A National Defense Strategy Based on Provably False Assumptions

- Assumptions Used by the DoD for GMD Performance Cannot Possibly be Known Hence, Actual Performance of the GMD is Unknowable
- The Record of “Proven Reliability” of the Navy’s SM-3 Interceptor Actually Shows that the SM-3 Will Be Highly Unreliable in Actual Combat Conditions
Main Points to be Made in This briefing (1 of 3)

- North Korea is only one successful flight test away from demonstrating an ICBM that would be able to deliver a nuclear warhead to almost anywhere in the Continental United States.

- Since the United States is building the wrong missile defense-systems to deal with this, it cannot be ruled out that the United States might eventually become vulnerable to nuclear-armed coercion.

- This potential future problem can be addressed as there are definitely boost-phase missile defense-systems that could provide for the strategic continental defence of the United States from both North Korean and Iranian long-range ballistic missiles.

- However, under current conditions, Boost-Phase ballistic missile defense systems will never be developed or built, as the Pentagon remains solely focussed on building unworkable exo-atmospheric missile defenses.

- Unless there is a serious evaluation of the true shortcomings of these exo-atmospheric defenses, they will remain the centerpiece of the Pentagon’s ballistic missile defense program, and bureaucratic politics will guarantee that there will be no money for the development of far more capable, reliable, and robust boost-phase missile defenses.

- There is also a very serious problem with the technical competence of upper level management in the Missile Defense Agency and elsewhere in the Pentagon.

Main Points to be Made in This briefing (2 of 3)

- This is manifest in the record of one technically naïve missile defense solution being substituted for another.

- For example, the adoption of post-boost phase missile defenses under the naïve belief that it takes time to deploy countermeasures after rocket motor shutdown. This reveals a startling lack of technical sophistication among managers at MDA.

- There is also a problem with the approach of political leadership to this problem, which has inadvertently encouraged repeated misrepresentations in the missile defense program by not holding individuals accountable for false statements they have and continue to make.

- In summary, none of the actions needed to get proper management into the missile defense program will be solved unless individuals and organizations are held accountable for misinformation that is being disseminated.

- It is ridiculous to call the missile defense testing that has been conducted to date by the Pentagon realistic.

- Even in orchestrated experiments, the systems fail catastrophically when anything unexpected happens (Note Ronald Kadish’s statement when the canister carrying the large balloon failed to open in the IFT-6).

- In the FTG-06, the scene recognition system of the Sea-Based X-band radar failed because of a trivial unplanned event when a rocket motor unexpectedly expelled small pieces of debris.
Main Points to be Made in This briefing (3 of 3)

- The SM-3 failures and the FTG-06 failure are simply another manifestation of the GMD and SM-3s profound vulnerabilities to decoys.

- This vulnerability was covered up in 1997 and 1998 when the IFT-1A and IFT-2 proof-of-concept experiments showed that the GMD could be fooled by a subset of the decoys flown in those experiments.

- It is now more than ten years later and the GMD has still not been tested against these decoys.

- Before the Gulf War of 1991 Patriot had a record of 17 successes out of 17 intercept tests. In the Gulf War of 1991 the actual intercept rate, defined as destruction of SCUD warheads, was almost certainly 0 out of 44.

- Most recently, the SM-3 has been misrepresented to the President as a “proven and reliable” missile defense system. In fact, the Pentagon’s own data shows the opposite.

- The Pentagon’s test data instead shows that the SM-3 is fragile and brittle, and unlikely to perform in combat at a level much higher than that allowed by chance.

- SM-3 Block IA kill vehicle will be even more vulnerable to countermeasures than the already fragile and vulnerable GMD kill vehicle, as the SM-3 kill vehicle cannot measure the temperature of objects (see MIT Lincoln Laboratory IR data).

What Needs to Be Done

- No sound technical decisions in missile defense will be possible unless there are science-based assessments of the true capabilities of missile defense systems.

- It is the responsibility of the technical community to stop playing political games with the truth.

- The community needs to confront the fact that strategic boost-phase ballistic missile defenses, which actually could provide for the strategic defense of the continental United States, will never be developed unless the truth about the fundamental limitations of exo-atmospheric defenses is addressed.

- The abysmal failure of the Pentagon to establish realistic standards for testing missile defense systems must be vigorously addressed and corrected.

- Real tests of the GMD and SM-3 systems must be done against simple realistic decoys of the kind that were flown in the IFT-1A and IFT-2 in 1997 and 1998.

- Unless the informed technical community takes responsibility as advisors to the nation, the next time the nation depends on missile defenses, one can only hope it will not be against nuclear-armed ballistic missiles.
Important Consequences of the Current Failures to Properly Address the Real Technical Issues Associated with Exo-Atmospheric Missile Defenses

- North Korea is only one successful flight test away from demonstrating an ICBM capable vehicle that could range the entire Continental United States. The United States could become vulnerable to such an ICBM threat, because it is building the wrong missile defense-systems to deal with it.

- There are definitely *boost-phase* missile defense-systems that could provide for the strategic continental defence of the United States from ICBM attack from North Korea and Iran.

- However, the new missile defense strategy de-emphasizes these defense-systems in favor of unproven, unworkable, and far more expensive systems.

- This failure to emphasize workable systems in favor of unworkable systems is a consequence of years of misrepresentations and coverups by the Missile Defense Agency, which has not been forcefully addressed by the nation’s political leadership.

- One consequence of these failures is that the US is poised to deploy systems that are easy to defeat, which will likely fail to deter, or actually stimulate, ballistic missile proliferation.

- If the current emphasis on exo-atmospheric systems continues without the imposition of proper testing standards and oversight, we can expect with near certainty that proliferators like North Korea and Iran will introduce highly effective countermeasures against these systems.

- These proliferators could, and likely would, sell these countermeasures to client states.

Important Additional Costs that are not Offset by Benefits Due to the Ongoing Failures to Properly Address the Real Technical Issues Associated with Exo-Atmospheric Missile Defenses

- The United States could damage its relations with allies and friends by pushing on them false and unreliable solutions to a real security problem.

- The United States will antagonize both Russia and China with massive defense deployments that have the appearance of being designed to be “flexibly” adaptable to deal with Russian and Chinese strategic forces.

- The negative effects of a costly and energetic US program that appears to be aimed at blunting Russia’s strategic retaliatory strike forces will sow distrust of the US within the Russian government and will create significant barriers to future arms reductions efforts.

- If arms reductions efforts with Russia come to a halt, this will have serious adverse effects on Russian and US efforts to maintain the viability of the Nonproliferation Treaty of 1968, which is already under considerable pressure due to the US-India Nuclear Deal.
Basic Issues to Be Discussed in this Briefing

1. Important aspects of the nation’s new nuclear strategy, laid out in the Nuclear Posture Review (NPR), rests on conclusions from the Department of Defense’s Ballistic Missile Defense Review (BMDR), released in February 2010.

2. The BMDR claims that all the fundamental technical problems associated with current missile defense systems have been solved.

3. This translates into the conclusion that the United States is now and for the foreseeable future able to defend itself from limited ICBM attacks.

4. It also translates into a conclusion that the United States can build sufficiently reliable and robust ballistic missile defenses that it will cause potential adversaries to deemphasize their reliance on ballistic missiles as instruments of intimidation.

5. However, as will be shown in this briefing, there are no new material facts to support any of the claims in the BMDR that suggest that the United States is now in a position to defend itself from limited ICBM attacks or that any of the fundamental unsolved problems associated with high-altitude ballistic missile defenses have been solved.

6. In fact, as this briefing will show, the most recent ballistic missile defense flight-test data released by the Department of Defense and the most recent failed test of the ground-based missile defense system in January show quite the opposite.

Relevant Information About the Arms Control Today Article and the Associated White Paper Being Provided to the Committee

- The PDF and printed versions of the Arms Control Today article includes copies of critical endnotes that contain powerful additional information that substantially expands upon the article’s content.

- The associated White Paper (which NRC information control was not provided to the Committee until today) contains a substantial amount of highly relevant additional information that could not be put into the original article because of space considerations. Endnote 1 in the Arms Control Today article refers to the White Paper.

- An expanded search of Missile Defense Agency documents show that they contain quite a bit of evidence that the Missile Defense Agency has not been accurately representing the record of accomplishments in ballistic missile flight tests.
Materials for the National Research Council Committee were sent by e-mail on May 5. After one week (May 12) I found out that the most informative part of the transmitted Committee materials had not been distributed.

The reason was that the National Research Council information system had the following questions:

1. The White Paper is unclassified; however, is it non-restrictive? Can it be released for unlimited distribution? If so, then the White Paper will be placed in the Public Access File that is associated with this study. (By the way, I checked for the White Paper on the Internet, but I did not find it.)

2. In regard to the unclassified MIT Lincoln Laboratory slides that are Figures 4, 5, 6, and the title page of Eric Evans’ brief (shown in the reference section), I need to know what process was used to vet these slides so that they are unclassified, non-restrictive, and public releasable for unlimited distribution.

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Relevant Information in this Briefing About Issues Discussed in the Arms Control Today Article and the Associated MIT White Paper

Misrepresentation of the SM-3 System as “Proven”

1. The Article and White Paper show that the Department of Defense’s owned ballistic missile defense flight test data contradicts the technical conclusions of the Ballistic Missile Defense Review.

2. In particular the two papers show that when the SM-3 interceptor was tested against rockets carrying warheads, and some times rockets not carrying warheads, the SM-3 kill vehicle almost always failed to hit the warhead section. This means that the warhead would not have been destroyed in 8 to 9 out of 10 intercept tests that were called successful by the Missile Defense Agency.

3. In testimony before the Congress in 1992 the Army was questioned about the greater than 90% intercept rate it had earlier claimed for the Patriot missile in the Gulf War of 1991. The army explained that its claims were not false because it defined intercept as an event where "a Patriot and a stud test in the sky." The same misleading statements about what constitutes successful intercepts are again being made by the Missile Defense Agency – this time with regard to the results of the SM-3 ballistic missile flight tests.
FTG-06 Failure and Its Relation to Vulnerabilities in the SM-3 Block IA Interceptor

4. The two papers explain how the most recent Ground-Based Missile Defense test on January 31, 2010, the FTG-06, failed because the defense-system encountered a special circumstance where it could not recognize the difference between the mock warhead and chunks of debris that were unexpectedly expelled by the spent rocket motor that launched the mock warhead.

5. They then explain how an adversary could intentionally create credible false targets that would without exception defeat the Ground-Based Missile Defense System in all circumstances.

6. The infrared and radar data released by MIT Lincoln Laboratory and discussed in the White Paper explains how chunks of debris that were unexpectedly expelled from a rocket motor in the FTG-06 caused the total collapse of the "scene recognition" process that is supposed to select the warhead from the cloud of deployment debris.

7. The MIT Lincoln Laboratory data shows that the SM-3 block IA kill vehicle will never be able to discriminate between a warhead and debris of the type encountered from the rocket motor in the FTG-6.

8. This is because the SM-3 Block IA kill vehicle cannot measure the temperature of the debris. This vulnerability will be eliminated with the Block IB kill vehicles and beyond, because these kill vehicles can observe targets in two wavelength bands, allowing them to distinguish hot objects from cooler near-room temperature objects.

Misrepresentations of GMD Proof of Concept Tests as Successful When in Fact they Failed

9. The two papers explain that the IFT-1A and IFT-2 experiments, performed more than 10 years ago, failed to demonstrate that the Ground-Based Missile Defense kill vehicle could discriminate between 0.6 m diameter balloons and cone shaped decoys because the infrared signals from these objects were essentially indistinguishable from the infrared signals from the warhead.

10. I have quite a lot of data on the IFT-1A and IFT-2 experiments I could share with the Committee if it wants to obtain information about the unambiguous failure of these critical proof-of-concept missile defense experiments and to analyze the implications for these failures for the current missile defense program.

11. The data shows that the Missile Defense Agency misrepresented the experiments to Congress as an unambiguous success when in fact the experiments failed.
Relevant Information in this Briefing About Issues Discussed in the Arms Control Today Article and the Associated MIT White Paper

Failure of Political Leadership to Hold Pentagon Officials Accountable When They Have Been Caught Lying or Turning a Blind Eye to Fraud

12. Two of the three most senior people involved in the IFT-1A and IFT-2 fraud, Lester B. Lyles and Keith Englander, hold high-level positions in the Pentagon in spite of their involvement in misrepresenting these critical proof-of-concept experiments as successes when in fact they failed.

13. One of them, Lester B. Lyles, was appointed by the Undersecretary of Defense for Acquisition, Technology and Logistics, Ash Carter, to be Vice Chair of the Defense Science Board.

14. Ash Carter was on the MIT Lincoln Laboratory Oversight Board when questions were being raised about Lincoln’s role in this fraud. In spite of the substantial publicity and documentary evidence of fraud, he took no steps to investigate.

Failure of the Pentagon to Set Proper Standards for What is Realistic Testing

15. The two papers describe how the Missile Defense Agency removed the credible subset of IFT-1A and IFT-2 decoys from all subsequent flight tests.

16. At that time, the New York Times published (attached after the endnotes in the file containing the article) a detailed and comprehensive article explaining how the Missile Defense Agency had rigged all of the then planned future flight tests by removing these decoys from all subsequent missile defense tests.

17. The Missile Defense Agency responded to the New York Times exposé by claiming that these decoys would be re-introduced into missile defense flight tests later in the testing. There is a very extensive public record of discussions with the press about the intention to fly against these decoys at a future time.

18. However, more than 10 years later, after the Ground-Based Missile Defense has been declared by the Ballistic Missile Defense Review as being able to defend the United States, the Ground-Based Missile Defense system has still not been tested against these decoys.

19. This lack of realistic testing is scandalous and an issue of overwhelmingly importance that needs to be addressed.
A Short Summary of Incidents of Misrepresentation in the Missile Defense Program

The Missile Defense Agency:
A Culture of Misrepresentation and a Repeated History of Being Caught

Other Concrete Examples

- The IFT-1A and IFT-2 demonstrated the capability to discriminate against decoys.
- Failure of FTG-06 does not indicate any fundamental problems.
- Failure to inform the President that the FTG-06 has profound implications for the SM-3 Block IA – it demonstrates the SM-3 Block IA is fundamentally vulnerable to small heated objects.
- False claims that the European Defense System will make it possible to defend Japan from an Iranian ICBM attack.
- False claims that GMD interceptors launched from Alaska can be used to defend Japan from an Iranian ICBM attack.
- False claims that the Russians are misrepresenting the possibility that interceptors placed in Poland can “theoretically” engage Russian ICBMs.
- False claims that the European Midcourse Radar will make it possible for the European Missile Defense to defend the United States and Northern and Western Europe from long-range ballistic missile attack.
- False claims to Congress that discrimination capabilities have been demonstrated. (Kadish and Lyles’ statements to Congress. Kadish’s statement on 60 Minutes II).

Question: What are the implications for the future accuracy of information about missile defense systems when there are so many people with a history of making false claims still involved with the Pentagon and the program?
What Are the Prospects for Building a Reliable, Robust, and Intimidating Boost-Phase Ballistic Missile Defense that Could Defend the Continental United States from Strategic Nuclear-Armed ICBM Attack?
Coverage Against Unha-2 – Like Large Liquid Propellant with 240 Second + Burn is Possible

5 km/sec Interceptor, ~500 km range in about 100 seconds, Unha-2 Ballistic Missile gets to about 400 km in about 240 seconds.
Stealthy Drone That Carries a Payload of 4500 pounds, Which Is More Than Enough to Accommodate Two 2000 pound Interceptors, or a Single Heavier Interceptor

This particular drone can carry a payload of 4500 pounds, which is more than enough to accommodate two 2000 pound interceptors, or a single heavier interceptor. The heavier interceptor might be more desirable for situations where an interceptor burnout speed in excess of 5 km/s is desired. Smaller interceptors would probably have burnout speeds of perhaps 4 to 4 1/2 km/s. These lower burnout speeds may well be adequate.

Estimate of the Radar Cross Section of a 50 Meter Wing Span B-2 Like Aircraft

Radar cross-sections that are less than 0.01 m² are certainly achievable. Such small radar cross-sections require not only that the aircraft have a shape that does not strongly reflect radar signals, but it also requires that the aircraft be covered with radar absorbing material. A bare skinned version of this aircraft would have a small radar cross-section, but it would still be roughly 10 times larger relative to a similarly shaped aircraft constructed with radar absorbing materials.
Iran is a Large Country, But Not Large Enough to Make It Easy to Test Long Range Missiles

Reports of Testing of the 2,000 Kilometer Range Sejjil-2 Ballistic Missile on Lofted Trajectories Are Easily Explained If Iran Wants to Test Only Within Its National Boundaries

Examples of Test Trajectories that Can Be Flown by a 2,000 Kilometer Range Ballistic Missile

4.11 km/sec Launch Trajectories
Locations Shown at 10 second Intervals
Minimum Energy = 40.51 Deg
1000 km Range = 61 Deg
1500 km Range = 72 Deg
1500 km Range = 21 Deg
Iran’s Launch of a 27 Kilogram Satellite on February 2/3, 2009 Was Obviously Chosen to Not Overfly Adjacent Countries

The First Stage of the Iran’s Satellite Launch Vehicle Fell Well Within Iran’s Borders and the Second Stage Went Into Orbit with the Small Satellite

<table>
<thead>
<tr>
<th>Stage Characteristics</th>
<th>First Stage</th>
<th>Second Stage</th>
<th>Third Stage</th>
</tr>
</thead>
<tbody>
<tr>
<td>Full Weight (kg)</td>
<td>22,300</td>
<td>3100</td>
<td>28</td>
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<tr>
<td>Structure Factor</td>
<td>0.1</td>
<td>0.09</td>
<td>0.03</td>
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<tr>
<td>Propellant Weight (kg)</td>
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<td>2821</td>
<td>2736</td>
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<td>Weight of Burned Fuel (kg)</td>
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<td>2736</td>
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<td>Empty Weight (kg)</td>
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<td>85</td>
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<tr>
<td>Burnout Weight (kg)</td>
<td>3033</td>
<td>364</td>
<td>85</td>
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<td>Fraction of Fuel Unburned</td>
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<td>0.03</td>
<td>0.03</td>
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<tr>
<td>Residual Unburned Fuel (kg)</td>
<td>803</td>
<td>85</td>
<td>85</td>
</tr>
<tr>
<td>Burn Time (sec)</td>
<td>137</td>
<td>274</td>
<td>85</td>
</tr>
</tbody>
</table>

Vehicle Locations Shown at 10 Second Intervals
- Satellite Weight (kg) = 27
- Launch Gross Weight (kg) = 25,427
- First Stage Burn Time (sec) = 137
- Second Stage Burn Time (sec) = 274
- Total Powered Flight Time (sec) = 411 (6 min, 51 sec)
Expected and Actual Flight Outcomes Associated with the North Korean Satellite Launch Attempt of April 4/5, 2009

**Stage 1 Characteristics:**
- Stage Full Weight (kg) = 74050
- Structure Factor = 0.08
- Stage Propellant Weight (kg) = 68126
- Weight of Burned Fuel (kg) = 65400
- Stage Empty Weight (kg) = 1585
- Stage Weight at Burnout (kg) = 1287
- Fraction of Fuel Unburned = 0.04
- Residual Unburned Fuel (kg) = 2872
- Burn Time (sec) = 117.6

**Stage 2 Characteristics:**
- Stage Full Weight (kg) = 13550
- Structure Factor = 0.095
- Stage Propellant Weight (kg) = 12283
- Weight of Burned Fuel (kg) = 11895
- Stage Empty Weight (kg) = 1655
- Stage Weight at Burnout (kg) = 1287
- Fraction of Fuel Unburned = 0.03
- Residual Unburned Fuel (kg) = 368
- Burn Time (sec) = 122

**Stage 3 Characteristics:**
- Stage Full Weight (kg) = 3100
- Structure Factor = 0.09
- Stage Propellant Weight (kg) = 2821
- Weight of Burned Fuel (kg) = 2736
- Stage Empty Weight (kg) = 276
- Stage Weight at Burnout (kg) = 364
- Fraction of Fuel Unburned = 0.03
- Residual Unburned Fuel (kg) = 85
- Burn Time (sec) = 274

Vehicle Locations Shown at 10 Second Intervals
- Upper Stage + Payload Weight (kg) = 4000
- Launch Gross Weight (kg) = 91,600
- First Stage Burn Time (sec) = 118
- Second Stage Burn Time (sec) = 122
- Third Stage Burn Time (sec) = 274
- Total Powered Flight Time (sec) = 514 (8 min, 34 sec)
Most Recent MDA Misrepresentation
The SM-3 is a “Ballistic Missile Defense System [that] has demonstrated 20 hit-to-kill *intercepts* [italics added] out of 24 at sea firing attempts.” **

**MDA Fact Sheet, November 24, 2009 09-MDA-5060

Results of SM-3 Flight Tests Derived from MDA’s Published Video Data
The Missile Defense Agency: A Culture of Misrepresentation and a Repeated History of Being Caught

Most Recent Concrete Example

Misrepresenting the SM-3 system test results to the press, and almost certainly to the President and the Secretary of Defense.

"There were subsequent views not publicly released to preclude potential adversaries from seeing exactly where the target was struck, so the authors were basing their assessment on incomplete information," Rick Lehner, a spokesman for the agency, told AOL News.


Incidents of Repetitive Misrepresentations by the Missile Defense Agency – (FM-6)

"There were subsequent views not publicly released to preclude potential adversaries from seeing exactly where the target was struck, so the authors were basing their assessment on incomplete information," Rick Lehner, a spokesman for the agency, told AOL News.

HIT ON WARHEAD IN THE FM-6 TEST ON DECEMBER 11, 2003 – ABSOLUTELY NO EVIDENCE OF SIGNIFICANT LATERAL ACCELERATION DURING HOMING
"There were subsequent views not publicly released to preclude potential adversaries from seeing exactly where the target was struck, so the authors were basing their assessment on incomplete information," Rick Lehner, a spokesman for the agency, told AOL News.

**HIT ON WARHEAD IN THE FM-6 TEST ON DECEMBER 11, 2003 – ABSOLUTELY NO EVIDENCE OF SIGNIFICANT LATERAL ACCELERATION DURING HOMING**

**CREDIBLE EVIDENCE OF REPETITIVE LYING BY THE MISSILE DEFENSE AGENCY – (FM-6)**

**WARHEAD MISS IN THE FTM-11 TEST ON DECEMBER 7, 2006 – ABSOLUTELY NO EVIDENCE OF SIGNIFICANT LATERAL ACCELERATION DURING HOMING**

**CREDIBLE EVIDENCE OF REPETITIVE LYING BY THE MISSILE DEFENSE AGENCY – (FTM-11)**
Credible Evidence of Repetitive Lying by the Missile Defense Agency – (FTM-11)

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WARHEAD MISS IN THE FTM-11 TEST ON DECEMBER 7, 2006 – ABSOLUTELY NO EVIDENCE OF SIGNIFICANT LATERAL ACCELERATION DURING HOMING

Incidents of Repetitive Misrepresentations by the Missile Defense Agency – (FM-6)

“...potential adversaries from seeing exactly where the target was struck, so the authors were basing their assessment on incomplete information,” Rick Lehner, a spokesman for the agency.”

Lateral Accelerations Required to Shift the Impact Point 1 Meter Within 1/30th of a Second

Distance = \( \frac{\text{acceleration} \times \text{time}^2}{2} = \frac{a \cdot t^2}{2} \)

\[ \text{acceleration} = \frac{2D}{t^2} = \frac{2 \times 1}{0.033^2} = 1800 \, \text{m/sec}^2 \]

\[ \text{Acceleration in Gs} = \frac{a}{g} = \frac{1800 \, \text{m/sec}^2}{9.8 \, \text{m/sec}^2} = 184 \, G \]

Required Rocket Thrust (Tonnes) = \( \frac{1800 \, \text{m/sec}^2 \times 25 \, \text{kg}}{1000 \, \text{kg/Tonne}} = 45 \, \text{Tonnes} \approx 3 \, \text{Times the Thrust of a SCUD-B Rocket Motor} \)
Other Problems with the Homing Process
The Kill Vehicle Must Hit the Warhead to Destroy It

Bullet Passes Through an Empty Container

Bullet Creates A Shock as It Passes Through the Material in a Filled Container
Predictions Made by the Missile Defense Agency for a Hit on US Satellite 193 that Misses and Hits a Full Hydrazine Tank in the Satellite

Actual Infrared Image of the Kill Vehicle Hit on US Satellite 193
Real World Event
Satellite Intercept – 20 FEB 08

• Objective
  - Protect against potential loss of life due to uncontrolled reentry of ~ 5,400 lb (2,450 kg) satellite
  - Destroy ~ 1,000 lbs (450 kg) hydrazine fuel tank

• Preparation
  - 3 Standard Missiles-3 (SM-3), radars and system software extensively modified to enable intercept

• Engagement
  - 1 SM-3 launched by USS Lake Erie northwest of Hawaii
  - Successful intercept occurred ~153 miles (250 km) above the earth verified by 3 different phenomenologies

• Post Intercept
  - Analysis (as of 25 FEB 08) shows vast majority of intercept debris has already burned up upon reentering the Earth’s atmosphere, or will do so shortly – there have been no reports of debris landing on earth
  - The 3 Aegis ships have already been reconfigured to support BMD mission

Truth or Consequences?
A RESPECTFUL SUGGESTION TO PRESIDENT OBAMA

The President should ask the Secretary of Defense, Robert Gates; the Undersecretary of Defense, Ashton Carter; the MDA Director, General Patrick O’Reilly; and the MDA Director for Engineering, Keith Englander, if they can verify the accuracy of Lehner’s statement.

If they cannot verify the accuracy of Lehner’s statement, the President should find who was involved in generating this false claim.

Once identified, these individuals should be fired.
What Caused the Failure of the X-Band Radar in the FTG-06 Test of the Ground-Based Missile Defense System?

Briefing on Theater Missile Defense Technology Provided to Military Officers Visiting the MIT Security Studies Program in 1999 for Command School Training

MIT Lincoln Laboratory
244 Wood Street
Lexington, MA 02420-9108

Missile Defense Technology
(Can BMD Systems Work?)

Eric D. Evans
MIT Lincoln Laboratory

Mini DTS Course

10 December 1999
Briefing on Theater Missile Defense Technology Provided to Military Officers Visiting the MIT Security Studies Program in 1999 for Command School Training

Potential Sources of TBM Natural/Countermeasure Debris

I. Non-Separated Payloads
   (a) Liquid Fuel (Little or No Debris)
   (b) Chaffing
   (c) Tank Fragments
      RV

II. Separated Payloads
   (a) RV, Tank with Reentry Breakup Debris
      Liquid or Solid Fuel
   (b) Tank Reentry Breakup Debris
      Fuel Debris Cloud
      Solid Fuel

III. Intentional Exo Tank Breakup
   (a) Fragment Cloud
      Fragmentation/Detonation
   (b) Segmentation
       Interceptor

IV. Intercept

TMD Countermeasure Concepts

- Tumbling target
  - Missiles or RV
- Multiple objects
  - Frag/Segmentation, CSOs
- Orientation control
  - RV pointing or spin-up
- Anti-cuing tactics
  - Depl. stage disposal
- Maneuvers
  - Evasive corkscrew, etc.
- Submunitions
  - Early Release, CW, BW
- Signature control
  - Low RCS, IR coatings
- Enveloping structure(s)
  - Extended targets...
- Masking
  - Chaff, Flares, Corner Cubes
- Decays
  - Radar, IR
- Jammers
  - Escort, barrage, repeaters
- Other
  - Suits, ARMs, EMP...

MIT Lincoln Laboratory
Radar Discrimination Capabilities

Narrowband
Mean Unresolved RCS vs Scintillation Frequency

Wideband
Mean Unresolved RCS vs Length

IR Seeker Discrimination Capabilities

One - Color IR Seeker
Metrics: Mean IR Signature Scintillation Rate

Two - Color IR Seeker
Metrics: Emissivity - Area Temperature
IFT-6 Target Complex as Seen By Distant Approaching EKV

Range of Observed Target Complex ~ 230 – 250 km for FOV 1 – 1.5°

- 2.2 Meter Diameter Balloon (Roughly Ten Times Brighter than the Mock Warhead)
- Mock Warhead
- Rocket Stage that Deployed the Mock Warhead and Balloon

The Inflated Balloon is Heated by the Sun and is 7 to 10 Times Brighter Than the Warhead at Infrared Wavelengths

The Kill Vehicle Has Been Programmed In Advance to Select the Least Bright Object It Is Supposed to See. As Long As Nothing Is Done to Cause Another Object to Be the Least Bright Object, the Kill Vehicle Will Correctly Select the Warhead

Statement Indicating that Top Management of the Ballistic Missile Defense Organization Knew About the Discrimination Problems Identified in the IFT-1A Experiment

"So the decoy is not going to look exactly like what we expected. It presents a problem for the system that we didn't expect,"

Statement of Lieutenant General Ronald Kadish, Director of the Ballistic Missile Defense Organization, while being filmed by 60 Minutes II after learning that the 2.2 meter balloon misdeployed (did not inflate properly) during the IFT-5 experiment
IFT-6 Target Complex as Seen By Distant Approaching EKV

Range of Observed Target Complex ~ 230 – 250 km for FOV 1 – 1.5°

2.2 Meter Diameter Balloon (Roughly Ten Times Brighter than the Mock Warhead)

Mock Warhead

Rocket Stage that Deployed the Mock Warhead and Balloon

In The IFT-5, The Balloon Failed to Inflate, So Only the Canister, Instead of the Hot Inflated Balloon, Would Have Been Observed By the Kill Vehicle.

Since the Cannister Has a very Small Signal in the Infrared, It Is Now the Least Bright Object Observed by the Kill Vehicle.

Hence, The Kill Vehicle Would Now Select the Cannister as the Warhead.

The Kill Vehicle Must Determine If a Balloon Contains a Warhead or If the Balloon Is Empty!

Balloons that Have Been Flown in Space

These Could Be Used as Decoys or to Surround Warheads Disguising Them as Balloons
The Kill Vehicle Must Determine Which of These Are Warheads and Which are Decoys from 500 Kilometers Range!

Objects Flown in the IFT-1A and IFT-2 NMD Tests

<table>
<thead>
<tr>
<th>LARGEBAL</th>
<th>Large Balloon (2.2 Meter Diameter Balloon)</th>
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</thead>
<tbody>
<tr>
<td>SCLR</td>
<td>Small Canisterized Light Replica (Balloon)</td>
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<tr>
<td>MEDBALA</td>
<td>Medium Balloon A (0.6 Meter Diameter Balloon)</td>
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<tr>
<td>MEDBALB</td>
<td>Medium Balloon B (0.6 Meter Diameter Balloon)</td>
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<tr>
<td>MEDRLR1</td>
<td>Medium Rigid Light Replica 1 (2 Meters Long &amp; 0.6 Meter Base)</td>
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<tr>
<td>MEDRLR2</td>
<td>Medium Rigid Light Replica 2 (2 Meters Long &amp; 0.6 Meter Base)</td>
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<td>MSLS</td>
<td>Mission Service Launch System (Rocket Upper Stage)</td>
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<td>SCTBA</td>
<td>Small Cannisterized Traffic Balloon A (Small Balloon)</td>
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<tr>
<td>SCTBB</td>
<td>Small Cannisterized Traffic Balloon B (Small Balloon)</td>
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<tr>
<td>MRV</td>
<td>Medium Reentry Vehicle (2 Meters Long &amp; 0.6 Meter Base)</td>
</tr>
</tbody>
</table>
Expected Brightness and Fluctuation in the Brightness of the Objects in the IFT-1A and IFT-2 NMD Tests

\[ J_{\text{Composite}} = A_1 \int_{\lambda_1}^{\lambda_2} J(\lambda) \, d\lambda + A_2 \int_{\lambda_1}^{\lambda_2} J(\lambda) \, d\lambda \]

False Claim that Ground-Based Missile Defense Interceptors Based in Poland Would Be Able to Defend Japan from ICBMs Launched from Iran
Which of These Two MDA Contradictory Claims Are True?

The Missile Defense Agency Needs to Tell Us Which of Their Contradictory Statements Are True

- If the Poland-Based Ground-Based Interceptors (GBI) are fast enough to defend Japan from Iranian long range ICBMs, then the GBIs are more than fast enough to intercept Russian ICBMs.

- Alternatively, if the Poland-Based Ground-Based Interceptors are not fast enough to intercept Russian ICBMs, than they are certainly not fast enough to defend Japan from Iranian long range ICBMs.
Relevant Observations:

- Radar in Czech Republic Not Used
- Intercept Achieved with FBX or Adjunct Radar Tracking from Eastern Turkey
- Interceptor Speed 40% Faster Than 6.3 km/sec Speed Claimed by US administration
- HOWEVER, MDA CONTINUES TO REVISE AND CHANGE ITS STATEMENTS ABOUT THE CHARACTERISTICS OF THE POLISH-BASED INTERCEPTORS
Missile Defense Agency Statement About the Poland-Based Interceptor Characteristics Revised to Explain Original Claim of 6.3 km/sec Burnout Speed

Relevant Observations:

- Predicted Interceptor Burnout Speed Drops from 9.4 km/sec to 7.5 km/sec
- Interceptor Can No Longer Achieve Defense-Coverage of Japan, As Claimed by Missile Defense Agency
- Interceptor Speed Still 20% Higher Than 6.3 km/sec Speed Originally Claimed by US administration
- Interceptor Still fast Enough to Achieve Intercepts Against Russian ICBMs, Although Only for Trajectories Towards the East Coast of the US

Hence, Defense Coverage Claimed by Missile Defense Agency Must Be Wrong!

Notional Intercept Trajectory for 9.4 km/sec Interceptor Launched from Poland

- Radar in Czech Republic Not Used
- Intercept Achieved with FBX or Adjunct Radar Tracking from Eastern Turkey
- Interceptor Speed 40% Faster Than 6.3 km/sec Interceptor Speed Claimed by US administration
Revised Interceptor Characteristics Indicates
Defense-Coverage Claimed by Missile Defense Agency Must Be Wrong!

- Radar in Czech Republic Not Used
- Intercept Achieved with FBX or Adjunct Radar Tracking from Eastern Turkey
- REVISED MDA Interceptor Parameters Give Burnout Speed of 7.5 km/sec, 15% Faster Than Originally Claimed by MDA Spokesman and Chief Scientist!

Capability Provided Versus Iranian Intermediate To Long-Range Ballistic Missiles

- NO STATEMENT ON MDA SLIDE THAT SAYS HOKKAIDO CAN BE DEFENDED WITH INTERCEPTORS FROM ALASKA!
- Can be covered by PAC-3, Aegis, THAAD or NATO-deployed systems
- BMD System W/Interceptor Field (Poland) + Midcourse Radar (Czech Republic) + Forward Based Radar
False Claims Made in Presentations to European (and Japanese?) Allies by Missile Defense Agency that US Proposed European Missile Defense Can Defend Northern Japan

**NO PLASIBLE WAY FOR DEFENSE SYSTEM TO OBTAIN PRECISION TRACKING DATA NEEDED TO GUIDE INTERCEPTORS FROM ALASKA!**

FBX in Eastern Turkey too Far from Missile Trajectory to Track the Deployed Warhead (RCS ~ 0.01 m²)

X-Band Radar in Czech Republic Below Radar Horizon

Launch from Iran

Sea-Based X-Band Radar Off Adak Below Radar Horizon

FBX in Eastern Turkey too Far from Missile Trajectory to Track the Deployed Warhead (RCS ~ 0.01 m²)

NO PLASIBLE WAY FOR DEFENSE SYSTEM TO OBTAIN PRECISION TRACKING DATA NEEDED TO GUIDE INTERCEPTORS FROM ALASKA!
False Claims Made in Presentations to European (and Japanese?) Allies by Missile Defense Agency that US Proposed European Missile Defense Can Defend Northern Japan

FBX in Eastern Turkey too Far from Missile Trajectory to Track the Deployed Warhead (RCS ~ 0.01m^2)

FBX in "Caspian-Sea Region" Only Radar That Could Provide Track Data for Intercept of the Postulated ICBM Warhead (RCS ~ 0.01m^2)

NO PLAUSIBLE WAY FOR DEFENSE SYSTEM TO OBTAIN PRECISION TRACKING DATA NEEDED TO GUIDE INTERCEPTORS FROM ALASKA!

False Claims to European Allies that Ground-Based Missile Defense Interceptors in Poland are Not Fast Enough to Engage Russian ICBMs
Which of These Two MDA Contradictory Claims Are True?

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- If the Poland-Based Ground-Based Interceptors (GBI) are fast enough to defend Japan from Iranian long range ICBMs, then the GBIs are more than fast enough to intercept Russian ICBMs.

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Interceptors Cannot Catch Russian Missiles

U.S. European Interceptor Site Cannot Affect Russian Strategic Capability
Engagement Event Timeline for Engagement of SS-25 from Vypolzovo with 2-Stage Missile Defense Interceptor

T=0 minutes
Interceptor Launch

T=0.17 minutes
Interceptor Burnout

T=0 minutes
Interceptor Launch

T=0.50 minutes
Infrared Satellite Detection

T=0 seconds
Launch Transient Detection?

T=5 minutes
Interceptor and warhead Collide

INTERCEPT!
T=5 minutes

T=2 minutes
T=1 minute
T=0 minutes
Interceptor Launch

T=170 sec
End of SS-25 Powered Flight, Also Point of Radar Acquisition

T=58 sec
Infrared Satellite Detection

800 sec
600 sec
400 sec

Time (sec) after Russian ICBM Launch

U.S. European Interceptor Site Cannot Affect Russian Strategic Capability
The North Korean Unha-2 Space-Launch Vehicle
The Evolution of Iranian and North Korean Rocket Vehicles

SCUD-B
- 3,786 kg Propellant
- Propellant Density: 1,223 kg/m³
- Oxidizer to Fuel Volume = 1.83
- Oxidizer to Fuel Weight = 3.4
- Actual Propellant Density = 1,266 kg/m³
- 0.944 of Fuel Tank Volume Filled

Shahab-1
- 16.5 m
- 12,400 kg

Shahab-2
- 14.0974
- 3,275 kg fuel for 298 sec burn and 3% residual fuel

Shahab-3
- 16.06 m
- 13,570 kg

Shahab-3M
- 15.5 m
- 14,900 kg

SCUD-B

SCUD-C

Nodong

First Stage Derived from Nodong
Second Stage Derived from SCUD and SA-5
Third Stage Derived from SS-21

First Stage Uses Cluster of Four Nodong Motors
Second Stage is an SS-N-6
Third Stage Same as the Second Stage from the Safir SLV

Video Frames Showing the Initial Acceleration at Launch of the Unha-2

Time After Launch: 0 seconds

Time After Launch: 2.27 seconds

Time After Launch: 3.00 seconds
Apparent Mock Up of Cluster of Four Nodong Rocket Motors Displayed by Iran, Possibly Replicating Motor Assembly from the North Korean Unha-2 First Stage

Source: Composite of two video frames constructed from http://www.youtube.com/watch?v=nZoNdfbhlI&feature=player_embedded

Observed Acceleration at Launch of the Unha-2

Unha-2 Launch Vehicle

Acceleration Rate = 0.34483 Gs
Estimated Dimensions of the Unha-2 Launch Vehicle

Rocket Components that *Might* Have Been Used to Construct the North Korean Unha-2
Expected and Actual Flight Outcomes Associated with the North Korean Satellite Launch Attempt of April 4/5, 2009

Vehicle Locations Shown at 10 Second Intervals

First Stage Impact Range = 500 to 700 km
Second Stage Impact Range = 3,150 to 4,000 km


First Stage Impact Range = 500 to 700 km
Second Stage Impact Range = 3,150 to 4,000 km
Powered and Free-Flight Profile of Notional Safir Ballistic Missiles Carrying Different Payloads

Vehicle Locations Shown at 10 Second Intervals
Vehicle Gross Weight Without Warhead (kg) = 27150
First Stage Burn Time (sec) = 151.6935
Second Stage Burn Time (sec) = 62

Stage 1 Characteristics:
Stage Full Weight (kg) = 22800
Structure Factor = 0.10088
Stage Propellant Weight (kg) = 20500
Weight of Burned Fuel (kg) = 19680
Stage Empty Weight (kg) = 2300
Stage Weight at Burnout (kg) = 3120
Fraction of Fuel Unburned = 0.04
Residual Unburned Fuel (kg) = 820
Burn Time (sec) = 151.6935

Stage 2 Characteristics:
Stage Full Weight (kg) = 4350
Structure Factor = 0.126
Stage Propellant Weight (kg) = 3800
Weight of Burned Fuel (kg) = 3610
Stage Empty Weight (kg) = 550
Stage Weight at Burnout (kg) = 740
Fraction of Fuel Unburned = 0.05
Residual Unburned Fuel (kg) = 190
Burn Time (sec) = 62

Strategic Great-Circle Distances From Iran and Korea to Parts of the United States

North Korea to the US

Iran to US
Range-Payload Capabilities of Two and Three Stage Variants of the North Korean Unha-2

**Range versus Payload of Two and Three-Stage Unha-2 Variants**

![Graph showing range versus payload for Unha-2 variants](image)

**Range Versus Payload for Iranian and North Korean Ballistic Missiles**

**Range Versus Payload of Two and Three Stage Taepodong-2 Ballistic Missiles and the SCUD-B, SCUD-C, Shahab-3 and Shahab-3M, and Safir**

![Graph showing range versus payload for ballistic missiles](image)
blast kills civilians, 4 foreigners in NW Pakistan

Iran has no problem with its neighbors
Iranian Simorgh Space Launch Vehicle
The MDA’s False Claims About the Range and Discrimination Capabilities of the European Midcourse Radar
Radar-Range Fans for US Proposed EMR and FBX Missile Defense Radars

Locations of Postulated ICBMs Launched from Iran to the Continental United States at One Minute Intervals

Radar-Range Fans for Vardo and US Proposed EMR and FBX Missile Defense Radars
Discrimination Ranges that Could Actually be ACHIEVED by the EMR and FBX Radars
Ground-Based Interceptor: Missile Defense Agency Claims It Achieves 6.3 km/sec Carrying a Payload of 120 – 130 kg, Pegasus Parameters Indicate a 6.3 km/sec Burnout Speed with a Roughly 900 – 1000 kg Payload

Evolution and Comparison of Launch Vehicles, ICBMs and the GBI Interceptor

- Pegasus Evolution
- GBI Evolution
- ICBMs

Comparison of Minuteman III and Midgetman ICBM

Launch Weight with 155 lb EKV = 47,655 lbs
Launch Weight with 132 lb EKV = 31,473 lbs
Ground-Based Interceptor
Achieves 8.5 to 8.7 km/sec Carrying a Payload of 220 to 155 lbs

The GBI Has a Higher Lift Capability than the US Midgetman ICBM!

The Ground-Based Interceptor
Can Carry a Full Minuteman III BUS and Three Warheads to 6,000+ Kilometers
False Claims to European Allies About the “Theoretical” Capabilities of the Europe-Based Missile Defense Components

Concerns Expressed by the Russians

Engagement With Russia


- April 3, 2006 (Moscow): Briefing of Russian officials by U.S. Embassy (Moscow) on DOD decision to resume consultations with Poland regarding the site of U.S. missile defense assets

- November 3, 2006 (Moscow): Dr. Cambone, Lt Gen Obering, DASD Green, Russian Minister of Defense Ivanov, Chief of General Staff Gen-Col Paluevskiy, Gen-Col Mazurkevich
  - Russians did not acknowledge Iran emerging threat as a rationale for deployment of U.S. missile defense assets
  - Believe Russia is real target
  - Russians “portrayed” lack of understanding and confusion on technical aspects of a deployed missile program and proposed architecture.

  U.S. committed to following-up with technical discussions to Russian counterparts

  - Ambassador re-committed that U.S. will follow-up with technical briefings/explanations regarding U.S. missile deployment

- February 9, 2007 (Seville): Secretary Gates and Minister of Defense Ivanov during NATO-Russia Council Ministerial meeting

  U.S. Has Offered Future Event Establishing Technical Experts Meeting (Spring 2007)
The interceptors planned for Poland are nearly identical to the three-stage interceptors based in the U.S. except that they are a two-stage variant that is quicker, lighter, and better suited for the engagement ranges and timelines for Europe. The silos that house the ground-based interceptors have substantially smaller dimensions (e.g., diameter and length) than those used for offensive missiles, such as the U.S. Minuteman III ICBM. Any modification would require extensive, lengthy, and costly changes that would be clearly visible to any observer.

The ground-based interceptors are comprised of a booster vehicle and an exoatmospheric kill vehicle (EKV). Upon launch, the booster flies to a projected intercept point and releases the EKV which then uses on-board sensors (with assistance from ground-based assets) to acquire the target ballistic missile. The EKV performs final discrimination and steers itself to collide with the enemy warhead, destroying it by the sheer kinetic force of impact.

Interceptor Upper Stage Loses 600 lbs of Propellant Relative to Commercial Rocket, Motor casing Becomes 600 lbs Heavier, Lower Rocket Stage Motors 5% Less Efficient Than Commercial Version.

Full and Empty Weights Plus Isp for Stages 1 and 2 from Taurus and Pegasus Commercial User’s Manuals Gives 8.9 km/sec Interceptor Burnout Speed.

The Constantly Changing Stories from the Missile Defense Agency

Four Contradictory Sets of Characteristics Describing the Performance of the Ground-Based Interceptor

1. Burnout Speed = 6.3 km/sec
2. Stage Full and Empty Weights Provided to the Associated Press by Colonel Rick Lehner, Spokesman for the Missile Defense Agency.
3. Full and Empty Weights Plus Isp for Stages 1 and 2 in “Response to Postol” (Still Gives 7.5 km/sec Interceptor! – Interceptor Upper Stage Loses 600 lbs of Propellant Relative to Commercial Rocket, Motor casing Becomes 600 lbs Heavier, Lower Rocket Stage Motors 5% Less Efficient Than Commercial Version)
4. Full and Empty Weights Plus Isp for Stages 1 and 2 from Taurus and Pegasus Commercial User’s Manuals Gives 8.9 km/sec Interceptor Burnout Speed.
The Constantly Changing Stories from the Missile Defense Agency

(2 of 2)

Multiple and Changing Explanations of How Defended Areas Are Expanded by the Addition of European Defense Components

(EMR in Czech Republic, Two-Stage Interceptors in Poland, and FBX at Unspecified Location)

1. Addition of Interceptors in Poland Makes It Possible to Defend Hokkaido, Japan!

2. 6.3 km/sec Interceptor Not Fast Enough to Defend Hokkaido!
   (Roughly 9 km/sec is Needed).

3. Czech Radar Could Play No Role in Defense of Hokkaido!

4. Alaska Radars Could Play No Role in Defense of Hokkaido!

5. Postol “Misinterpreted” Missile Defense Agency Slides!
   Interceptors from Alaska Are Used to Defend Hokkaido!

6. Forward-Based X-Band Radar Might Be Able to Provide Tracking for Interceptors If It Is Deployed in the Caspian Sea!

Details Associated with the Contradictions and False Claims Being Made By the US Missile Defense Agency About the Two-Stage Poland-Based Interceptor
Data from Press Statements by Spokesman and Chief Scientist for the Missile Defense Agency, Colonel Rick Lehner and Mr. Keith Englander
Provided Stage Weights for the Orbital Sciences Two-Stage Ground-Based Interceptor

**STATMENTS MADE BY MDA TO THE PRESS:**

- Launch Weight = 47,400 lbs
- First Stage Weight = 37,800 lbs
- Second Stage Weight = 9,500 lbs
- Kill Vehicle Weight = 155 lbs
- Burnout Speed = 6.3 km/sec

**ANALYTIC RESULTS:**

- **Assumptions:**
  The shroud weighs 200 lbs, and the Pegasus-derived rocket motor fuel weights and specific impulses are exactly those from the *AIAA International Reference Guide to Space Launch Systems*.

- **Expected Launch Weight of GBI** = 37,800 + 9,500 + 155 + 200 = 47,655 lbs.

- **The vehicle weight stated by Lehner is 47,400 lbs**

- **If one assumes a vehicle with a Launch Weight of 49,730 lbs, a payload of 2075 + 155 = 2230 lbs,** the burnout speed is **6.30 km/sec**.

- **The same vehicle carrying a 155 lb payload achieves a burnout speed of 9.37 km/sec.**

- **If the vehicle payload is 255 lbs, to accommodate a 100 lb vibration isolation and mounting adapter, (and/or endo/exo heatshield protection for EKV) the burnout speed is then 9.11 km/sec**

**CONCLUSION**

US Interceptors will have sufficient speed to engage all Russian ICBMs launched from West of the Urals against all targets in the continental United States

---

**Data on Ground-Based Interceptor Launch Gross Weight, Stage Weights and Burn Times**

Provided by MDA Spokesman, Rick Lehner, and MDA Chief Scientist, Keith Englander

**Orion 50SXLG Rocket Motor**

<table>
<thead>
<tr>
<th>Source</th>
<th>Full Weight (lbs)</th>
<th>Propellant (lbs)</th>
<th>Empty Weight (lbs)</th>
<th>Burn Time (sec)</th>
<th>$I_o$ (sec$^{-1}$)</th>
<th>Length (m)</th>
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<td>Taurus</td>
<td>??</td>
<td>33,120</td>
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<td>68.4</td>
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<td>8.94</td>
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<td>34,398</td>
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**Orion 50XL Rocket Motor**

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<th>Source</th>
<th>Full Weight (lbs)</th>
<th>Propellant (lbs)</th>
<th>Empty Weight (lbs)</th>
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<td>1,425?</td>
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**Two-Stage GBI Launch Weight = 21,400 kg (47,400 lbs)**

GBI Carries No Ballast
Ability of Polish-Based Interceptors to Engage Russian ICBMs from Tatischevo

The Record of Initial Integrated Flight (IFT’s) Tests 1A Through 9
Rigging of the Test Program to Avoid the Simplest of the Baseline Threats

Scintillating Targets Removed from Test Program

Original Plans to Fly Ten or More Objects in IFT-3 and IFT-4 Experiments
### IFT Targets Selections

#### As of 05/05/00 (U)

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<th>Date (mm/dd)</th>
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<th>LB</th>
<th>CSB-1</th>
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<th>MB</th>
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<th>SCLR</th>
<th>LBB</th>
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**Note:** Configuration controlled by NMD JPQ – Do not alter this document.

**OTA TESTS**

#### Flight Path Conditions of IFT-1A Through IFT-10 Experiments
Integrated Flight Test-10 failed, but was supposed to be an attempt to demonstrate an intercept at night.

**Intercept Conditions**
- **Altitude:** 230 km
- **Location:** 680 km from Kwajalein
- **Speeds at Intercept:** 2.1 km/sec and 6.5 – 6.6 km/sec

**Specialized Alignment of Large Balloon, Mock Warhead, and Upper Stage During All Integrated Flight Tests**

**IFT-6 Target Complex as Seen By Distant Approaching EKV**

Range of Observed Target Complex ~ 230 – 250 km for FOV 1 – 1.5°

- **2.2 Meter Diameter Balloon** (Roughly Ten Times Brighter than the Mock Warhead)
- **Mock Warhead**
- **Rocket Stage that Deployed the Mock Warhead and Balloon**

* Integrated Flight Test-10 failed, but was supposed to be an attempt to demonstrate an intercept at night.
"So the decoy is not going to look exactly like what we expected. It presents a problem for the system that we didn't expect,"

Statement of
Lieutenant General Ronald Kadish,
Director of the Ballistic Missile Defense Organization,
while being filmed by 60 Minutes II after learning that
the 2.2 meter balloon misdeployed (did not inflate properly)
during the IFT-5 experiment
IFT-6 Target Complex as Seen By Distant Approaching EKV

Range of Observed Target Complex ~ 230 – 250 km for FOV 1 – 1.5°

Balloon Canister is Now the Least Bright Object: Hence, the Balloon Canister Looks Like the Warhead

2.2 Meter Diameter Balloon (Roughly Ten Times Brighter than the Mock Warhead)

Mock Warhead

Rocket Stage that Deployed the Mock Warhead and Balloon

~3 km

~3.5 km

False Targets Cloud Created in Army Ballistic Missile Development Agency Test Using a Titan II ICBM on January 10, 1975, Signature of Fragmented Tanks (SOFT),