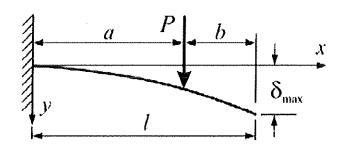


Results Cont.





Material properties of spring steel

ρ	E
kg/m³	- Gra
8000	193

Key Valve Geometric Information

Dport	D _{valve}	a a		thickness	stopper	\mathbf{x}_{tr}
mm	mm	mm	mm	mm	mm	mm
5.9	7	14	18	.15	1.8	1.24

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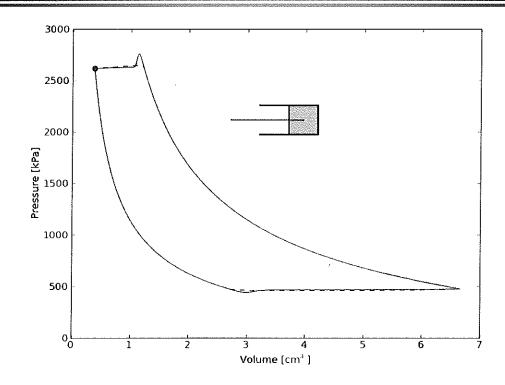
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Valve Influence







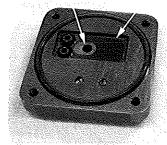
Appendix

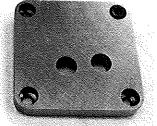


Discharge Port

Suction Valve

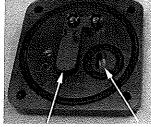
Valve Body Side 1





Valve Plenum Side 1

Valve Body Side 2



Valve Plenum Side 2

Discharge Valve and Stopper

Suction Port

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Nomenclature



- ρ Density, $\frac{kg}{m^3}$
- θ Crank Angle, rad
- ω Rotational Speed, $\frac{rad}{sec}$
- V Volume, m^3
- \dot{m} Mass flow rate, $\frac{kg}{\sec}$
- h Enthalpy, kJ/kg
- u Internal energy, kJ/kg
- \dot{Q} Heat transfer, kW
- T Temperature, K
- A Area, m^2

- x Displacement, m
- γ Specific Heat Ratio, -
- V Gas Velocity, $\frac{m}{\text{sec}}$
- μ Viscosity, Pa sec
- U Piston Velocity, $\frac{m}{\sec}$
- g Leakage gap, m
- M- Moving Mass, kg
- C_D Drag Coefficient, -
- k Stiffness, $\frac{N}{m}$
- E Modulus of elasticity, N_{m^2}
- I Bending moment of inertia, m^4



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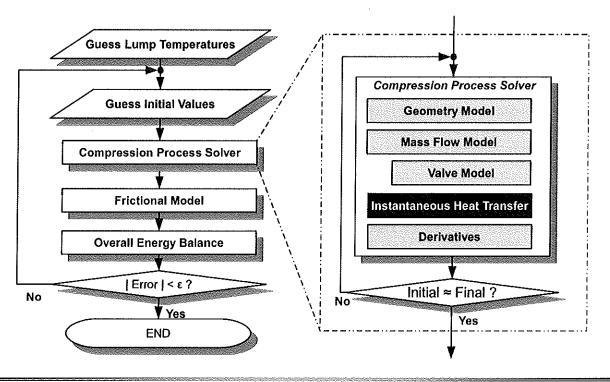


Heat Transfer



Heat Transfer





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Need for Heat Transfer



• Looking at the compression process equations

$$\frac{d\rho}{d\theta} = \frac{1}{V} \left[-\rho \frac{dV}{d\theta} + \frac{1}{\omega} (\dot{m}_{im} - \dot{m}_{out}) \right]$$

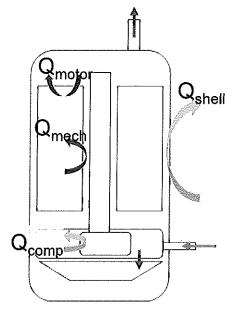
$$\frac{dT}{d\theta} = \frac{-\rho h \frac{dV}{d\theta} - \left(uV + \rho V \frac{\partial u}{\partial \rho} \right) \frac{\partial \rho}{\partial \theta} + \frac{1}{\omega} \left(\dot{Q} + \dot{m}_{in} h_{im} - \dot{m}_{out} h_{out} \right)}{\rho V \frac{\partial u}{\partial T}}$$

 Need to calculate heat transfer at each instant throughout the compression process





- In general, the most significant heat transfer paths to the refrigerant involve convection
 - » Forced convection between refrigerant and interior compressor walls, Q_{comp}
 - » Free convection between shell and surroundings, Q_{shell}
- Also need to consider heat addition to refrigerant in shell due to motor and mechanical losses, Q_{motor} and Q_{mech}



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Compressor Heat Transfer Models

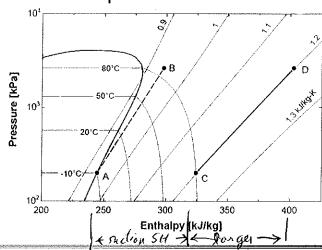


- Heat transfer increases the gas temperature and specific volume (reduces density) thus reducing the compressor efficiency
- This relationship is evidenced in the equation for reversible

flow work,

$$W_{rev} = \int v dP$$

 The P-h diagram also illustrates that increased superheat increases compression work



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 Convective heat transfer is modeled by Newton's law of cooling:

$$\dot{Q} = h_c A \left(T_w - T_g \right)$$

Where positive Q indicates heat transfer to the gas.

A = Surface area, m²

 h_c = Heat transfer coefficient, W/m²-K

 T_q = Temperature of refrigerant, K

Tw = Temperature of wall, K from overall en arealfalunce

Q = Heat transfer rate, W

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Compressor Heat Transfer Models



- The heat transfer coefficient determines the resistance to heat transfer based on characteristics of the flow and geometry
- In general, for forced convection,

$$h_c = a \frac{k}{D_h} \operatorname{Re}^b \operatorname{Pr}^c$$

» k = Thermal conductivity of the fluid, W/m-k

» D_h = Hydraulic diameter of flow path, m

» Re = Reynolds number

» Pr = Prandtl number





 The Reynolds number is based on fluid properties, the flow speed, and the channel diameter (ratio of inertial and viscous forces):

$$Re = \frac{\rho u D_h}{\mu}$$

$$Re = \frac{$$

 The Prandtl number only depends on fluid properties (ratio of momentum and thermal diffusivity):

$$\Pr = \frac{\mu C_p}{k} \qquad \qquad \begin{array}{ll} C_p = & \text{Constant pressure specific heat, J/kg-K} \\ k = & \text{Thermal conductivity, W/m-K} \\ \mu = & \text{Dynamic viscosity, Pa-s} \end{array}$$

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Compressor Heat Transfer Models



- The exact correlation for heat transfer coefficient depends on the compressor type. Examples:
 - » Reciprocating

$$h_c = 0.053 \frac{k}{D_h} \text{Re}_L^{0.8} \text{Pr}^{0.6}$$

(Adair et al. 1972)

» Scroll (spiral HX corelation)

$$h_c = 0.023 \frac{k}{D_h} \text{Re}^{0.8} \text{Pr}^{0.4} \left[1.0 + 1.77 \left(\frac{D_h}{r_{aver}} \right) \right]$$

(Chen et al. 2002)







- For a reciprocating compressor, the heat transfer model involves the following steps:
 - » Use the known density and temperature to calculate the required fluid properties (Pr, k, μ)
 - » Calculate the hydraulic diameter of the flow path,

$$D_h = \frac{4A_c}{P}$$

- » Estimate the velocity of the gas in the chamber, for example as one-half of the piston speed
- » Calculate the Reynolds number,

 $Re = \frac{\rho u D_h}{u}$

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Compressor Heat Transfer Models



- Reciprocating compressor example continued:
 - » Calculate the convection coefficient using the correlation from Adair et al. (1972)

$$h_c = 0.053 \frac{k}{D_h} \text{Re}_L^{0.8} \text{Pr}^{0.6}$$

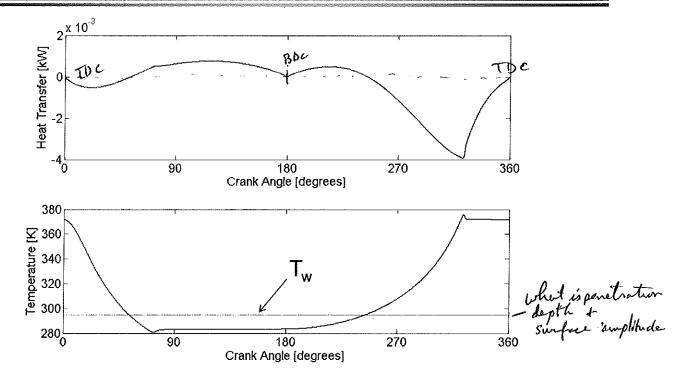
» Use the wall temperature calculated from the overall, lumped heat transfer model (discussed later) or an initial guess for wall temperature to calculate the instantaneous heat transfer rate:

$$\dot{Q} = h_c A \left(T_w - T_g \right)$$



Heat Transfer Results





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Compressor Heat Transfer Models



- Reciprocating compressor example continued:
 - » The average heat transfer over one crank rotation can be calculated by numerically integrating the instantaneous heat transfer:

$$Q_{total} = \sum_{\theta=0^{\circ}}^{360^{\circ}} \frac{\dot{Q}}{\omega} \Delta \theta$$

» This total heat transfer rate will be used in the overall, lumped energy balance to update the temperatures of the lumped elements such as the cylinder walls



Nomenclature



 $A = Surface area, m^2$

 C_n = Constant pressure specific heat, J/kg-K

 D_h = Hydraulic diameter of flow path, m

h = Specific enthalpy, J/kg

 h_c = Heat transfer coefficient, W/m²-K

k = Thermal conductivity of the fluid, W/m-K

 \dot{m} = Mass flow rate, kg/s

 m_{cv} = Mass in control volume, kg

Nu = Nusselt number

P = Pressure, Pa

Pr = Prandtl number

Ö = Heat transfer rate, W

 r_{aver} = Average radius of curvature of chamber, m

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Nomenclature



Re = Reynolds number

t = Time, seconds

 T_a = Temperature of refrigerant, K

 T_w = Temperature of wall, K

a = Average velocity of gas in chamber (for Reynolds number), m/s
 OR specific internal energy (for energy balance), J/kg

 $V = Volume, m^3$

v = Specific volume, m³/kg

Greek symbols

 θ = Crankshaft angle, degrees

 μ = Dynamic viscosity, Pa-s

 ω = Rotational speed of crank, degree/second

 ρ = Density, kg/m³



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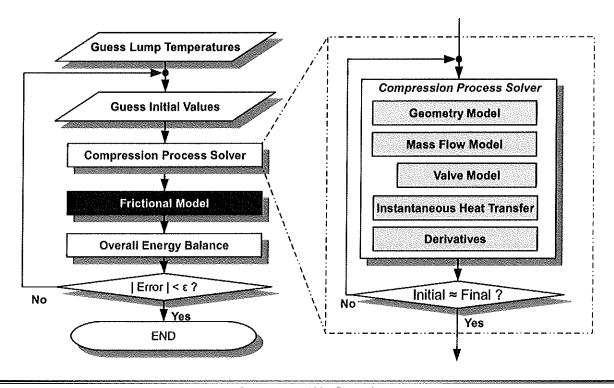


Friction & Mechanical Losses



Frictional Model





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Frictional Losses



 Numerous models available, with varying levels of empiricism

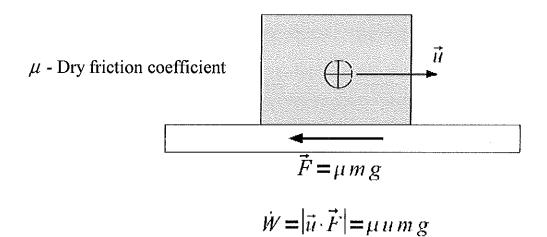
Mechanistic	og 1931	Empirical
Detailed Dynamics and Frictional	Correlation of semi- empirical parameters	Correlation Of Mechanical Losses
Model	$\dot{W}_{ML} = f\left(au, \dot{W}_{gas}, \ldots\right)$	$\dot{W}_{ML} = f\left(T_s, T_d, \ldots\right)$



Frictional Losses



Mechanistic Simple Example



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Frictional Losses



Adham ring



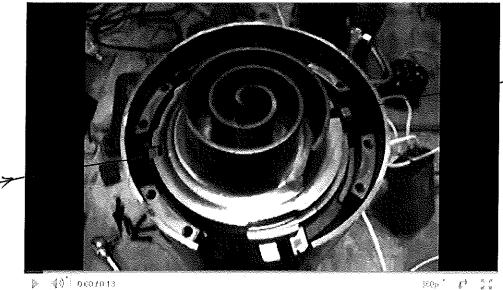
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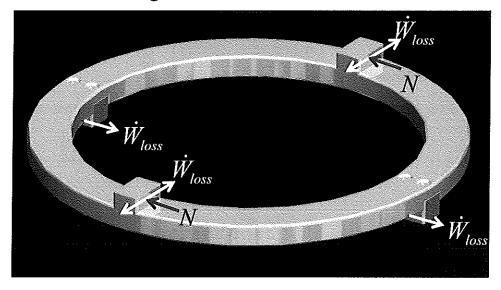
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Frictional Losses



 Mechanistic ML Scroll Compressor Oldham Ring



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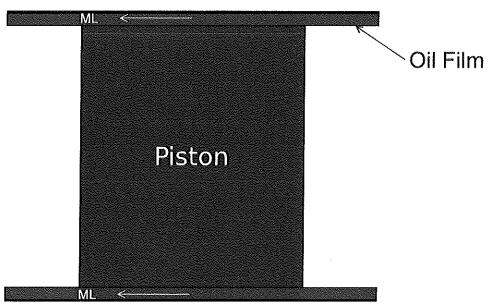
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Frictional Losses



Example: Piston – Cylinder Frictional Losses





Frictional Losses



• Example:

» Frictional shear between piston and cylinder wall

d = 2 cm

(diameter piston)

L = 2 cm

(length piston)

 $x_{max} = 2 \text{ cm}$

(piston stroke)

» $\delta = 20 \mu m$

(gap width)

 $\eta = 0.0086 \text{ Pa-s}$

(oil viscosity)

 $\omega = 377 \text{ rad/s}$

(60 Hz)

 $u_{max} = 7.54 \text{ m/s}$

 $(\omega^* x_{max})$

 $u_{avg} = 4.8 \text{ m/s}$

 $(2/\pi^*u_{max})$

» Assume Couette flow and "unwrapped" bearing

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Frictional Losses



• Area:

»
$$A = \pi DL$$

= $\pi (0.02 \text{ m})(0.02 \text{ m}) = 0.001257 \text{ m}^2$

Shear Force

$$F = \frac{\eta A u_{avg}}{\delta}$$

$$= \frac{(0.0086 \text{ Pa} - \text{s})(0.00126 \text{ m}^2)(4.8 \text{ m/s})}{20 \text{ x} 10^{-6} \text{ m}} = 2.6 \text{ N}$$

Frictional Power

$$W = Fu_{avg}$$

= (2.6 N)(4.8 m/s) = 12.48 W



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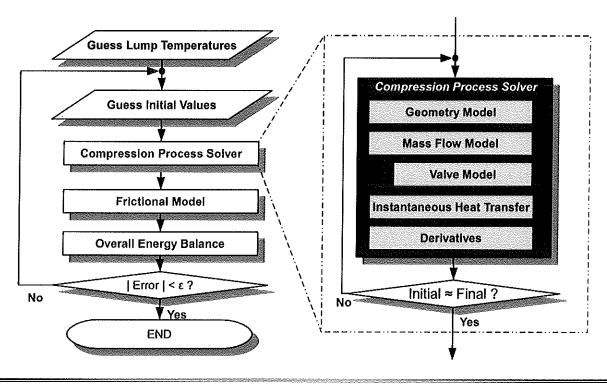




Numerical Methods / Solver







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Numerical Methods



Basic Problem:

- » At step *i* of the revolution we know temperature, pressure, density, etc. (initially guessed)
- » The derivatives of the two independent properties are then calculated with respect to the crank angle using the
- » Using derivatives, find temperature, pressure, density, be solve sural based etc. at the next step
- » Repeat for the entire revolution
- » Same method applies to multiple control volumes





Vector of partial differential equations to integrate for the control volume

$$\mathbf{x} = \begin{bmatrix} T \\ \rho \end{bmatrix} \qquad \mathbf{y} = \begin{bmatrix} \frac{dT}{d\theta} \\ \frac{d\rho}{d\theta} \end{bmatrix}$$

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Numerical Methods



Simple Euler

$$\mathbf{x}_{i+1} = \mathbf{x}_{i} + \Delta \theta \frac{dT}{d\theta} \Big|_{i}$$

$$\mathbf{x}_{i+1} = \mathbf{x}_{i} + \Delta \theta \frac{dT}{d\theta} \Big|_{i}$$

$$\rho_{i+1} = \rho_{i} + \Delta \theta \frac{d\rho}{d\theta} \Big|_{i}$$

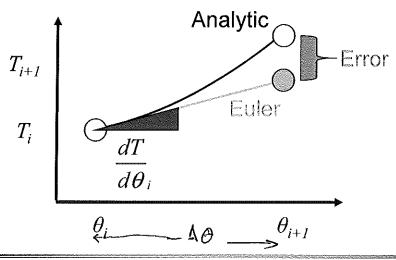
Vector Form

Variable Form





Simple Euler



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Numerical Methods



- Other Potential Solver Methods (Chapra, 2006)
 - » Heun
- 2nd orde peditor-corrector
- » Runge-Kutta (RK4)
- » Adaptive Runge-Kutta (RK4/RK5)
- » Implicit Euler
- » Semi-Implicit Backward Euler
- » ...
- » Among many others





Runge-Kutta (RK4)

$$\mathbf{k}_{1} = \mathbf{f} \left(\theta_{n}, \mathbf{y}_{n} \right)$$

$$\mathbf{k}_{2} = \mathbf{f} \left(\theta_{n} + \frac{1}{2} \Delta \theta, \mathbf{y}_{n} + \frac{1}{2} \mathbf{k}_{1} \Delta \theta \right)$$

$$\mathbf{k}_{3} = \mathbf{f} \left(\theta_{n} + \frac{1}{2} \Delta \theta, \mathbf{y}_{n} + \frac{1}{2} \mathbf{k}_{2} \Delta \theta \right)$$

$$\mathbf{k}_{4} = \mathbf{f} \left(\theta_{n} + \Delta \theta, \mathbf{y}_{n} + \mathbf{k}_{3} \Delta \theta \right)$$

$$\mathbf{x}_{n+1} = \mathbf{x}_{n} + \frac{1}{6} \Delta \theta \left(\mathbf{k}_{1} + \mathbf{k}_{2} + \mathbf{k}_{3} + \mathbf{k}_{4} \right)$$

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Numerical Methods



- Simple Euler v. RK4
 - Simple Euler is 1st order O(h)
 - RK4 is 4th order O(h4)
 - To maintain same local accuracy, need ~40 times fewer steps, but need to call derivative function 6 times per step, overall speedup can be significant
 - Both struggle with numerical stability is one of big phobellarge in compressor modeling





Recip - stiffregion at beginning and end.

- Adaptive Runge-Kutta
 - RK4 with embedded error estimate
 - Take big steps when you can, take baby steps when necessary
 - Chapra (2006) has good coverage and analysis

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References



- Chapra, S.C., Canale, R.P., 2006, Numerical Methods for Engineers 5th Edition, McGraw-Hill
 - » Excellent resource for a wide range of numerical methods, including adaptive RK4/RK5
- Kreyszig, E., 2006, Advanced Engineering Mathematics 9th Edition, John Wiley
 & Sons
 - » Also provides solid background in numerical methods





Overall Energy Balance



Overall Energy Balance



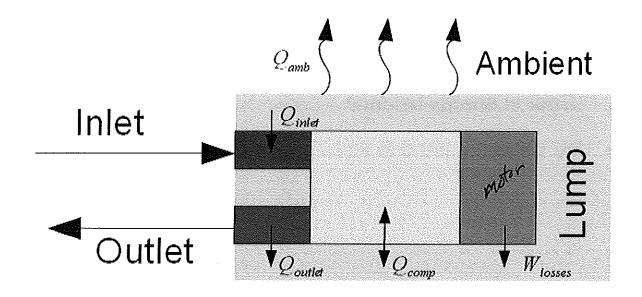
Basic Idea:

» Apply a steady-state energy balance to certain components, or grouped components in a lumped model, to be used in further heat transfer analysis



Overall Energy Balance





Use new for-Ropphon for multiple lines

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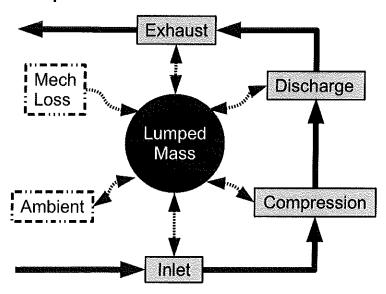
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Energy Balance



Single Lump





Energy Balance



• Energy balance for lump

$$r_{EB} = Q_{inlet}(T_{lump}) + Q_{outlet}(T_{lump}) + Q_{comp}(T_{lump}) + Q_{amb}(T_{lump}) + W_{loss}$$

$$\begin{tabular}{l} Error in energy balance \end{tabular}$$

ullet Use secant method to find T_{lump} by driving r_{EB} to zero

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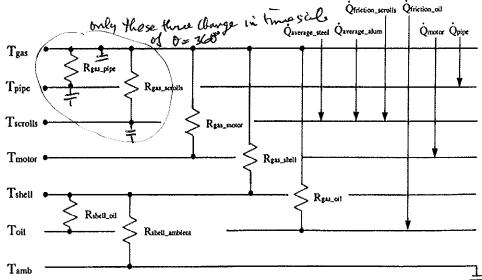


Energy Balance



Ay Wall penetration depth is may bettle effect on Ms

Av Hanging Multiple Lumps (Chen, 2000)





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Conclusion



- Presentation covered the fundamentals of compressor modeling:
 - Discussed the mass and energy balance equations that govern the state of the refrigerant during the compression process
 - Introduced models for leakage (including flow through valves), heat transfer, and mechanical losses that improve the accuracy of the temperature and pressure predictions
 - Explained numerical solution methods for calculating the temperature and pressure variations over a crankshaft rotation
 - Discussed an overall energy balance on the compressor to predict the temperatures of different components that will influence the refrigerant temperature
- Compressor modeling concepts were applied to a simplified reciprocating compressor to demonstrate the modeling process



Conclusion



- For a real compressor design, experimental testing would be required to validate the reciprocating compressor model accuracy
- However, the reciprocating compressor model illustrates that modeling is a relatively fast and cost effective tool for predicting compressor performance
 - Model can be used to study the effect of design modifications and operating conditions on compressor performance
 - Compressor model can be incorporated into a larger system model
- The example model can be modified to fit any compressor design by updating the geometry equations, the leakage paths under consideration, and the mechanical loss model

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