) 10A + \rho \left(-V^2 + \frac{1}{10} V^2 \right) A$$

$$\boxed{p_2 - p_1 = \frac{9}{100} \rho V^2 A}$$

Control Volume :

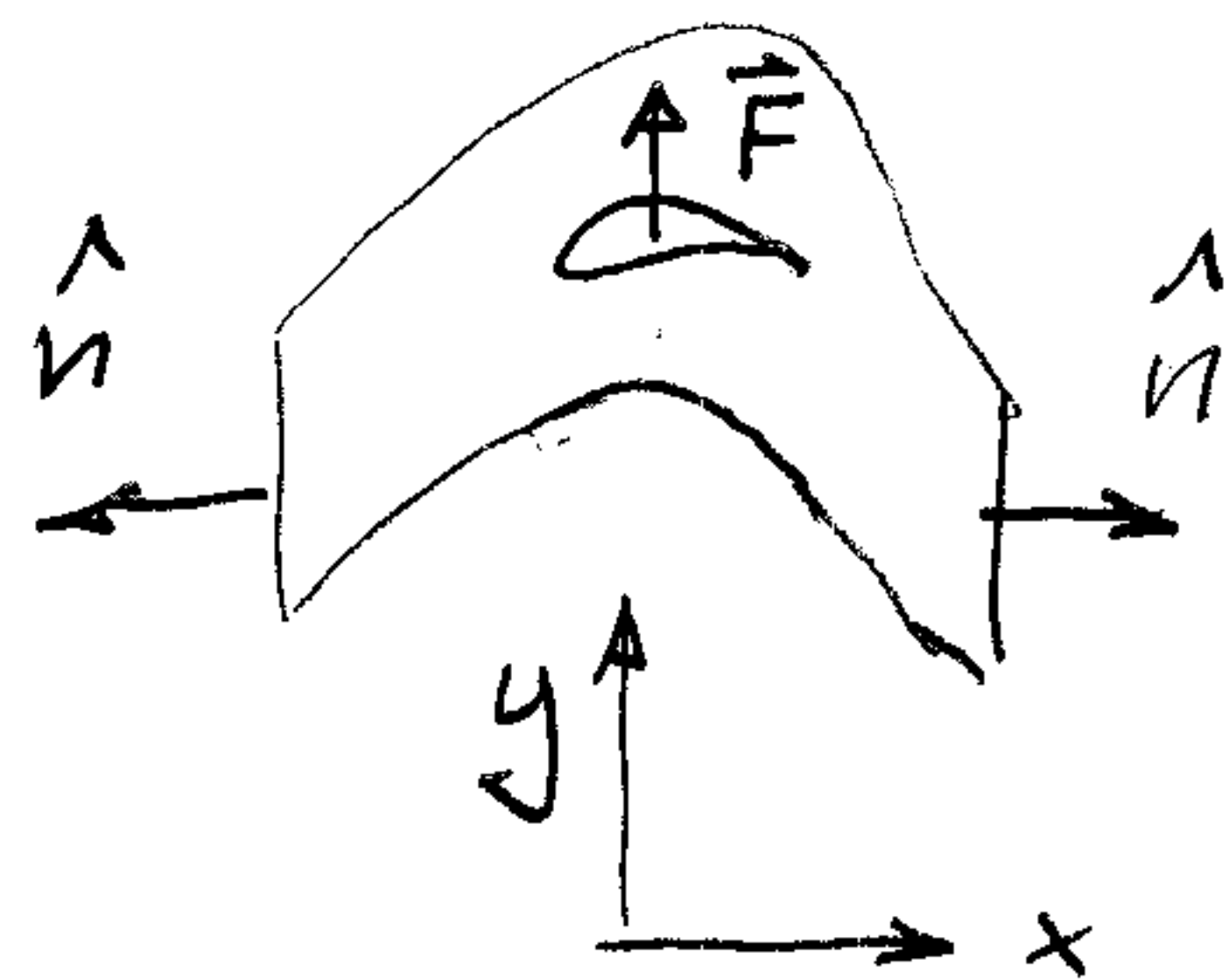
By mass conservation, $V_2 = V$



Because flow is periodic, $\vec{V}_3 = \vec{V}_4$, $p_3 = p_4$
 And since $\hat{n}_4 = -\hat{n}_3$, then sides ③ and ④ will cancel in momentum integral.

$$\oint_{①} (\rho \hat{n} + \rho \vec{V} \cdot \hat{n} \vec{V}) dA + \oint_{②} (\rho \hat{n} + \rho \vec{V} \cdot \hat{n} \vec{V}) dA = -\vec{F}$$

By symmetry, $\oint_{①} \rho \hat{n} dA + \oint_{②} \rho \hat{n} dA = 0$



$$\oint_{①} (\rho \vec{V} \cdot \hat{n} \vec{V}) dA + \oint_{②} \rho \vec{V} \cdot \hat{n} \vec{V} dA = -\vec{F}$$

$$-\rho V \frac{\sqrt{2}}{2} \cdot V \begin{bmatrix} \sqrt{2}/2 \\ \sqrt{2}/2 \end{bmatrix} h + \rho V \frac{\sqrt{2}}{2} \cdot V \begin{bmatrix} -\sqrt{2}/2 \\ -\sqrt{2}/2 \end{bmatrix} = -\vec{F}$$

$\vec{F} = \begin{bmatrix} 0 \\ \rho V^2 h \end{bmatrix}$	Vertical Force
---	----------------

13-782 500 SHEETS FILLER 3 SQUARE
 42-381 50 SHEETS EYE-EASE 3 SQUARE
 42-382 100 SHEETS EYE-EASE 6 SQUARE
 42-389 200 SHEETS EYE-EASE 6 SQUARE
 42-392 100 RECYCLED WHITE 5 SQUARE
 42-398 200 RECYCLED WHITE 5 SQUARE
 Made in U.S.A.

