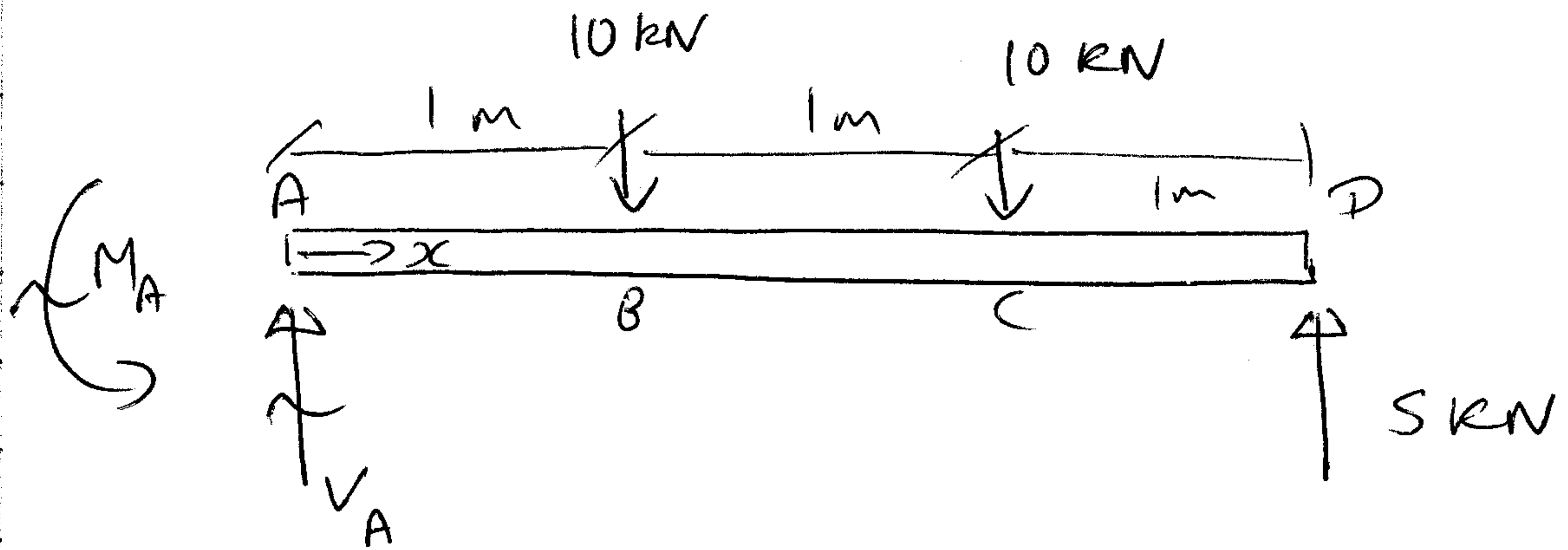


M3

a)

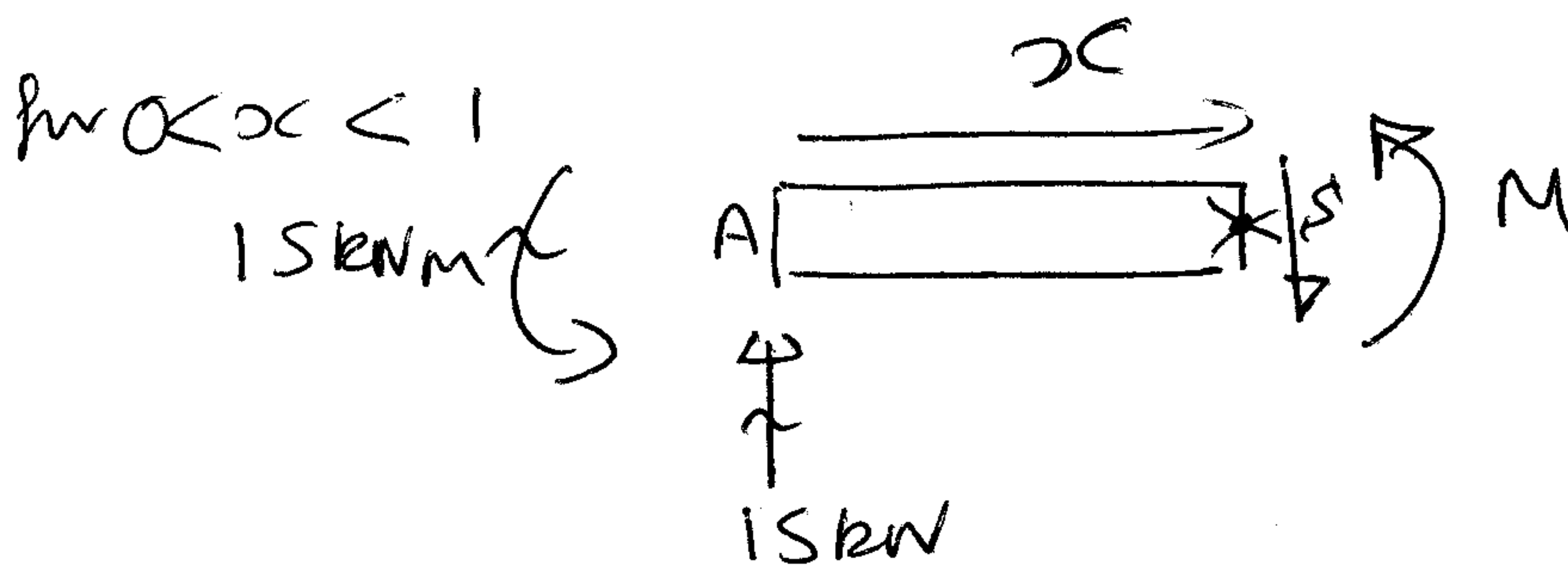


$$\sum F_y \uparrow = 0 \quad V_A - 10 - 10 + S = 0$$

$$V_A = 15 \text{ kN} \leftarrow$$

$$\sum M = 0: \quad M_A - 10 \times 1 - 10 \times 2 + S \times 3 = 0$$

$$M_A = 15 \text{ kNm} \leftarrow$$

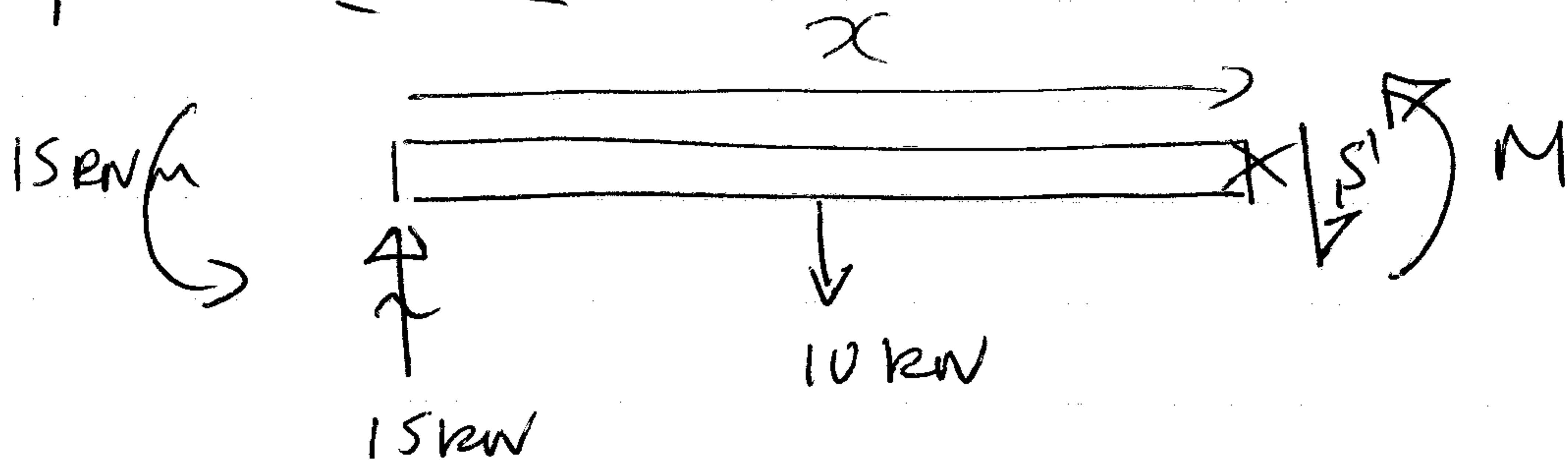


$$\sum F_y \uparrow = 0 \quad 15 - S = 0 : \quad S = 15 \text{ kN}$$

$$\sum M = 0: \quad M - 15x + 15 = 0$$

$$M = 15x - 15 \text{ (kNm)}$$

for $1 \leq x \leq 2$



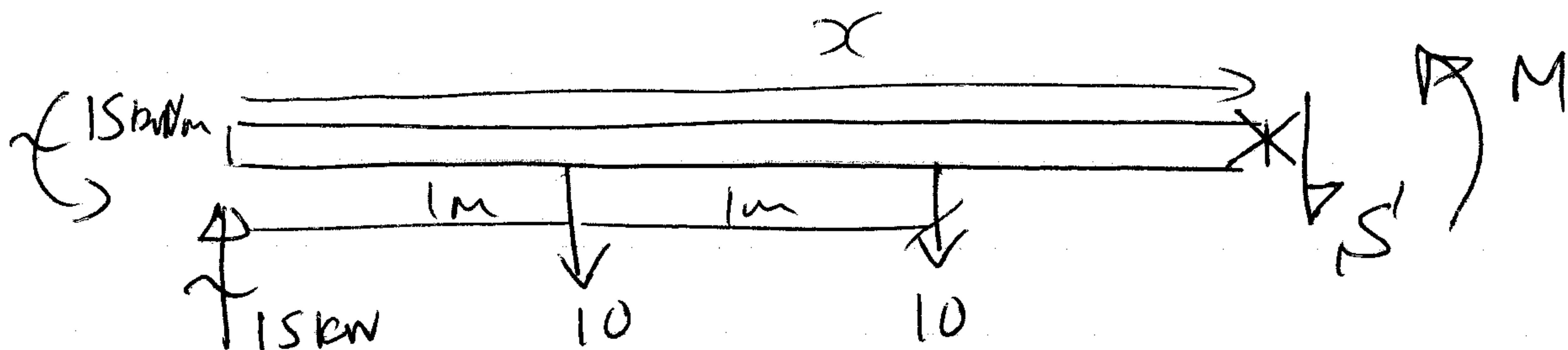
$$\sum F_y \uparrow = 0 \quad 15 - 10 - S' = 0 \quad S' = 5 \text{ kN}$$

$$\sum (M_x = 0) \quad M - 15x + 10(x-1) + 15 = 0$$

$$\quad \quad \quad -5x + 5$$

$$M = 5x - 5$$

for $2 \leq x \leq 3$



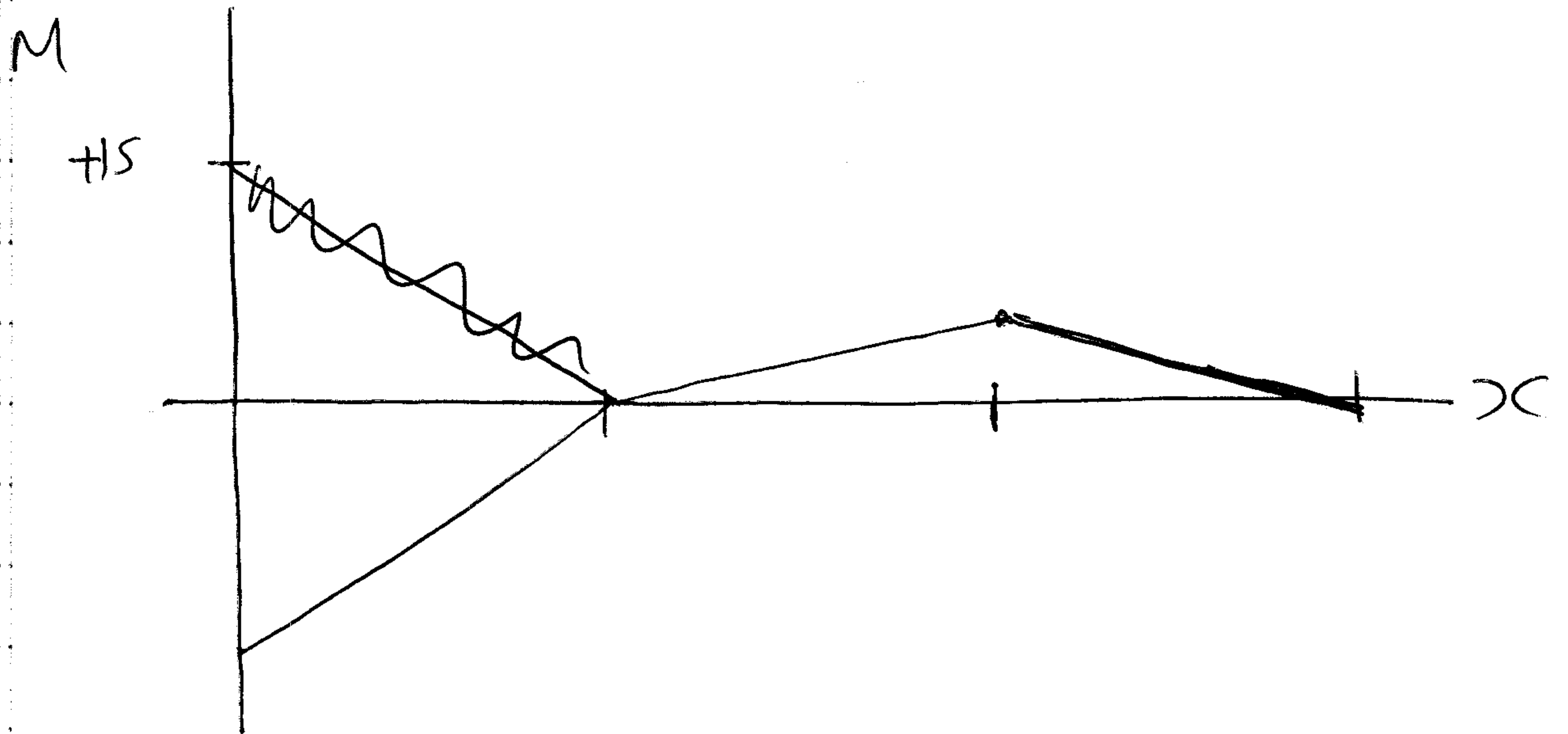
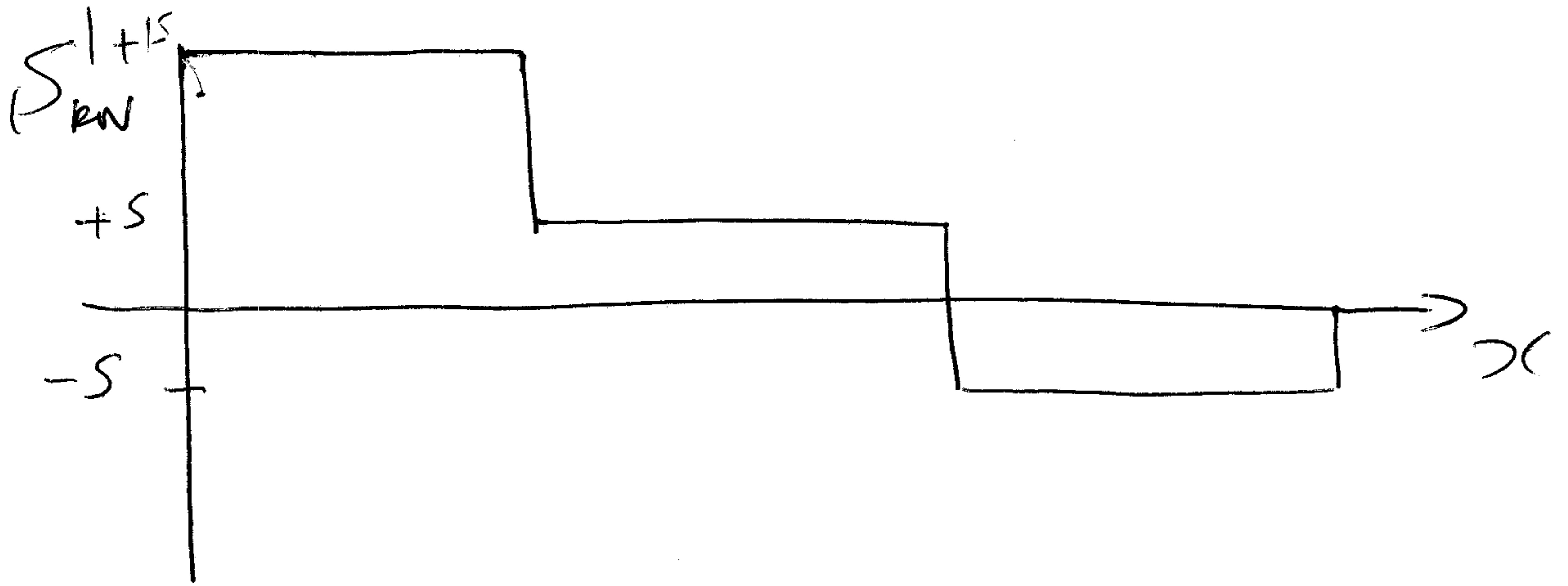
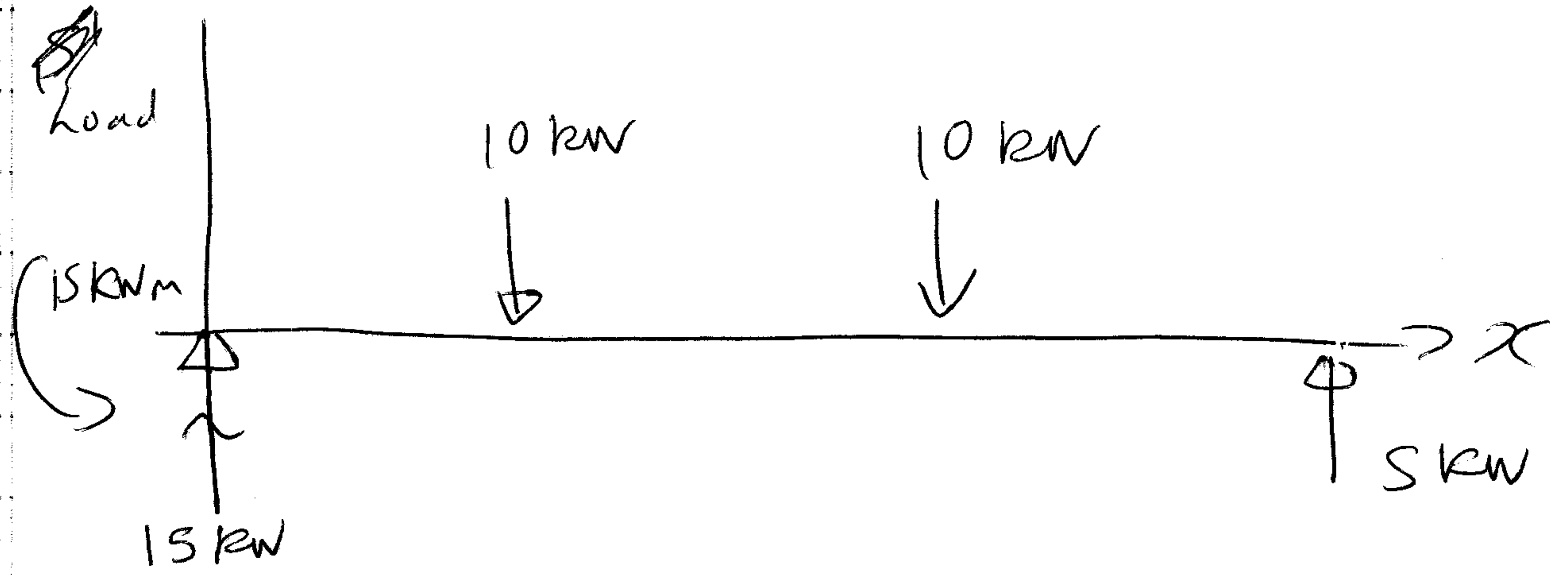
$$\sum F_y \uparrow = 0 \quad 15 - 10 - 10 - S' = 0 \quad S' = -5 \text{ kN}$$

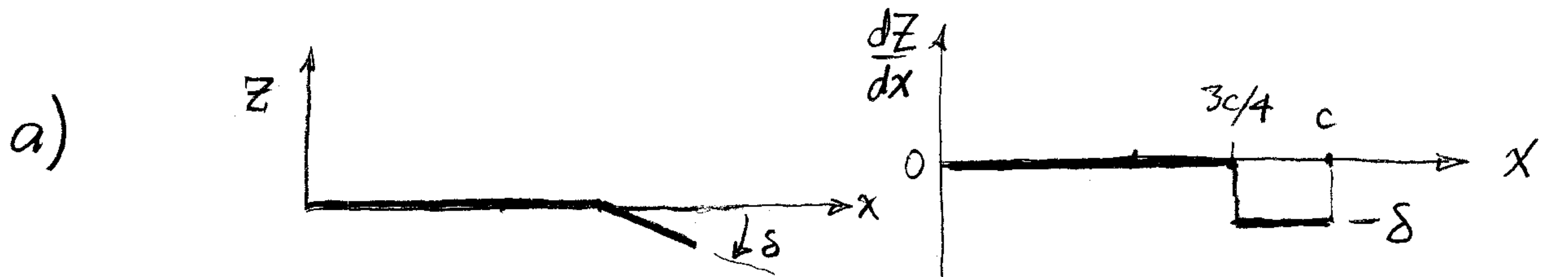
$$\sum (M_x = 0) \quad M + 15 - 15x + 10(x-1) + 10(x-2) = 0$$

$$\quad \quad \quad +15 \quad -15x \quad +10x \quad +10x \quad -10 \quad -20$$

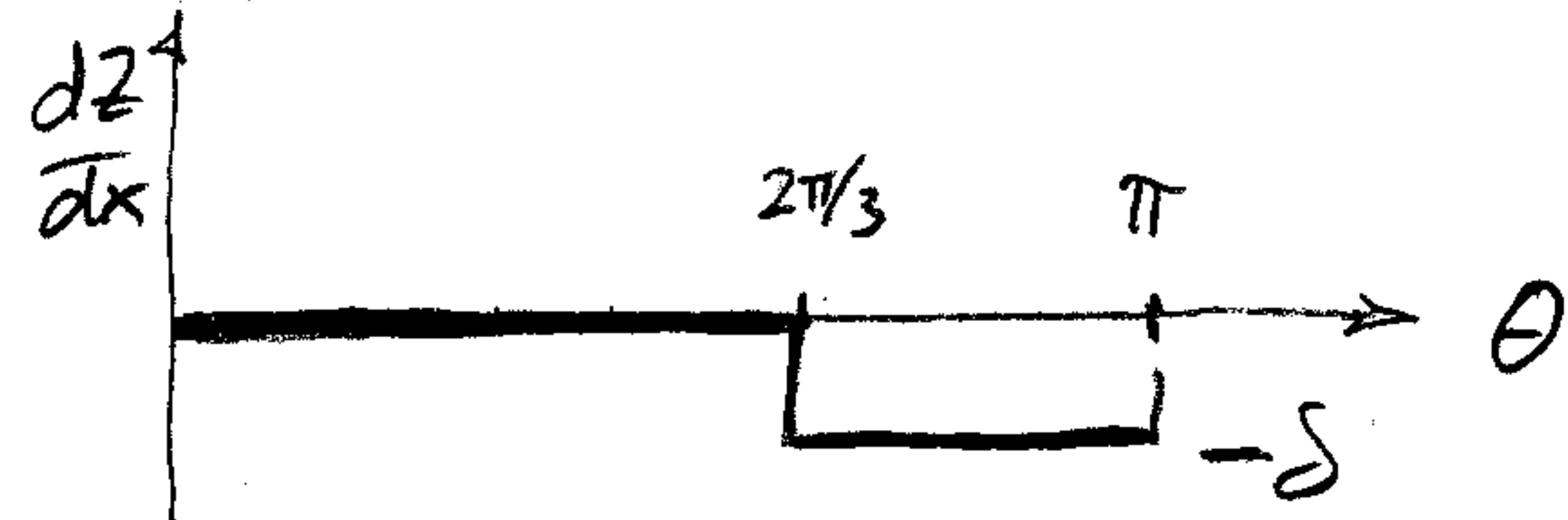
$$M = 15 - 5x$$

Plotting





$$\frac{x_h}{c} = 0.75, \quad \theta_h = \arccos\left(1 - 2\frac{x_h}{c}\right) = \frac{2\pi}{3}$$



$$b) \quad A_0 = \alpha - \frac{1}{\pi} \int_0^{\pi} \frac{dz}{dx} d\theta = \alpha - \frac{1}{\pi} \int_{\frac{2\pi}{3}}^{\pi} -\delta d\theta = \alpha - \frac{1}{\pi} \left(\pi - \frac{2\pi}{3}\right)(-\delta) = \alpha + \frac{1}{3}\delta$$

$$A_1 = \frac{2}{\pi} \int_0^{\pi} \frac{dz}{dx} \cos\theta d\theta = \frac{2}{\pi} \int_{\frac{2\pi}{3}}^{\pi} -\delta \cos\theta d\theta = \frac{2}{\pi} (-\delta) \left(\sin\theta\right) \Big|_{\frac{2\pi}{3}}^{\pi} = 0.551 \delta$$

$$A_2 = \frac{2}{\pi} \int_0^{\pi} \frac{dz}{dx} \cos 2\theta d\theta = \frac{2}{\pi} \int_{\frac{2\pi}{3}}^{\pi} -\delta \cos 2\theta d\theta = \frac{2}{\pi} (-\delta) \left(\frac{\sin 2\theta}{2}\right) \Big|_{\frac{2\pi}{3}}^{\pi} = -0.276 \delta$$

$$C_L = \pi(2A_0 + A_1) = 2\pi(\alpha + 0.609\delta)$$

$$C_{m,c/4} = \frac{\pi}{4}(A_2 - A_1) = -0.649\delta$$

$$c) \quad \left[\begin{array}{l} \frac{\partial C_L}{\partial \delta} = 2\pi \cdot 0.609 = 3.826 \\ \frac{\partial C_m}{\partial \delta} = -0.649 \end{array} \right]$$

a) Assuming L' is uniform across the span,

$$D'_i = L' \alpha_i = L' \frac{W}{V_\infty}$$

Given: $W = \frac{L}{2\rho V_\infty b^2}$

$$D_i = D'_i b = L' b \frac{W}{V} = L \frac{W}{V} = \frac{L^2}{2\rho V_\infty^2 b^2} = \frac{L^2}{\frac{1}{2}\rho V^2} \frac{1}{4b^2}$$

$$C_{D_i} = \frac{D_i}{\frac{1}{2}\rho V_\infty^2 S} = \frac{L^2}{(\frac{1}{2}\rho V_\infty^2)^2 S^2} \cdot \frac{S}{4b^2} = \frac{C_L^2}{4R}$$

b) $C_L = 2\pi \alpha_{\text{eff}} = 2\pi(\alpha - \alpha_i)$

ignore $\alpha_{L=0}$, no effect on $\frac{dC_L}{d\alpha}$

but we have $C_{D_i} = \alpha_i C_L$, or $\alpha_i = \frac{C_{D_i}}{C_L} = \frac{C_L}{4R}$

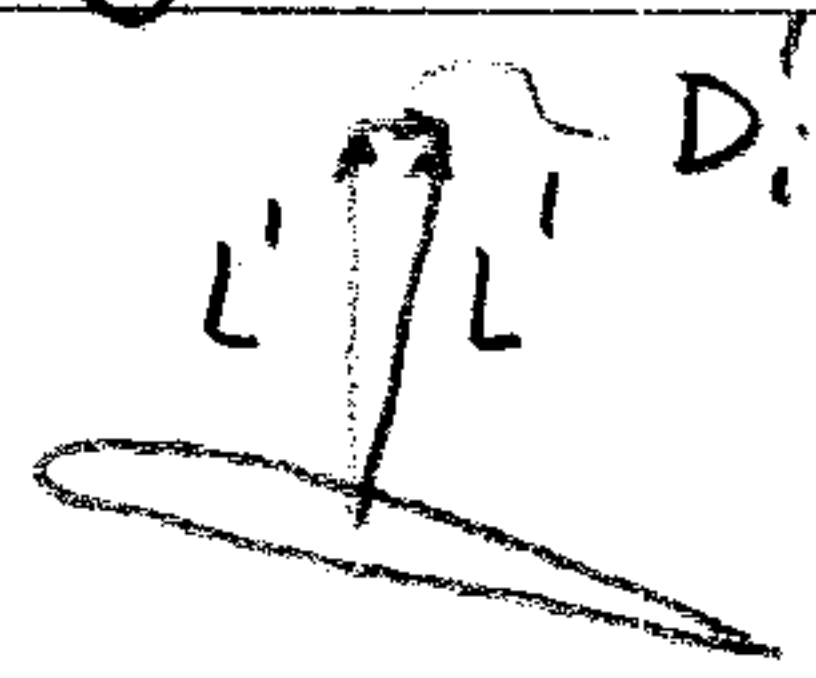
$$\Rightarrow C_L = 2\pi\left(\alpha - \frac{C_L}{4R}\right)$$

$$C_L\left(1 + \frac{\pi}{2R}\right) = 2\pi\alpha$$

$$C_L(\alpha) = \frac{2\pi}{1 + \frac{\pi}{2R}} \alpha$$

$$\frac{dC_L}{d\alpha} = \frac{2\pi}{1 + \frac{\pi}{2R}}$$

small than 2-D value of 2π
by factor of $\frac{1}{1 + \frac{\pi}{2R}}$



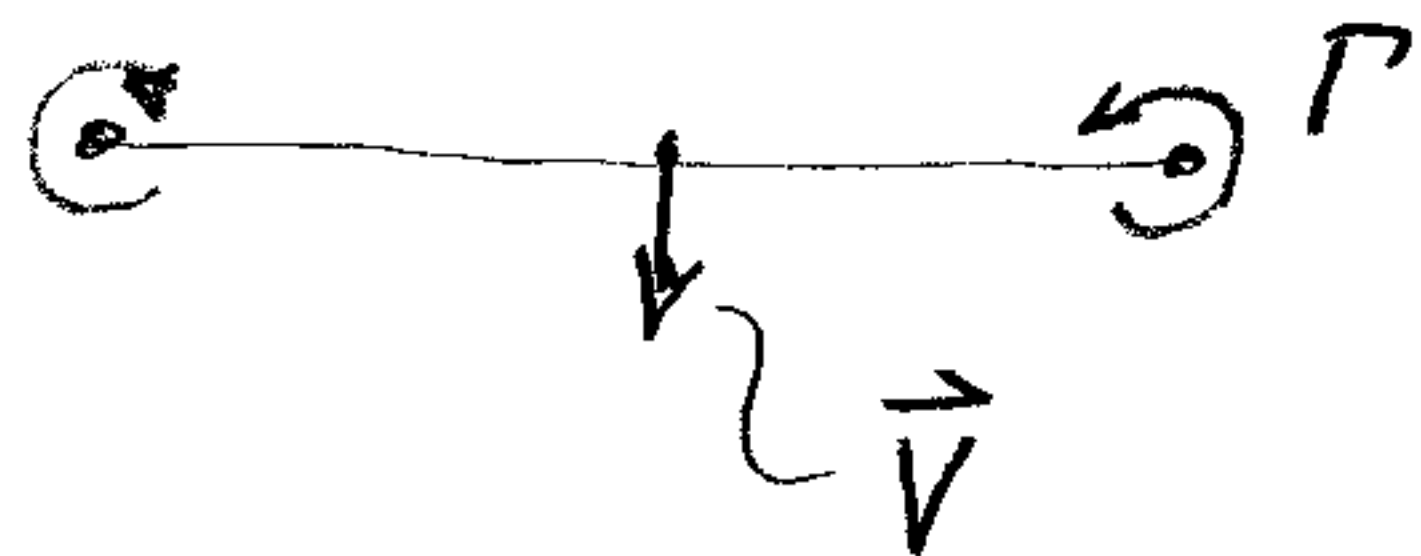
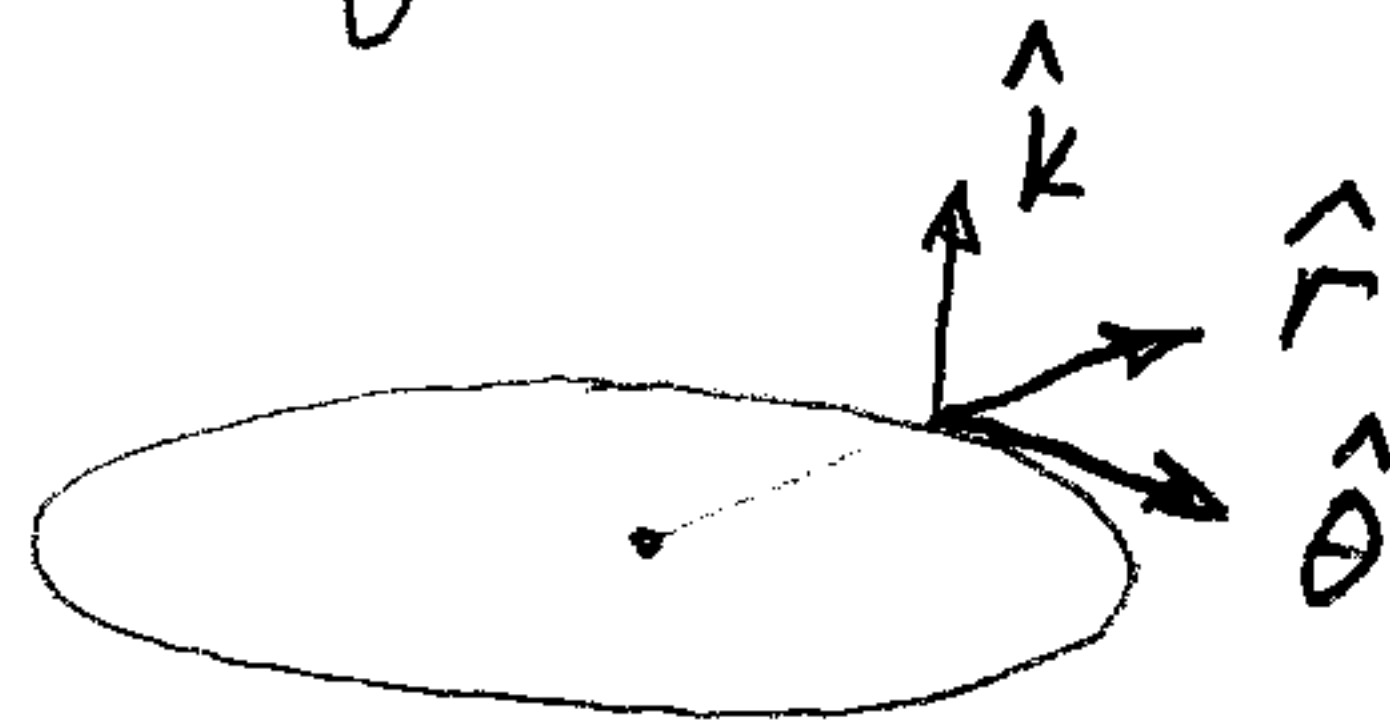
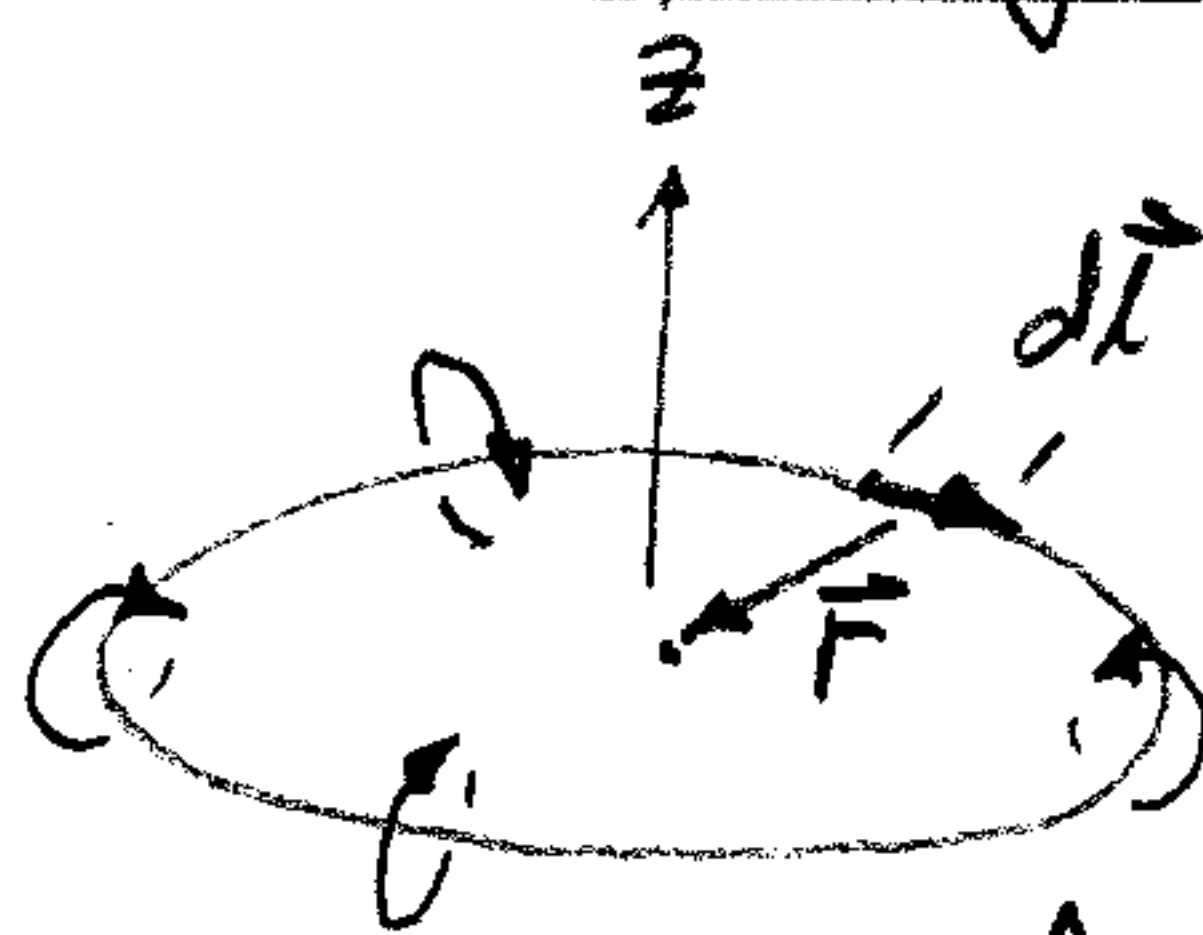
(Anderson p 416)

$$1. \vec{V} = \frac{\Gamma}{4\pi} \int \frac{d\vec{l} \times \vec{r}}{r^3}$$

$$d\vec{l} = R d\theta \hat{\theta}, \quad \vec{r} = -R \hat{r}, \quad r = R$$

$$d\vec{l} \times \vec{r} = -R^2 d\theta \hat{\theta} \times \hat{r} = -R^2 d\theta \hat{k}$$

$$\vec{V} = \frac{\Gamma}{4\pi} \int_0^{2\pi} \frac{-R^2 d\theta \hat{k}}{R^3} = -\frac{\Gamma}{2R} \hat{k}$$



2. Now we have

$$d\vec{l} = R d\theta \hat{\theta}, \quad \vec{r} = -R \hat{r} + A \hat{k}$$

$$r^3 = (R^2 + A^2)^{3/2}$$

$$d\vec{l} \times \vec{r} = (-R^2 (\hat{\theta} \times \hat{r}) + RA (\hat{\theta} \times \hat{k})) d\theta = (-R^2 \hat{k} - RA \hat{r}) d\theta$$

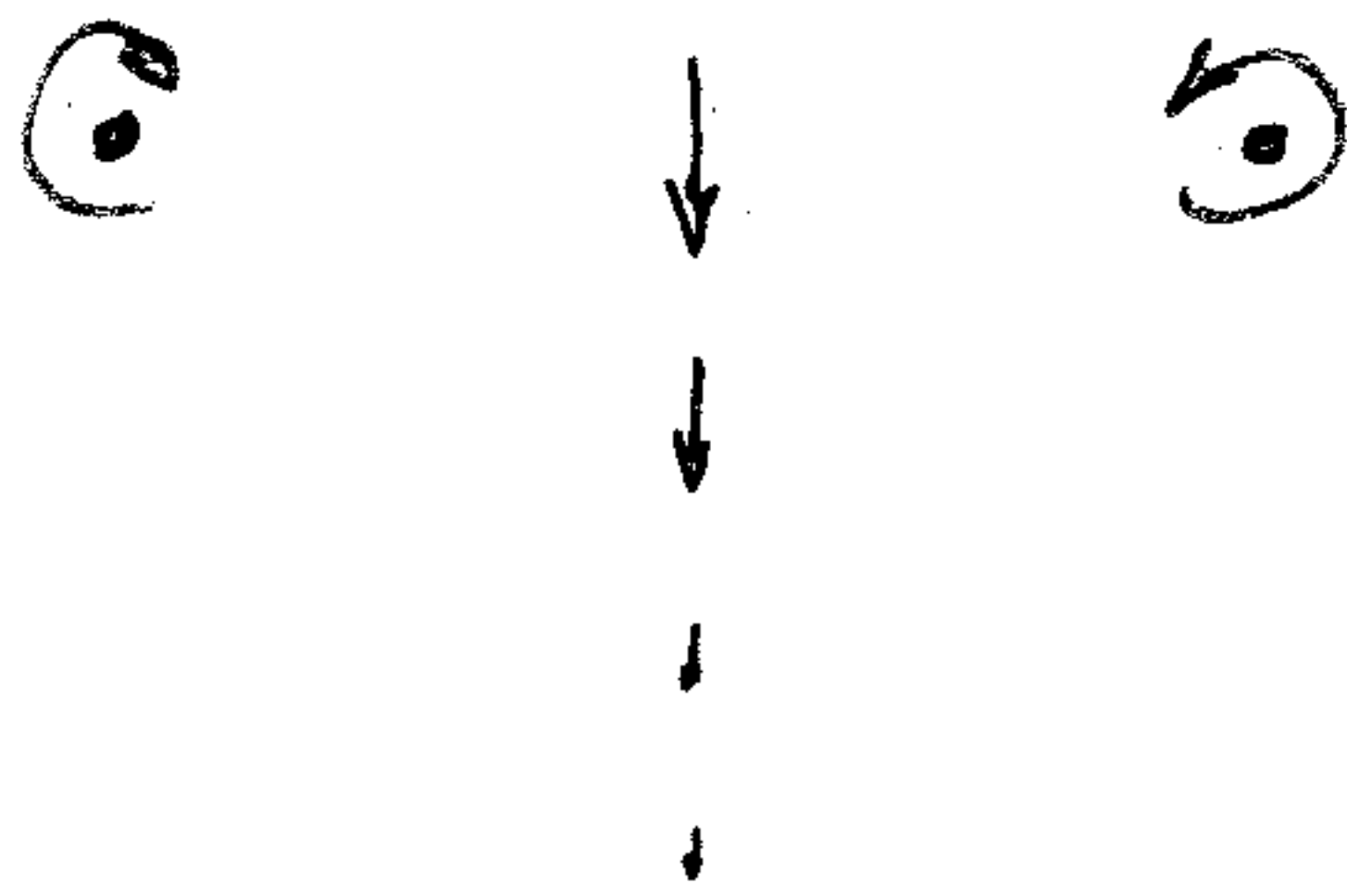
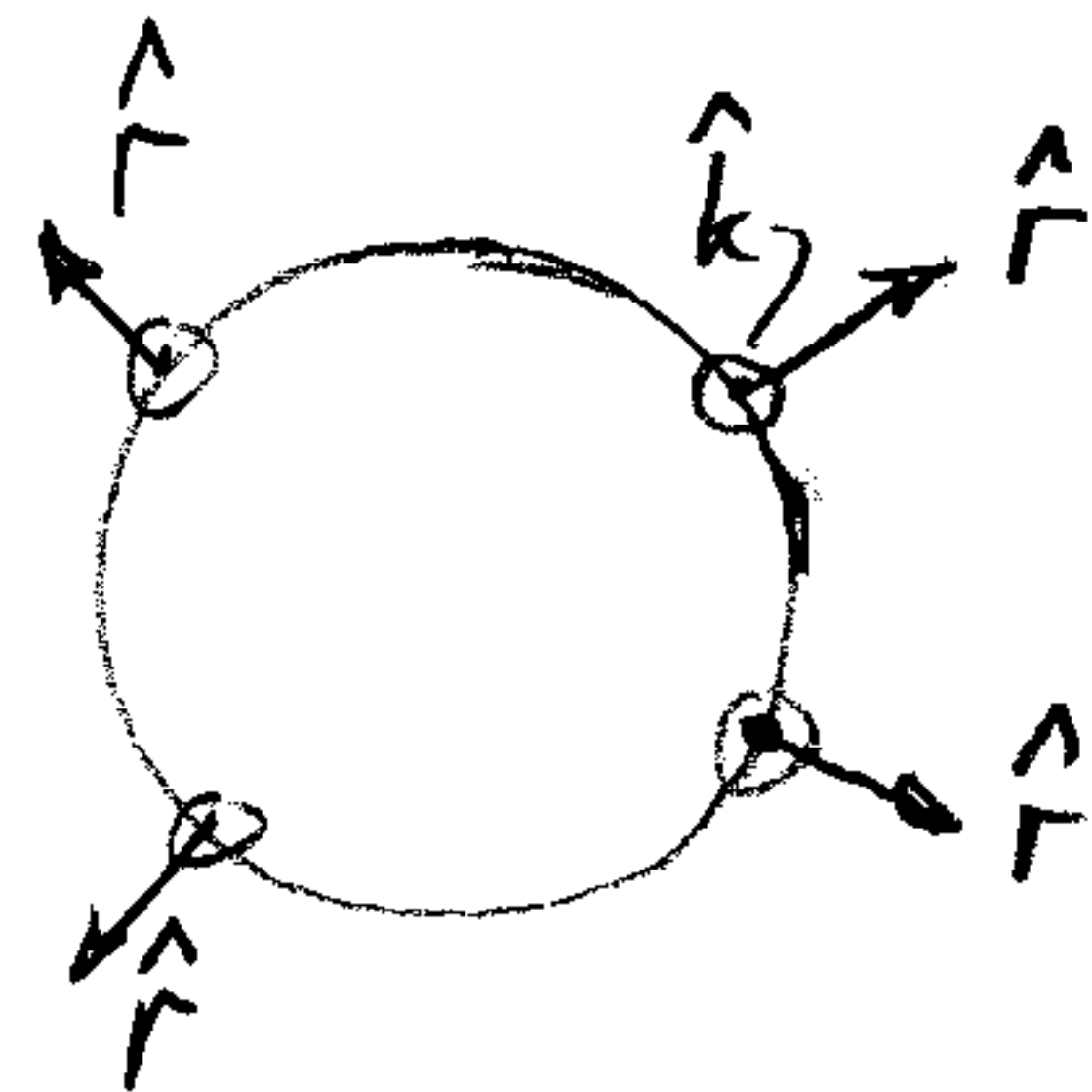
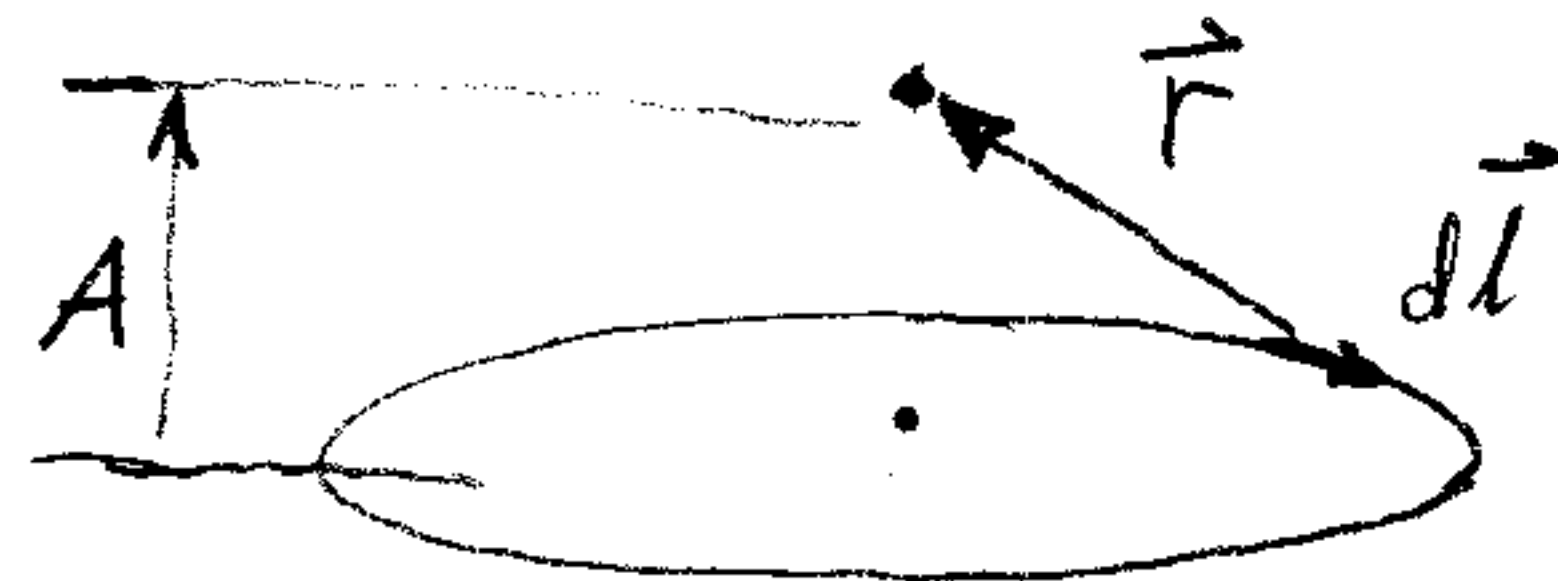
Note that \hat{k} is constant, but \hat{r} depends on θ (varies around circle)

$$\vec{V} = \frac{\Gamma}{4\pi} \int_0^{2\pi} \frac{-R^2 \hat{k} - RA \hat{r}}{(R^2 + A^2)^{3/2}} d\theta$$

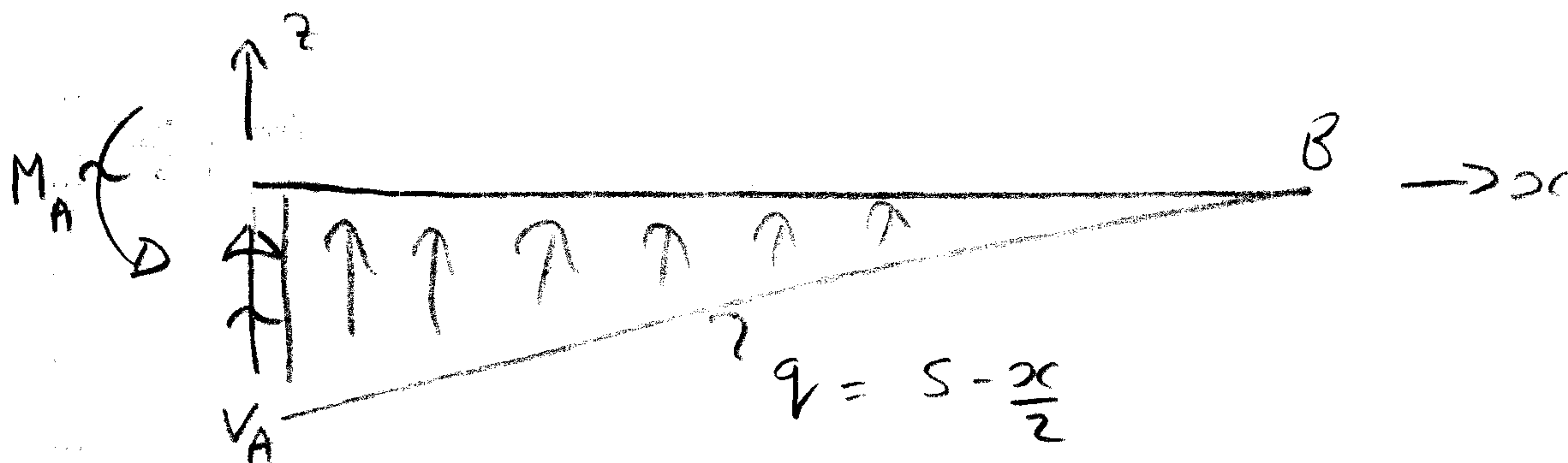
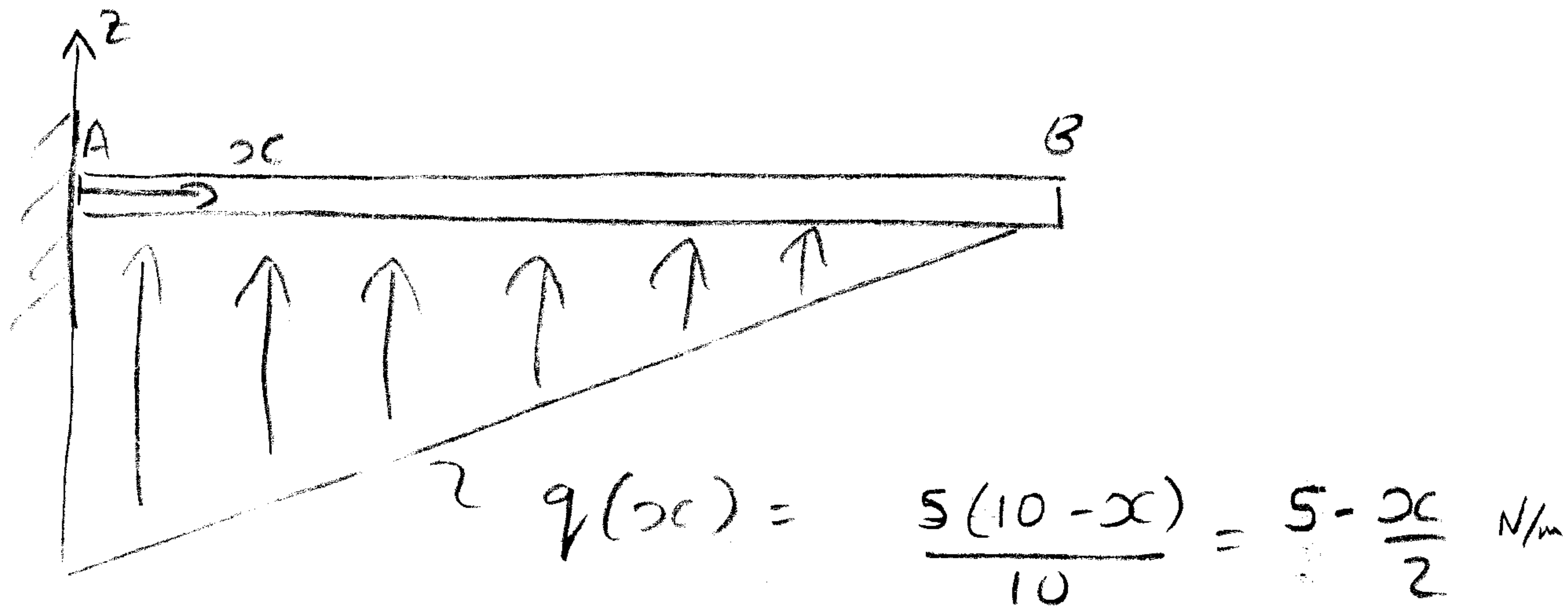
$$\vec{V} = -\frac{\Gamma}{2} \frac{R^2}{(R^2 + A^2)^{3/2}} \hat{k} + \frac{\Gamma}{4\pi} \frac{-RA}{(R^2 + A^2)^{3/2}} \int_0^{2\pi} \hat{r} d\theta$$

But we note that $\int_0^{2\pi} \hat{r} d\theta = 0$, since \hat{r} cancels when integrated around perimeter.

$$\vec{V} = -\frac{\Gamma}{2} \frac{R^2}{(R^2 + A^2)^{3/2}} \hat{k}$$



M4



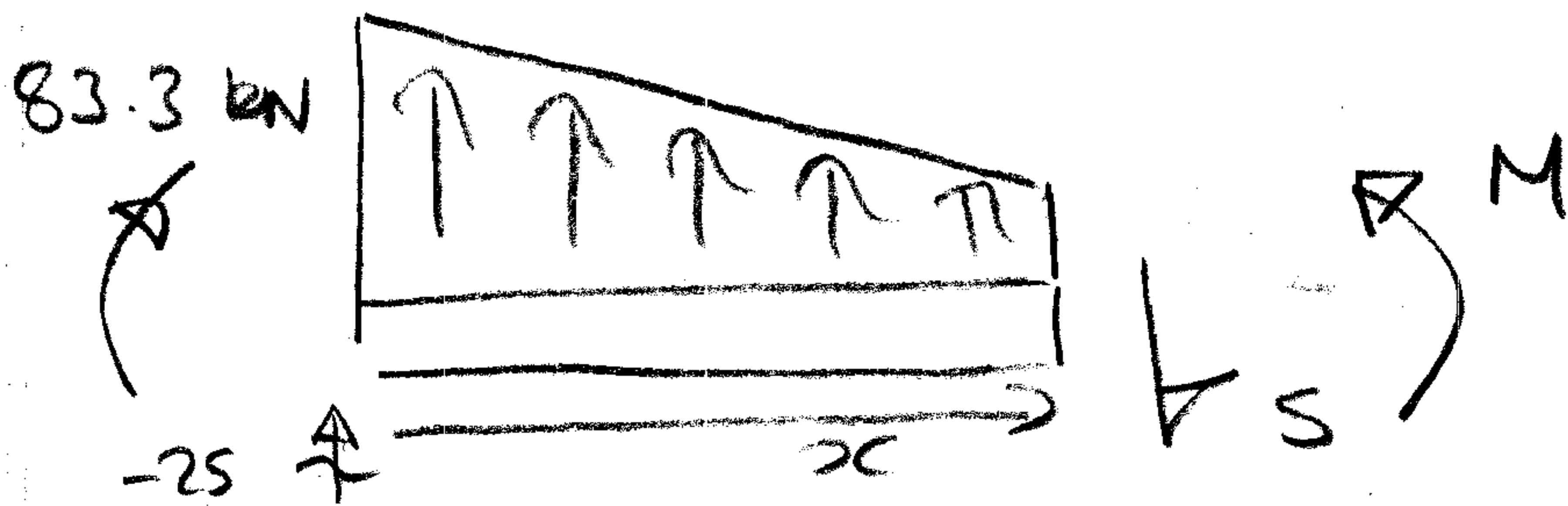
$$\sum F_z \uparrow = 0 \quad V_A + \int_0^{10} 5 - \frac{x}{2} dx = 0$$

$$V_A = - \left[5x - \frac{x^2}{4} \right]_0^{10} = -25 \text{ kN}$$

$$\left(\sum M_A = 0 \right) : M_A + \int_0^{10} \left(5 - \frac{x}{2} \right) x dx = 0$$

$$M_A = - \left[\frac{5x^2}{2} - \frac{x^3}{6} \right]_0^{10} = -83.3 \text{ kNm} \leftarrow$$

Moment distribution



$$\sum F_2 \uparrow = 0:$$

$$-25 - \int_0^x \left(5 - \frac{x}{2}\right) dx = 0$$

$$\int_0^x \left(5 - \frac{x}{2}\right) dx - 25 = 5x - \frac{x^2}{4} - 25 = 0$$

$$\sum (M_A = 0) \quad M - 83.3 + \int_0^x \left(5 - \frac{x}{2}\right) x dx - \int_0^x dx = 0$$

$$M - 83.3 + \frac{5x^2}{2} - \frac{x^3}{6} - 5x + x = 0$$

$$M = -\frac{1}{12}x^3 + \frac{5}{2}x^2 - 25x + 83.3$$

Max bending stress at root $x=0$ $z = \pm \frac{100 \text{ mm}}{2}$

$$M_{\text{max}} = 83.3 \text{ kNm @ } x=0$$

$$I = \frac{1}{12}bh^3$$

$$\sigma_{\text{max}} = \frac{6 M_{\text{max}} \frac{h}{2}}{\frac{1}{12}bh^3} = \frac{6 M_{\text{max}}}{bh^2}$$

$$= \frac{6 \times 83.3 \times 10^3}{50 \times 10^{-3} \times (100 \times 10^{-3})^2} = 1 \text{ GPa !! (high)}$$

M4 tip deflection, from moment curvature relation

$$M = EI \frac{d^2 w}{dx^2}$$

Integrate moment twice \rightarrow

$$EI \frac{dw}{dx} = -\frac{1}{12} \frac{20^4}{4} + \frac{5}{2} \frac{20^3}{3} - \frac{25 \cdot 20^2}{2} + 83.3x + A$$

$$\frac{dw}{dx} = 0 \text{ @ } x=0 \Rightarrow A = 0$$

$$EI w = -\frac{1}{12} \frac{20^5}{20} + \frac{5}{2} \frac{20^4}{12} - \frac{25 \cdot 20^3}{6} + \frac{83.3 \cdot 20^2}{2} + B$$

$$w = 0 \text{ @ } x=0 \Rightarrow B = 0$$

$$EI w = -\frac{20^5}{240} + \frac{5 \cdot 20^4}{24} - \frac{25 \cdot 20^3}{6} + \frac{83.3 \cdot 20^2}{2}$$

at tip $x = 0$

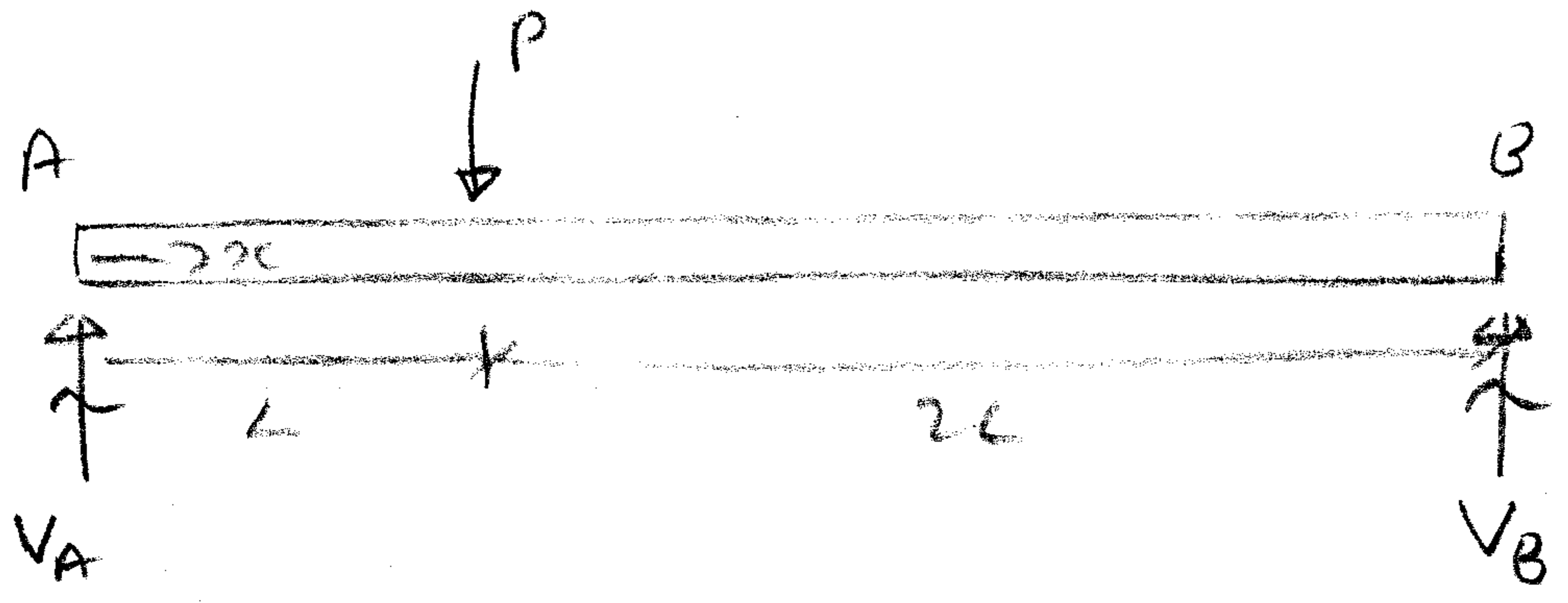
$$\delta = \frac{1}{EI} \left[\frac{-10^5}{240} + \frac{5 \cdot 10^4}{24} - \frac{25 \cdot 10^3}{6} + \frac{83.3 \cdot 10^2}{2} \right] = \frac{1650 \times 10^3}{EI}$$

$$E = 70 \times 10^9, \quad I = \frac{1}{12} \times 50 \times 10^{-3} \times (100 \times 10^{-3})^3 = 4.17 \times 10^{-6}$$

$$\delta = \frac{1650 \times 10^3}{70 \times 10^9 \times 4.17 \times 10^{-6}} = 5.66 \text{ m. } \equiv$$

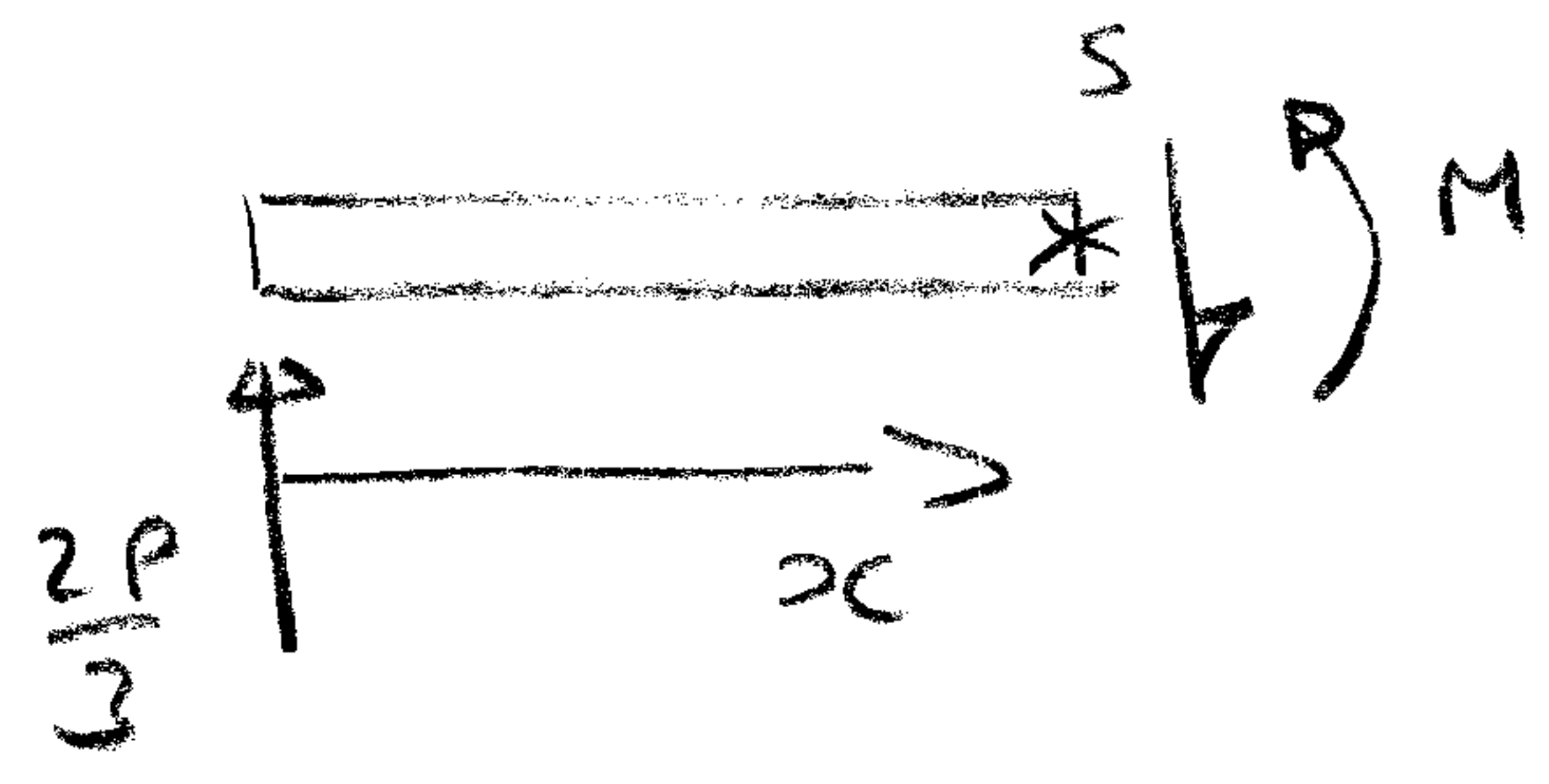
clearly rather large - not small
deflection

MS



$$\sum (M_A = 0) : 3LV_B - PL = 0 \quad \therefore V_B = \frac{P}{3}, \quad V_A = \frac{2P}{3}$$

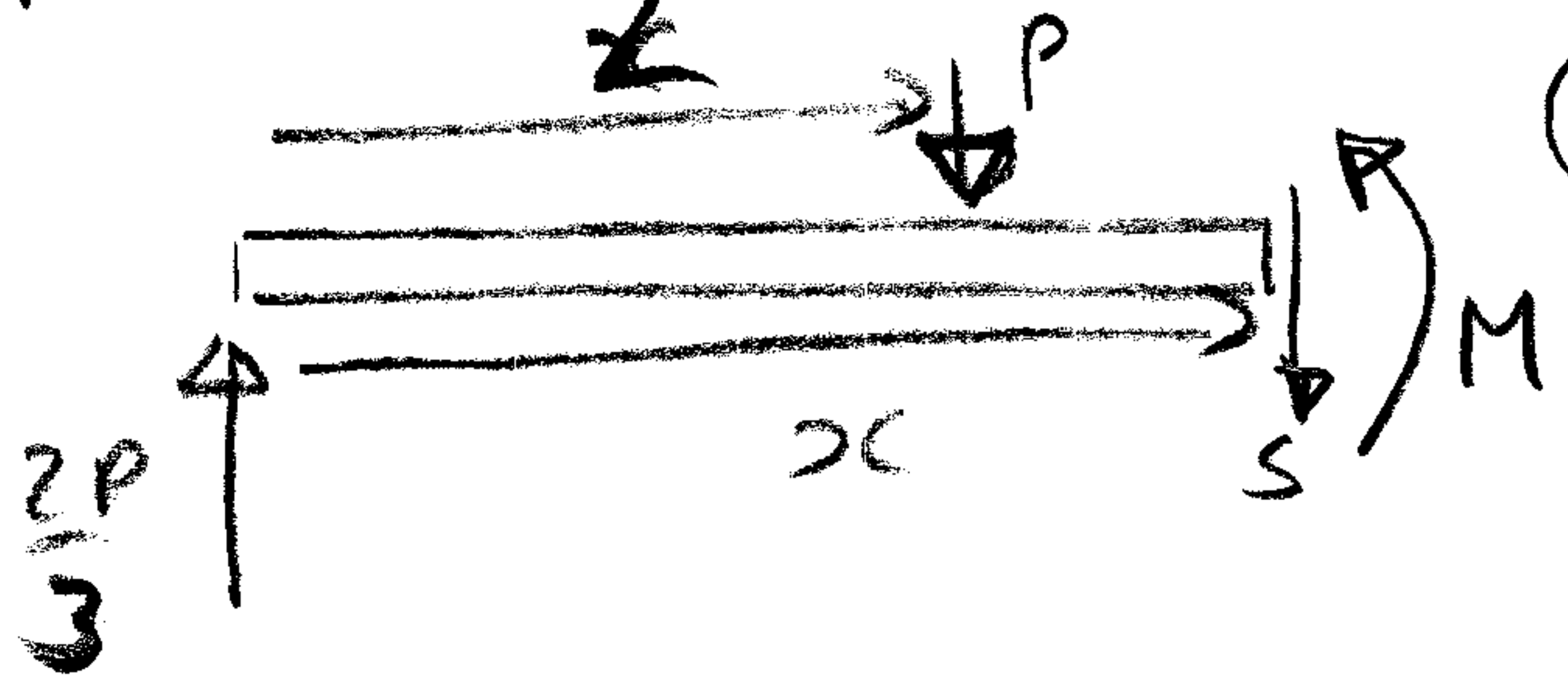
for $0 < x < L$



$$\left(\sum M_x = 0 : M - \frac{2P}{3}x = 0 \right)$$

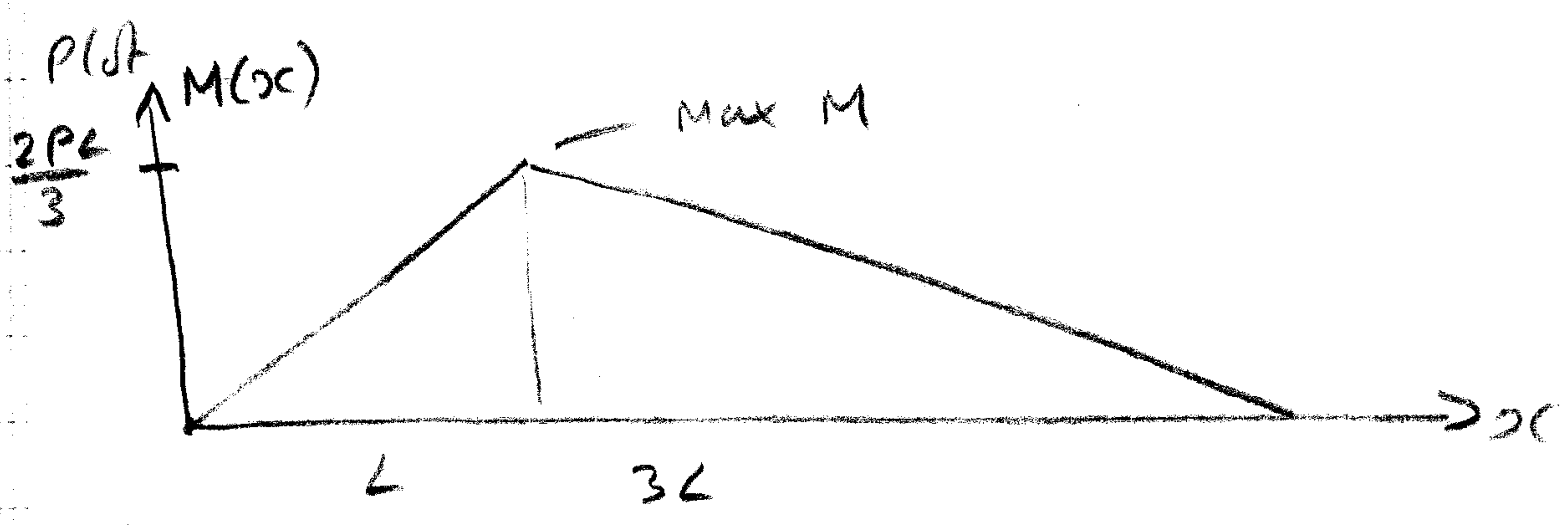
$$M = \frac{2Px}{3}$$

for $L < x < 3L$



$$\left(\sum M_x = 0 : M + P(x-L) - \frac{2P}{3}x = 0 \right)$$

$$M = -\frac{P}{3}x + PL$$



deflection from
①

$$0 < x < L$$

$$EI \frac{d^2 w^1}{dx^2} = \frac{2Px}{3}$$

$$EI \frac{dw^1}{dx} = \frac{2Px^2}{6} + A$$

$$EI w^1 = \frac{2Px^3}{18} + Ax + B$$

$$EI \frac{d^2 w}{dx^2} = M$$

②

$$L < x < 3L$$

$$EI \frac{d^2 w^2}{dx^2} = -\frac{Px}{3} + PL$$

$$EI \frac{dw^2}{dx} = -\frac{Px^2}{6} + PLx + C$$

$$EI w^2 = -\frac{Px^3}{18} + \frac{PLx^2}{2} + Cx + D$$

Apply boundary conditions

$$\text{@ } x=0$$

$$w^1 = 0$$

$$x=3L \quad w^2 = 0$$

$$\Rightarrow B = 0,$$

$$-\frac{27PL^3}{18} + \frac{9PL^3}{2} + 3C \cancel{L} + D = 0$$

$$= 3PL^3 + 3C \cancel{L} + D = 0 \quad \text{①}$$

$$\text{@ } x=L$$

$$\frac{dw^1}{dx} = \frac{dw^2}{dx}$$

$$\Rightarrow \frac{2PL^2}{6} + A = -\frac{PL^2}{6} + PL^2 + C$$

$$A - C = \frac{PL^2}{2} \quad \text{②}$$

$$\textcircled{a} \quad x = L$$

$$w^1 = w^2$$

$$\frac{2PL^3}{18} + AL = -\frac{PL^3}{18} + \frac{PL^3}{2} + CL + D$$

$$(A - C)L = \frac{PL^3}{3} + D \quad \textcircled{3}$$

$$\text{Substitute for } A - C = \frac{PL^2}{2} \quad \textcircled{2}$$

$$\Rightarrow \frac{PL^3}{2} = \frac{PL^3}{3} + D \Rightarrow D = \frac{PL^3}{6} \Leftarrow$$

$$\text{Substitute in } \textcircled{1} \quad 3PL^2 + 3Cx + \frac{PL^2}{6} = 0$$

$$C = \frac{1}{3} \left(-3PL^2 - \frac{PL^2}{6} \right) = -\frac{19}{18} PL^2 \Leftarrow$$

$$\therefore A = \frac{PL^2}{2} + \frac{19}{18} PL^2 = \frac{14}{9} PL^2 \Leftarrow$$

\therefore for $0 < x < L$

for $L < x < 3L$

$$w = \frac{1}{EI} \left(\frac{2Px^3}{18} + \frac{14PL^2}{9} x \right) \quad \left| \quad w = \frac{1}{EI} \left(-\frac{Px^3}{18} + \right.$$

for $L < x < 3L$

$$w = \frac{1}{EI} \left(-\frac{Px^3}{18} + \frac{PLx^2}{2} - \frac{19PL^2x}{18} + \frac{PL^3}{6} \right) \quad \checkmark$$

max deflection when $\frac{dw}{dx} = 0$

between $0 < x < L$?

$$\frac{dw}{dx} = \frac{1}{EI} \left(\frac{6Px^2}{18} + \frac{28PL^2x}{9} \right) = 0 \quad \text{Not in } 0 < x < L$$

between $L < x < 3L$ or $1 < \frac{x}{L} < 3$

$$\frac{dw}{dx} = \frac{PL^2}{EI} \left(-\frac{3Px^2}{18L^2} + \frac{2PLx}{2L} - \frac{19PL^2}{18} \right) = 0$$

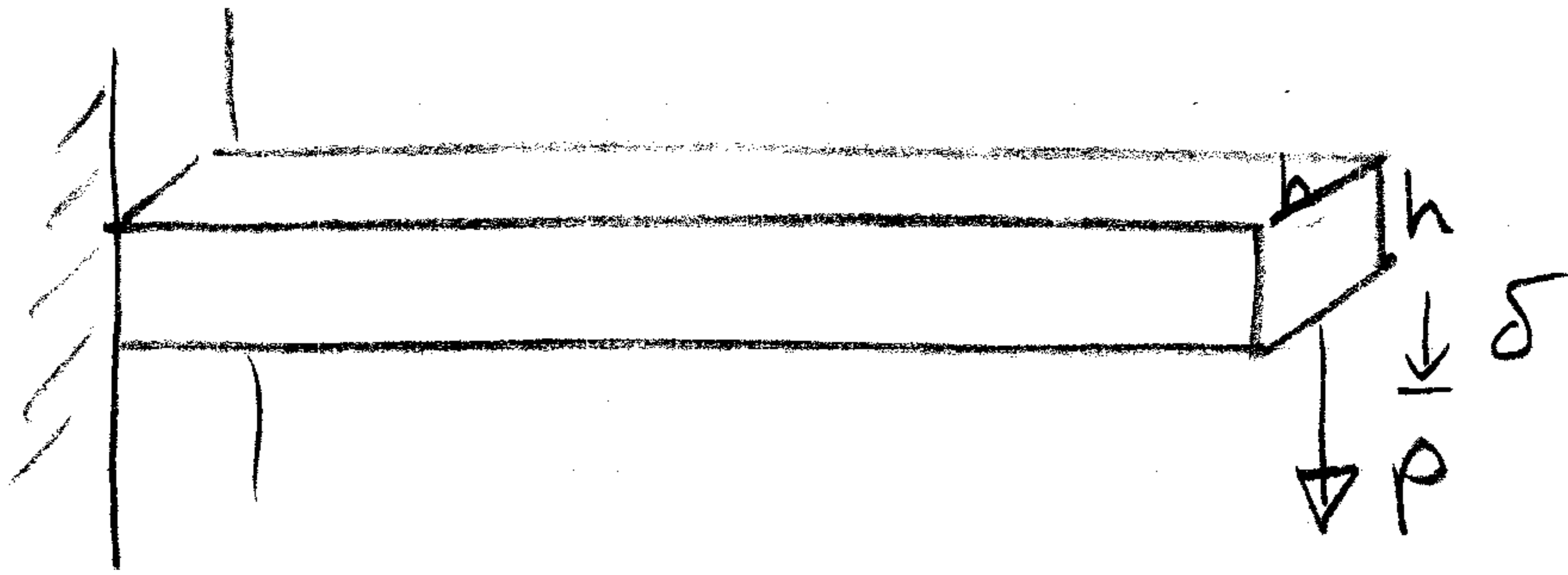
$$\text{for } \frac{-b \pm \sqrt{b^2 - 4ac}}{2a} = \frac{-1 \pm \sqrt{1 - 4 \times \frac{1}{6} \times \frac{19}{18}}}{-\frac{2}{6}}$$

Root between L and $3L = x = 1.37L$

$$= \delta = \frac{PL^2}{EI} \left(\frac{-1}{5} \right) = -\frac{PL^2}{5EI} \quad \ominus$$

Note this does not occur at $x = L$ where
 $M = \text{Maximum}$.

M6



For cantilever beam $\delta = \frac{PL^3}{3EI}$, $\sigma_{max} = \frac{PL}{I} \frac{h}{2}$

$$I = \frac{1}{12} h^4$$

Mass of beam = $\rho LA = \rho L h^2 = m$
density

Max stiffness of beam = $\frac{P}{\delta} = \frac{3EI}{L^3} = \frac{3}{12} \frac{h^4 E}{L^3}$

eliminate h^2 $h^2 = \frac{m}{\rho L}$

\therefore stiffness = $\frac{P}{\delta} = \frac{1}{4} \left(\frac{m}{\rho L} \right)^2 E = K$

max stiffness for given mass (or vice-versa)

requires max $\frac{E}{\rho^2}$ \leftarrow

For strength: $P_{max} = \frac{2\sigma_f I}{Lh} = \frac{\sigma_f h^3}{6L}$

$$P_{max} = \frac{\sigma_f h^3}{6L}$$

mass of beam, $m = \rho h^2 L$

eliminate h

$$P_{max} = \frac{\sigma_f}{6L} \left(\frac{m}{\rho L} \right)^{3/2}$$

so max P_{max} for given mass requires

max $\frac{\sigma_f}{\rho^{3/2}}$ \leftarrow

	E/GPa	σ_f / MPa	$\rho / kg/m^3$	E/ρ^2	$\sigma_f/\rho^{3/2}$
steel	193	1435	7900	3092	2044
Al	71	350	2800	9056	2362
Ti	120	850	4500	5926	2815
CFRP	70	700	1500	3111	12050 \leftarrow
wood	12	50	600	3333	3402
Si	410	300	3000	4500 $\leftarrow?$	1826

- c) a) SIC has the highest E/e^2 , but may be impractical because of its low toughness
better choices might be wood or CFRP
- b) CFRP has the highest $\sigma_f/e^{3/2}$

Other factors - toughness, environmental durability, cost

- d) I & box sections are structurally efficient because they move material away from the neutral axis of the beam.

The key issue in selecting the shape of a beam is the ability to manufacture the shape in question.