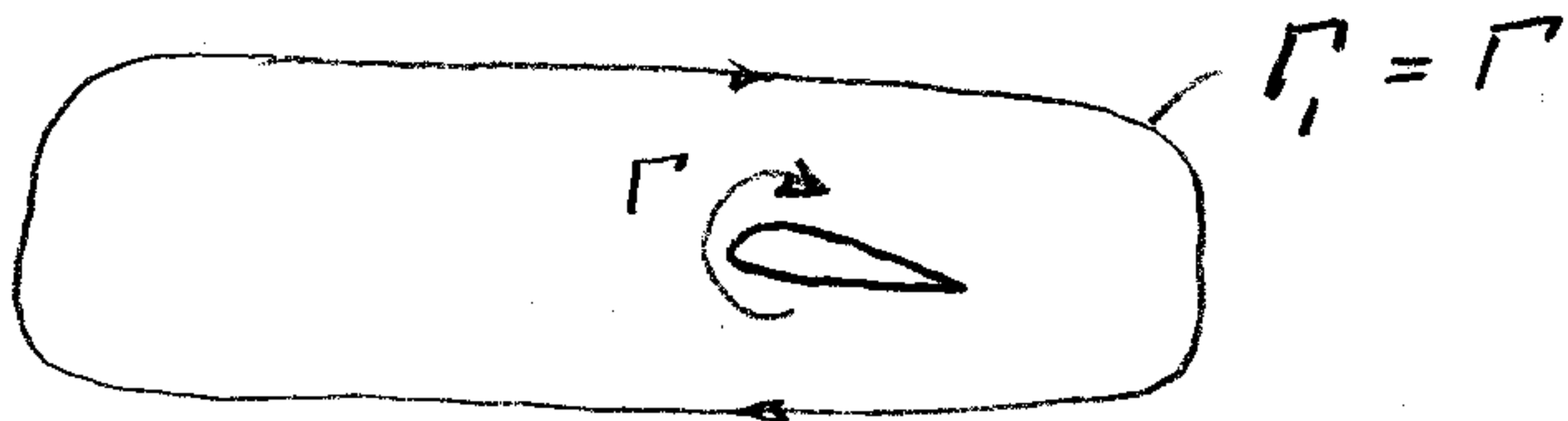
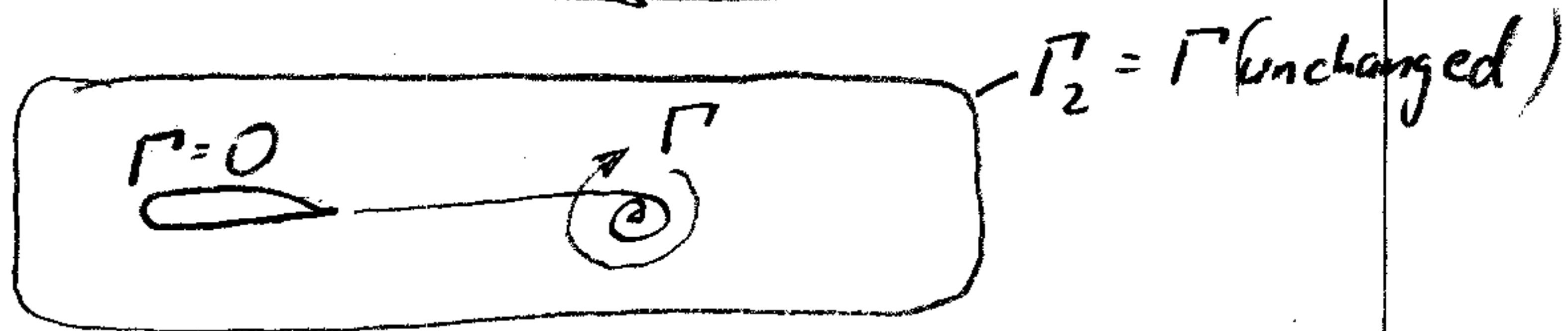


F1a. Before touchdown:



After touchdown:



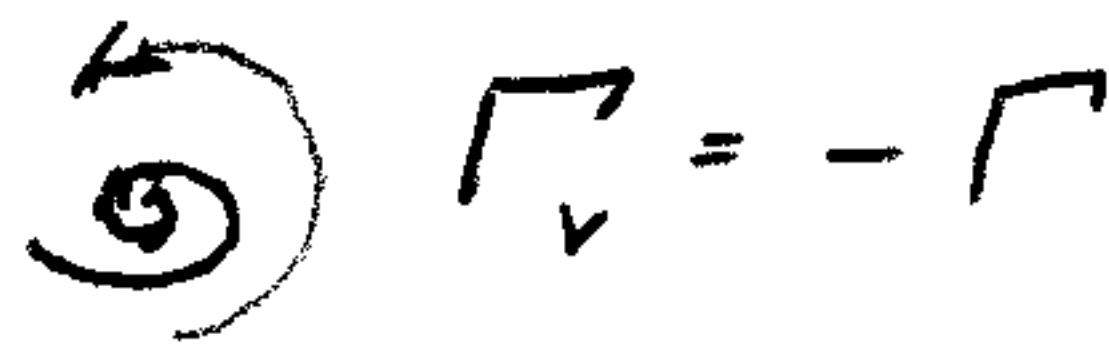
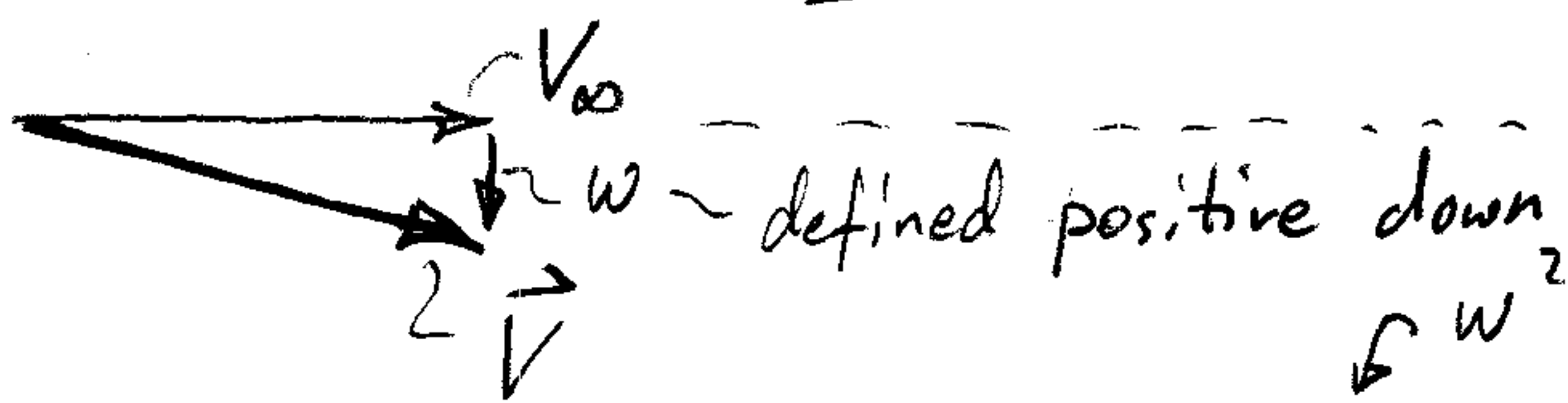
Airfoil sheds a vortex which contains all the airfoil's initial circulation.
Airfoil is left with zero circulation.

F1b. Since initial circulation is zero,

must have $\Gamma_v = -\Gamma$



Velocity seen by airfoil



$$w = \frac{\Gamma}{2\pi d}, \quad |\vec{V}|^2 = V_\infty^2 + \left(\frac{\Gamma}{2\pi d}\right)^2 \approx V_\infty^2 \text{ if } w \ll V_\infty$$

Net lift force/span is perpendicular to apparent velocity.

$$F' = \rho |\vec{V}| \Gamma \approx \rho V_\infty \Gamma$$



Take components \perp and \parallel to \vec{V}_∞

$$L' = F' \frac{V_\infty}{|\vec{V}|} \approx F' = \rho V_\infty \Gamma \rightarrow C_l = \frac{L'}{\frac{1}{2} \rho V_\infty^2 c} = \frac{2\Gamma}{c V_\infty}$$

$$D' = F' \frac{w}{|\vec{V}|} \approx F' \frac{w}{V_\infty} = \rho w \Gamma \rightarrow C_d = \frac{D'}{\frac{1}{2} \rho V_\infty^2 c} = \frac{2\Gamma w}{c V_\infty V_\infty} = C_l \frac{w}{V_\infty}$$

since $w \sim \frac{1}{d} \sim \frac{1}{\text{time}}$

C_d decreases as $\frac{1}{\text{time}}$

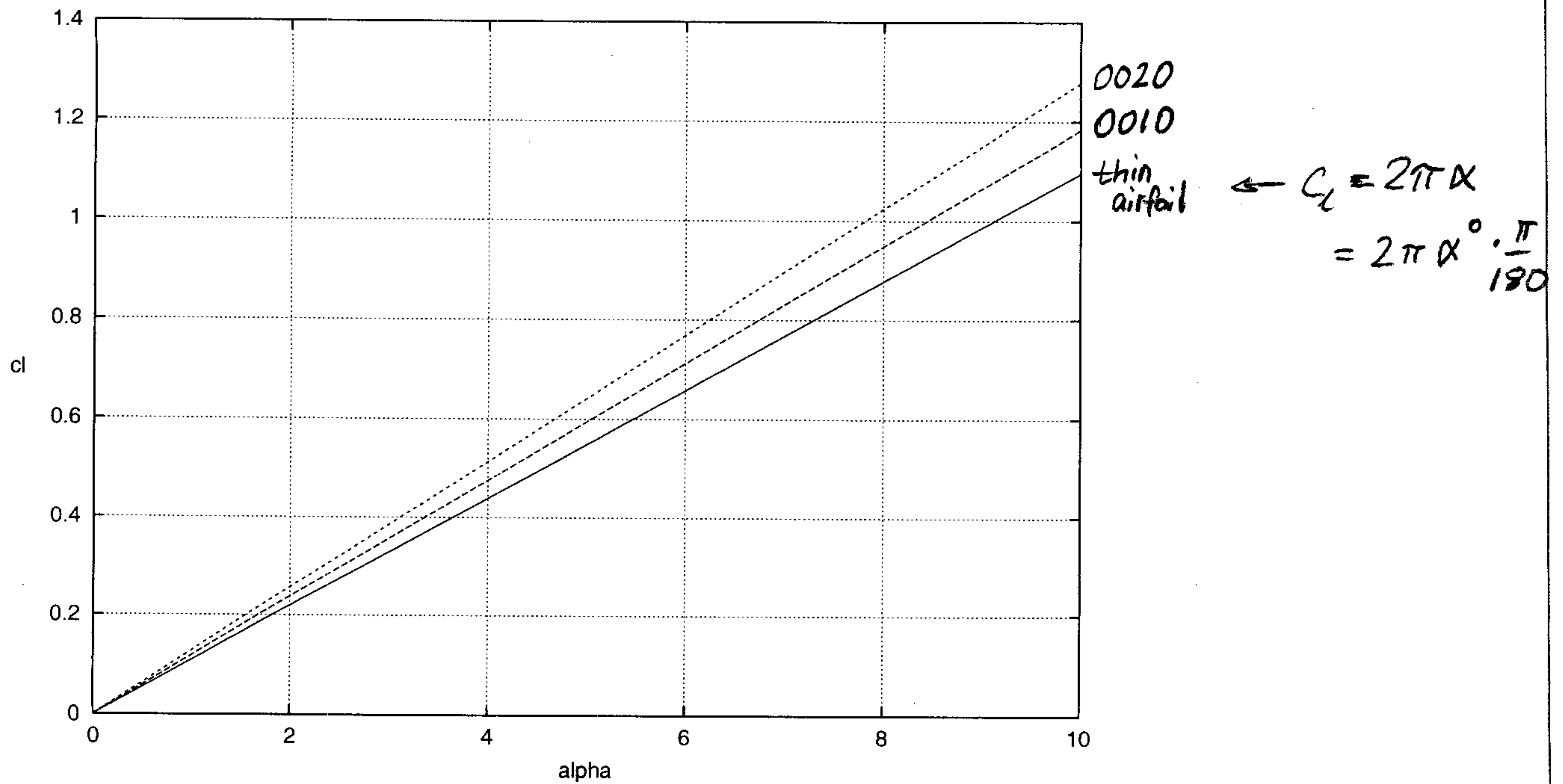
$$\text{or } C_d = \frac{2\Gamma}{c V_\infty} \cdot \frac{\Gamma}{2\pi d V_\infty} = \frac{1}{4\pi} \frac{c}{d} C_l^2$$



13-782
42-381
42-382
42-383
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42-393
42-394
42-395
42-396
42-397
42-398
42-399
42-400
500 SHEETS, FILLER, 5 SQUARE
50 SHEETS EYE-EASE, 5 SQUARE
100 SHEETS EYE-EASE, 5 SQUARE
200 SHEETS EYE-EASE, 5 SQUARE
100 RECYCLED WHITE, 5 SQUARE
200 RECYCLED WHITE, 5 SQUARE
Made in U.S.A.

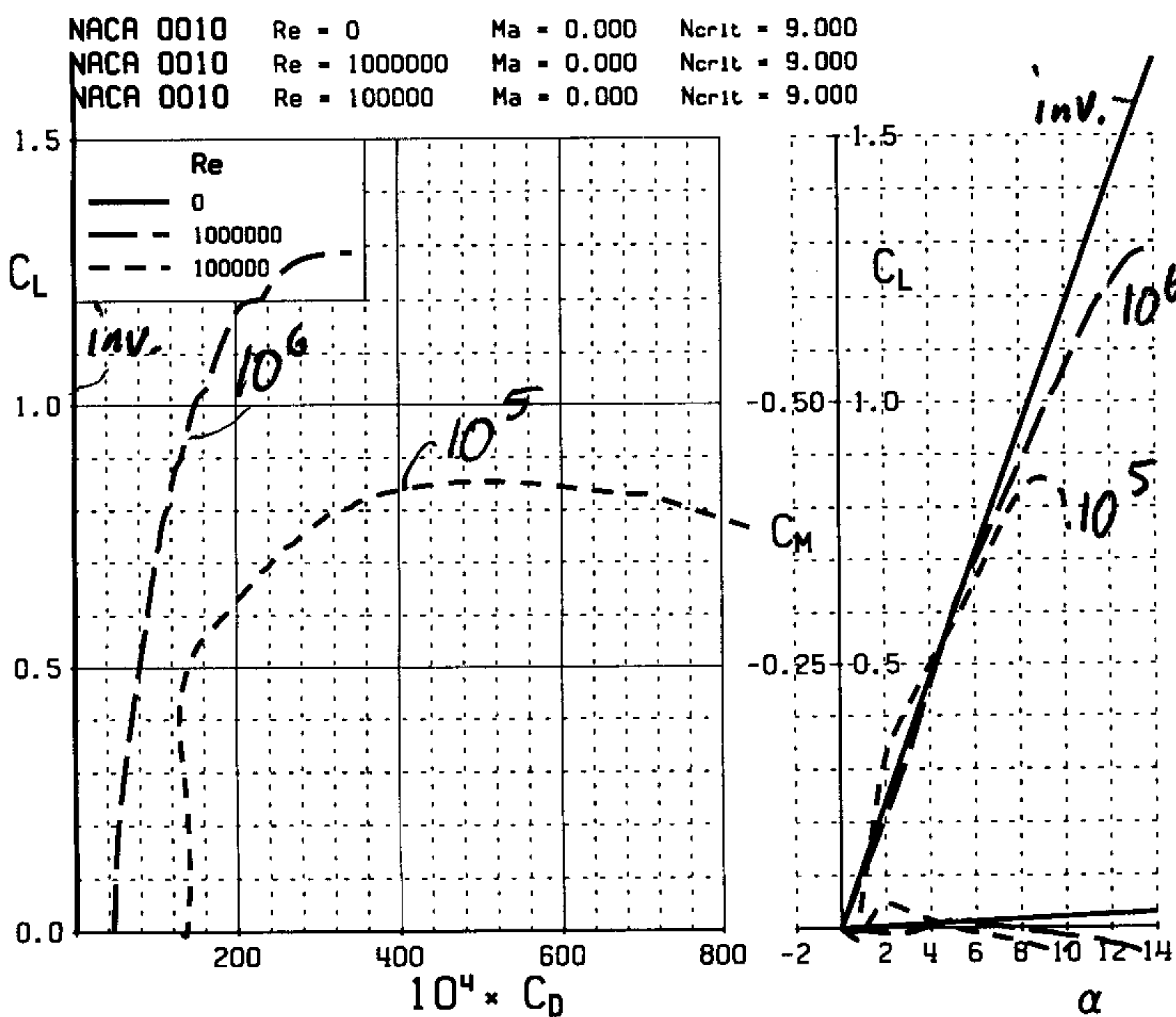


F2a.



As airfoil thickness goes to zero, the panel results approach the result from thin airfoil theory (assumptions become more valid).

F2b.



Viscous results approach inviscid results as Re increases (as viscosity gets smaller)

Unified problems M1-M3

Solutions

M1 ... Minimum mass structure with a strength requirement, need to maximize σ_f / ρ

... compute σ_f / ρ for available materials

	σ_f / MPa	$\rho / \text{kg/m}^3$	$\text{MPa} / \text{kg/m}^3$
... Steel	220	7900	27.8
... Al	380	2800	125
... Ti	850	4500	188
... CFRP	700	1500	467
... Wood	30	600	50
... SiC	300	3000	100

... CFRP works best for bars in truss.

... Now decide on truss configuration

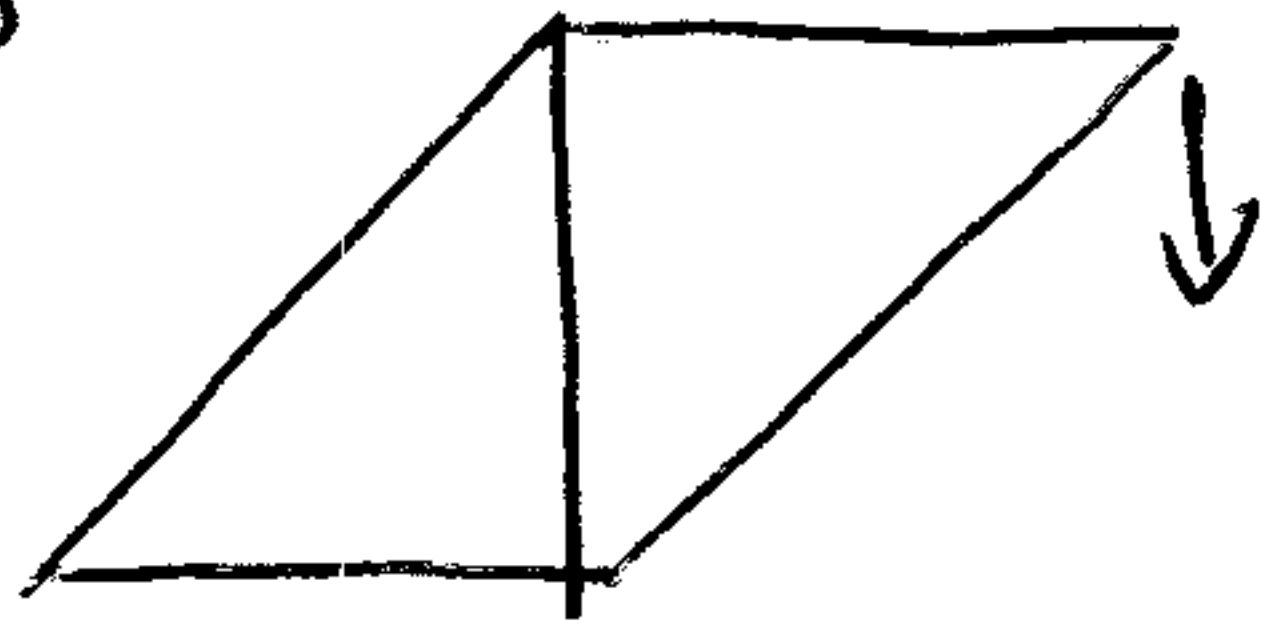
... Design considerations:

... Minimize number of bars (simplicity is good).

... Aim to have all bars carrying similar loads

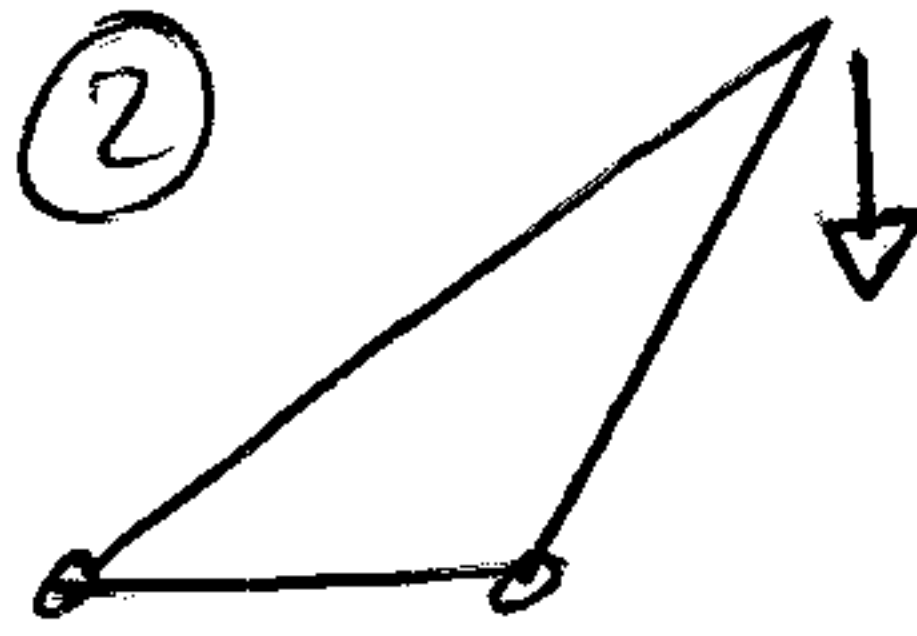
Candidates

①

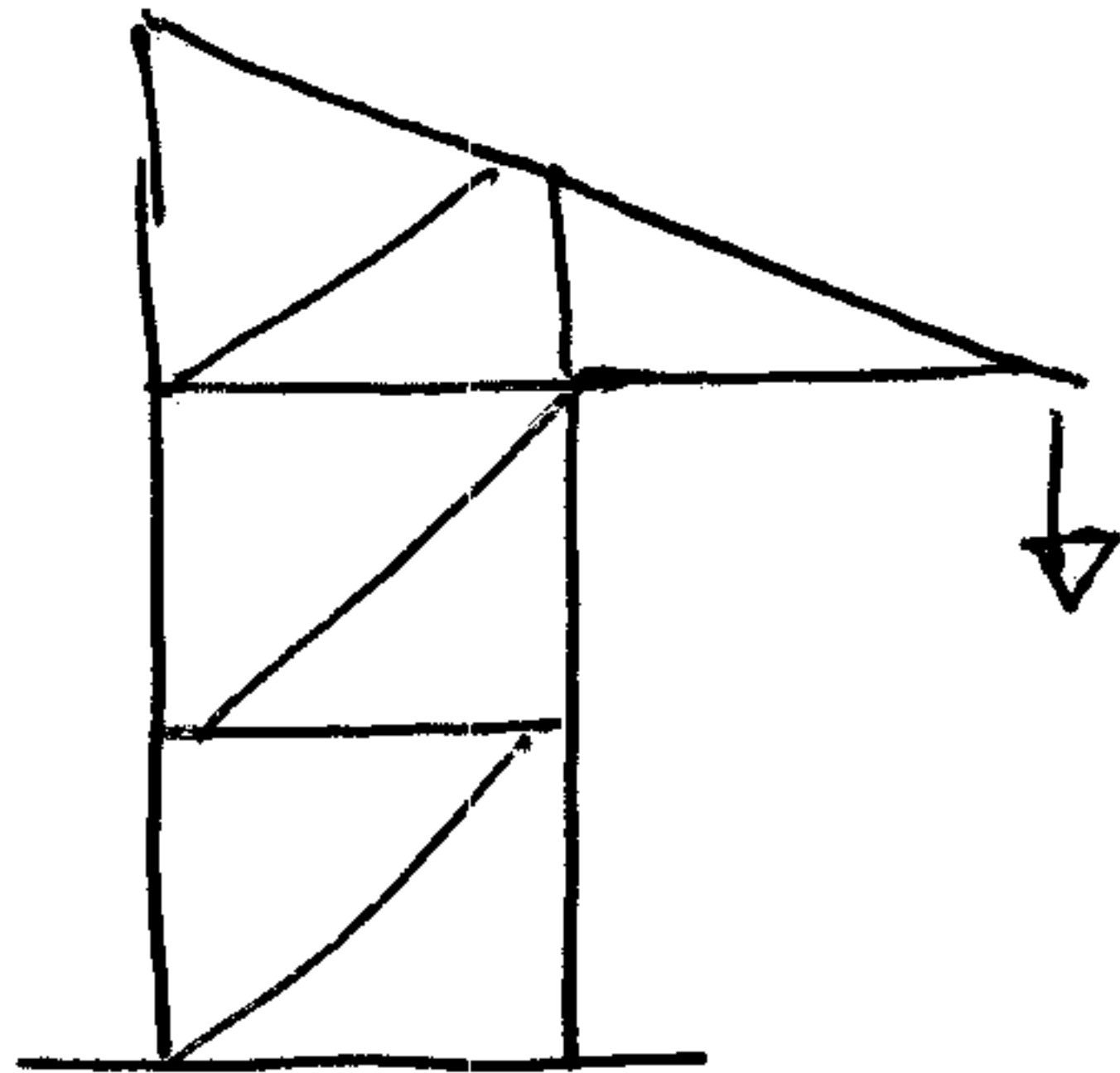


or

②

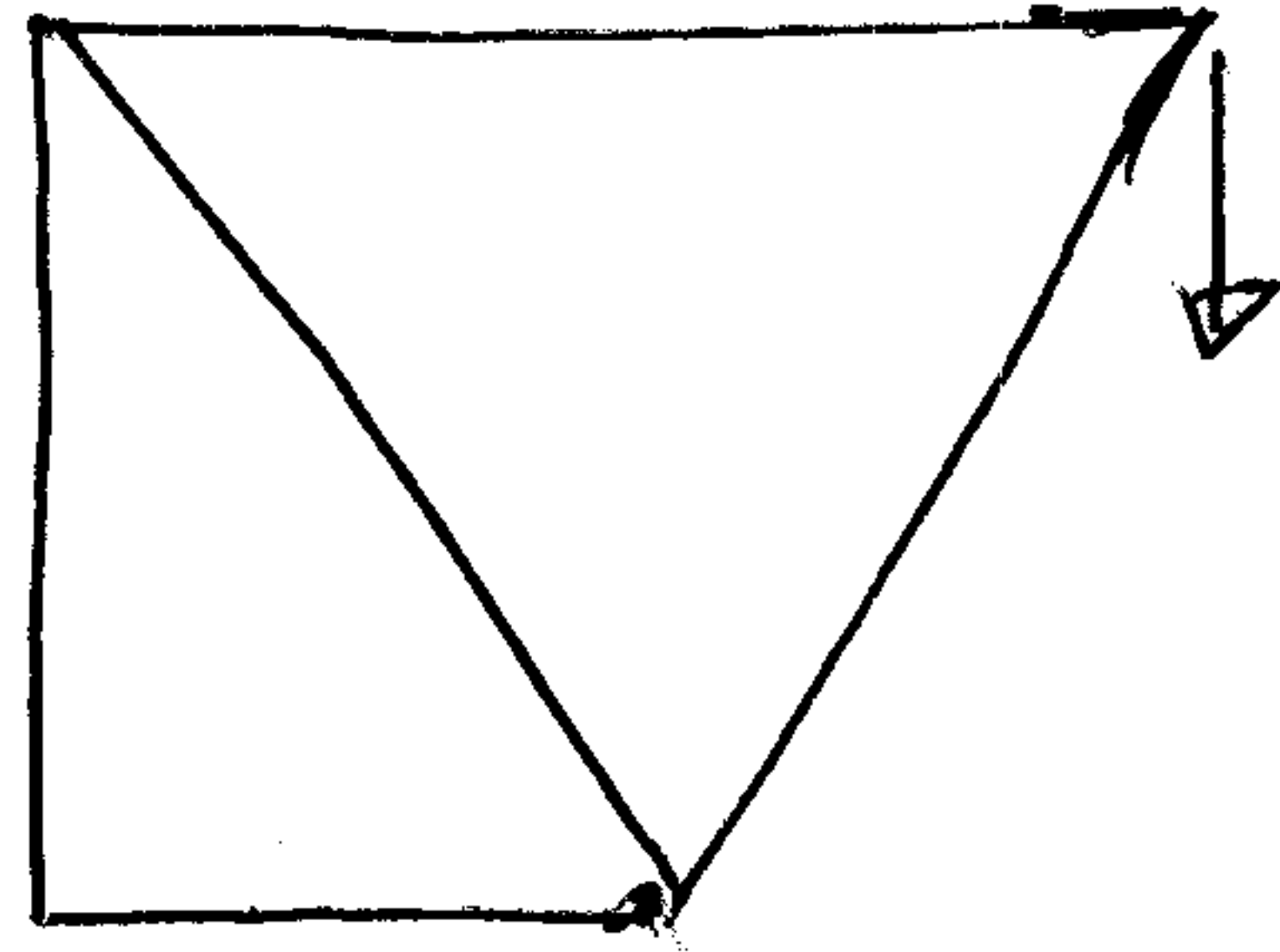


or



③

or

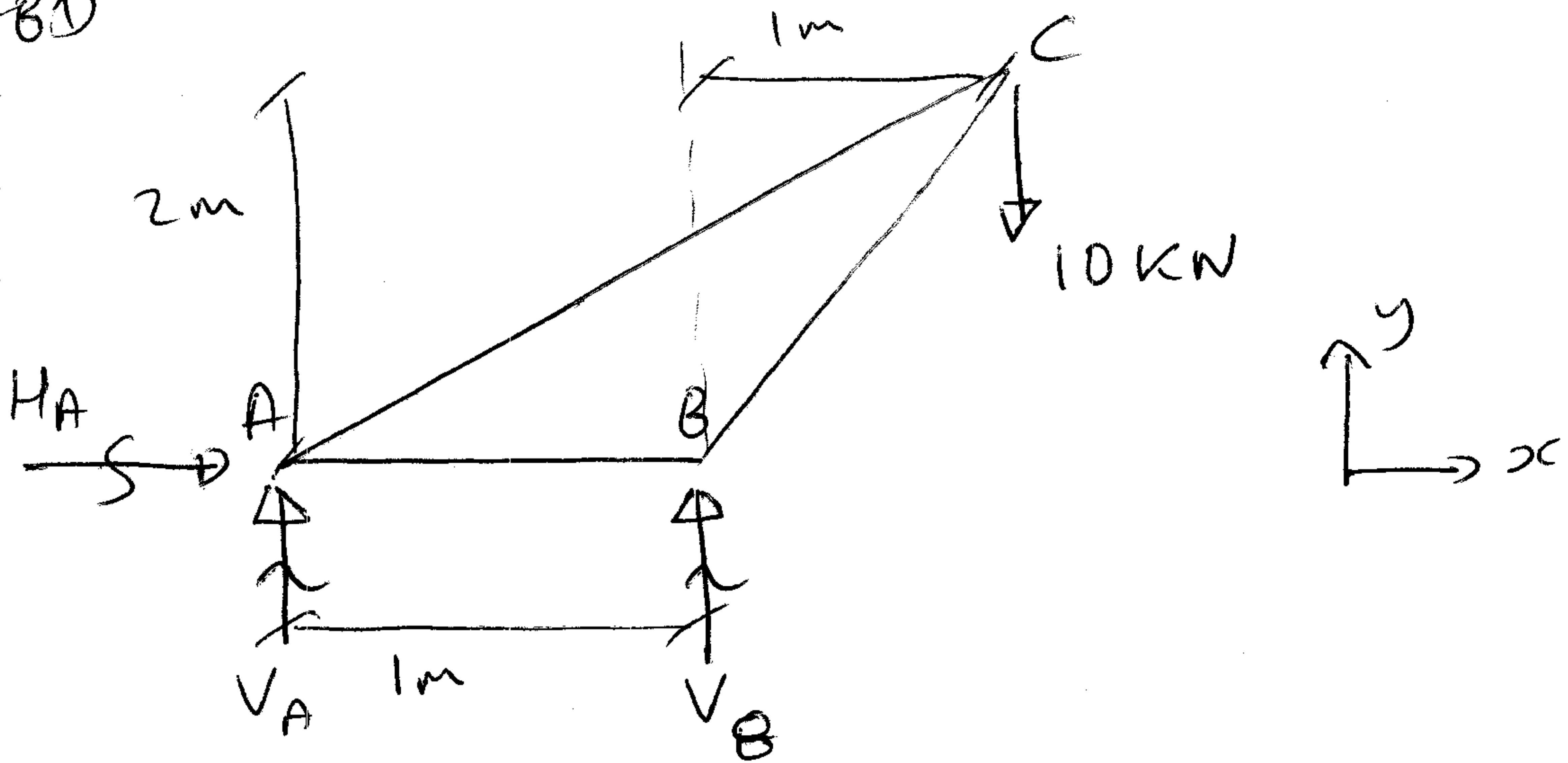


② looks simplest - smallest number of bars
∴ probably lowest mass

suspect ① may have more bars at some
others - ∴ more efficient

In any case go with ②

FBD



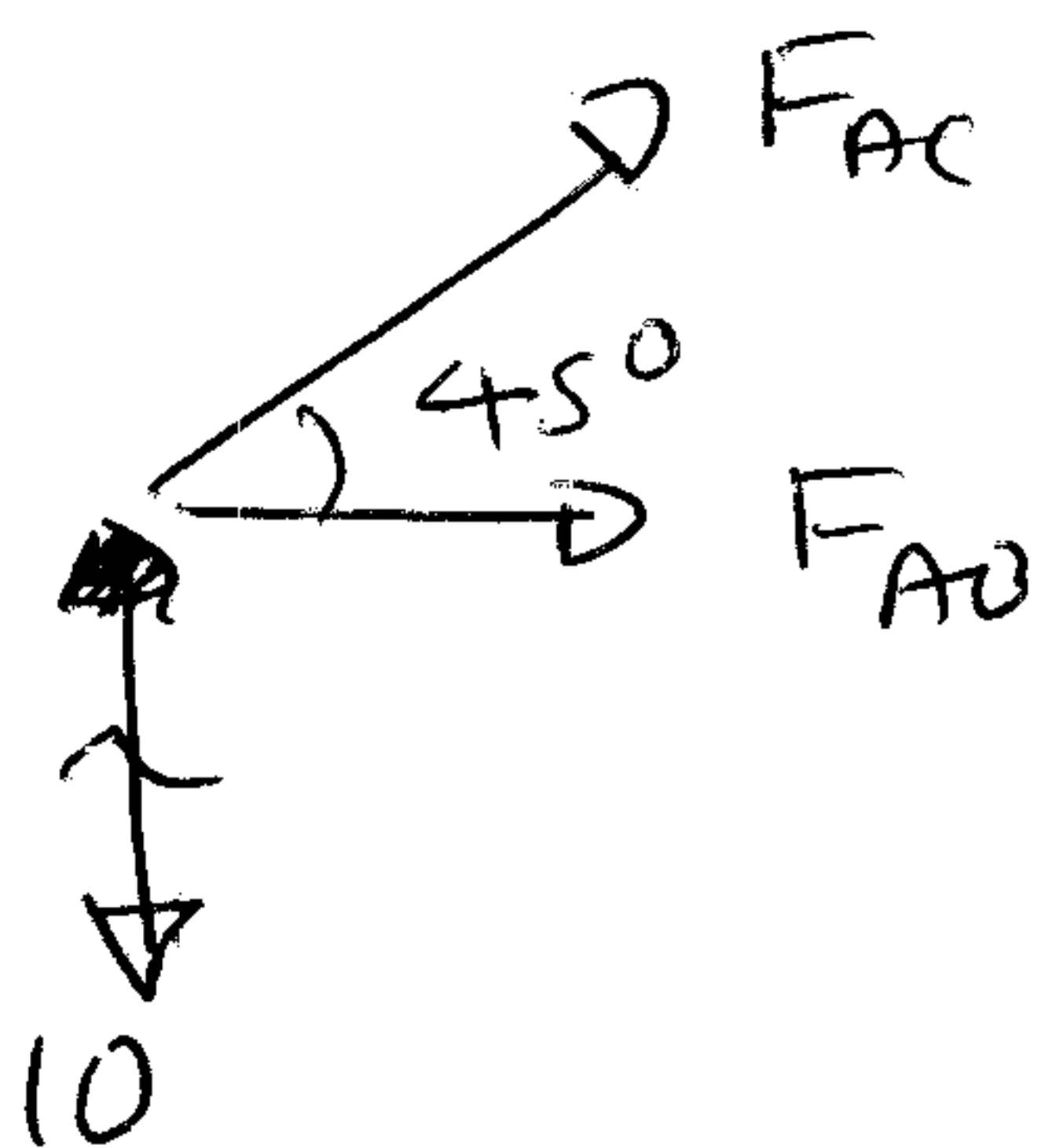
$$\rightarrow \sum F_x = 0 : H_A + 0 = 0 \Rightarrow H_A = 0$$

$$\uparrow \sum F_y = 0 : V_A + V_B - 10 = 0$$

$$\sum (M)_A = 0 : 1 \cdot V_B - 10 \cdot 2 = 0 : V_B = +20 \text{ kN}$$

$$\therefore V_A = -10 \text{ kN} \Leftarrow$$

M.O.J @ A



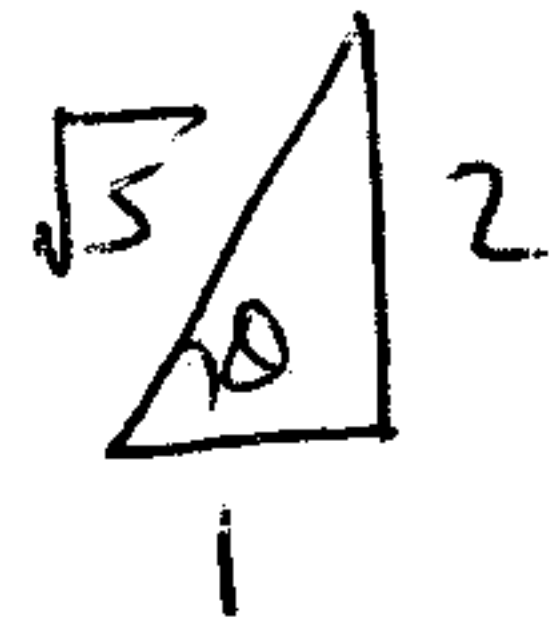
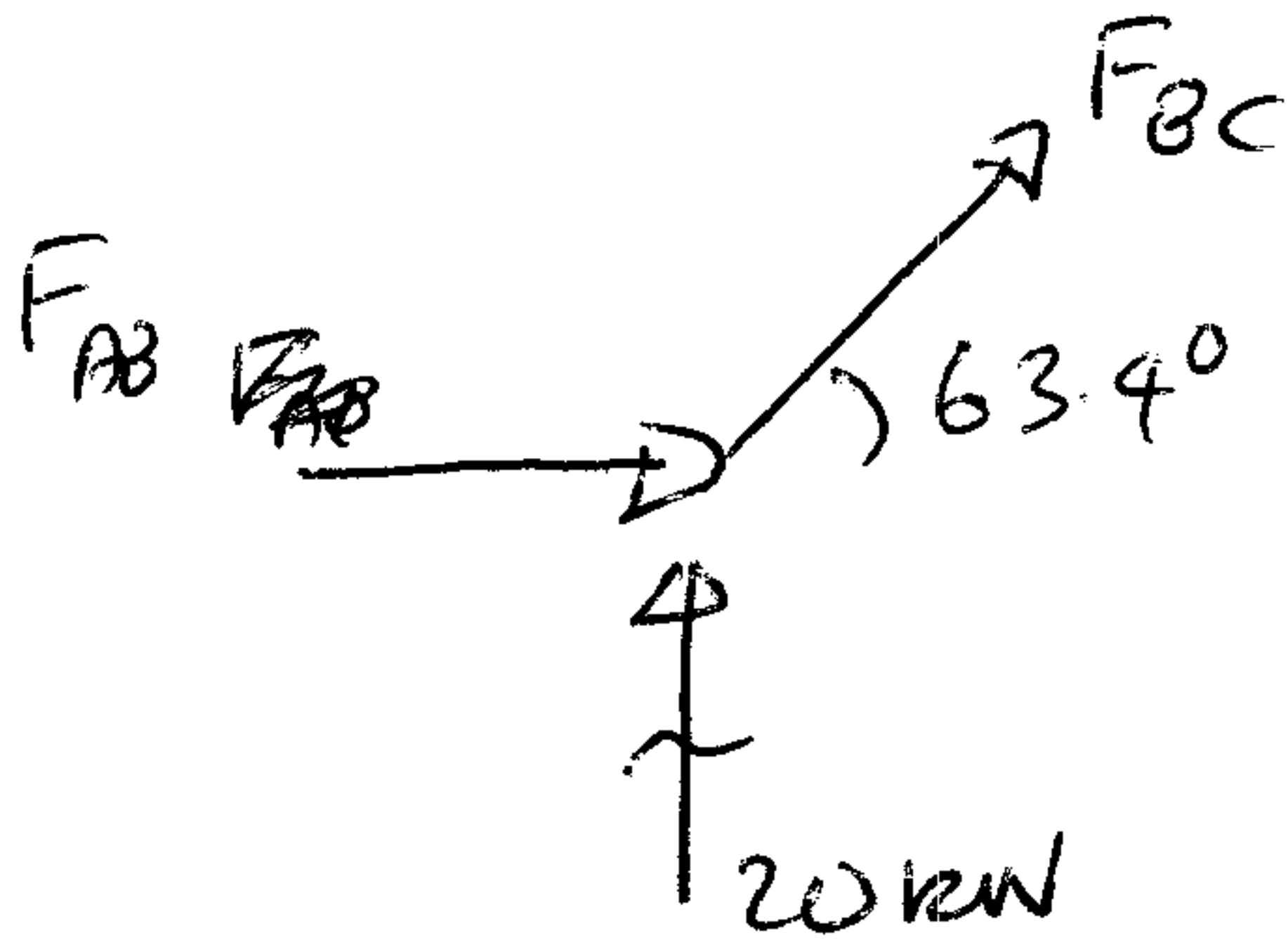
$$\sum F_y \uparrow = 0 : F_{AC} \sin 45^\circ - 10 = 0$$

$$F_{AC} = +10\sqrt{2} \text{ kN} \Leftarrow$$

$$\sum F_x \Rightarrow = 0 : F_A \cos 45^\circ + F_{AB} = 0$$

$$F_{AB} = -10 \text{ kN} \Leftarrow$$

Mom @ B



$$\sin \theta = \frac{2}{\sqrt{5}}$$

$$\sum F_y \uparrow = 0 \quad F_{BC} \sin \theta + 20 = 0$$

$$F_{BC} = \frac{-20 \cdot \sqrt{5}}{2} = -22.4 \text{ kN}$$

Critical bar is BC - carries highest load
 \therefore this determines cross-section

$$\frac{22.4 \times 10^3}{\sigma_f} = A_{\text{crit}}$$

$$\frac{22.4 \times 10^3}{700 \times 10^6} = 31.9 \times 10^{-6} \text{ m}^2 = 31.9 \text{ mm}^2$$

Total length of bars

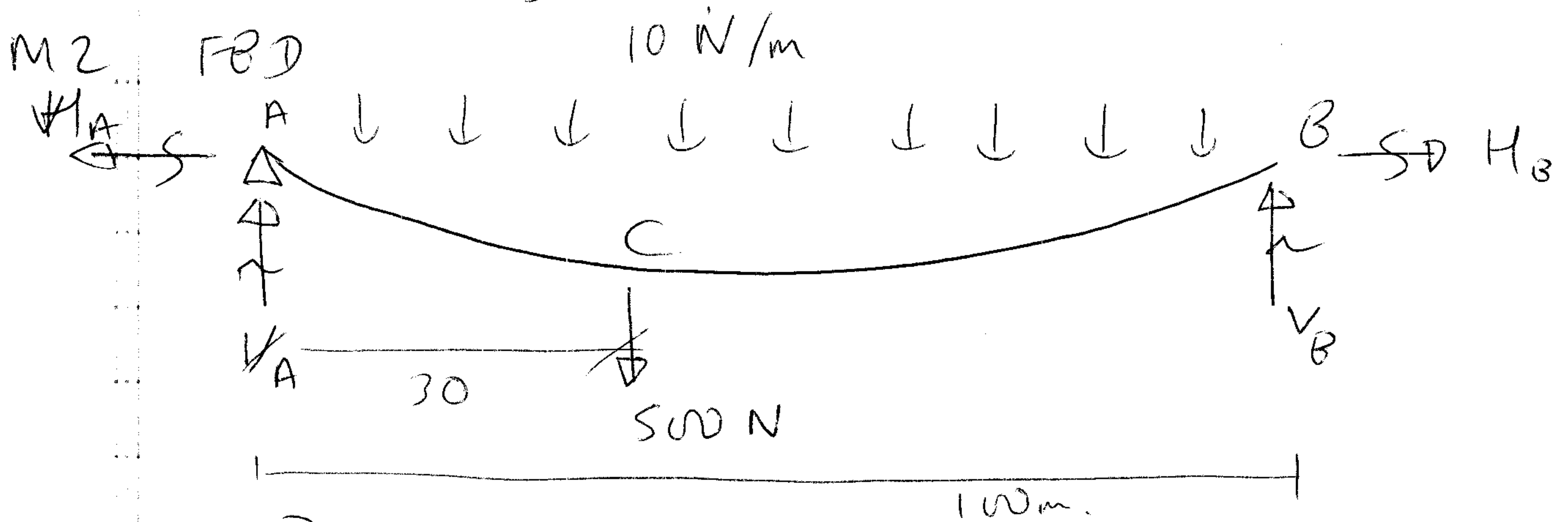
$$= L_{AC} + L_{AB} + L_{BC}$$

$$2\sqrt{2} + 1 + \sqrt{5} = 6.06 \text{ m}$$

$$\therefore \text{Total mass} = 1500 \times 6.06 \times 31.9 \times 10^{-6} = 0.29 \text{ kg}$$

i.e. $\frac{1}{2}$ a lb! seems light!

Assume weight of cable is constant per horizontal length.



$$\sum \vec{F}_x = 0 \quad -H_A + H_B = 0 \quad H_A = H_B$$

$$\sum F_y \uparrow = 0 \quad V_A + V_B - 100 \times 10 - 500 = 0$$

-1500

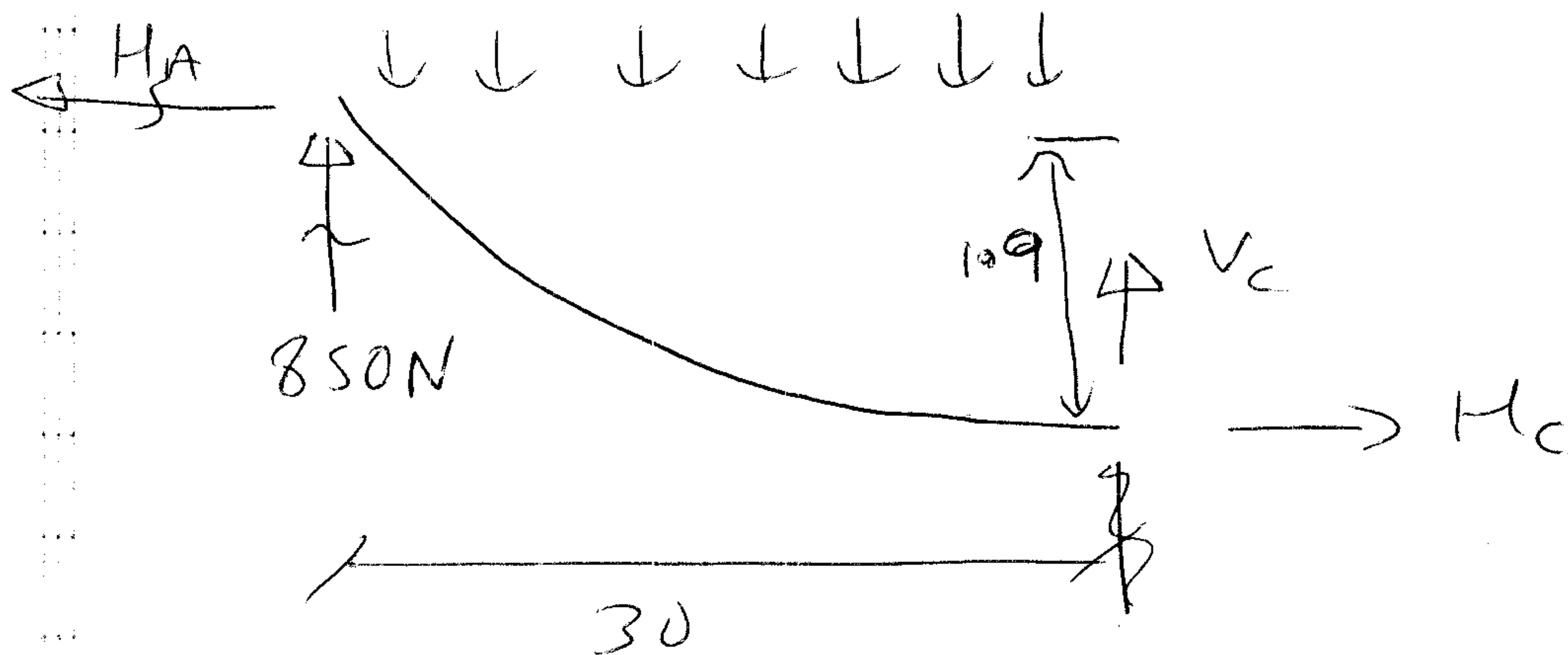
$$\left(\sum \mathcal{M}_A = 0 \right) \quad V_B \cdot 100 - 500 \times 30 - 1000 \times 50 = 0$$

$$V_B = \cancel{60} 650 \text{ N}$$

$$V_A = 1500 - 650 = 850 \text{ N}$$

App Structure is apparently statically indeterminate

Apply method of sections, just to left of C.



$$\sum \vec{F}_x = 0: -H_A + H_c = 0 \quad (\text{Tension in cable constant})$$

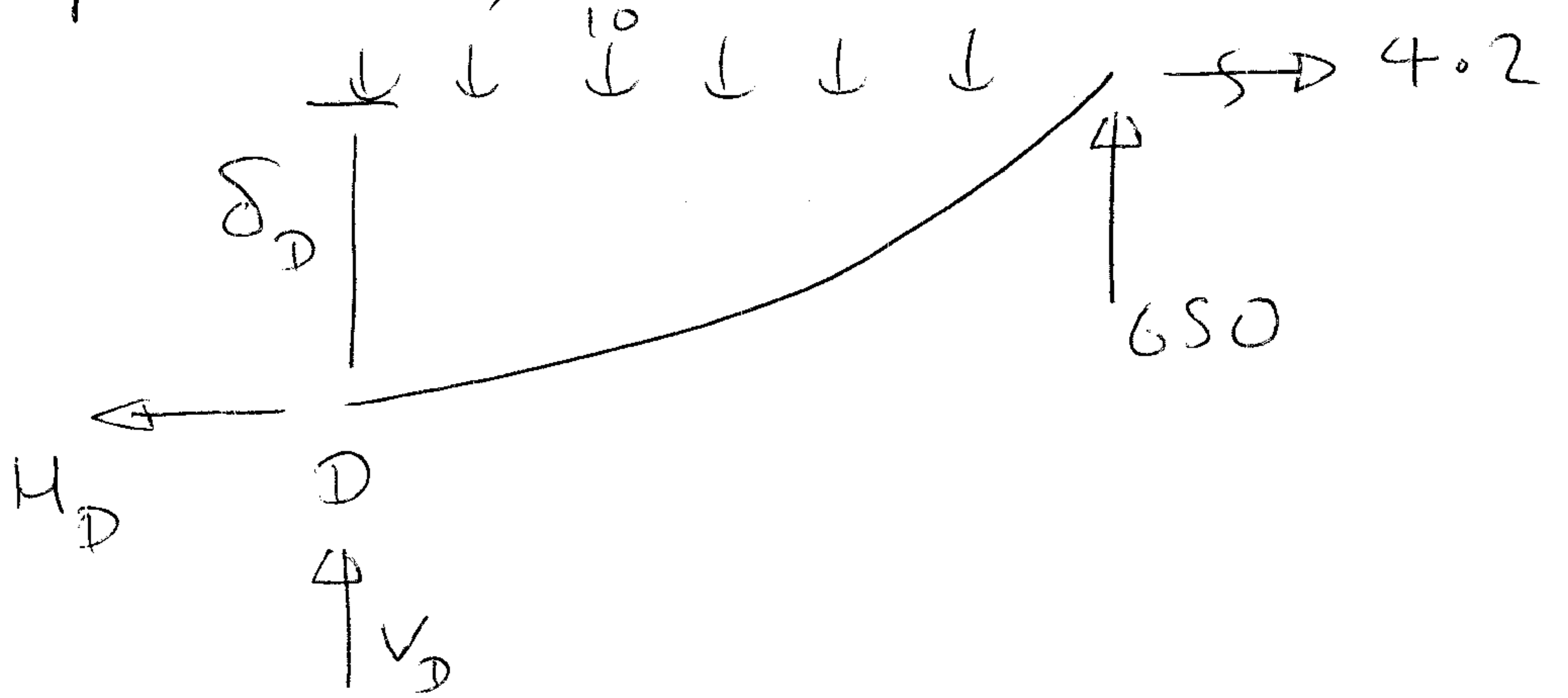
$$\sum F_y \uparrow = 0 \quad 850 - 10 \times 30 + V_c = 0$$

$$V_c = -550 \text{ N}$$

$$\sum (M_A = 0): -30 \times 10 \times 15 + V_c \cdot 30 + H_c \cdot 10.9 = 0$$

$$\frac{-4500 - 550 \times 30}{-10.9} = H_c = \overset{11.0}{\cancel{11.08}} \text{ kN} \leftarrow$$

At center of cable, consider RHS



~~$$\sum F_y \uparrow = 0: V_D - 10 \times 50 + 650 = 0$$~~

~~$$V_D = -150 \text{ N}$$~~

~~$$\sum M_D = 0: 650 \times 50 - 4.2 \times 10^3 \cdot \delta_D = 0$$~~

~~$$\delta_D = \frac{650 \times 50}{4.2 \times 10^3} =$$~~

$$\sum (M_D = 0) : 650 \times 50 - 10 \times 50 \times 25 - 11.05 \times 10^3 \delta_D = 0$$

$$\delta_D = \frac{650 \times 50 - 500 \times 25}{11.05 \times 10^3} = 1.8 \text{ m} \quad \Leftarrow$$

b) Consider only horizontal component of tension in cable (much larger than vertical)

$$H = 11 \text{ kN} \quad \therefore \sigma = \frac{11 \times 10^3}{1000 \times 10^{-6}} = 11.0 \text{ MPa}$$

$$\text{Young's modulus} = 2 \text{ GPa}$$

$$\therefore \text{Strain} = \frac{11 \times 10^6}{2 \times 10^9} = 5.5 \times 10^{-3} = 5500 \mu \text{m} \quad \Leftarrow$$

$$\text{Change in length} = 5500 \times 10^{-6} \times 100 = 0.55 \text{ m}$$

= 0.55 m \rightarrow This is on the same order as the dip of the cable so it is likely to result in an appreciable change in geometry which would need to be accounted for.

Note 2 GPa is a low modulus - equivalent to Nylon or polyester rope.