

## 1. ACE Core

Consider the unsteady partial differential equation

$$\frac{\partial u}{\partial t} = \frac{\partial^2 u}{\partial x^2} + f(x, t) \quad (1)$$

with a given forcing function  $f(x, t)$  and initial condition  $u(x, t = 0) = u_0(x)$ . The spatial domain is **periodic** in  $[0, 2\pi]$ . It is discretized by a **non-uniform grid**  $0 = x_0 < x_1 < \dots < x_{K-1} < 2\pi$ .

In this question, we focus on spatial discretization. A forward Euler time integration scheme is assumed throughout the problem.

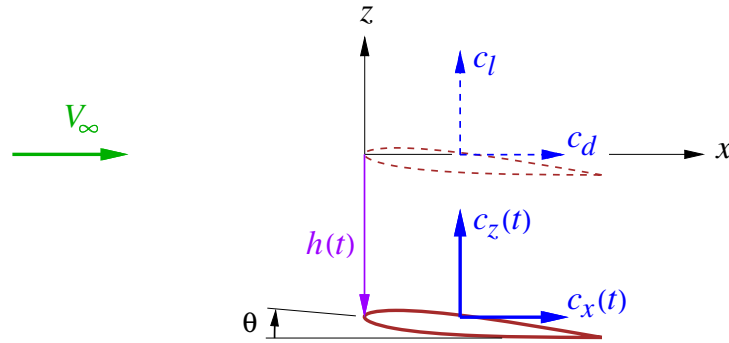
- (a) Derive a consistent finite difference scheme for the PDE, using forward Euler for time discretization. Please only use a stencil of three grid points  $x_{i-1}, x_i$  and  $x_{i+1}$ . Show the order of accuracy of your spatial approximation.
- (b) Consider the  $L_\infty$ -norm of the numerical solution when  $f = 0$ . What **sufficient** conditions can you impose to ensure  $L_\infty$ -stability?
- (c) Consider the numerical solution at a fixed time  $t_f$ . Does the error in the numerical solution go to zero as  $N$  (the number of time steps) goes to infinity,  $\Delta t$  goes to zero (keeping  $N\Delta t = t_f$ ), and  $K$  (the number of spatial grid points) goes to infinity? Why or why not? If not, what sufficient conditions can you impose to ensure that the numerical solution goes to zero?

## 2. Fluid Mechanics

A 2D airfoil of chord  $c$  is placed in an incompressible flow which has freestream speed  $V_\infty$  along the horizontal  $x$  axis. The airfoil maintains a small angle  $\theta$  above the  $x$  axis, while undergoing an unsteady sinusoidal vertical displacement

$$h(t) = h_1 \sin \omega t$$

where by convention  $h$  is positive down, and  $h_1$  is the half-amplitude of the displacement.



- (a) What are the physical boundary condition on the surface of the airfoil for  
 i) viscous flow, and ii) inviscid flow? Assume that the normal vector  $\hat{\mathbf{n}} = n_x \hat{\mathbf{x}} + n_z \hat{\mathbf{z}}$  is known everywhere on the airfoil surface.
- (b) If  $h_1 = 0$ , then this situation reduces to the usual steady 2D airfoil case (shown dashed), with angle of attack  $\alpha = \theta$ . For  $\alpha \ll 1$ , the drag and lift coefficients (always defined parallel and normal to the airfoil's onset flow), are then given by:

$$\frac{D'}{\frac{1}{2}\rho V_\infty^2 c} \equiv c_d = c_{d0} \quad (8)$$

$$\frac{L'}{\frac{1}{2}\rho V_\infty^2 c} \equiv c_\ell = 2\pi\alpha \quad (9)$$

For the more general case  $h_1 \neq 0$ , what are the requirements on the problem parameters  $\rho, V_\infty, c, h_1, \omega$  such that the flow is quasi-steady, and equations (8) and (9) are valid?

- (c) Assuming the requirements above hold, determine the instantaneous force coefficients

$$c_x(t) \equiv \frac{F_x(t)}{\frac{1}{2}\rho V_\infty^2 c}$$

$$c_z(t) \equiv \frac{F_z(t)}{\frac{1}{2}\rho V_\infty^2 c}$$

along the horizontal and vertical axes  $x, z$ . Also determine the corresponding time-averaged force coefficients  $\bar{c}_x$  and  $\bar{c}_z$  defined over any integer number of oscillation periods. You may use small-angle approximations.

- (d) Consider the case where the condition(s) for the quasi-steady assumption in part (a) do not quite hold. In which direction will the resulting  $c_d(t)$  and  $c_\ell(t)$  forces change from their quasi-steady values you obtained for part (b)?

### 3. Numerical Linear Algebra

An iterative method is used to solve the following linear system,

$$Ax = b \tag{10}$$

where  $A \in R^{N \times N}$ , and  $x, b \in R^N$ . You may assume that  $A$  is a full rank matrix. The iterative method is of the following form,

$$Mx^{n+1} = Nx^n + b \tag{11}$$

where  $A = M - N$ . You may assume that  $M$  is also a full rank matrix.

- (a) Define the error  $e^n \equiv x - x^n$  (i.e.  $x$  being the exact solution). Determine  $G$  which is the matrix relating  $e^n$  and  $e^{n+1}$ , specifically,

$$e^{n+1} = Ge^n \tag{12}$$

Please express your answer for  $G$  in terms of  $M$  and  $A$  (not  $N$ ).

- (b) Where in the complex plane must the eigenvalues of  $M^{-1}A$  be such that  $\|e^n\| \rightarrow 0$  as  $n \rightarrow \infty$  for any initial guess  $x^0$ ?
- (c) The residual at iteration  $n$  is defined as  $r^n \equiv b - Ax^n$ .  $H$  is the matrix relating  $r^n$  and  $r^{n+1}$ , specifically,

$$r^{n+1} = Hr^n \tag{20}$$

How is  $H$  related to  $G$ ? How are the eigenvalues of  $H$  related to the eigenvalues of  $G$ ?

- (d) A difficult aspect of iterative methods is to determine when to stop iterating based on a desired error level. For example, suppose that the desired error tolerance is  $\tau_e > 0$  such that  $\|e^n\|/\|x\| \leq \tau_e$  and  $\|\cdot\|$  is a vector norm. Since the exact solution  $x$  and therefore the error  $e^n$  are unknown, this criteria cannot generally be enforced. Alternatively, suppose that a residual criteria is used as a stopping criteria. Specifically, the iterative method is terminated when  $\|r^n\|/\|b\| \leq \tau_r$  for some positive  $\tau_r$ . Determine lower and upper bounds on  $\|e^n\|/\|x\|$  in terms of  $\|r^n\|/\|b\|$  and  $\kappa(A) = \|A\| \|A^{-1}\|$  (i.e. the condition number of  $A$  where the matrix norms are those induced by the vector norms).

#### 4. Optimization Methods

Consider the least squares problem

$$\underset{\theta}{\text{minimize}} \|X\theta - y\|_2^2, \quad (30)$$

corresponding to the over-constrained linear relationship

$$\begin{bmatrix} X \\ \end{bmatrix} \begin{bmatrix} \theta \\ \end{bmatrix} \approx \begin{bmatrix} y \\ \end{bmatrix}.$$

The analytical solution  $\theta^*$  to this problem satisfies

$$(X^T X)\theta^* = X^T y. \quad (31)$$

For intuition, you may assume the problem size  $X \in \mathbb{R}^{10,000 \times 100}$ ,  $y \in \mathbb{R}^{10,000}$ . In many settings, it is desirable that  $\theta$  not be too large. One way to achieve this is *ridge regression*, which penalizes  $\|\theta\|_2^2$ :

$$\underset{\theta}{\text{minimize}} \|X\theta - y\|_2^2 + \lambda \|\theta\|_2^2. \quad (32)$$

For now, assume  $\lambda > 0$  is a given parameter.

- (a) Determine the analytical optimum  $\theta^*$  to (32).
- (b) Assume you have available a reliable and efficient black box for solving (30). The inputs to this program are  $X$  and  $y$ , and the output is  $\theta^*$ . You want to use this program (unmodified) to solve (32), given a value for  $\lambda$ . How will you do it? Give expressions for  $X_{\text{aug}}$  and  $y_{\text{aug}}$ , the inputs you will provide to the solver.

In typical ridge regression solutions, many entries in  $\theta^*$  are close to, but not equal to, 0. Another regularizer, the *Lasso*, has sparsity-inducing effects: it drives some of the entries in  $\theta^*$  all the way to 0. The Lasso formulation is

$$\begin{aligned} &\underset{\theta}{\text{minimize}} \|X\theta - y\|_2^2 \\ &\text{subject to } \|\theta\|_1 \leq t, \end{aligned} \quad (33)$$

for some given  $t > 0$ , where  $\|\theta\|_1$  is the sum of the absolute values of the entries in  $\theta$ .

- (c) Formulate Lasso as a quadratic program (QP), an optimization problem with the form

$$\begin{aligned} &\underset{x}{\text{minimize}} \quad x^T Q x + c^T x \\ &\text{subject to} \quad A x \leq b. \end{aligned}$$

Give expressions for  $x$ ,  $Q$ ,  $c$ ,  $A$ , and  $b$ .

- (d) Sketch, on a plot, how the optimal cost  $\|X\theta - y\|_2^2$  of (33) changes as  $t$  is varied. Be sure to identify any salient features and values.