

$$a) C_D = c_d + \left(\frac{C_L^2}{\pi R}\right) \equiv C_{Di}, \quad \frac{C_L}{C_D} = \frac{C_L}{c_d + C_L^2/\pi R}$$

$$\text{Maximize } \frac{C_L}{C_D} \rightarrow \text{set } \frac{d}{dC_L} \left(\frac{C_L}{C_D}\right) = 0$$

$$\frac{d}{dC_L} \left(\frac{C_L}{c_d + C_L^2/\pi R}\right) = \frac{(c_d + C_L^2/\pi R) - C_L(2C_L/\pi R)}{(c_d + C_L^2/\pi R)^2} = \frac{c_d - C_L^2/\pi R}{(\quad)^2} = 0$$

$$\Rightarrow c_d = \frac{C_L^2}{\pi R} \quad \text{or} \quad \boxed{C_L = \sqrt{c_d \pi R}} \quad \text{at max } \frac{C_L}{C_D}$$

$$\text{At this point, } \boxed{C_D = c_d + \frac{(\sqrt{c_d \pi R})^2}{\pi R} = 2c_d} \quad C_{Di} = c_d$$

$$b) \text{ Set } \frac{d}{dC_L} \left(\frac{C_L^{3/2}}{C_D}\right) = 0$$

$$\frac{d}{dC_L} \left(\frac{C_L^{3/2}}{c_d + C_L^2/\pi R}\right) = \frac{\frac{3}{2} C_L^{1/2} (c_d + C_L^2/\pi R) - C_L^{3/2} (2C_L/\pi R)}{(c_d + C_L^2/\pi R)^2} = C_L^{1/2} \frac{\frac{3}{2} c_d - \frac{1}{2} \frac{C_L^2}{\pi R}}{(\quad)^2} = 0$$

$$\Rightarrow 3c_d = \frac{C_L^2}{\pi R} \quad \text{or} \quad \boxed{C_L = \sqrt{3c_d \pi R}} \quad \text{at max } \frac{C_L}{C_D}$$

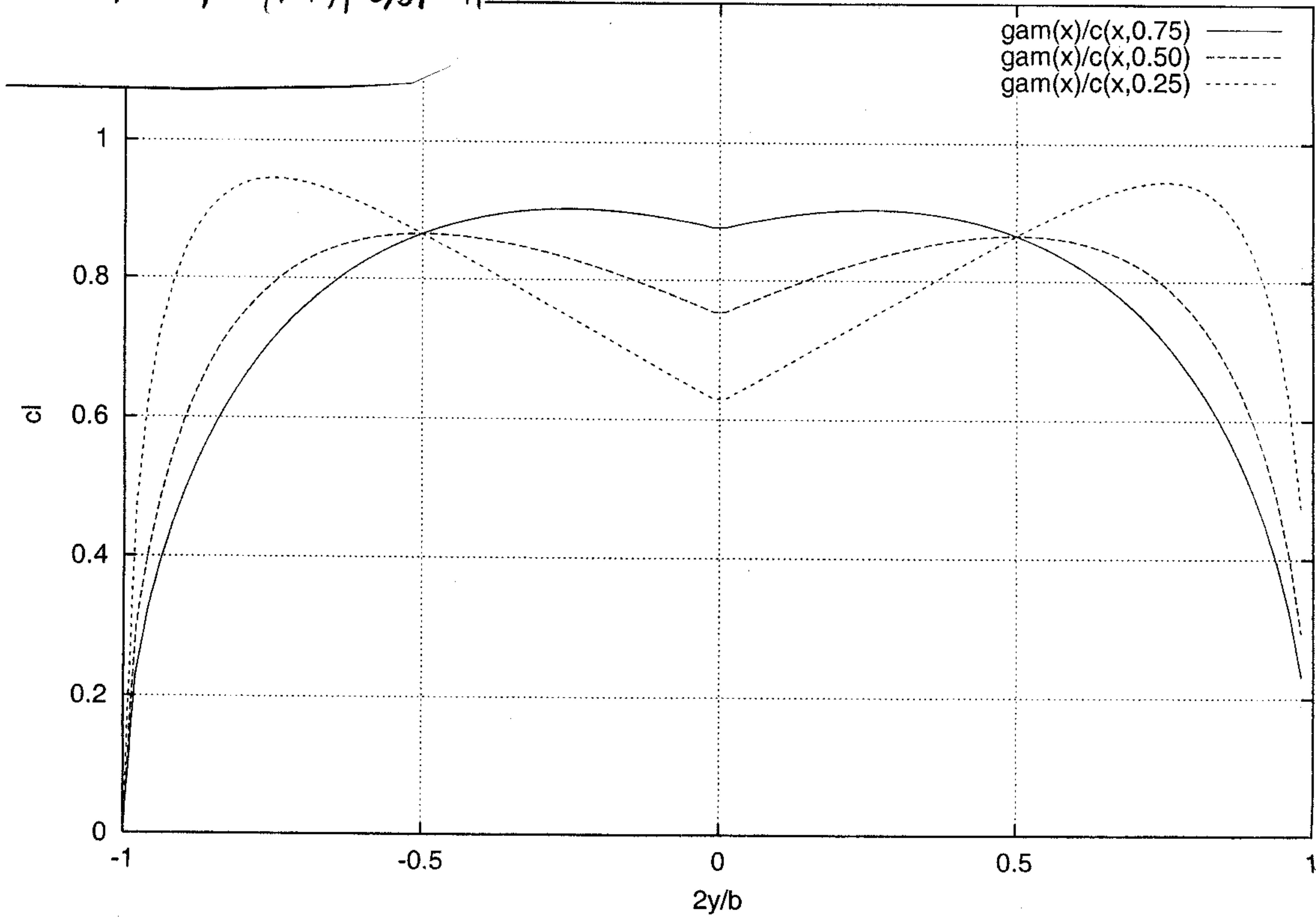
$$\text{At this point, } \boxed{C_D = c_d + \frac{(\sqrt{3c_d \pi R})^2}{\pi R} = 4c_d} \quad C_{Di} = 3c_d$$

a) Linear taper from  $C_r$  to  $C_t$ :  $C(y) = C_r + (C_t - C_r) \left| \frac{2y}{b} \right|$   
 or  $C(y) = \frac{2}{1+r} C_{avg} \left[ 1 - (1-r) \left| \frac{2y}{b} \right| \right]$

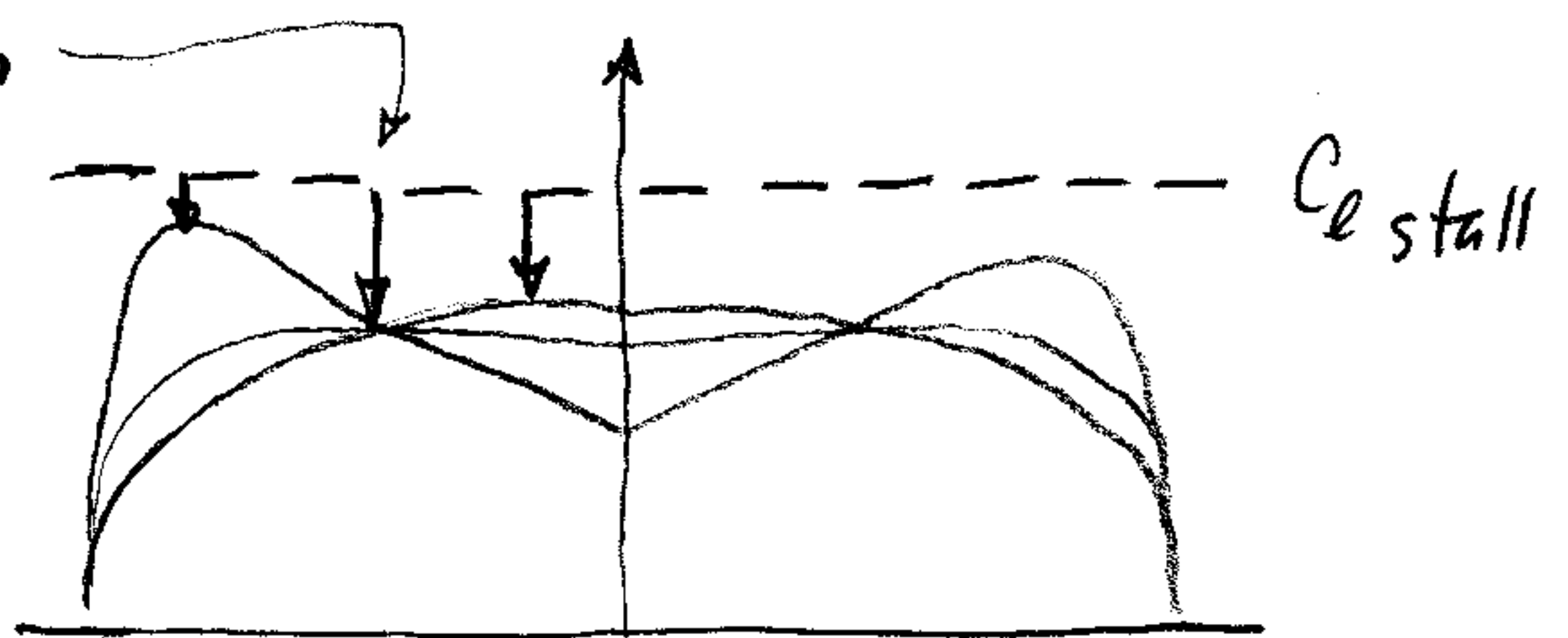


b)  $\Gamma(y) = \frac{1}{2} V C(y) C_z(y) \Rightarrow C_z(y) = \frac{2\Gamma(y)}{V_\infty C(y)} = \frac{2\Gamma_0}{V_\infty C_{avg}} \frac{1+r}{2} \frac{\sqrt{1 - (2y/b)^2}}{1 - (1-r)|2y/b|}$

Plots of  $\frac{\sqrt{1 - (2y/b)^2}}{1 - (1-r)|2y/b|}$

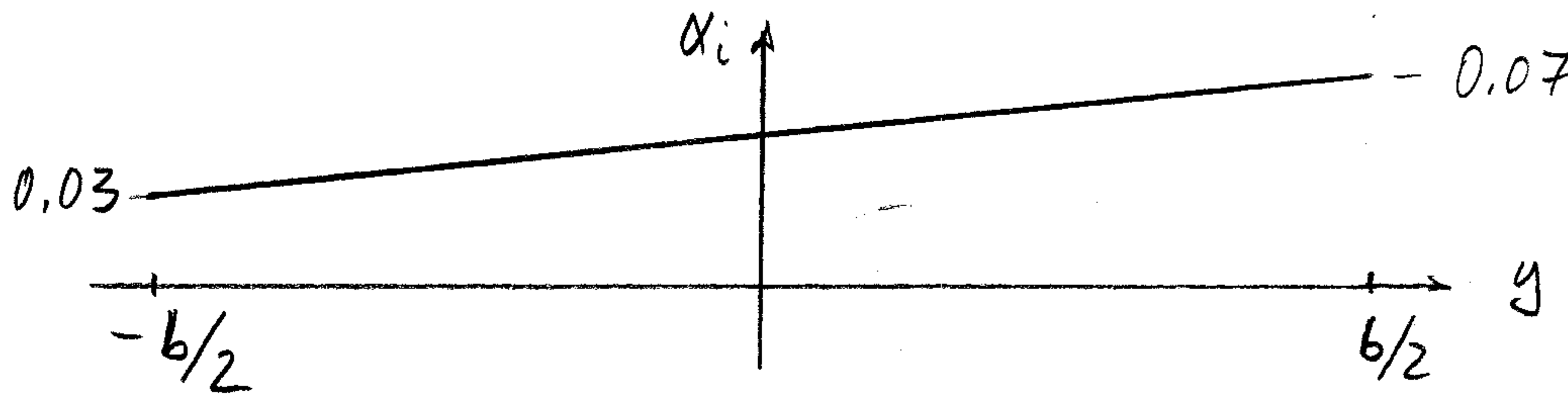


The middle  $r=0.50$  case has the smallest  $(C_l)_{max} / C_L$ , so it has the largest stall margin



$$a) \alpha_i = \sum n A_n \frac{\sin n\theta}{\sin \theta} = A_1 + A_2 \frac{\sin 2\theta}{\sin \theta} = A_1 + A_2 2 \cos \theta$$

but since  $\cos \theta = \frac{2y}{b}$ ,  $\alpha_i = 0.05 + 0.02 \cdot \left(\frac{2y}{b}\right)$



$$b) M_{roll} = \int_{-b/2}^{b/2} \rho V_{\infty} \Gamma y dy$$

Using  $\Gamma = 2b_{\infty} V_{\infty} (A_1 \sin \theta + A_2 \sin 2\theta)$

$$y = \frac{b}{2} \cos \theta$$

$$dy = -\frac{b}{2} \sin \theta d\theta$$

$$M_{roll} = \int_{\pi}^0 \rho V_{\infty} 2b V_{\infty} (A_1 \sin \theta + A_2 \sin 2\theta) \frac{b}{2} \cos \theta \left(-\frac{b}{2} \sin \theta d\theta\right)$$

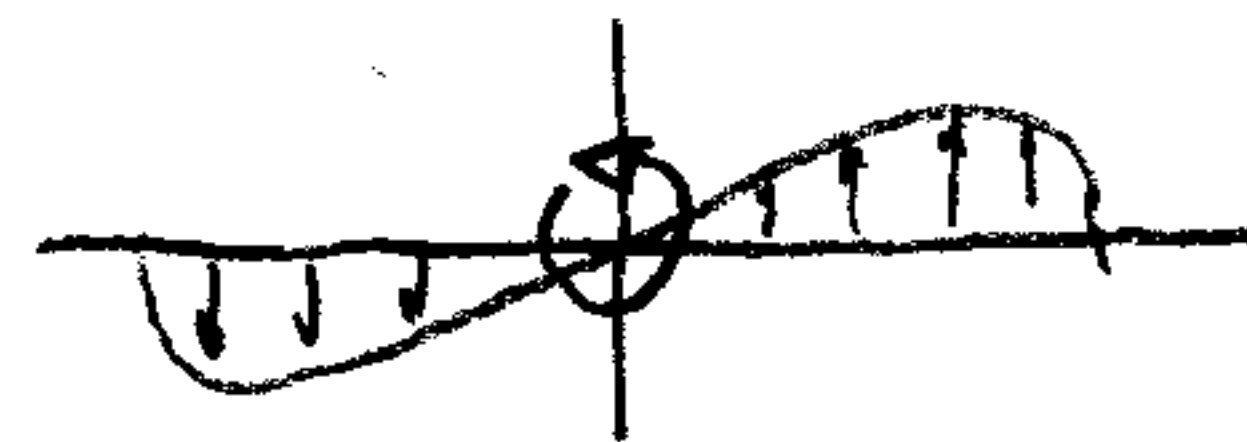
$$= \frac{1}{2} \rho V_{\infty}^2 b^3 \int_0^{\pi} (A_1 \sin \theta + A_2 \sin 2\theta) \frac{1}{2} \sin 2\theta d\theta$$

$$= \frac{1}{2} \rho V_{\infty}^2 b^3 \left[ \frac{1}{2} A_1 \int_0^{\pi} \sin \theta \sin 2\theta d\theta + \frac{1}{2} A_2 \int_0^{\pi} \sin 2\theta \sin 2\theta d\theta \right]$$

since  $\cos \theta \sin \theta = \frac{1}{2} \sin 2\theta$

$$M_{roll} = \frac{\pi}{8} \rho V_{\infty}^2 b^3 A_2$$

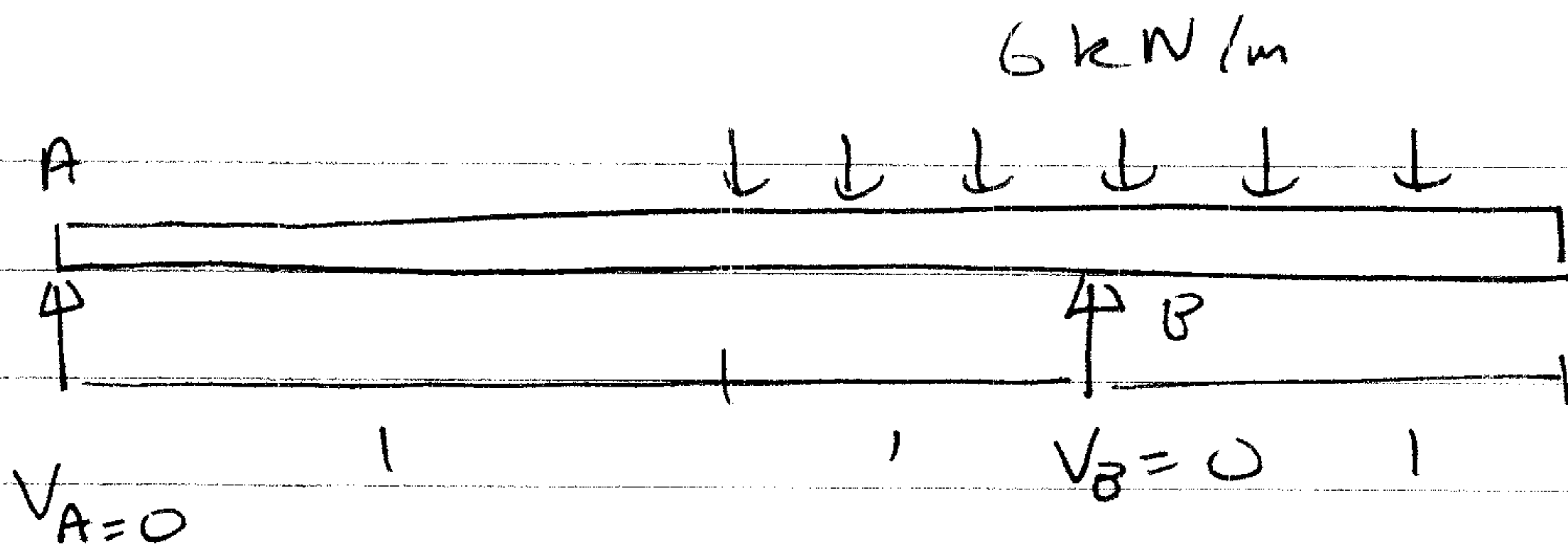
only  $A_2$  term contributes to rolling moment



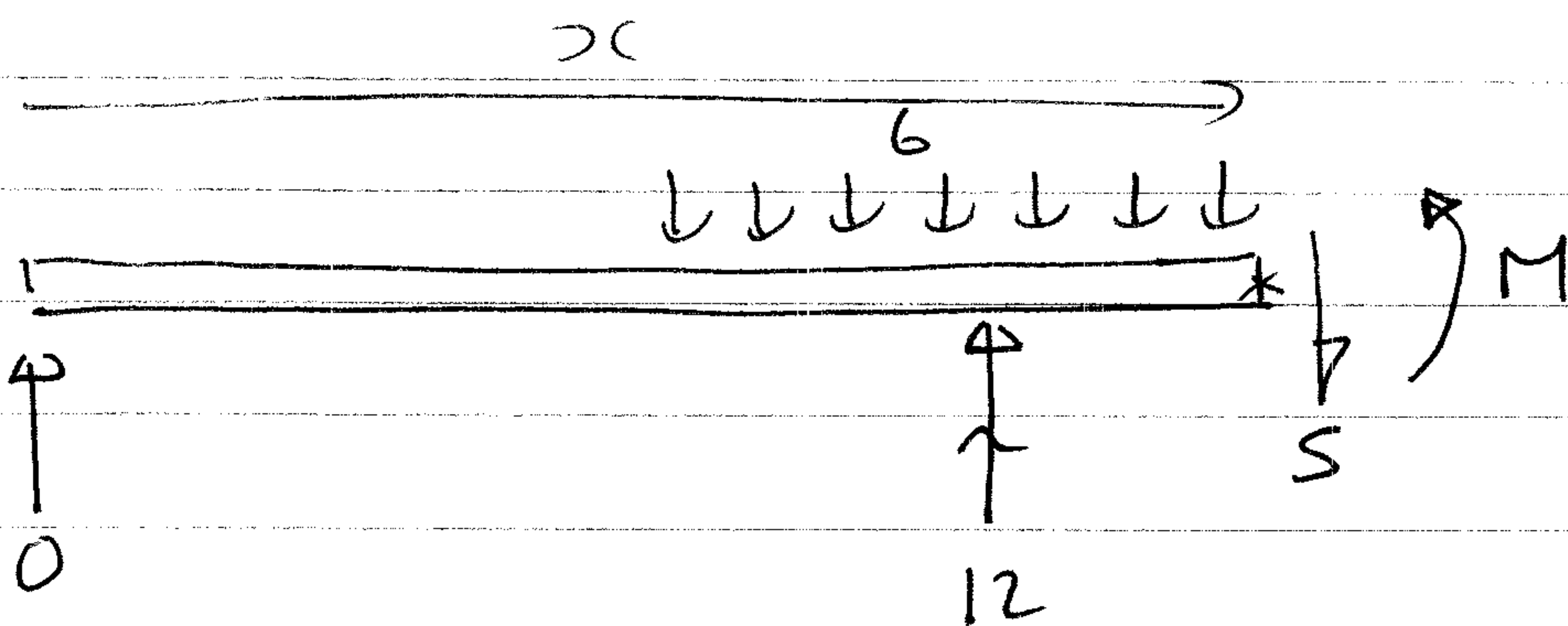


M7/M8

a)



Apply Macaulay's method



$$\sum M_x = 0: M + \frac{6\{x-1\}^2}{2} - 12\{x-2\} = 0$$

$$M = -\frac{3\{x-1\}^2}{2} + 12\{x-2\}$$

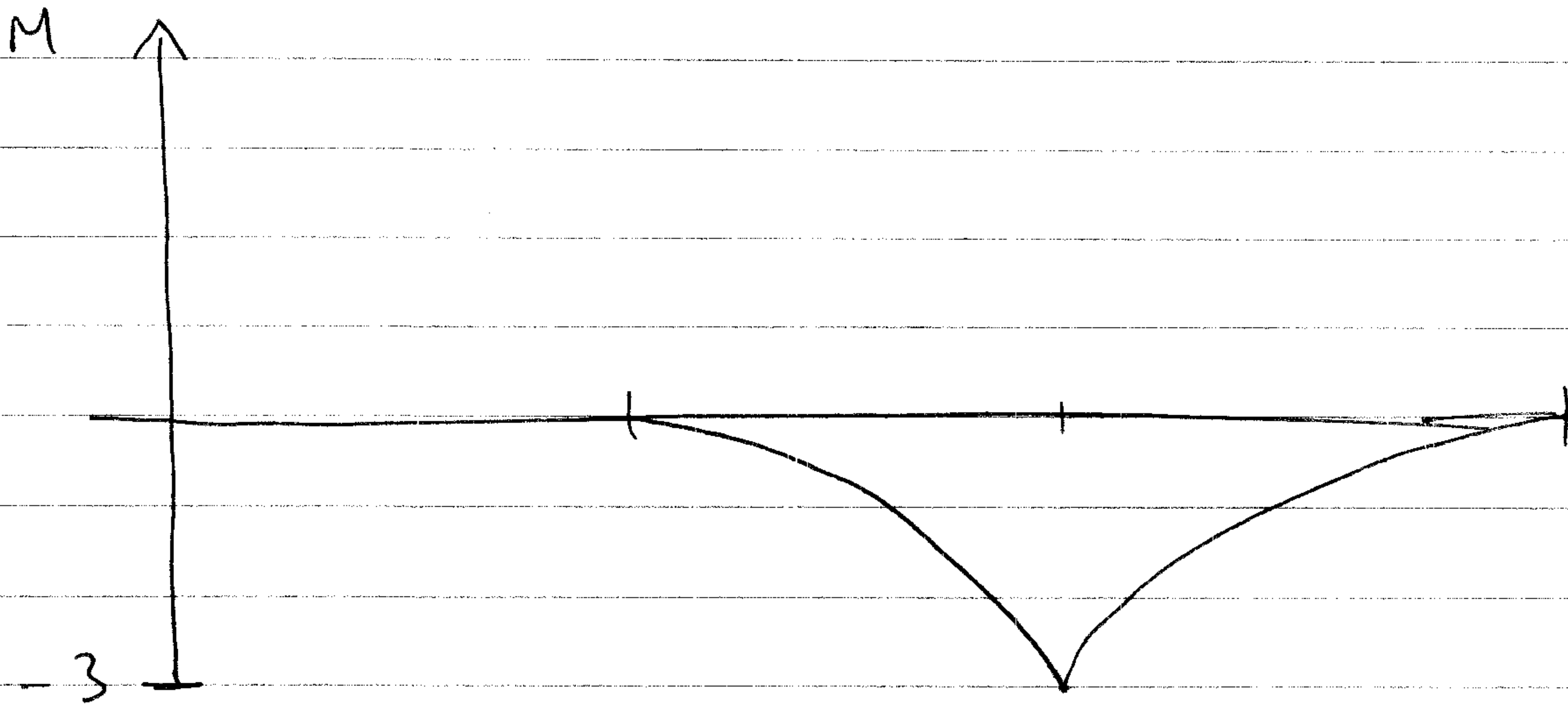
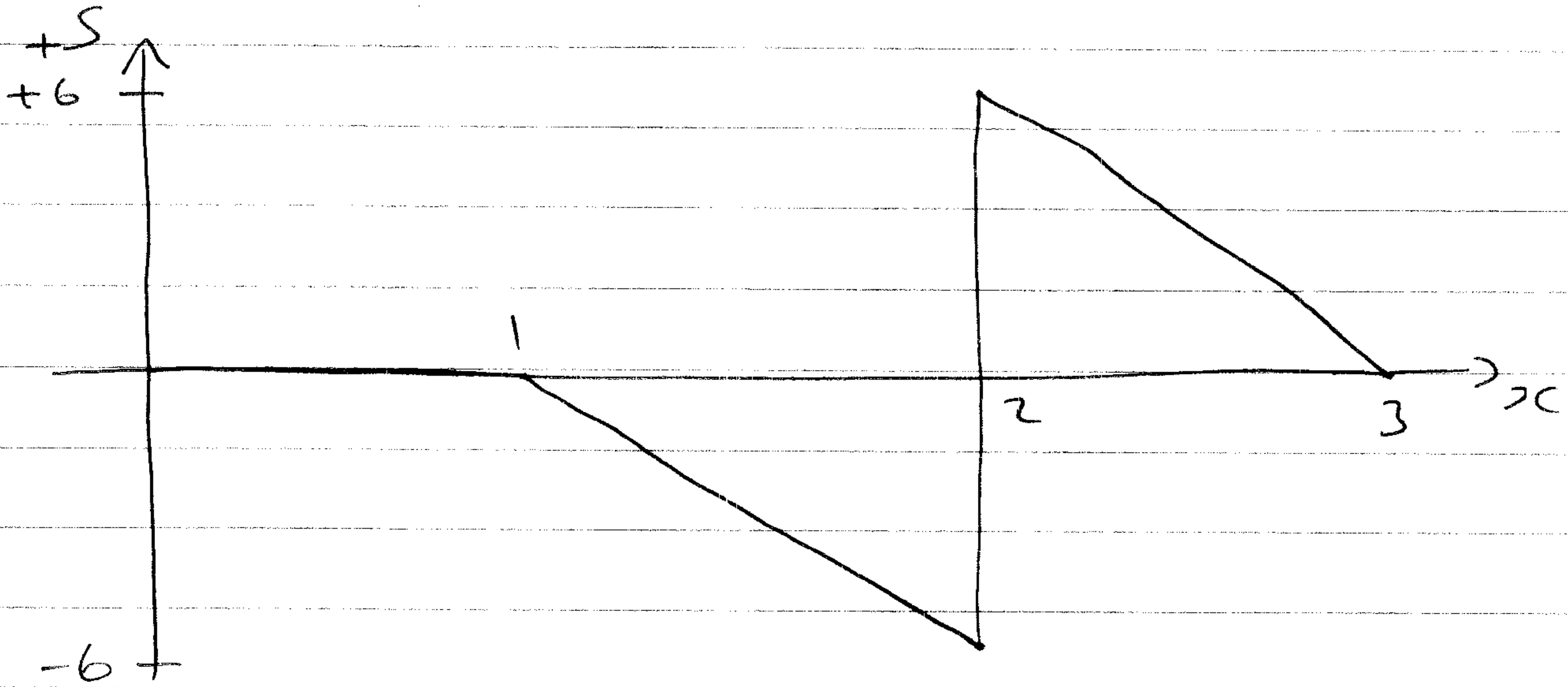
~~$$\sum F_z = 0:$$~~

~~$$-S - 6\{x-1\} + 12 = 0$$~~

~~$$S = -6\{x-1\} + 12$$~~

from  $MP = 2EI \frac{d^2y}{dx^2}$        $S = \frac{dM}{dx}$

$$S = \frac{dM}{dx} = +6\{x-1\} + 12\{x-2\}^0$$



6)

$$EI \frac{d^2 w}{dx^2} = -3 \{x-1\}^2 + 12 \{x-2\}$$

$$EI \frac{dw}{dx} = \frac{-3 \{x-1\}^3}{3} + \frac{12 \{x-2\}^2}{2} + A$$

$$EI w = -\frac{\{x-1\}^4}{4} + 2 \frac{\{x-2\}^3}{3} + Ax + B$$

apply boundary conditions

$$w = 0 \quad @ \quad x = 0, \quad x = 2.$$

$$@ \quad x = 0 \quad \Rightarrow \quad B = 0 \quad \{x-1\}, \{x-2\} = 0$$

$$@ \quad x = 2 : \quad -\frac{1}{4} + A \cdot 2 = 0$$

$$\therefore A = \frac{1}{8}$$

$$EI w = -\frac{\{x-1\}^4}{4} + 2 \frac{\{x-2\}^3}{3} + \frac{x}{8} \leftarrow$$

maximum deflection occurs either

for  $0 < x < 2$  or at  $x = \underline{3}$



$$\text{for } 0 < x < 1 \quad EI \frac{dw}{dx} = \frac{A}{8} \quad \therefore \text{no min/max}$$

$$\text{for } 1 < x < 2$$

$$EI \frac{dw}{dx} = - \{x-1\}^3 + \frac{A}{8}$$

$$-x^3 + 3x^2 - 3x + \frac{9}{8} + \frac{1}{8} = 0$$

$$\text{Solution } x = 1.04$$

$$\Rightarrow EIw = \frac{-(0.04)^4}{4} + \frac{1.04}{8} = \underline{+0.13} \text{ EI} \Leftarrow$$

$$w = \frac{0.13}{EI} \Leftarrow \text{ @ } x = 1.04 \text{ m.}$$

This is the maximum up bending

$$w = \frac{0.13 \times 10^3 \text{ kNm}}{3.54 \times 10^6} = 36.7 \text{ } \mu\text{m} \Leftarrow$$

for  $x = 3 \text{ m}$

$$EIW = -\frac{(2^4)}{4} + 2(1)^3 + \frac{3}{8}$$

$$EIW = -4 + 2 + \frac{3}{8}$$

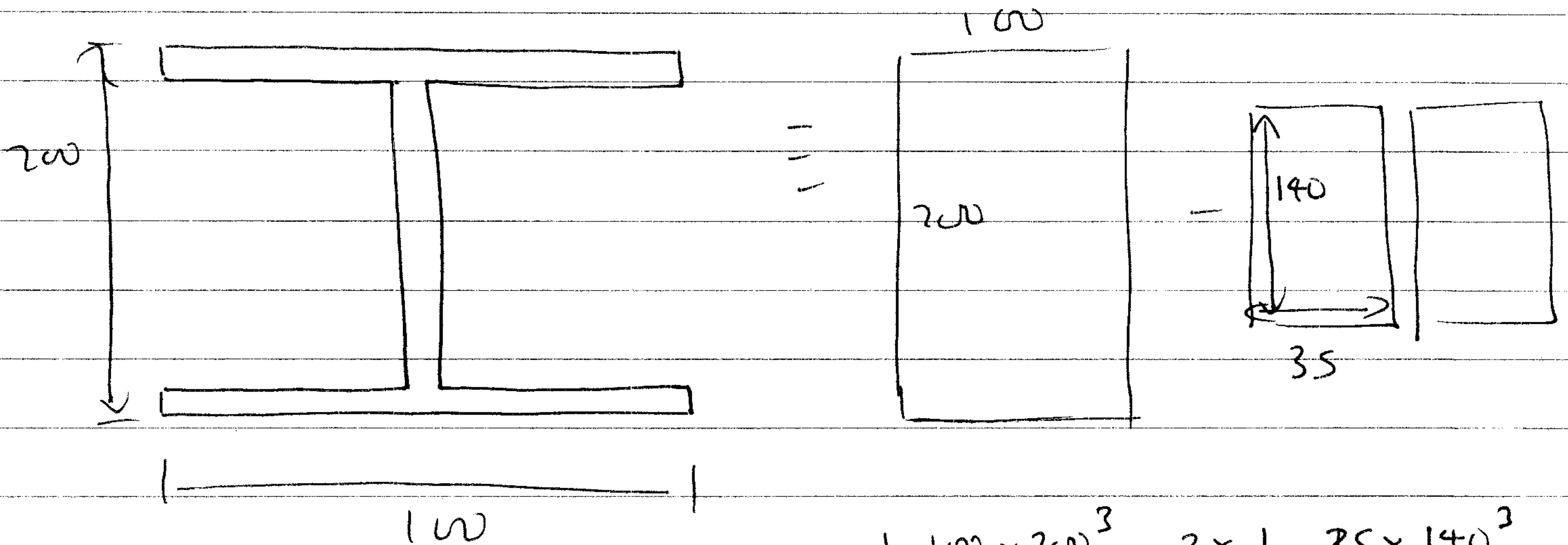
$$W = \frac{-13}{8EI} \leftarrow$$

$$= \frac{-1.625}{EI} \Leftarrow$$

This is the maximum downward bending @  $x = 3 \text{ m}$ .

$$W = -459 \mu\text{m} \Leftarrow \quad (0.459 \text{ mm})$$

Calculation of  $I$



$$= \frac{1}{12} 100 \times 200^3 - 2 \times \frac{1}{12} \times 35 \times 140^3$$

$$= 50.7 \times 10^6 \text{ mm}^4 \Leftarrow$$

$$EI = 70 \times 10^9 \times 50.7 \times 10^6 \times 10^{-12} = 3.54 \times 10^6 \text{ Nm}^2 \Leftarrow$$

$$= -459$$



c)

$$\sigma_{xx} = -\frac{Mz}{I}$$

maximum at max bending moment,  
max  $z$  @  $x = z$ ,  $M = 3 \text{ kNm}$   
@  $\frac{h}{2} = \pm 100 \text{ mm}$

$$\therefore \sigma_{xx} = \frac{3 \times 10^3 \times 100 \times 10^{-3}}{50.7 \times 10^6 \times 10^{-12}} = 5.9 \text{ MPa}$$

No danger of yield due to tensile stresses

$$\sigma_{xz} = -\frac{SQ}{Ib}$$

$$Q = \int_z^{h/2} z \, dA$$

will be maximum at center of beam. i.e. for  $z = 0$

$$Q = \int_0^{70} z \cdot 30 \, dz + \int_{70}^{100} z \cdot 100 \, dz$$

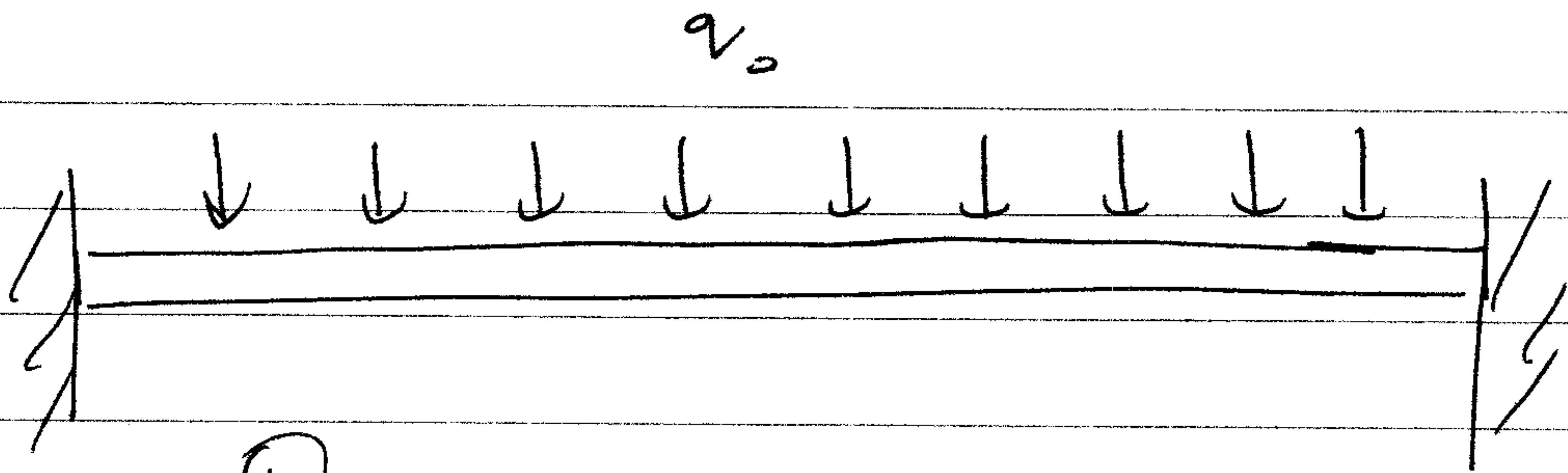
$$= \left[ \frac{30z^2}{2} \right]_0^{70} + \left[ \frac{100z^2}{2} \right]_{70}^{100} = 328.5 \times 10^3 \text{ mm}^3$$

$$\therefore \sigma_{xz} = -\frac{6 \times 10^3 \times 328.5 \times 10^3 \times 10^{-9}}{50.7 \times 10^6 \times 10^{-12} \times 30 \times 10^{-3}} = 1.29 \times 10^6 \text{ MPa}$$

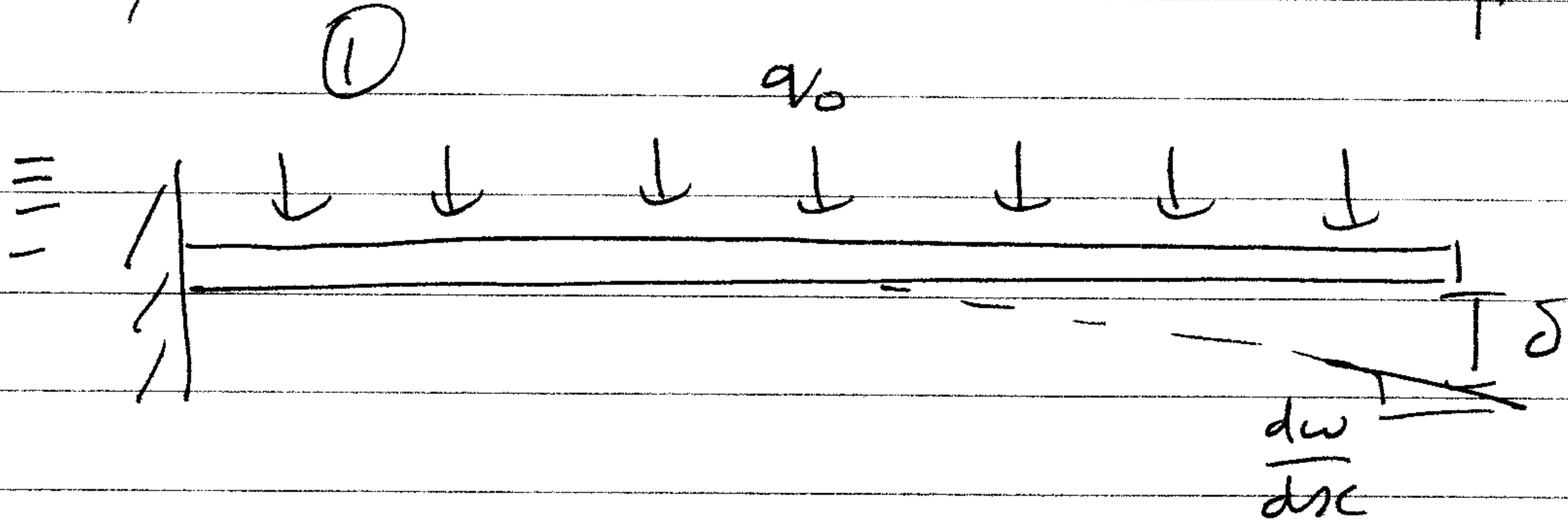
Note I beam Shear stress  $\approx$  bending stress

Yield is not a problem.  $\Leftarrow$

M9



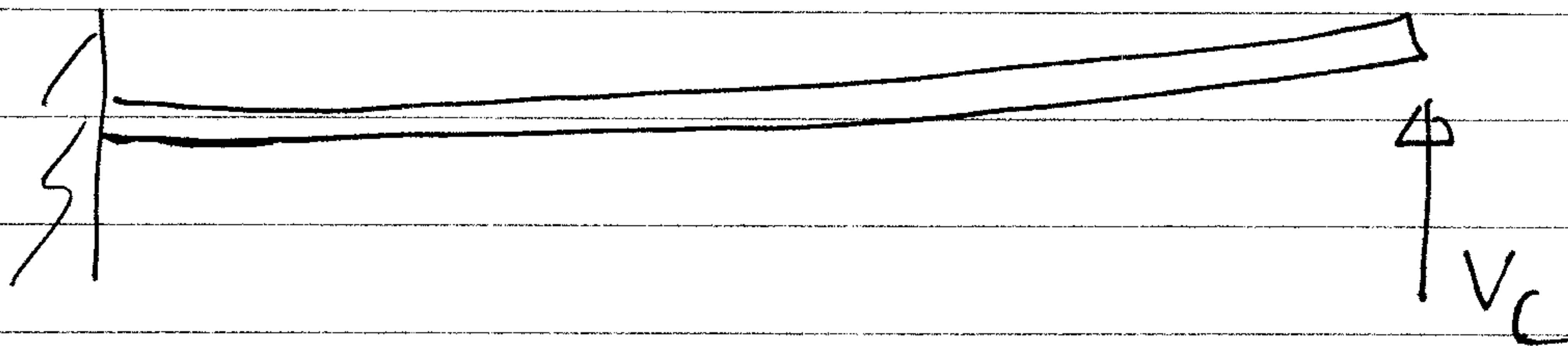
Statically indet.



$$\delta^1 = \frac{q_0 L^4}{8EI}$$

$$\frac{dw^1}{dsc} = \frac{q_0 L^3}{6EI}$$

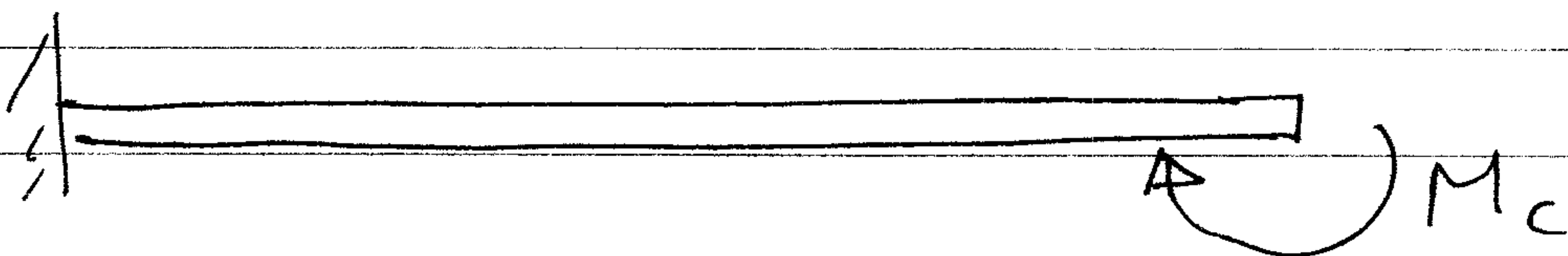
②



$$\delta^2 = -\frac{V_c L^3}{3EI}$$

$$\frac{dw^2}{dsc} = -\frac{V_c L^2}{2EI}$$

③



$$\delta^3 = \frac{M_c L^2}{2EI}$$

$$\frac{dw^3}{dsc} = \frac{M_c L}{EI}$$

Match B.C.'s at RHE,  $x = L$   $w = 0$   $\frac{dw}{dx} = 0$

$$w = 0: \delta' + \delta^2 + \delta^3 = 0$$

$$\frac{L^2}{EI} \left( \frac{q_0 L^2}{8} - \frac{V_c L}{3} + \frac{M_c}{2} \right) = 0$$

$$12M_c - 8V_c L + 3q_0 L^2 = 0 \quad (1)$$

$$\frac{dw}{dx} = 0 \quad \frac{d\omega^1}{dx} + \frac{d\omega^2}{dx} + \frac{d\omega^3}{dx} = 0$$

$$\frac{L}{EI} \left( \frac{q_0 L^2}{6} - \frac{V_c L}{2} + M_c \right) = 0$$

$$6M_c - 3V_c L + q_0 L^2 = 0 \quad (2)$$

Multiply (2) by 2 and subtract from (1)

$$0 - 2V_c L + 2q_0 L^2 = 0$$

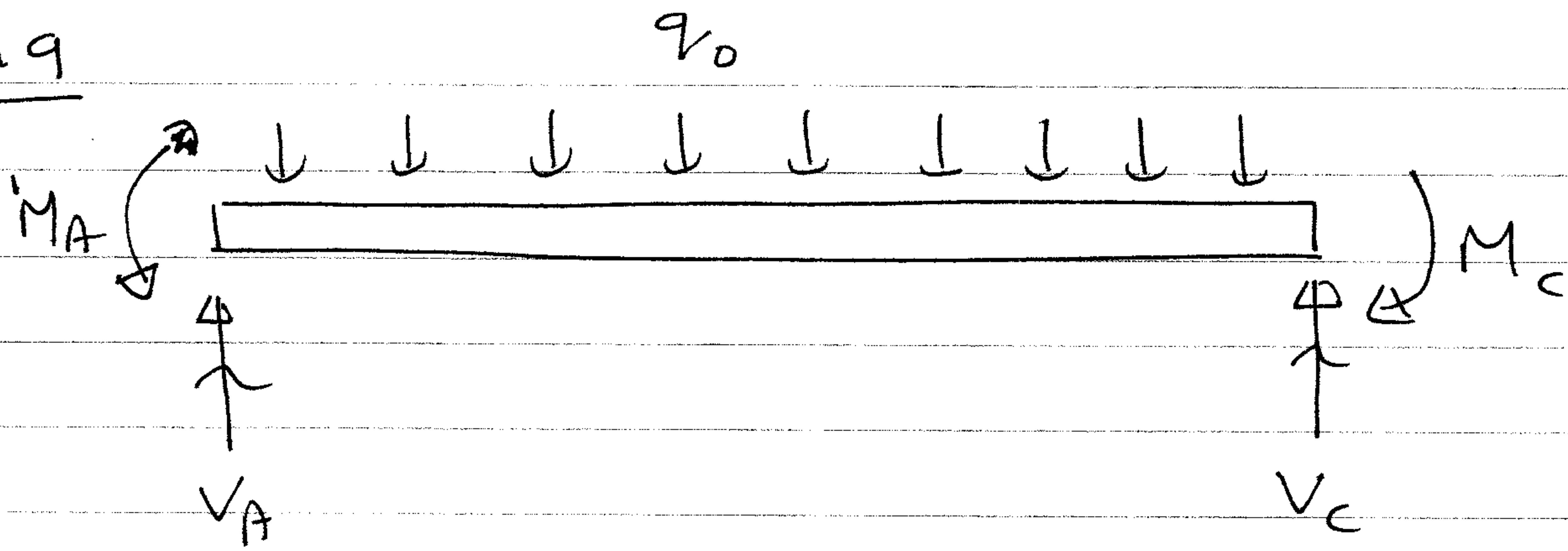
$$V_A = V_c = \frac{q_0 L}{2} = \frac{q_0 L}{2} \in (!!)$$

Substitute back into (2)

$$6M_c - 3 \frac{q_0 L^2}{2} + q_0 L^2 = 0 : M_c = M_A = \frac{q_0 L^2}{12} \in$$

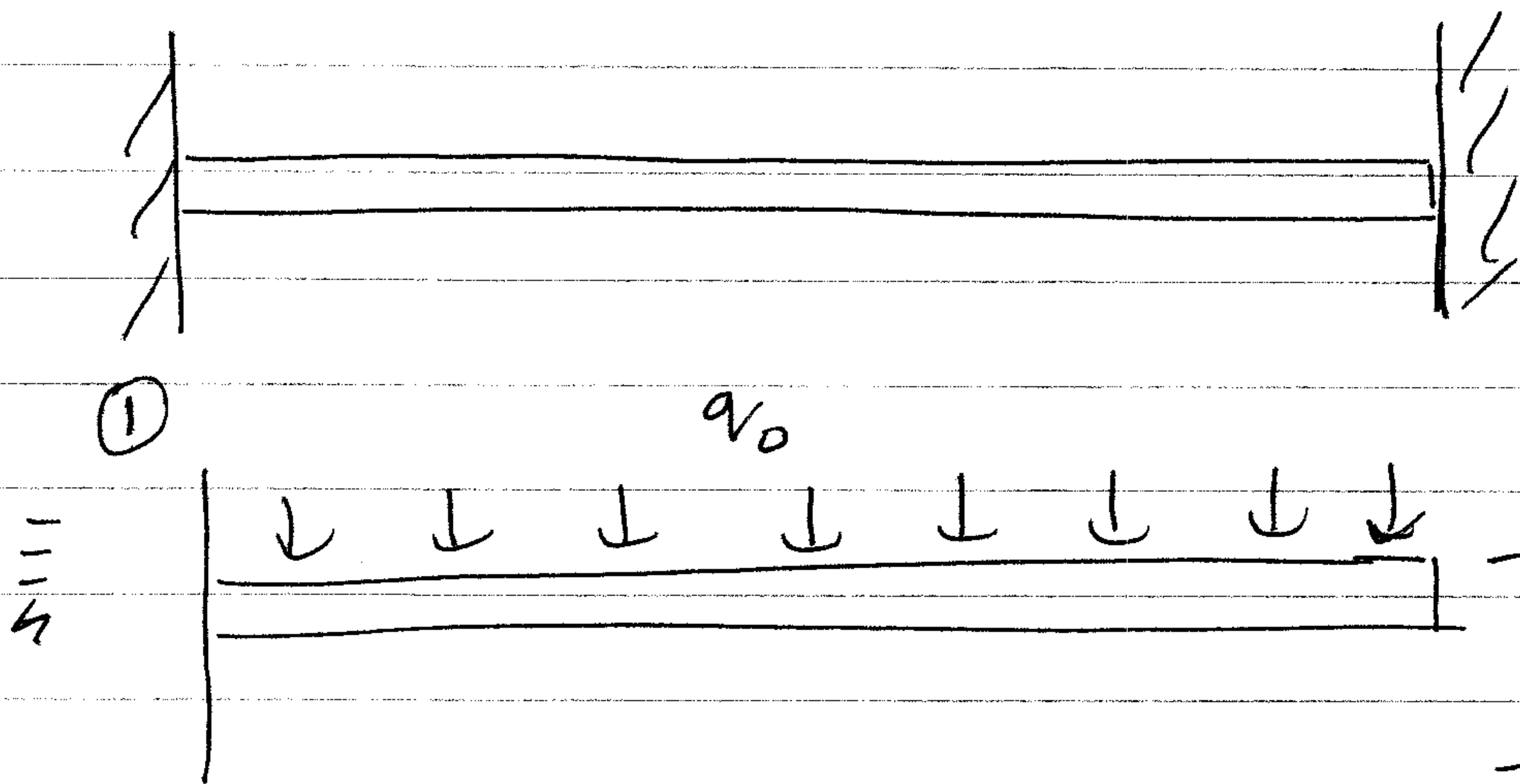


Mq



By symmetry  $M_A = M_C$   
 $V_A = V_C$ .

But-



$$\delta = \frac{q_0 L^4}{8EI} \frac{dw}{dx}$$

$dw$