

Which of the following is NOT an example of Aerodynamics in action?

1. Thrust of a rocket
2. Curve of a baseball
3. Drag of a car
- 4.* Buoy floating down a river
5. Sailboat underway
6. Lift on an aircraft

Or ...

7. All of the above involve Aerodynamics

There is a thin boundary layer on a flat wall. The pressure and speed just above the boundary layer are p_∞ and V_∞ . What is the pressure p at the wall?

1. $p = p_\infty + \frac{1}{2}\rho V_\infty^2$

2.* $p = p_\infty$

3. There's no way to know for sure.

An airfoil at a certain angle of attack has $M' = -10$, $L' = 0$. This is an example of ...

- 1.* This is a pure aerodynamic moment (couple)
2. An impossible situation

The PS02 airfoil with 1m chord has:

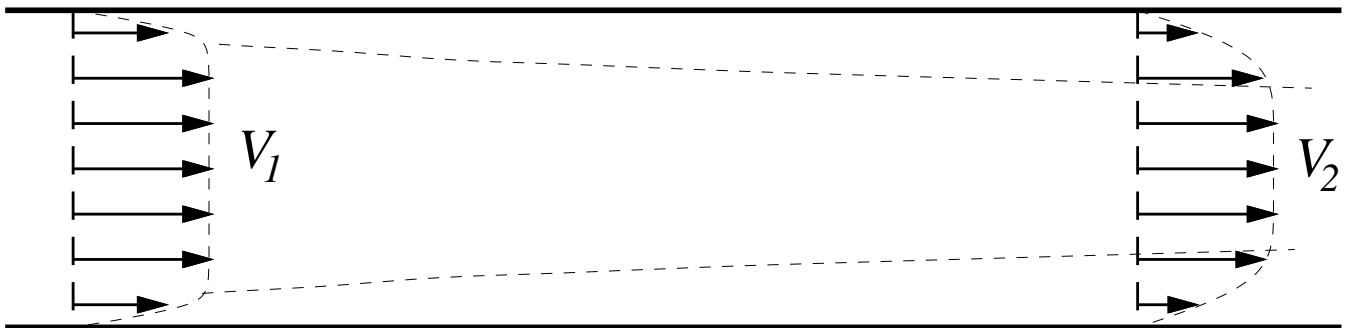
$$\text{Re} = 10^6 \quad c_d = 0.006$$

What is the diameter of a round cylinder with nearly the same drag/span D' in the same flow?

1. 1 mm
- 2.* 5 mm
3. 20 mm
4. 100 mm
5. None of the above

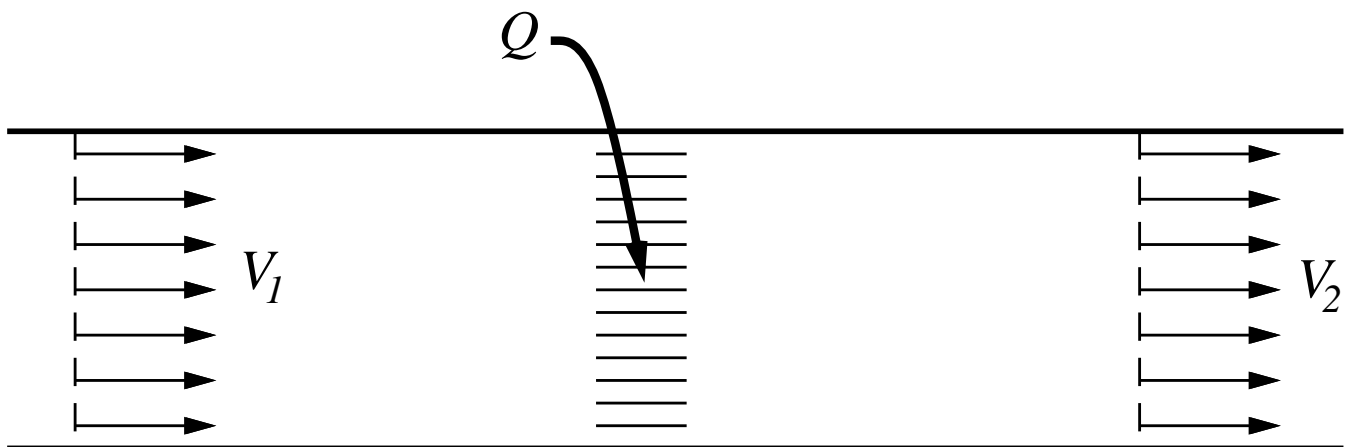
Boundary layers grow on the walls of the low-speed constant-area channel. How do the centerline velocities V_1 and V_2 compare?

- 1.* $V_1 < V_2$
2. $V_1 = V_2$
3. $V_1 > V_2$
4. No way to tell for sure



A heater is placed in a slow-flowing channel. How do the two velocities V_1 and V_2 compare?

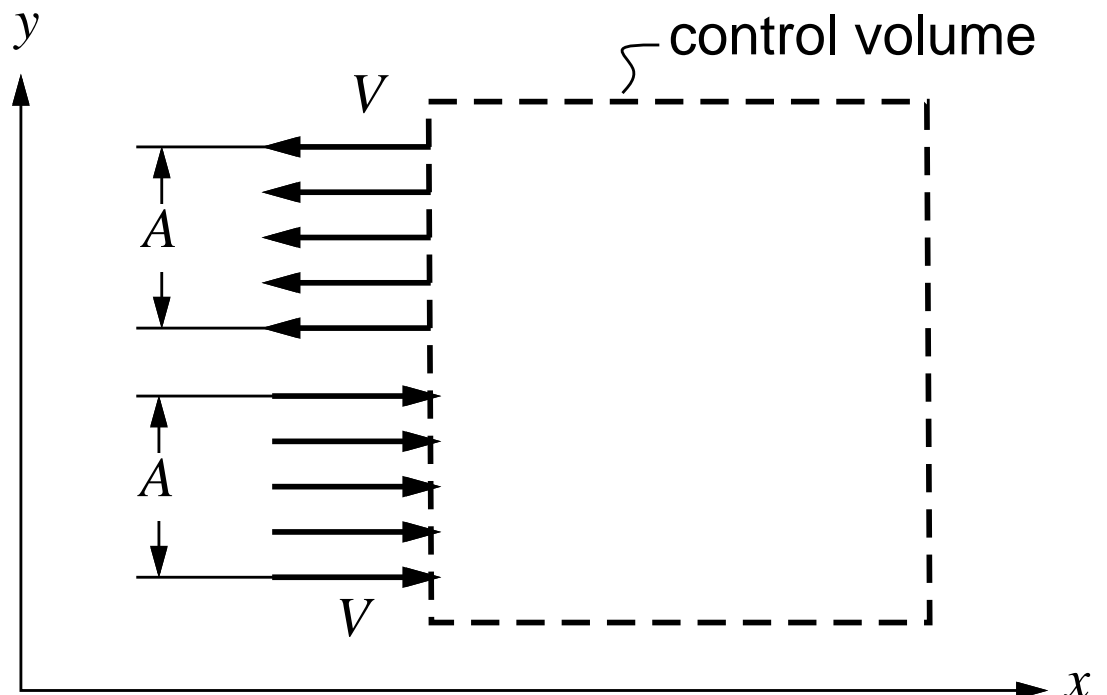
- 1.* $V_1 < V_2$
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Two fluid jets of the same density ρ flow as shown. What is the mass flow integral for the control volume?

$$\oint \rho \vec{V} \cdot \hat{n} \, dA$$

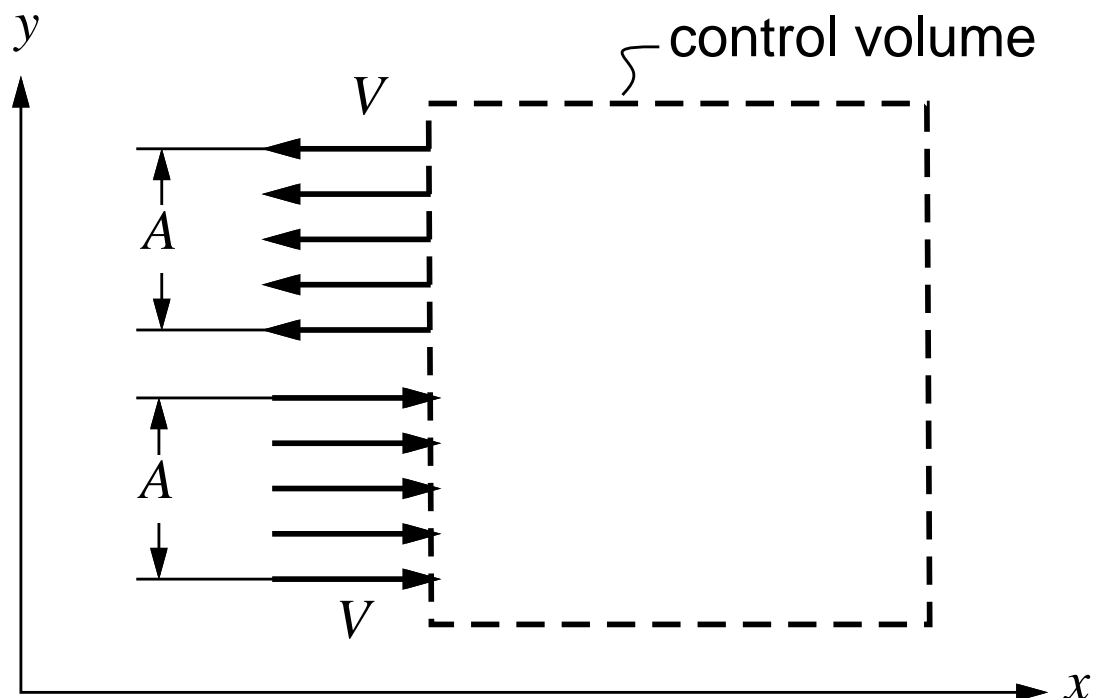
1. $2\rho VA$
2. $2\rho VA \hat{i}$
3. $\rho VA/2$
- 4.* 0



Two fluid jets of the same density ρ flow as shown. What is the momentum flow integral for the control volume?

$$\oint \rho (\vec{V} \cdot \hat{n}) \vec{V} \, dA$$

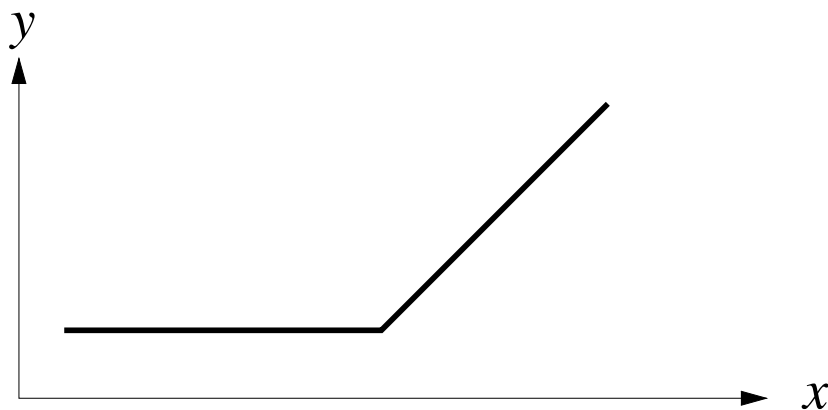
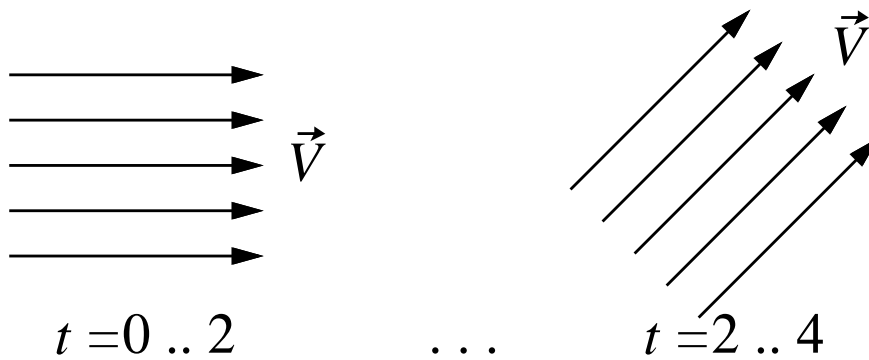
1. $\rho V^2 A \hat{i}$
2. $2\rho V^2 A \hat{i}$
- 3.* $-2\rho V^2 A \hat{i}$
4. 0



A uniform velocity field points in the x direction for times $t = 0 \dots 2$, and 45° up from the x direction for times $t = 2 \dots 4$.

What is the curve in the x - y plot?

1. Streamline
- 2.* Pathline
3. Streakline



The force on the body \vec{R} is computed in two ways:

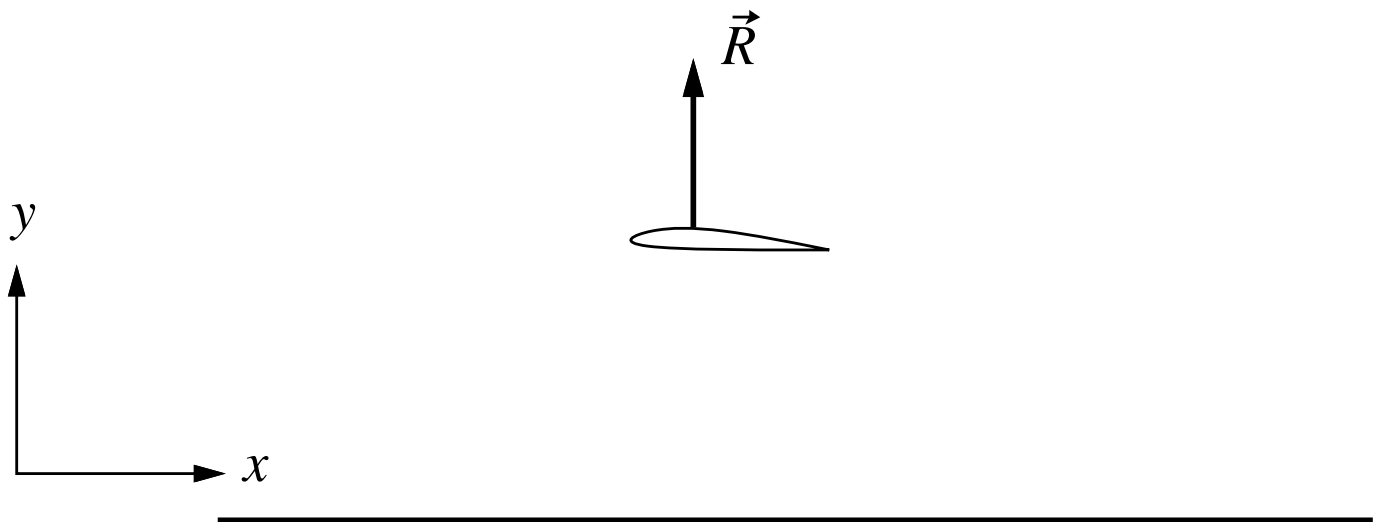
- a) Using the true pressure p
- b) Using the corrected pressure p_c

What can you say about the difference $\vec{R}_p - \vec{R}_{p_c}$?

1. It's zero.
- 2.* Depends only on the body volume.
3. Depends on the body shape in a complicated way.

A wing with lift force $\vec{R} = L\hat{j}$ flies overhead. What is the resulting force applied to the ground?

1. 0
2. $L\hat{j}$
- 3.* $-L\hat{j}$



Traffic leaves a toll gate located at $x = 0$. At some location x , every car's speed is $u(x)$. What is a car's acceleration at location x ?

1. u^2/x
2. du/dt
- 3.* $u \, du/dx$
4. Cannot be determined from the given information

