

# **Frontiers & Challenges in CAE Simulations**

**Klaus-Jürgen Bathe**

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Barcelona, Spain, September 2018**

# Content of Presentation

- **Philosophy adopted in our research and developments**
- **Analysis of shells – the MITC3+ and MITC4+ elements**
- **3D solids – the 3D-MITC8 element**
- **The enriched subspace iteration scheme for frequency solutions**

- **Direct time integration**
- **Multiphysics problems, FSI with EM**
- **Molecular structures – DNA and Proteins**
- **Overlapping finite elements and the “AMORE paradigm”**
- **Concluding remarks**

# **Our research and development 'philosophy':**

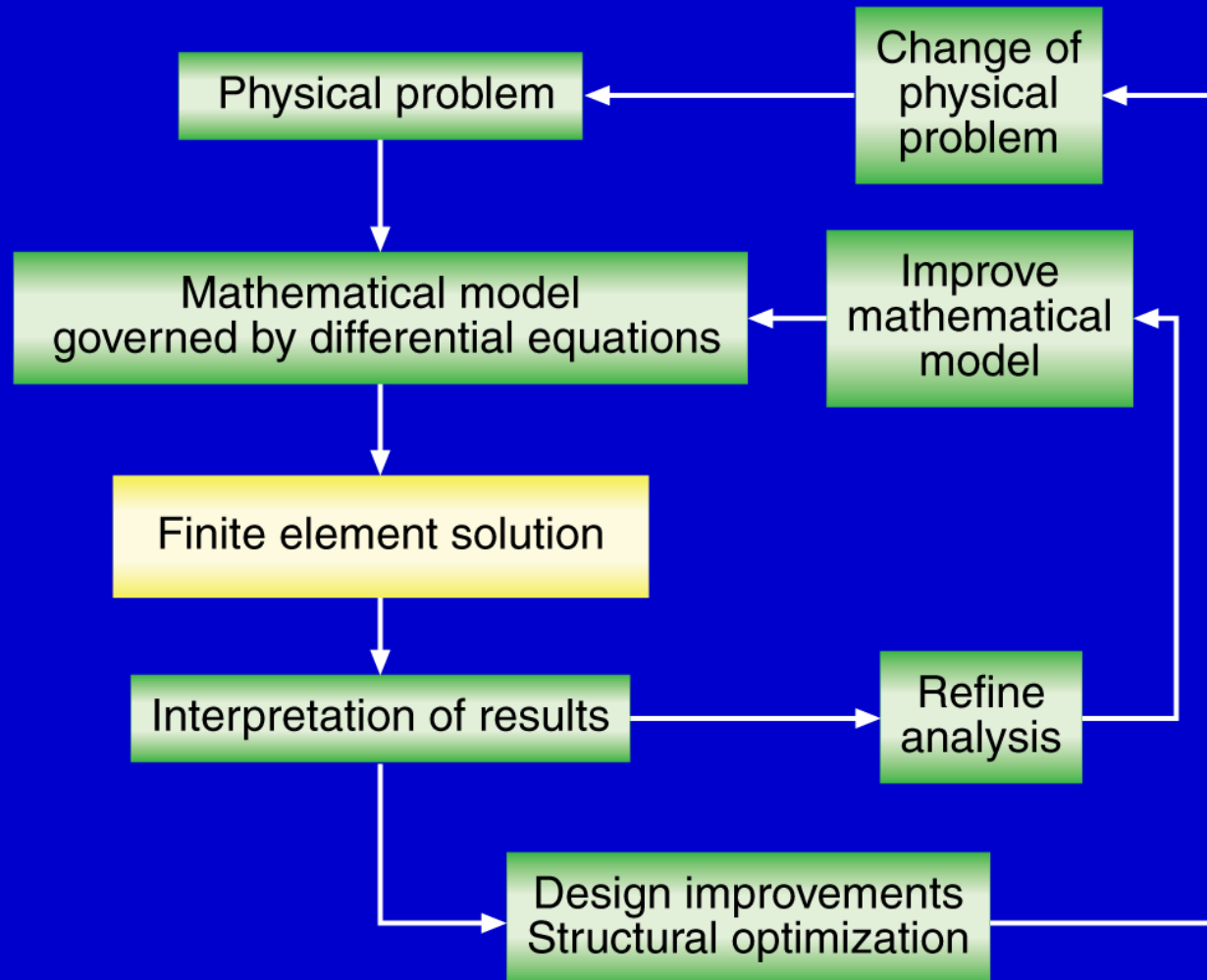
**To develop “reliable and efficient procedures”  
that can be used in general to --**

**advance simulations as practiced in  
industry and the sciences;**

**simulations of “structures” in multiphysics  
environments**



# The process of modeling for analysis



**Our focus: The analysis of problems involving ---**

**Solids and shells, general structures, from the km-scale to the nano-scale**

**Multi-physics media governed by the Navier-Stokes and Maxwell's equations**

**The fully coupled response**

# The analysis of shells

“General shell elements” widely used  
do not include the through -the-  
thickness stress (and strain ), e.g.  
the MITC4 element

But in some analyses this stress  
can be important --- like metal forming,  
contact problems

# The analysis of shells:

## Testing of elements

Considering the solution scheme ---  
convergence should be optimal for all problems  
and independent of the shell thickness

$$\| \text{error} \| \approx c h^k$$

# A general mixed formulation is:

Find  $U \in \Sigma$  and  $N \in \Xi$  such that

$$\tilde{A}(U, V) + B(N, V) = F(V)$$

$$\forall V \in \Sigma$$

$$B(Q, U) - t^2 C(Q, N) = 0$$

$$\forall Q \in \Xi$$

Choose  $\Sigma_h$  and  $\Xi_h$  for the finite element spaces

**The inf-sup condition is on the bilinear form B !**

## For the discretization, we

- must have no spurious element mode, patch tests satisfied, element spatial isotropy
- must have consistency
- want optimal convergence, inf-sup condition is satisfied

Want a scheme with no (artificial numerical) stabilization factors

**The question of what norm to use:**

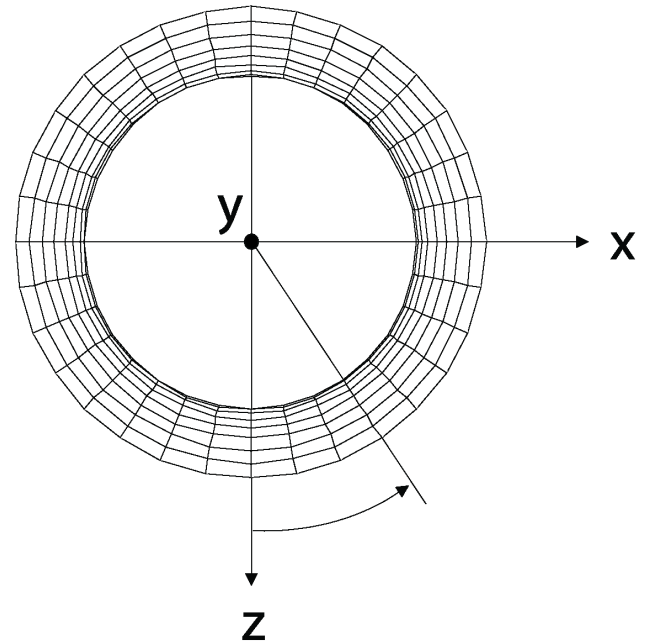
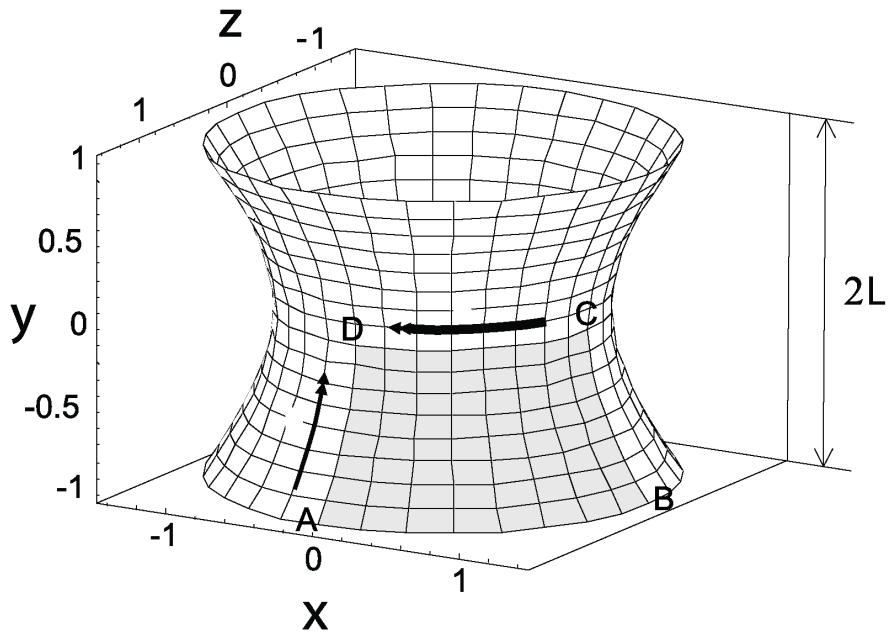
**Point values are meaningless**

**Difference in energies is not good for mixed formulations**

**‘s-norm’ (F Hiller, KJ Bathe) is valuable, it seems to be the only norm that can be used for all types of shell problems**

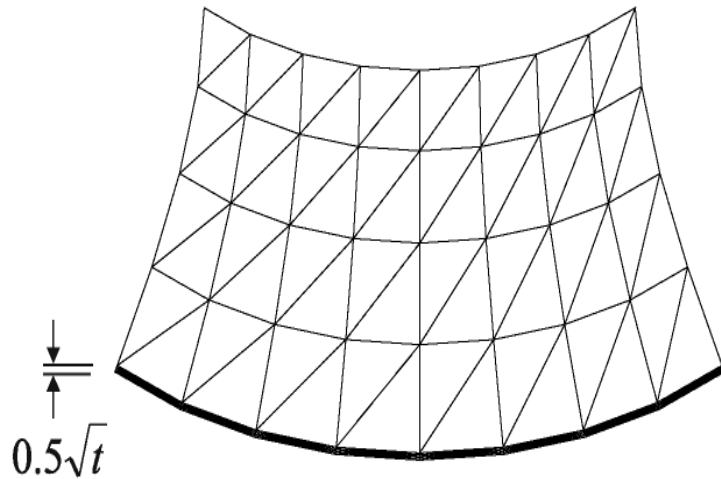
$$\|\cdot\|_s^2 = \int_{\Omega} (\text{stress error}) \cdot (\text{strain error}) dV$$

**We have used this norm extensively**

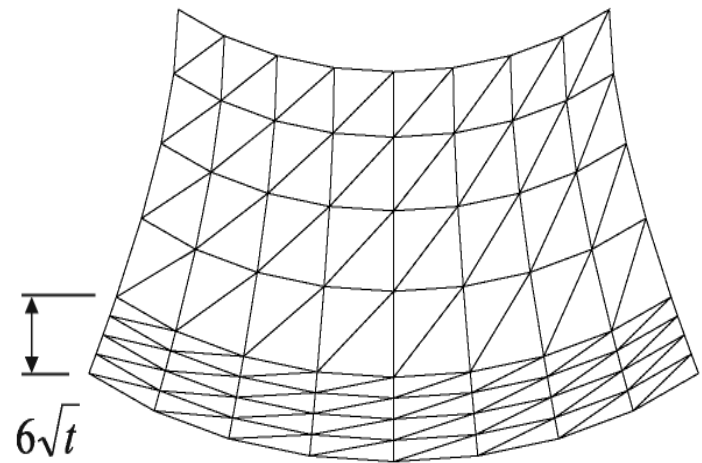


**The hyperbolic shell test problem,  
free and fixed ends**





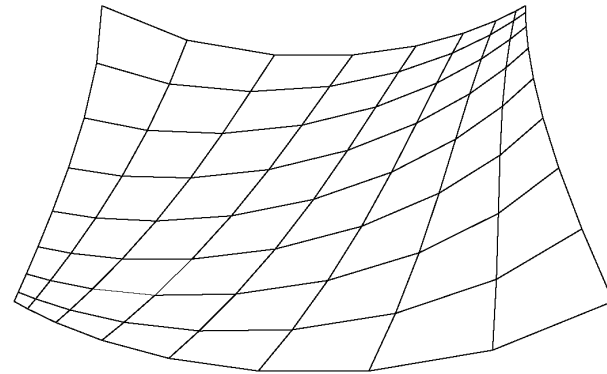
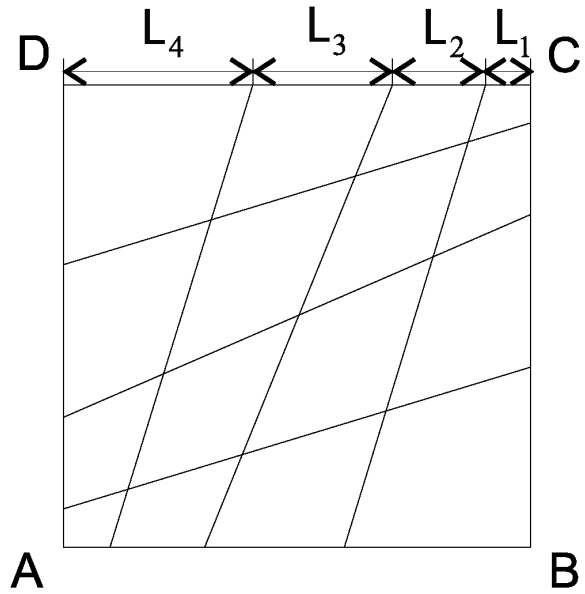
(a)



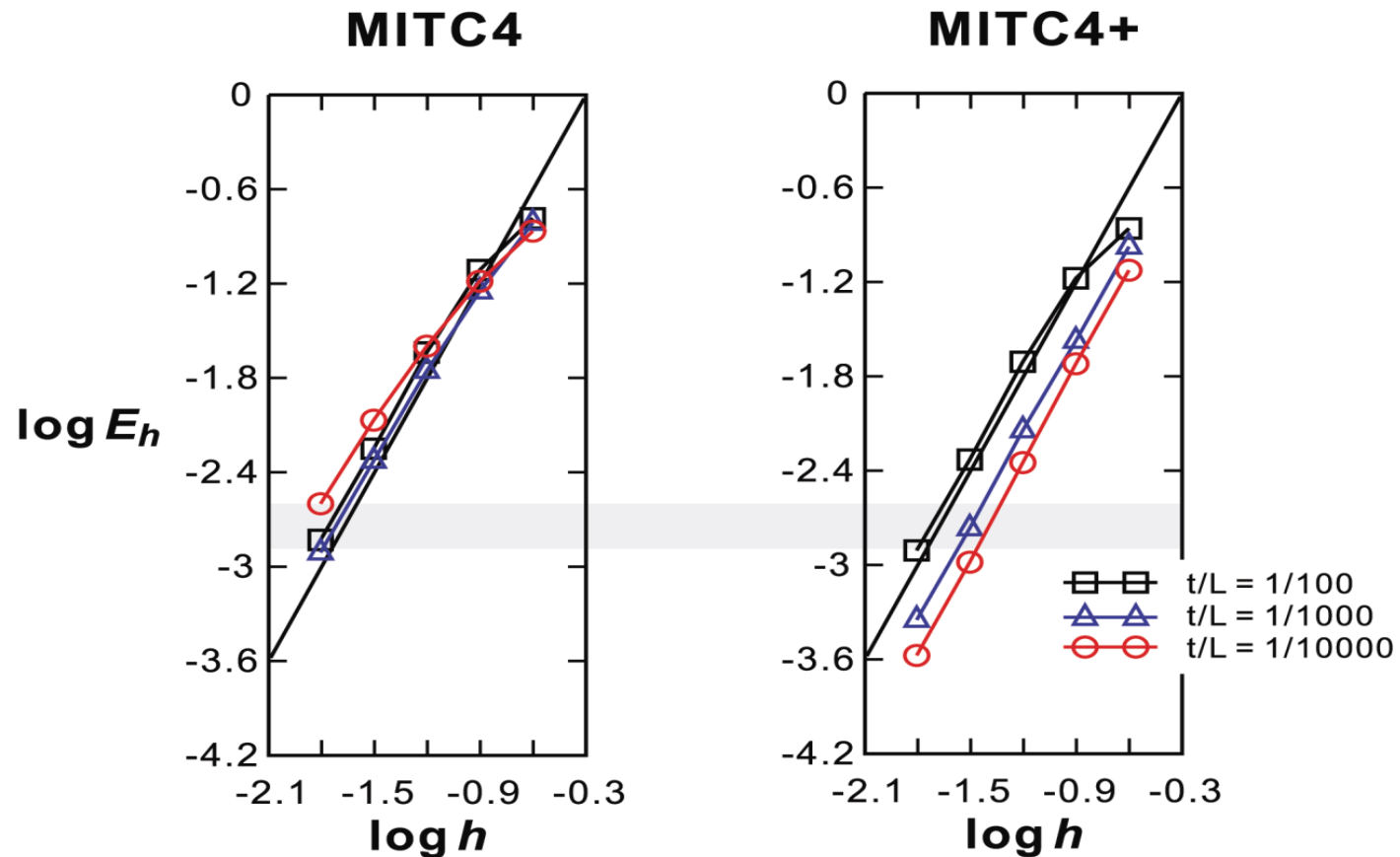
(b)

**Use of 3 and 4- node shell elements, with boundary layers**

**Meshing used for 1/8<sup>th</sup> of the structure**



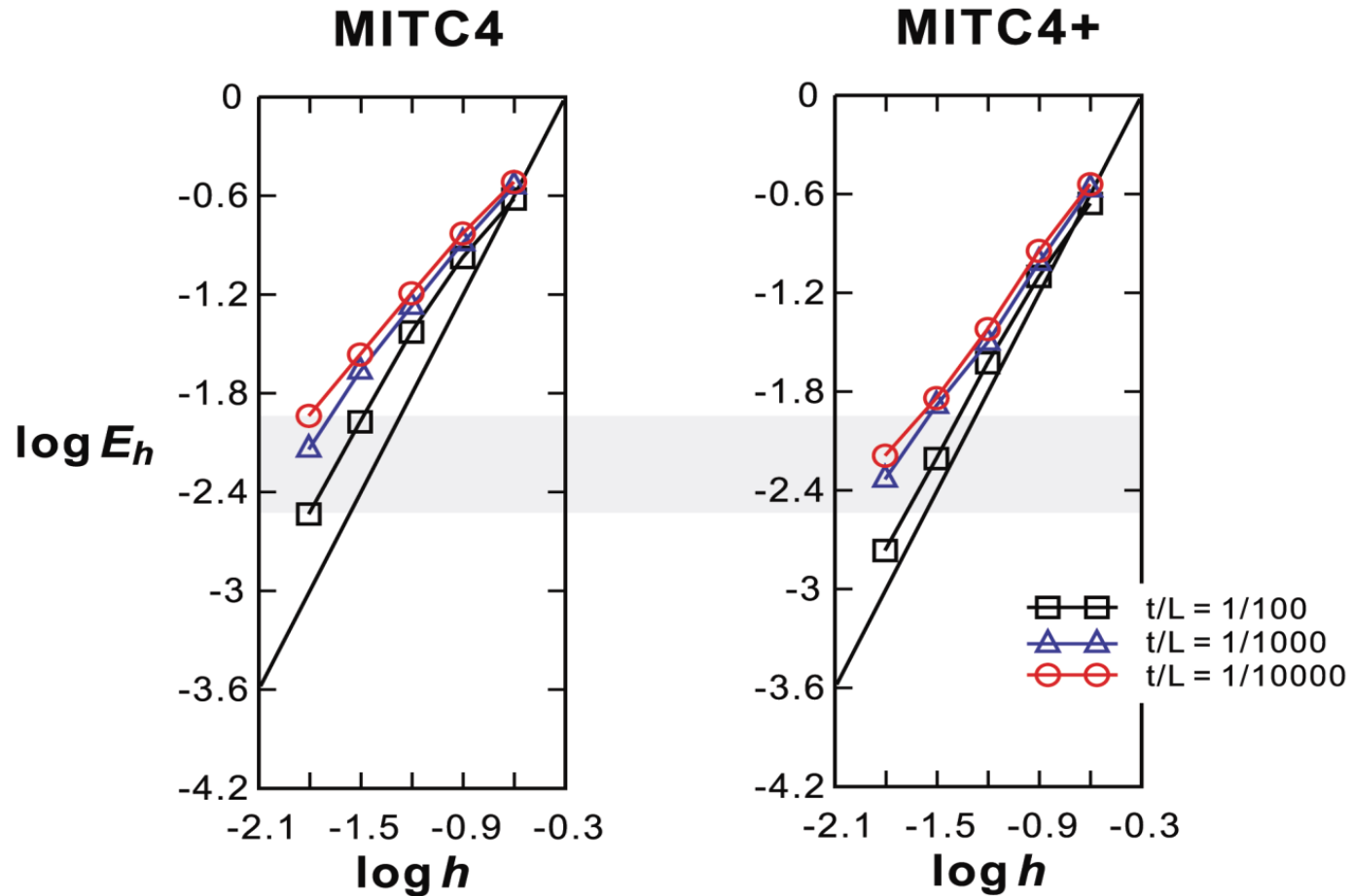
**Distorted meshes used**



(a)

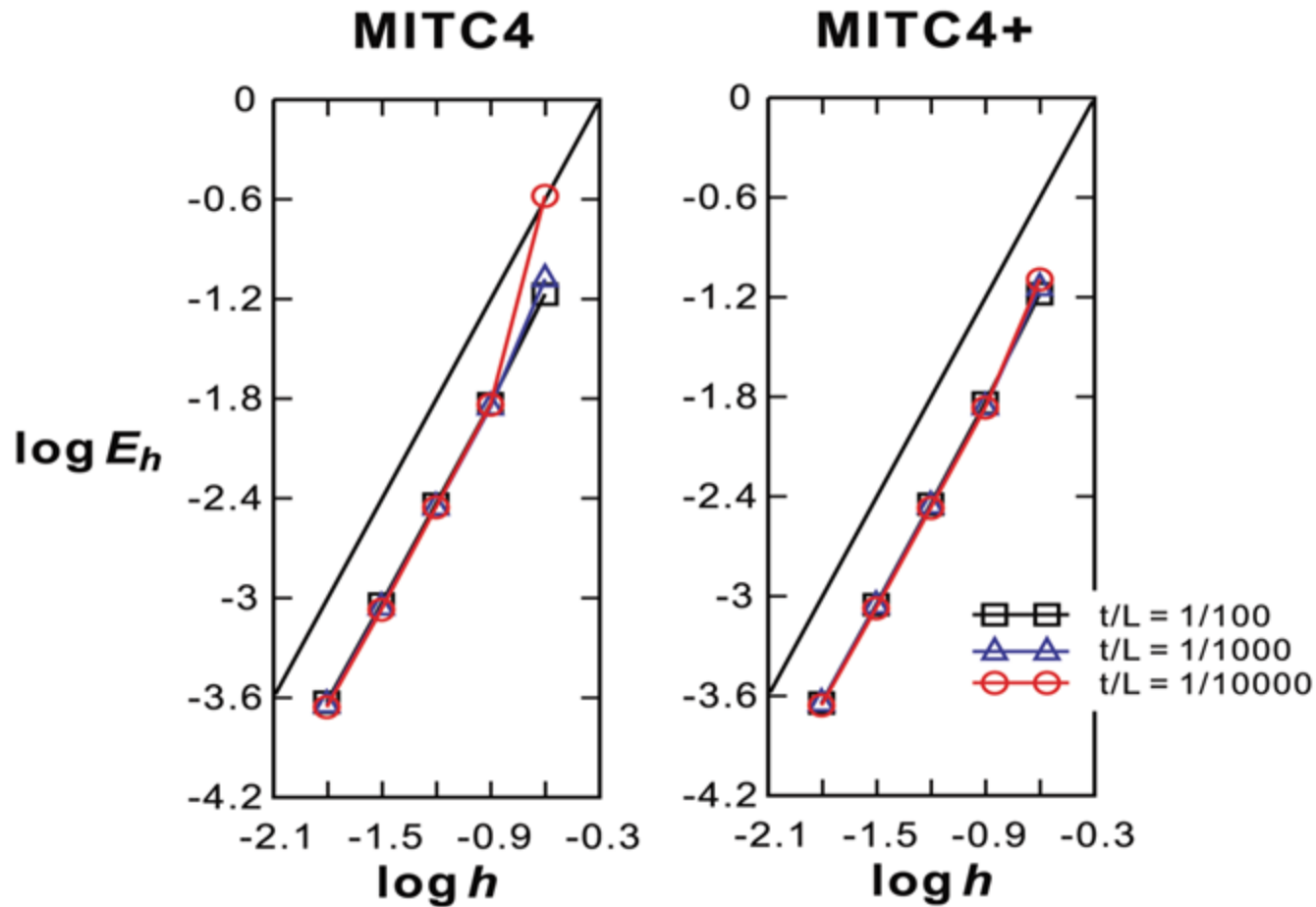
**Clamped shell results; regular meshes**

**Y Ko, PS Lee, KJ Bathe. A new 4-node ... C & S 2017**



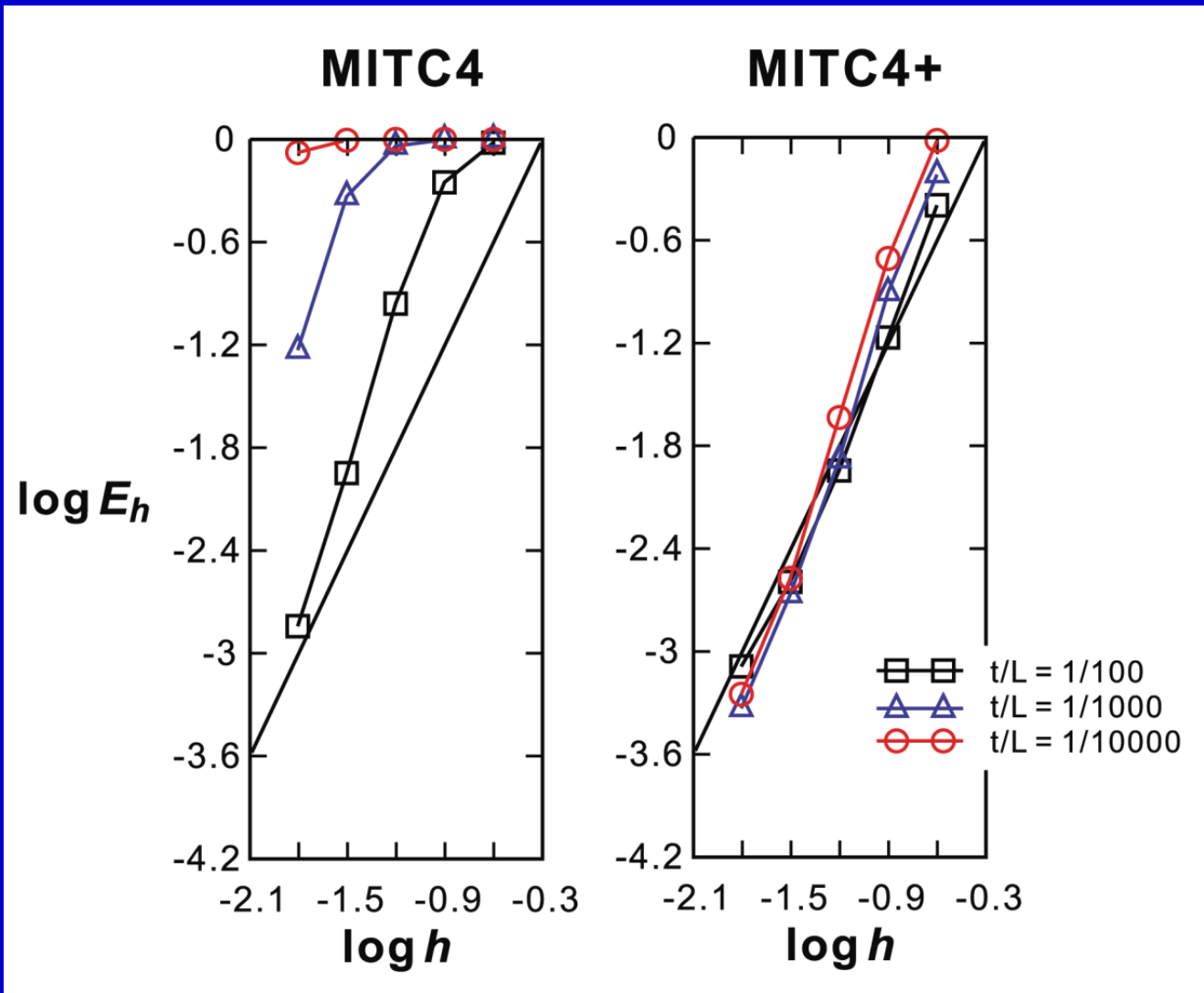
(b)

**Clamped shell results; distorted meshes**

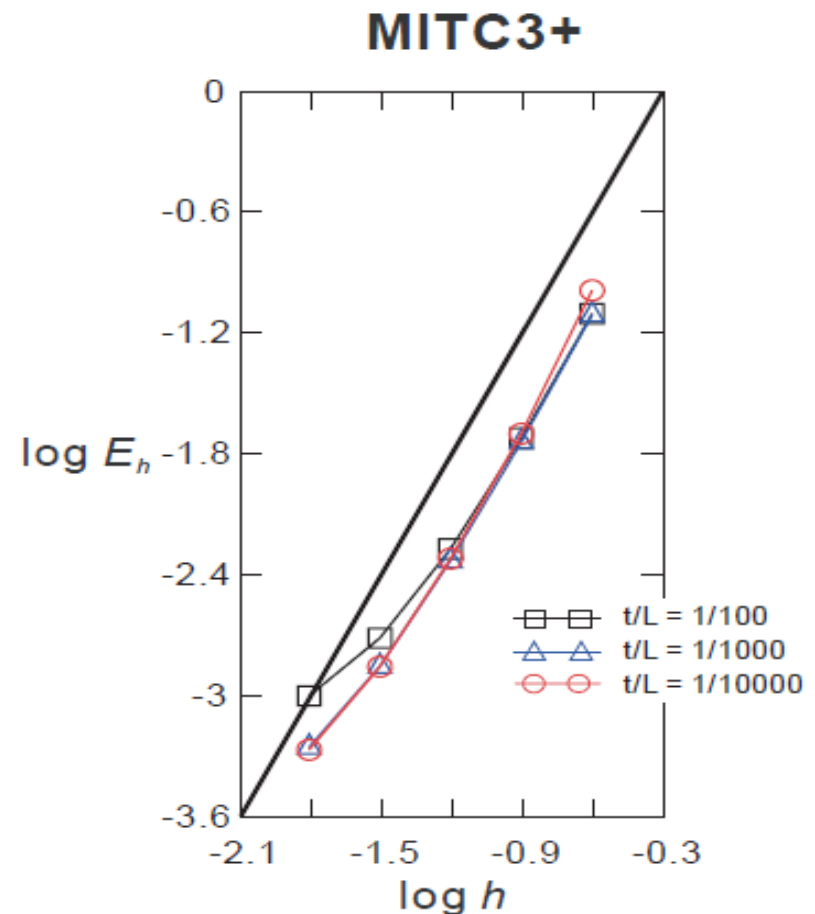
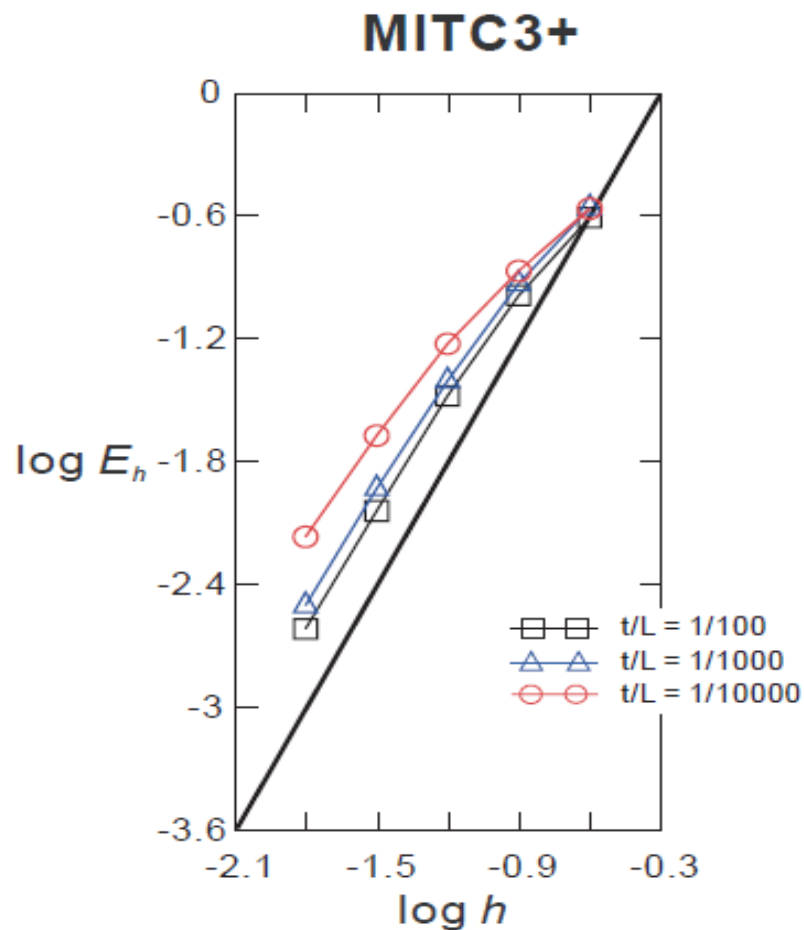


(a)

**Free shell results; regular meshes**



**Free shell results; distorted meshes**



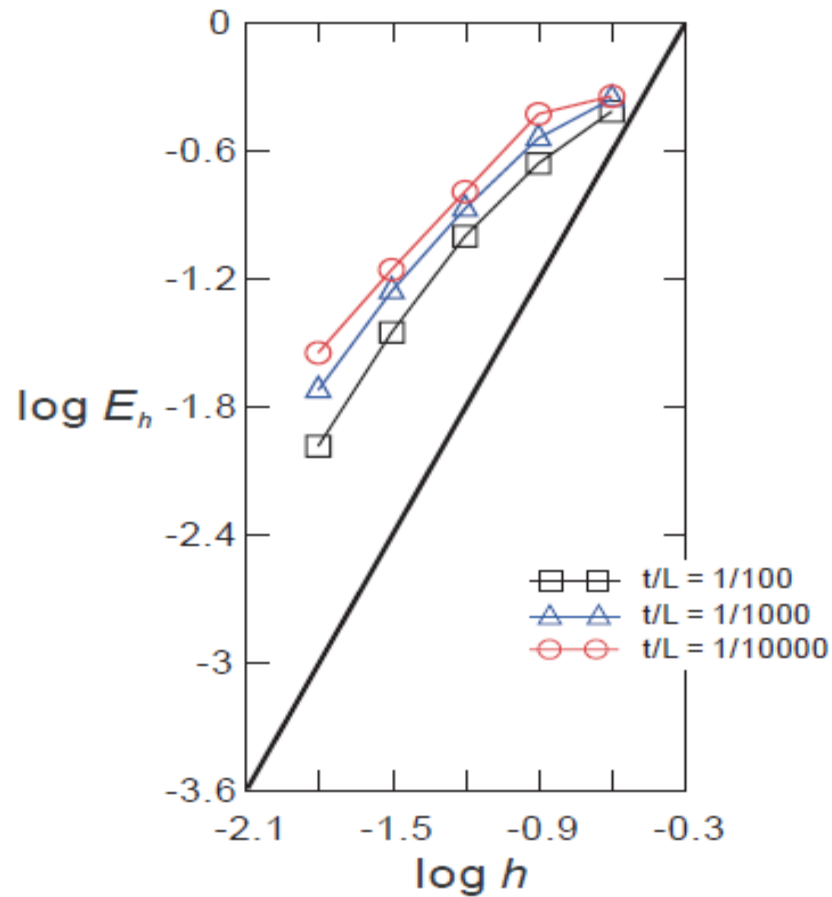
**Regular meshes**

**left: fixed structure, right: free structure**

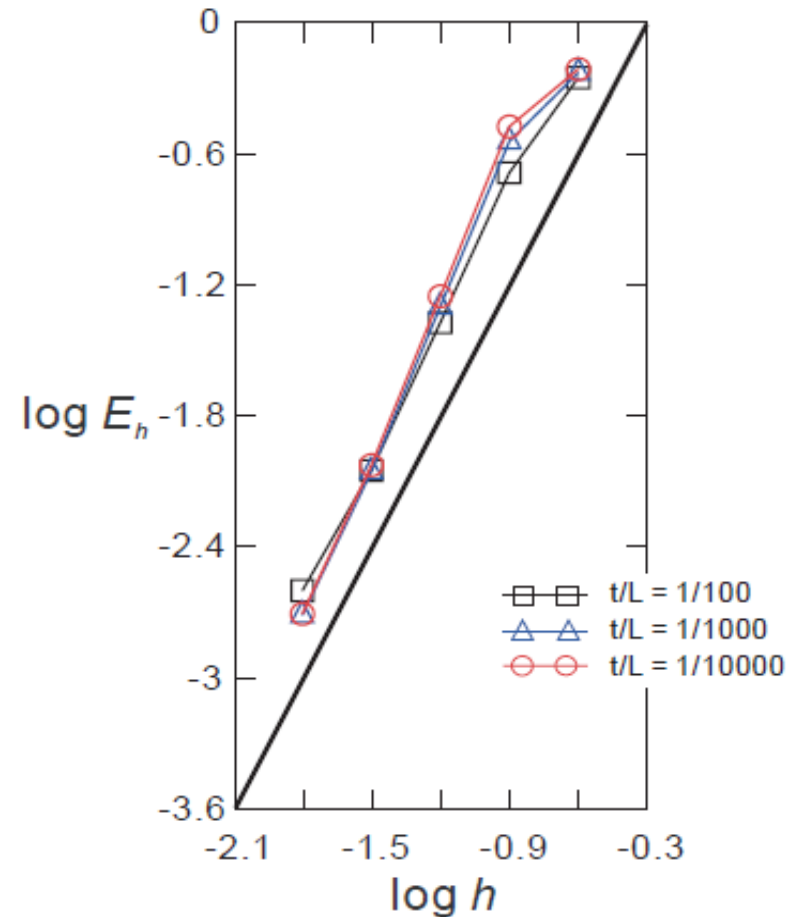
Y Lee, PS Lee, KJ Bathe. The MITC3+ ... . C & S 2014

H Jun, K Yoon, PS Lee, KJ Bathe. The MITC3+ ... . CMAME 2018

### MITC3+



### MITC3+



**Distorted meshes**  
**left: fixed structure, right: free structure**



# Observations –

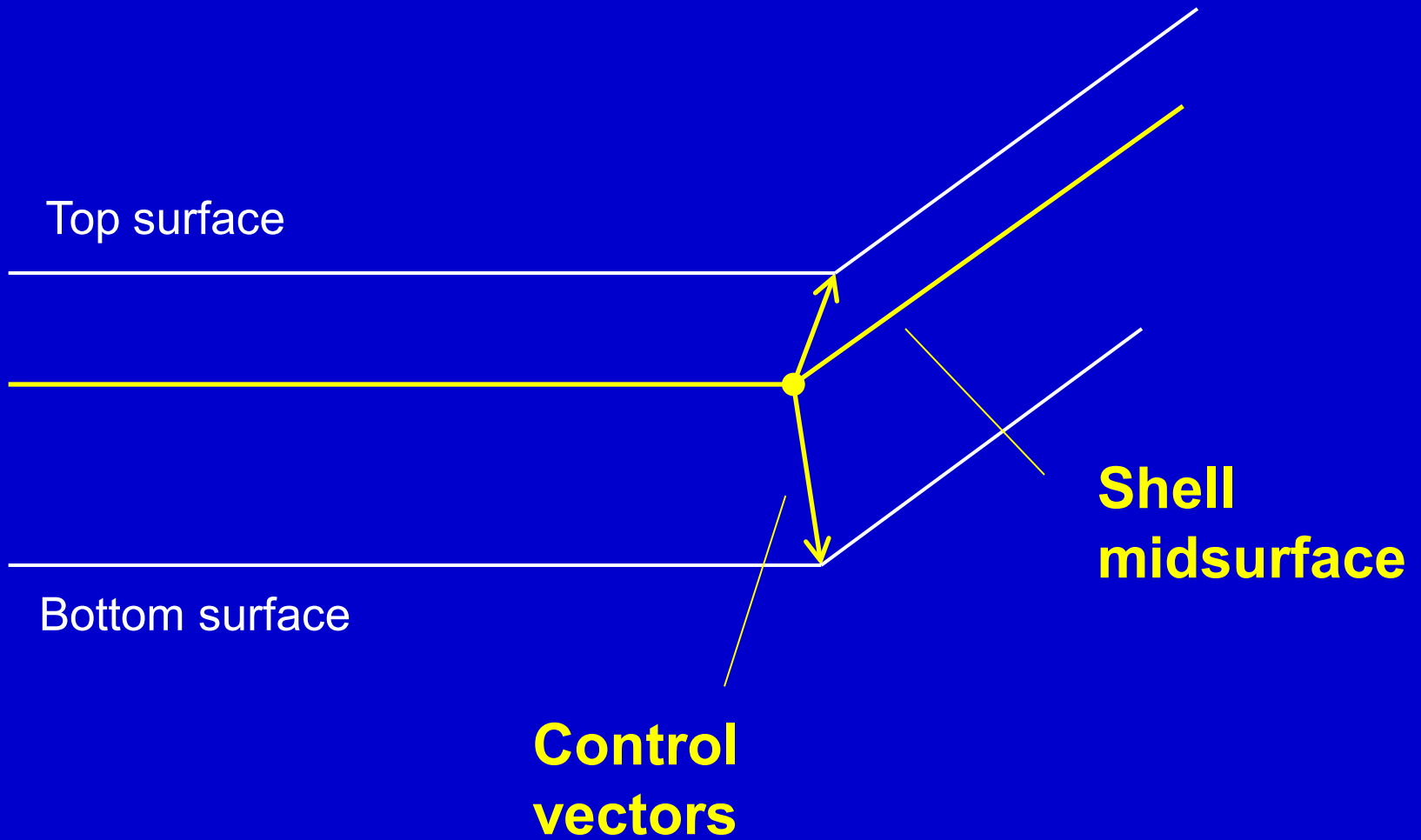
The MITC4+ and MITC3+ elements are practically optimal

but do not include the through-the-thickness stress and strain

In some analyses this stress can be important --- like in metal forming, contact problems, ....

A 3D-shell element can then be effective

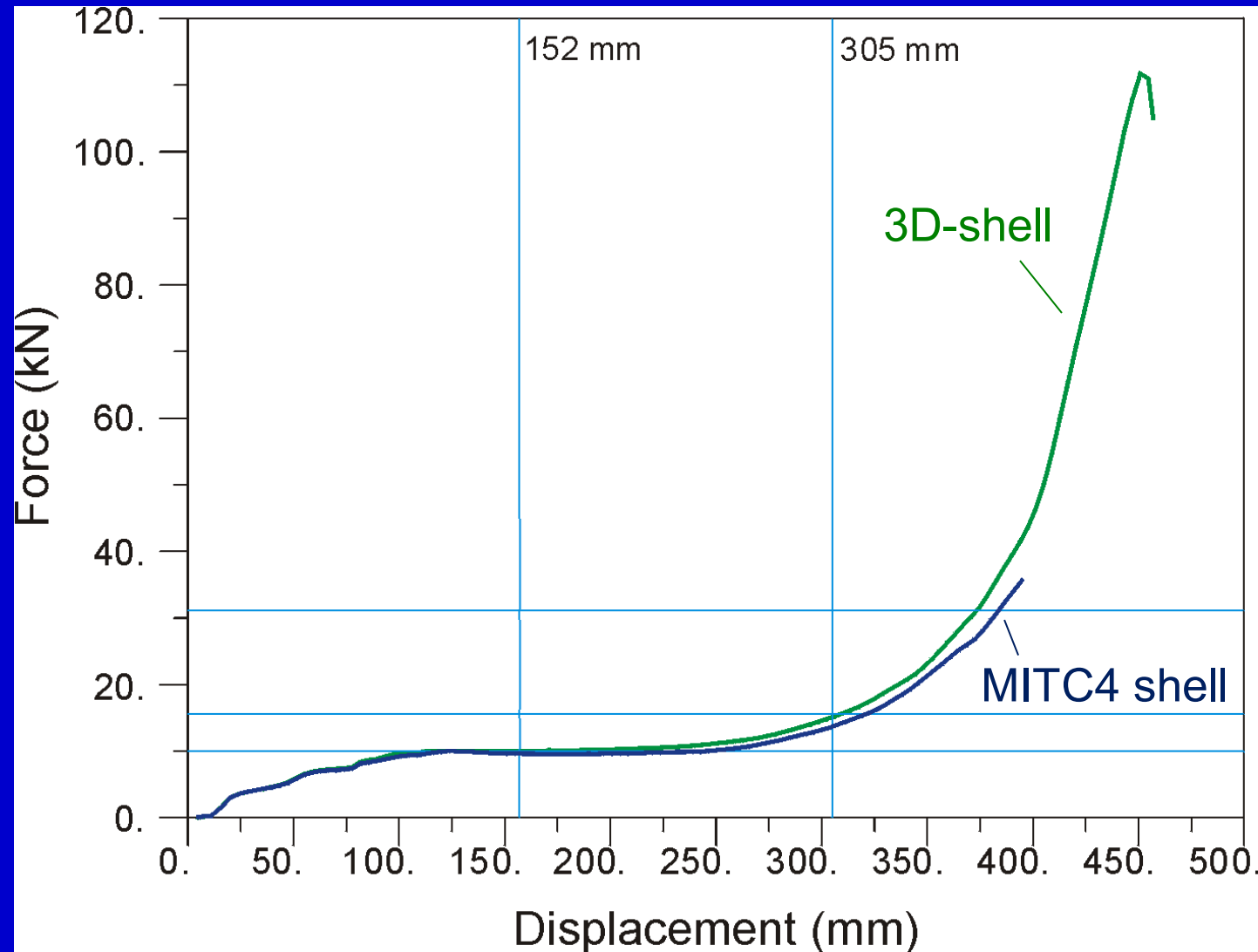
# 3D-Shell Element





**Door crush problem**

# Door Crush Problem

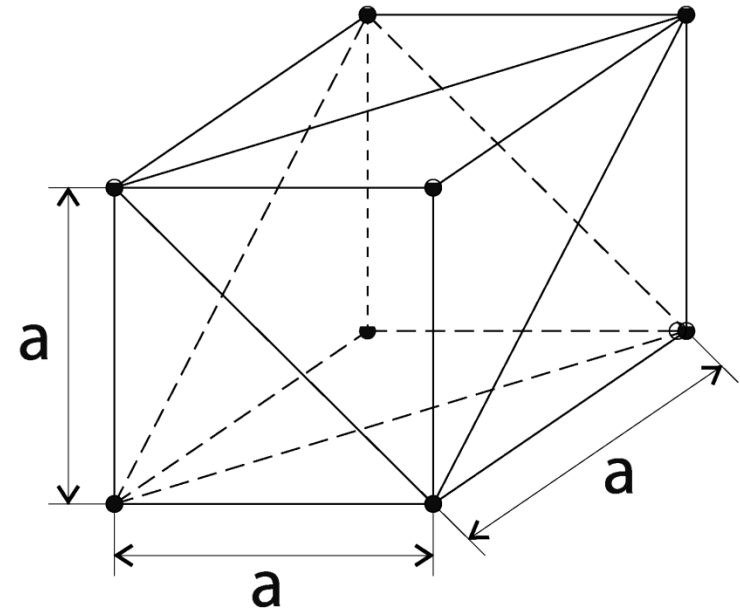
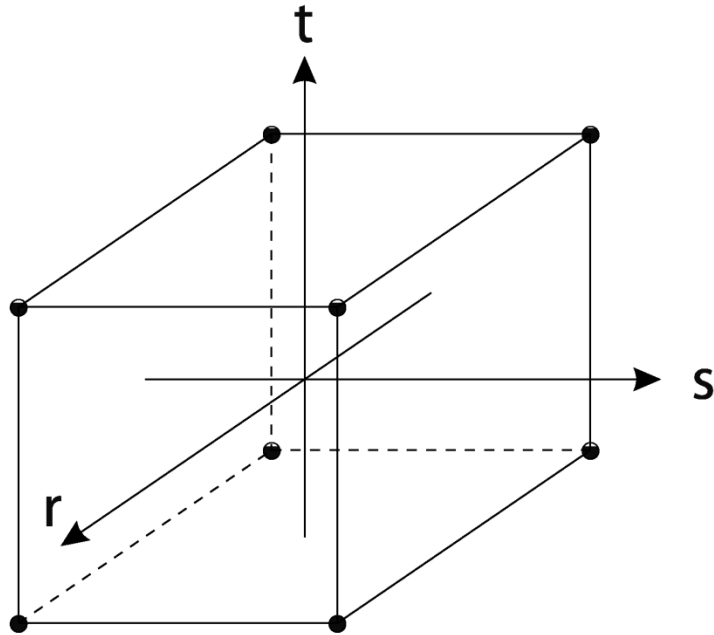


# **A new 3D solid element, the 3D-MITC8 element**

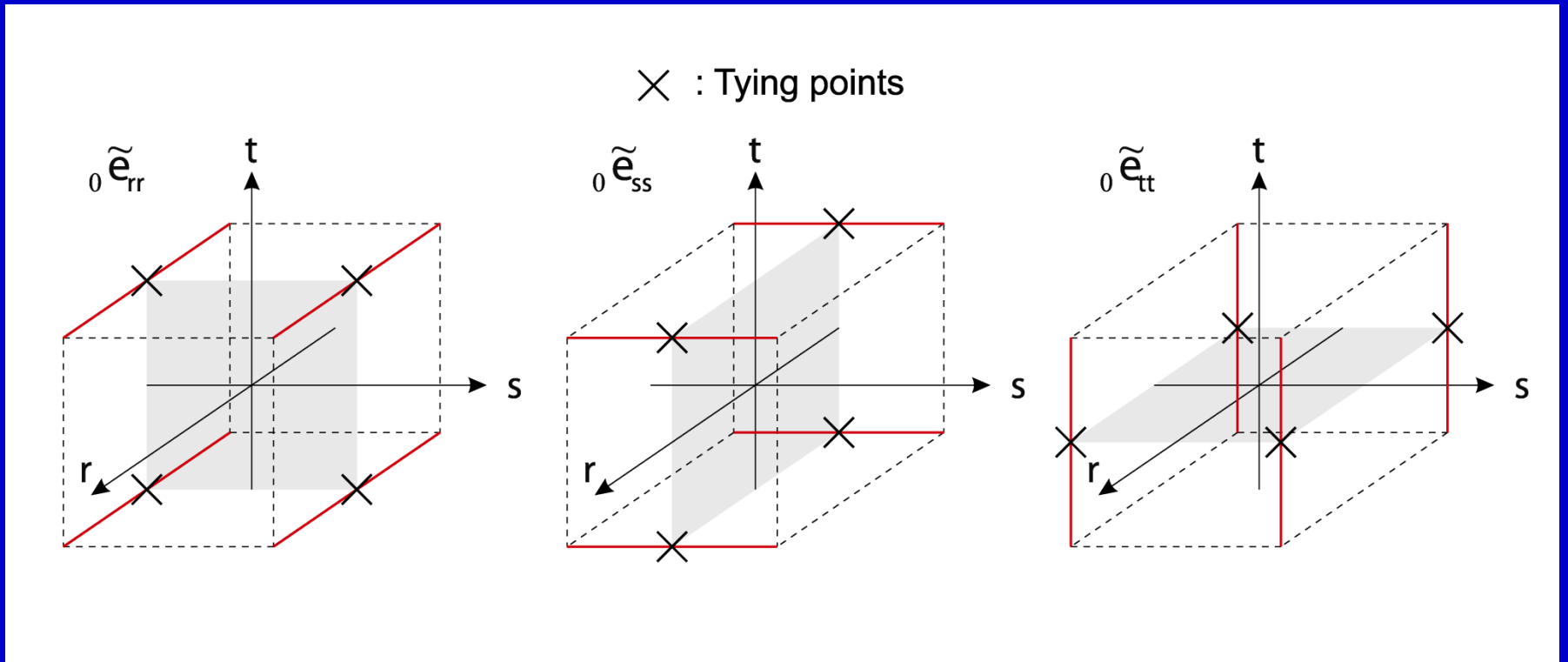
**Eight-node element using MITC technique,  
described by only displacement DOF**

**The element is isotropic, passes the patch  
tests, and can be used in linear and non-  
linear analyses**

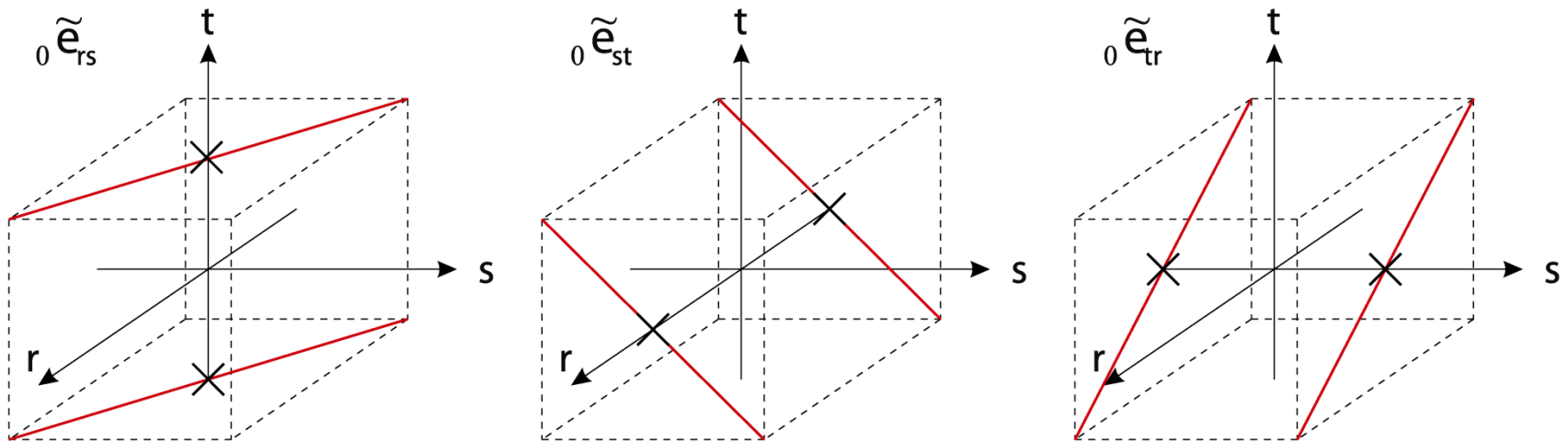
**The element displays no spurious mode in  
nonlinear analysis as does the H8I9 element**



**The domain of the 8-node element  
idealized by a stable truss structure**



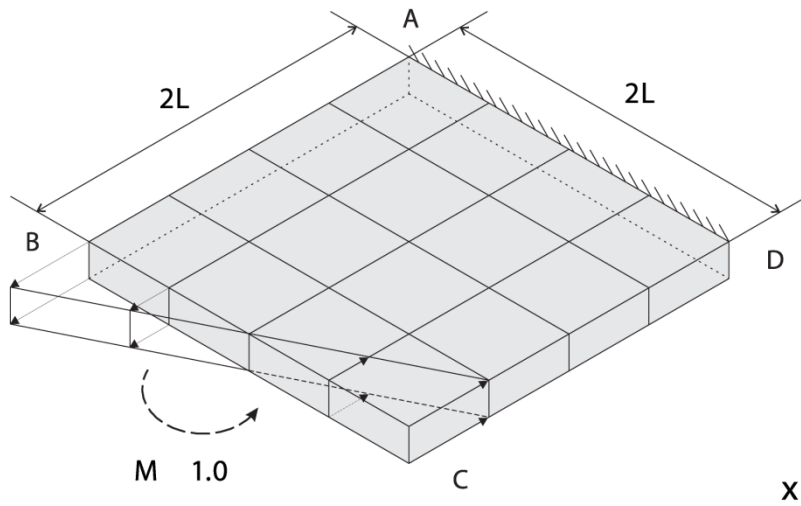
**2-node truss elements at edges for normal stress components and their interpolations**



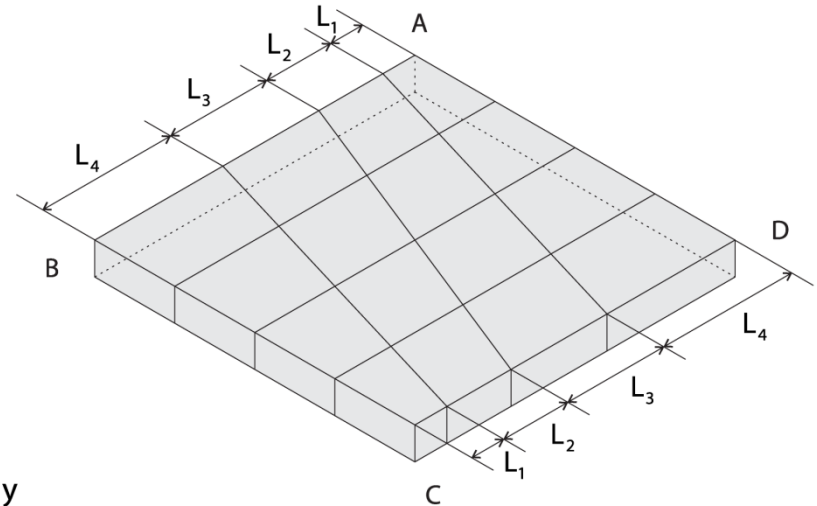
**2-node diagonal truss elements for shear stress components and their interpolations**



# Illustrative solutions

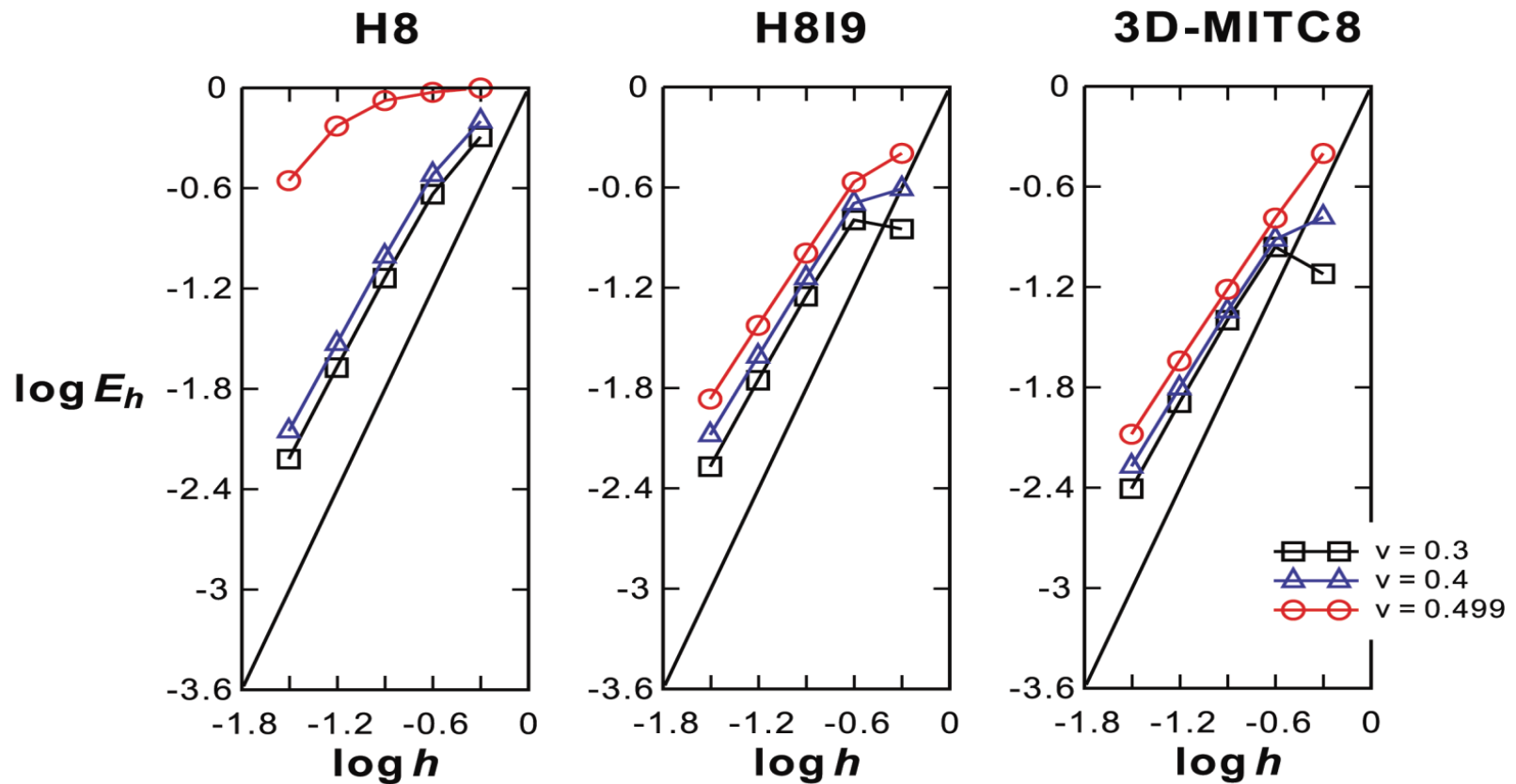


(a)

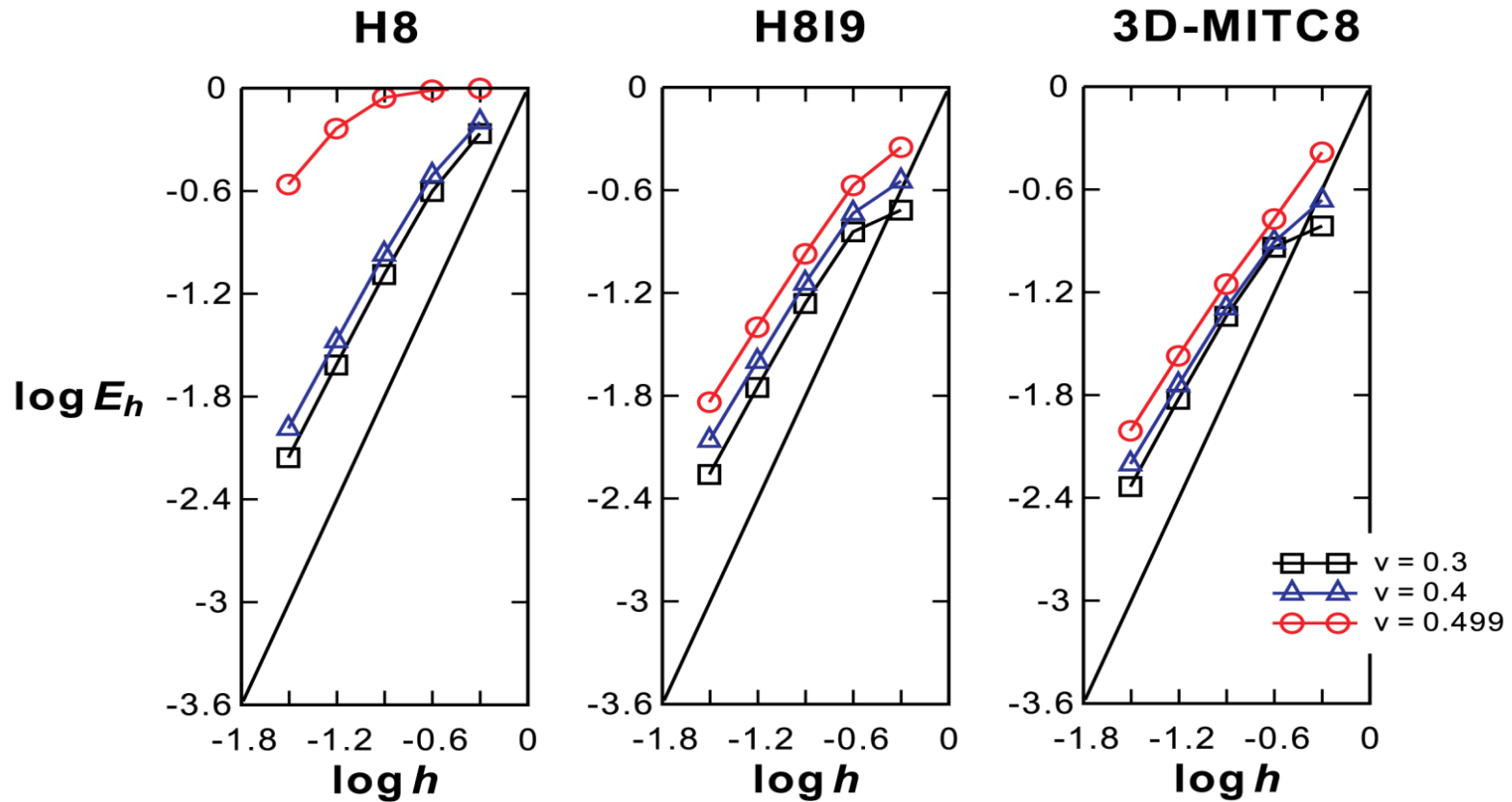


(b)

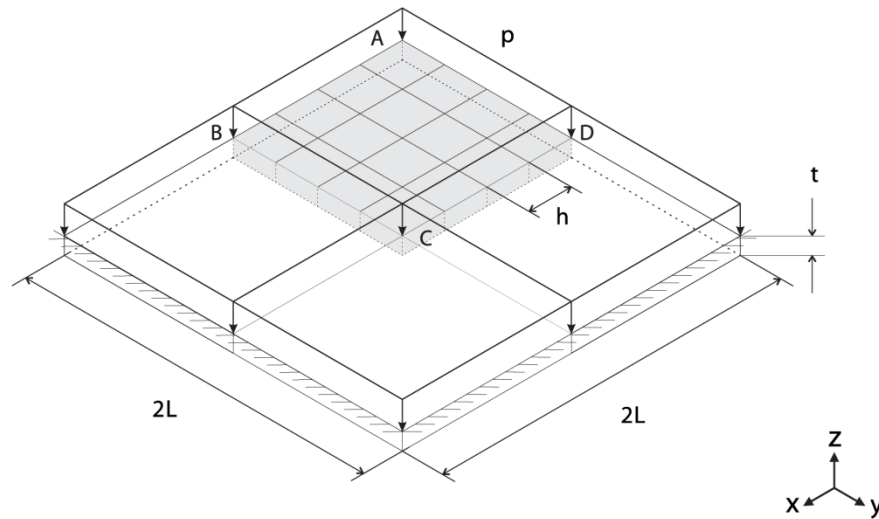
**Analysis of cantilever plate subjected to in-plane moment, test for volumetric locking**



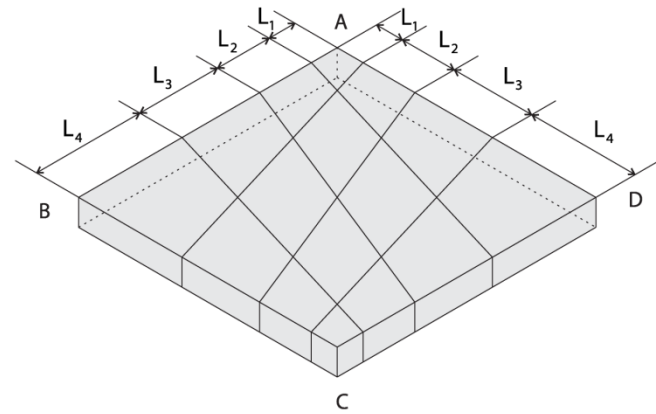
**Analysis of cantilever plate, regular meshes**



**Analysis of cantilever plate, distorted meshes**

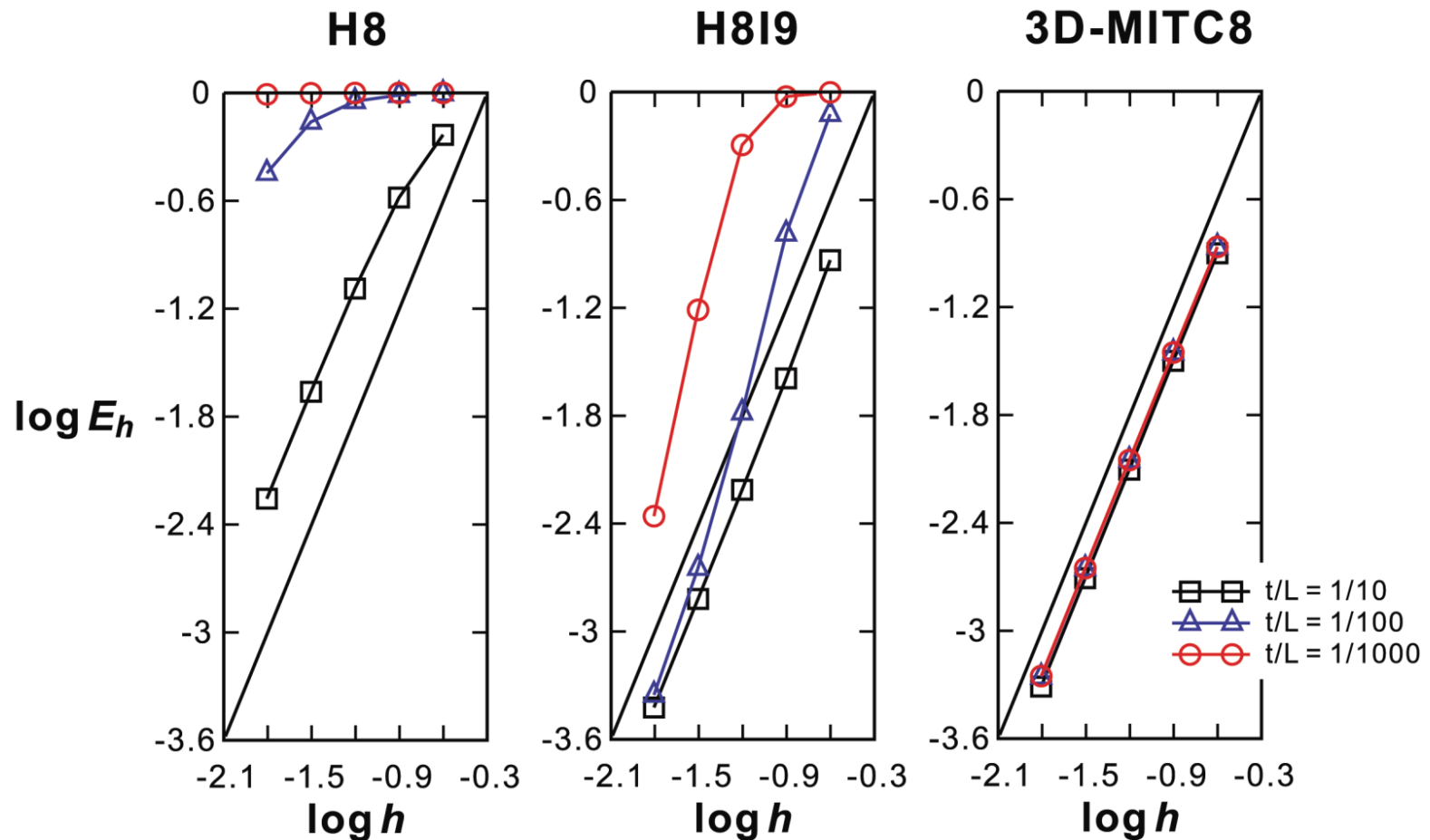


(a)



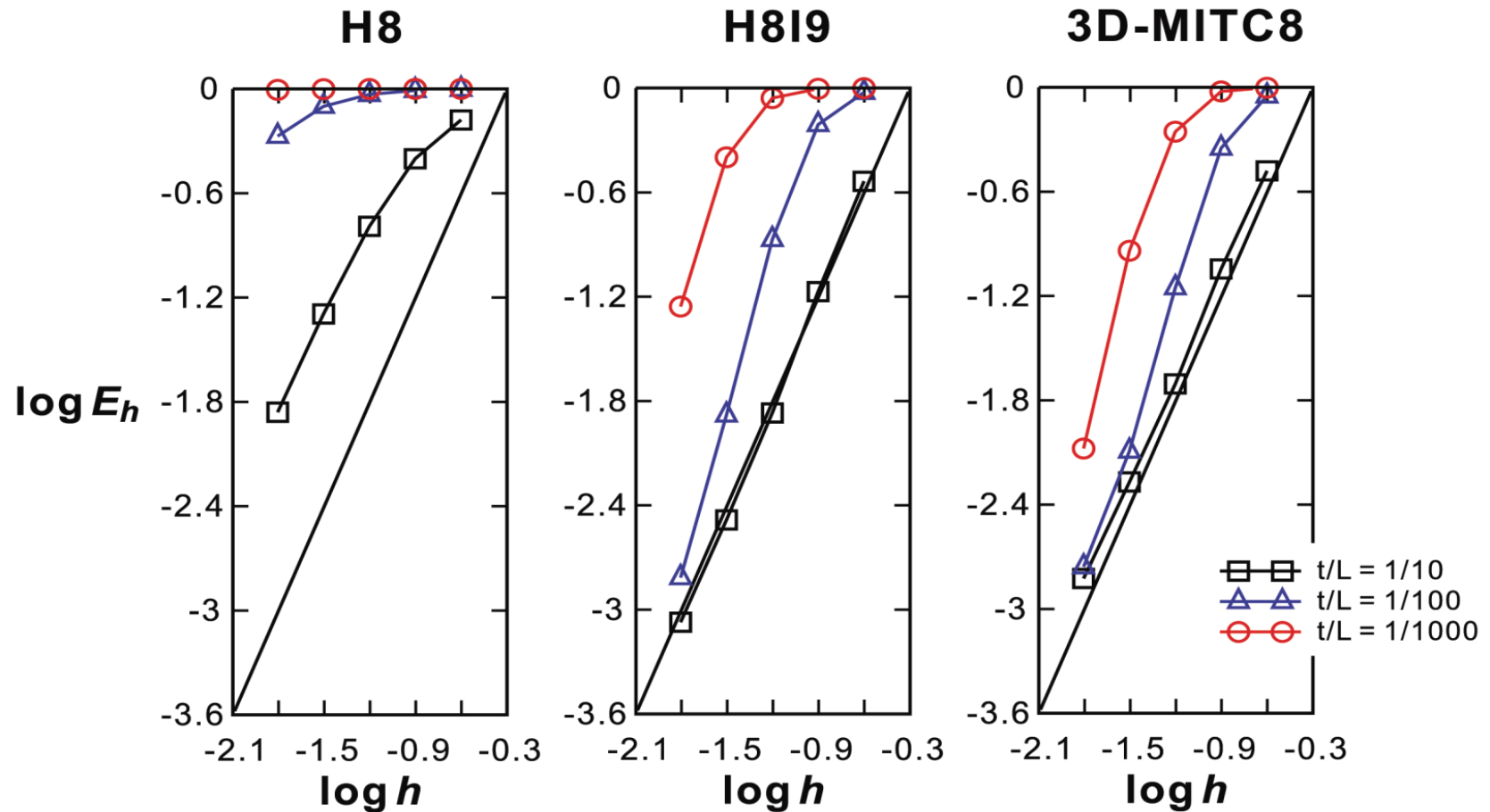
(b)

**Analysis of clamped plate subjected to transverse pressure, test for shear locking**



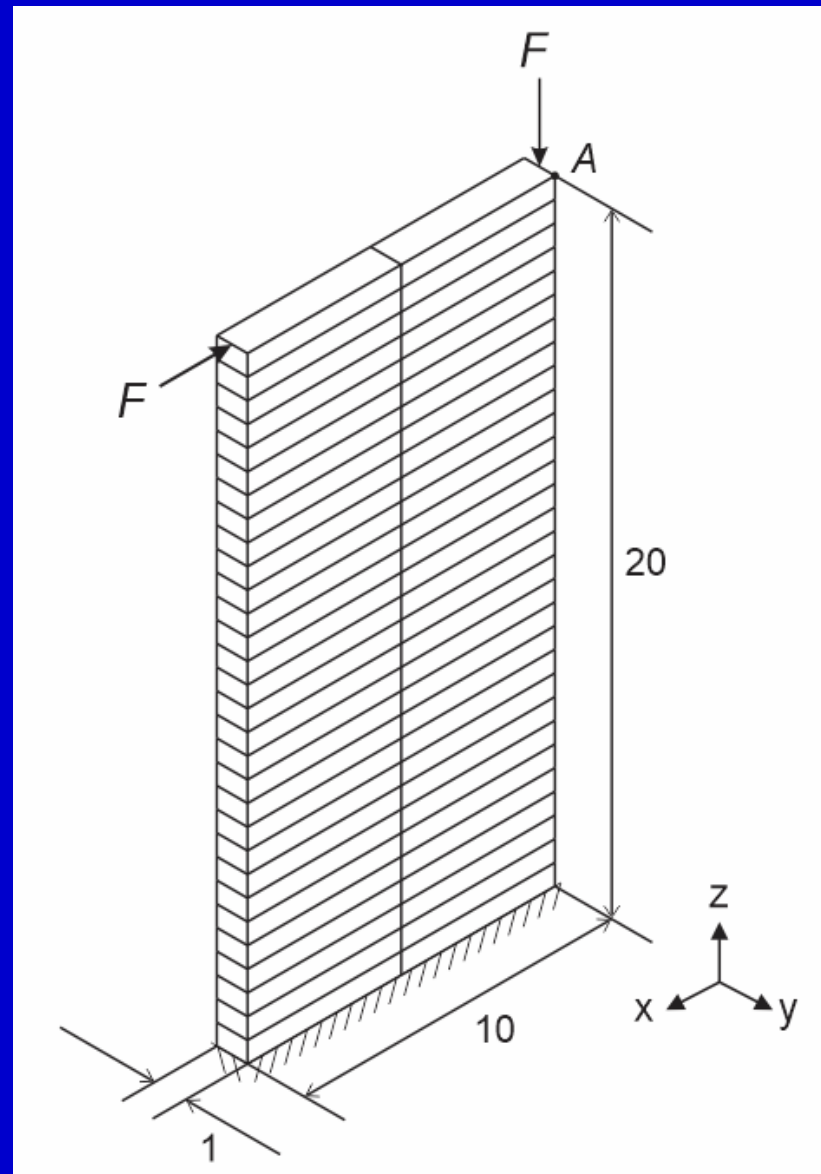
(a)

**Analysis of clamped plate, regular meshes**

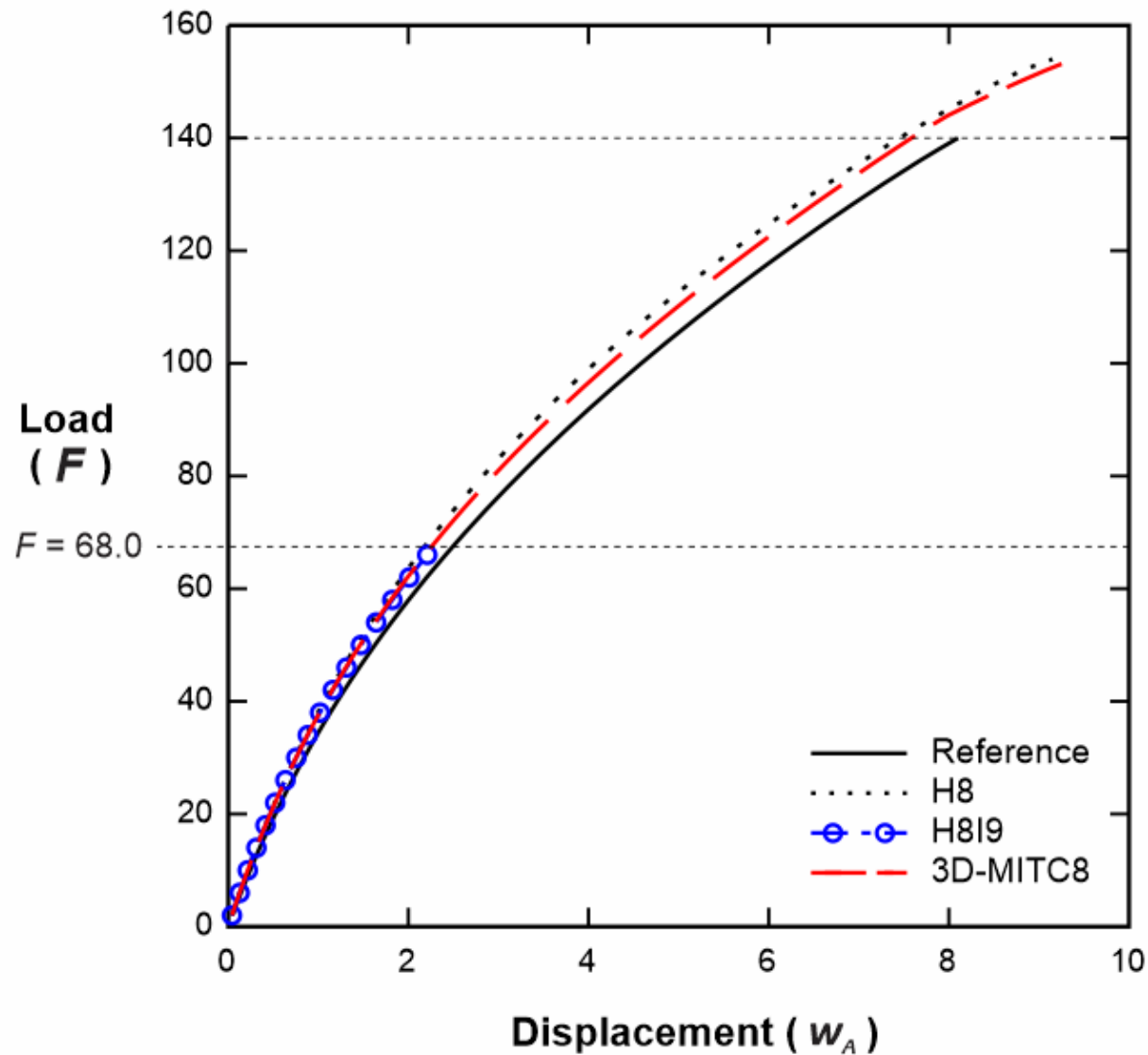


(b)

**Analysis of clamped plate, distorted meshes**



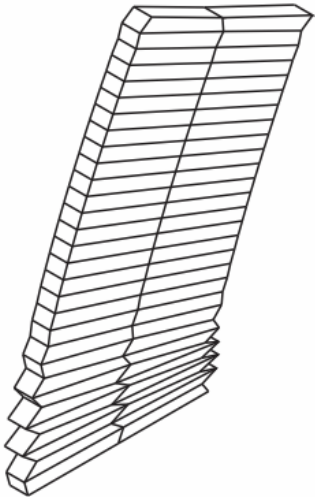
**Large displacement analysis of cantilever**



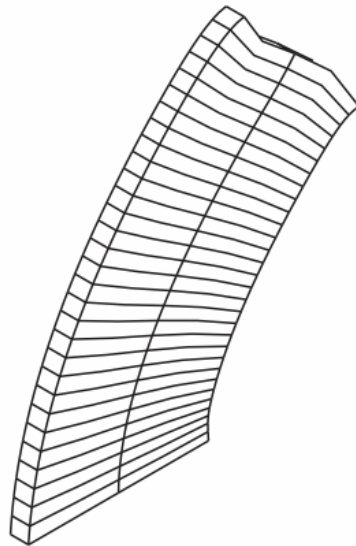
**Response of cantilever**



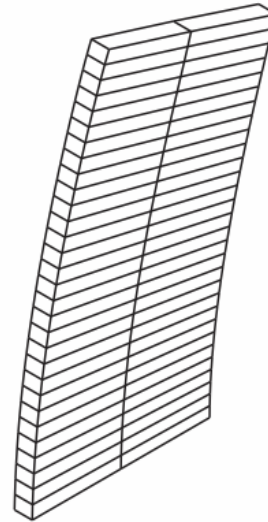
H8I9  
 $F = 68.0$



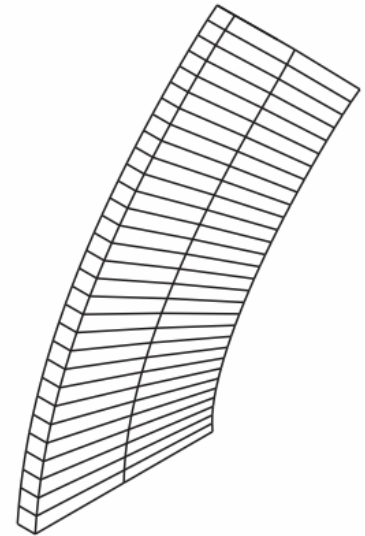
Reference  
 $F = 140.0$



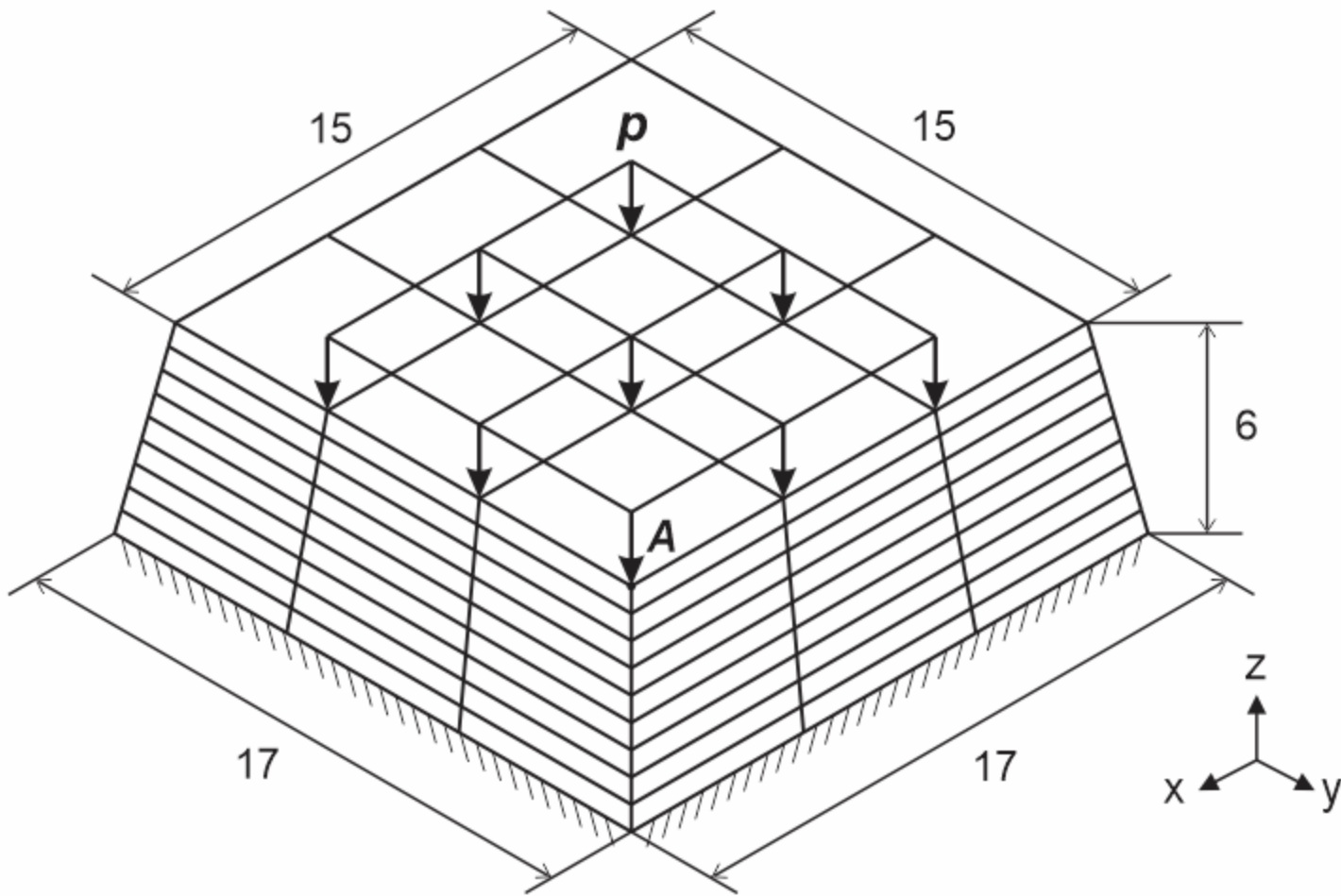
3D-MITC8  
 $F = 68.0$



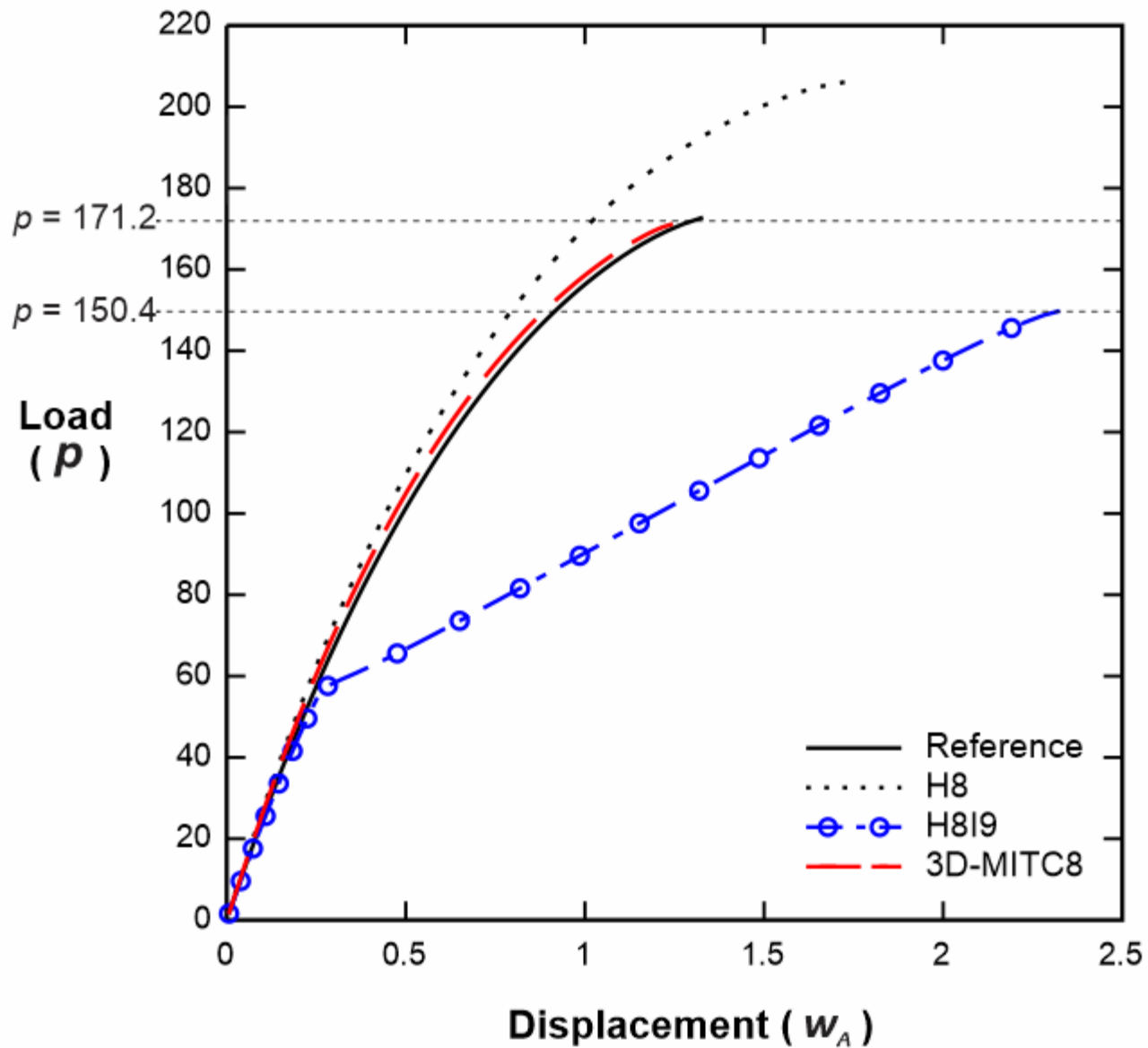
3D-MITC8  
 $F = 140.0$



**Deformed shapes of cantilever**



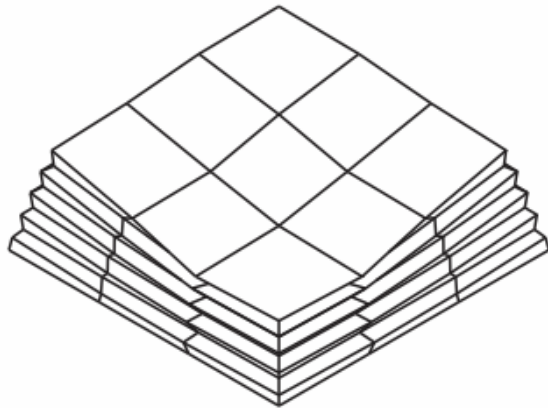
**Analysis of rubber block**



**Response of rubber block**

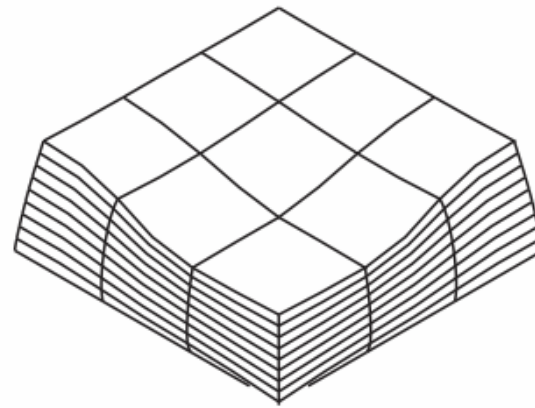
H8I9

$$p = 150.4$$



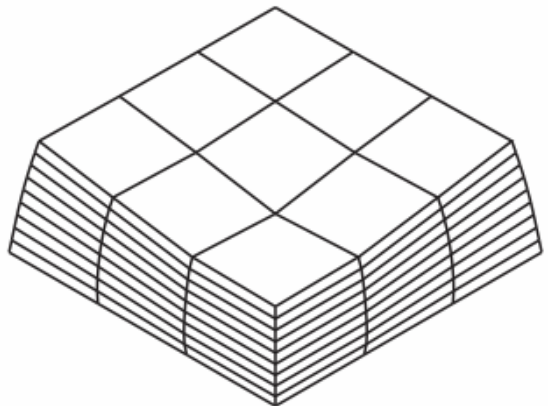
Reference

$$p = 171.2$$



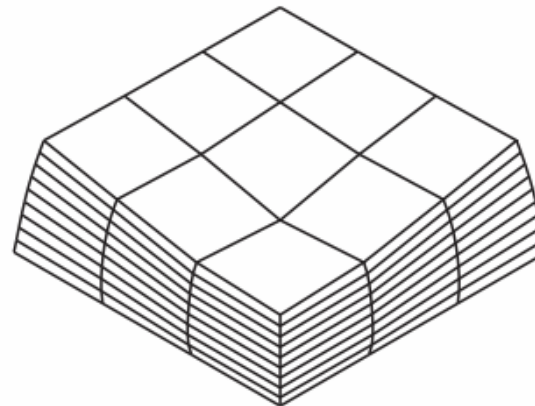
3D-MITC8

$$p = 150.4$$



3D-MITC8

$$p = 171.2$$



**Deformed shapes of rubber block**

# Inf-sup testing of elements

**We want**  $\exists \gamma_A > 0$

$$\inf_{\mathbf{w}_h \in V_h} \sup_{\mathbf{v}_h \in V_h} \frac{\mathbf{w}_h^T \mathbf{A}_h \mathbf{v}_h}{\|\mathbf{w}_h\| \|\mathbf{v}_h\|} \geq \gamma_A$$

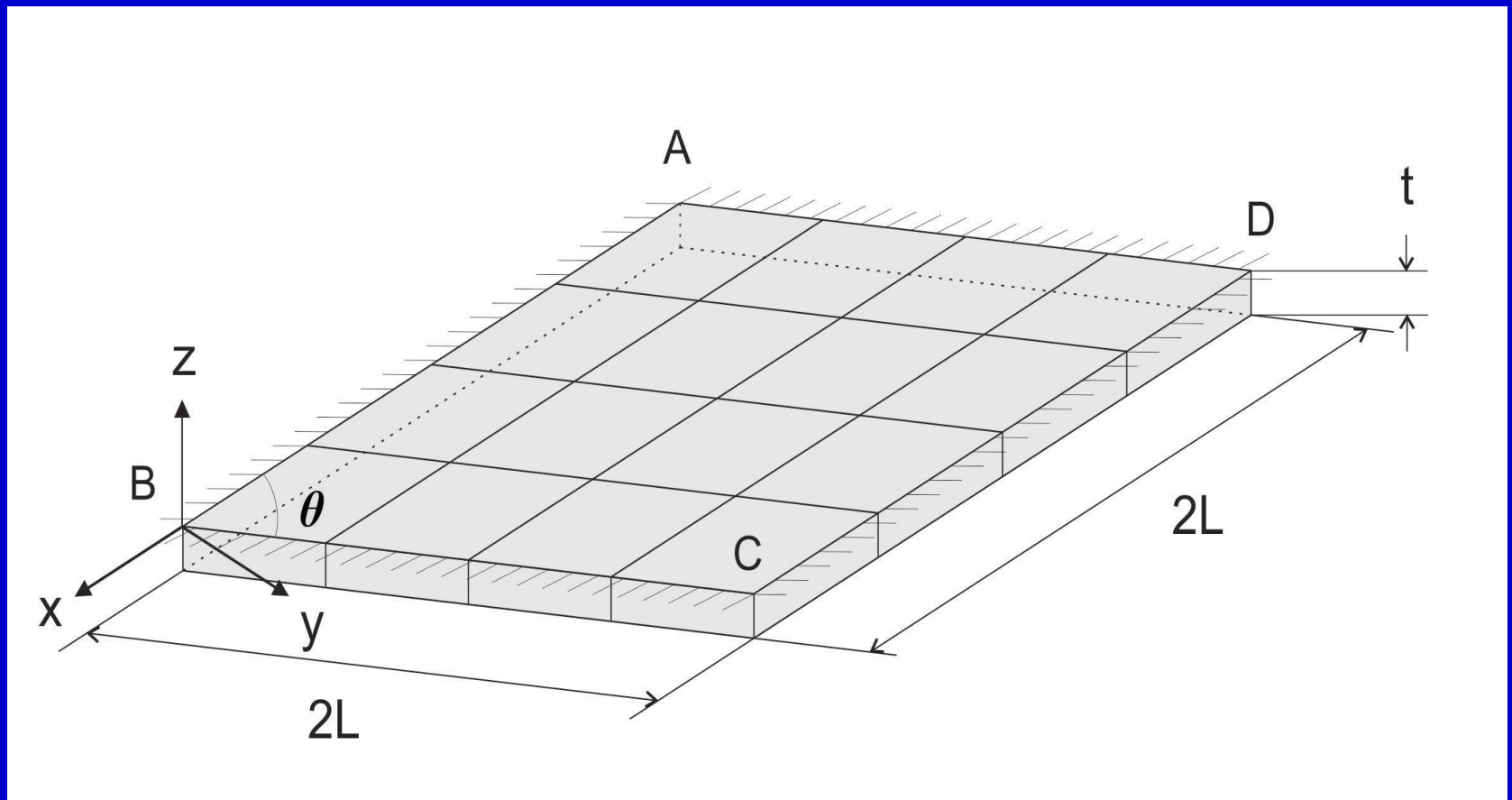
**and find the smallest eigenvalue of**

$$\mathbf{A}_h \mathbf{v}_h = \lambda_k \mathbf{G}_h \mathbf{v}_h$$

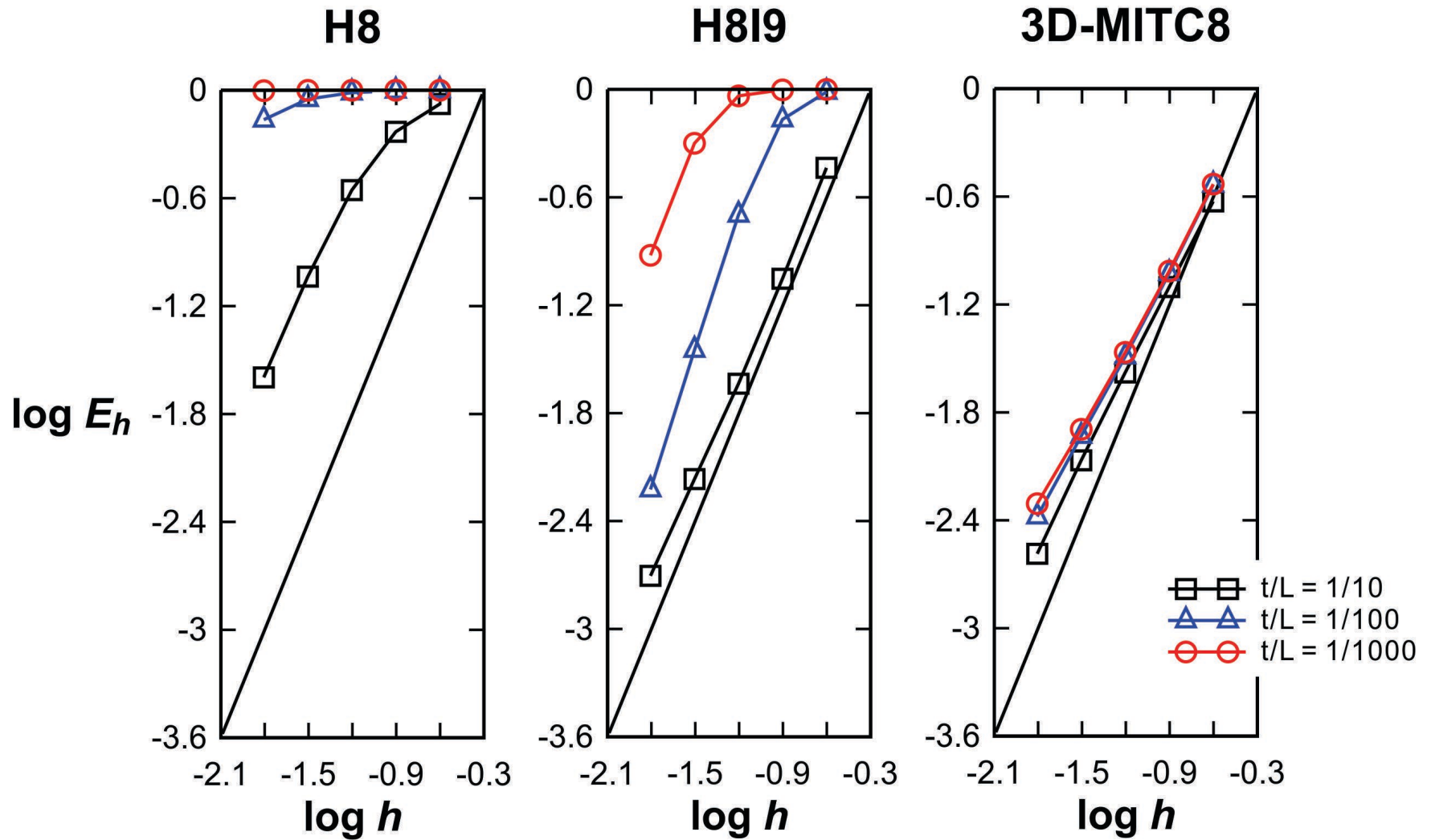
**Since  $A_h$  is the stiffness matrix ( not the matrix of the bilinear form B, which we do not have explicitly ) we really test for coercivity**

**But we still deem to get insight into the stability of the discretization scheme**

**The numerical results show that indeed we do obtain good insight, but more mathematical analysis would be valuable**

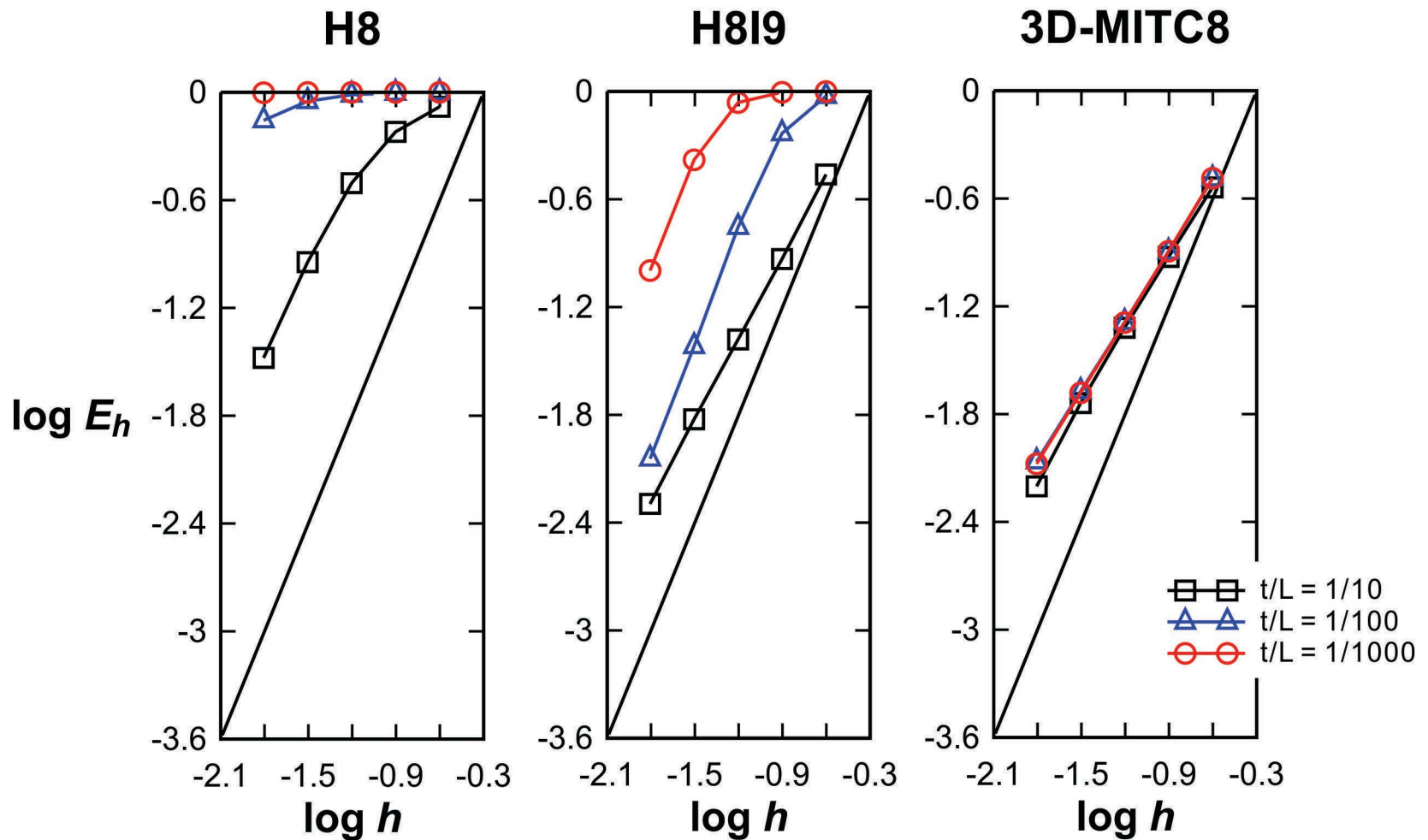


**Clamped skew plate, angle of skew  $\theta$**

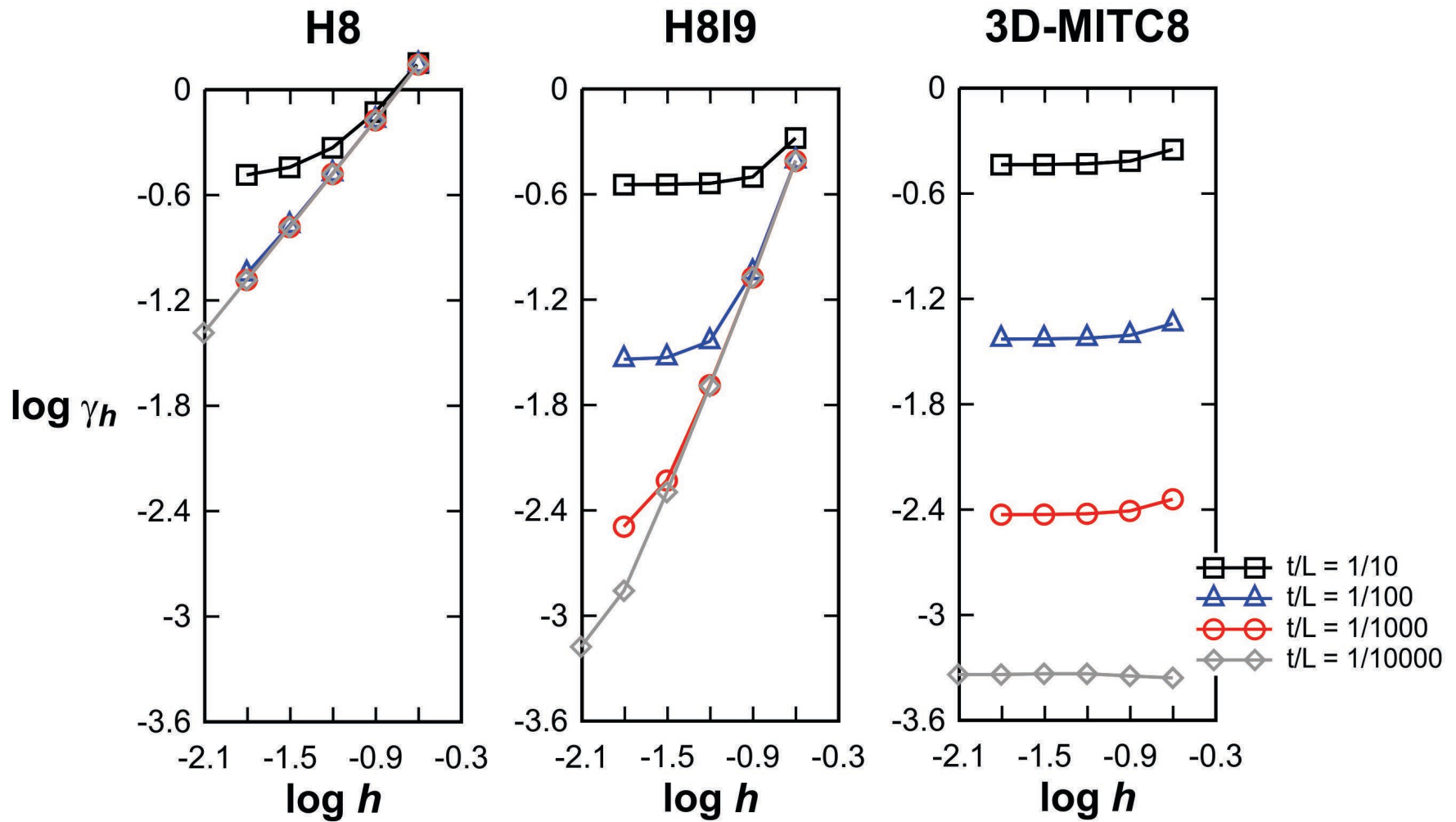


s-norm for applied uniform pressure,  $\theta = 60$  degrees

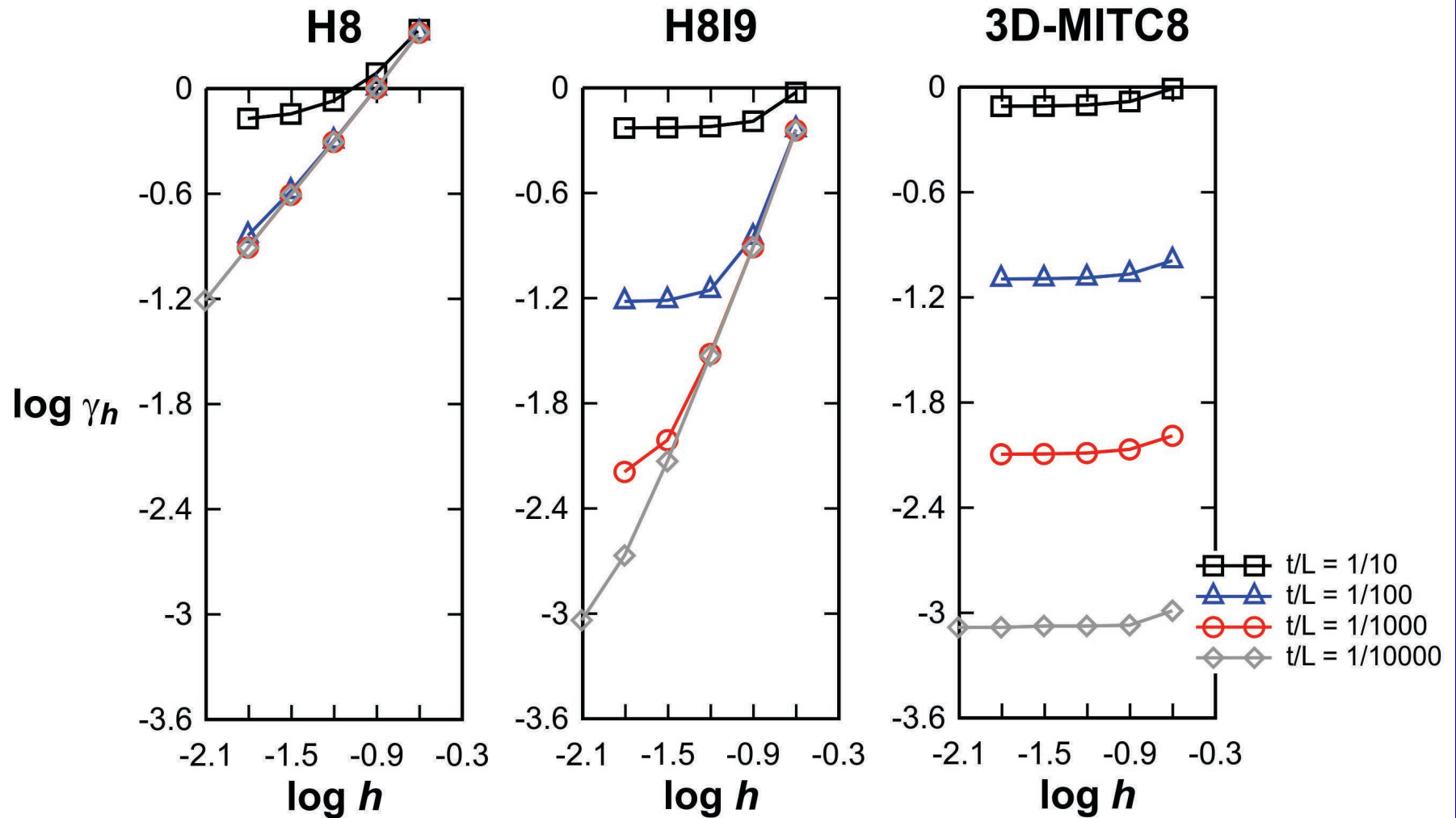




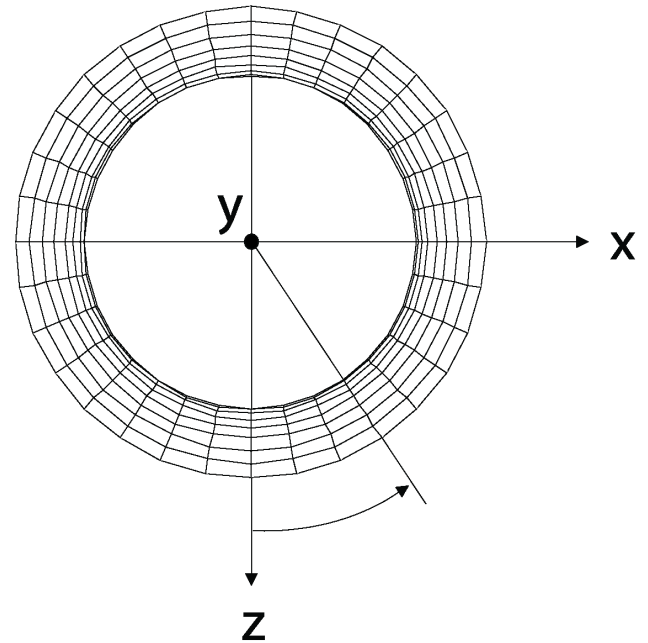
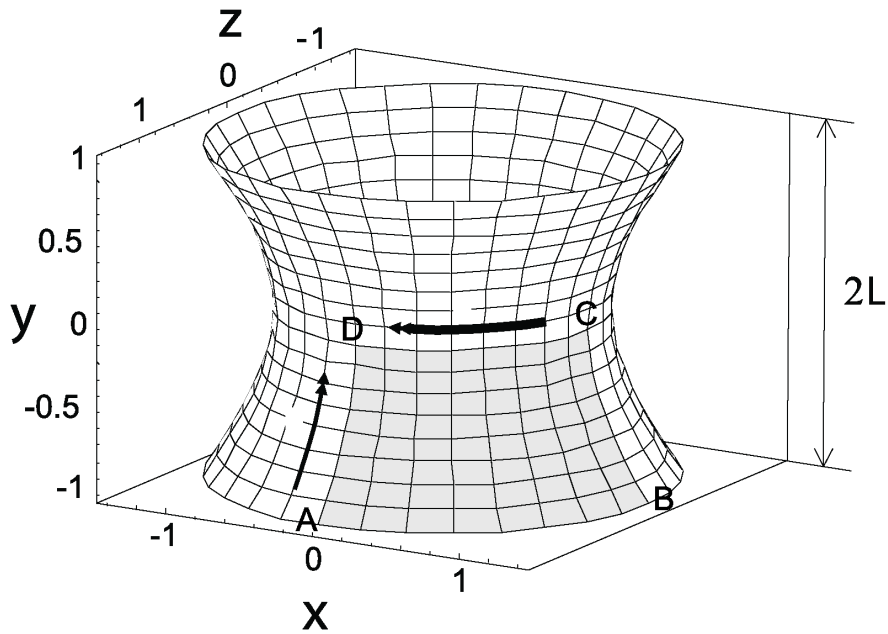
s-norm for applied uniform pressure,  $\theta = 30$  degrees



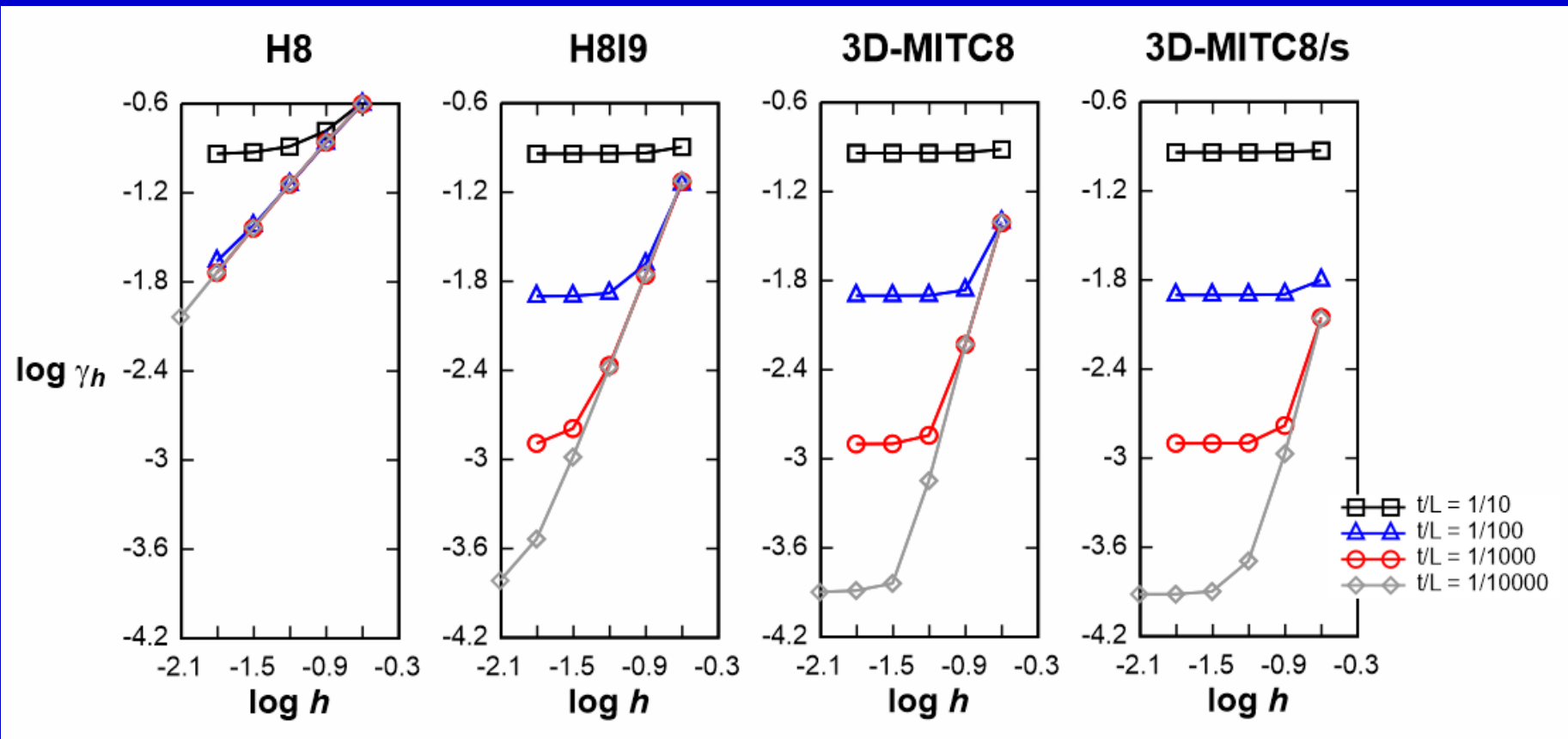
**Inf-sup test,  $\theta = 60$  degrees**



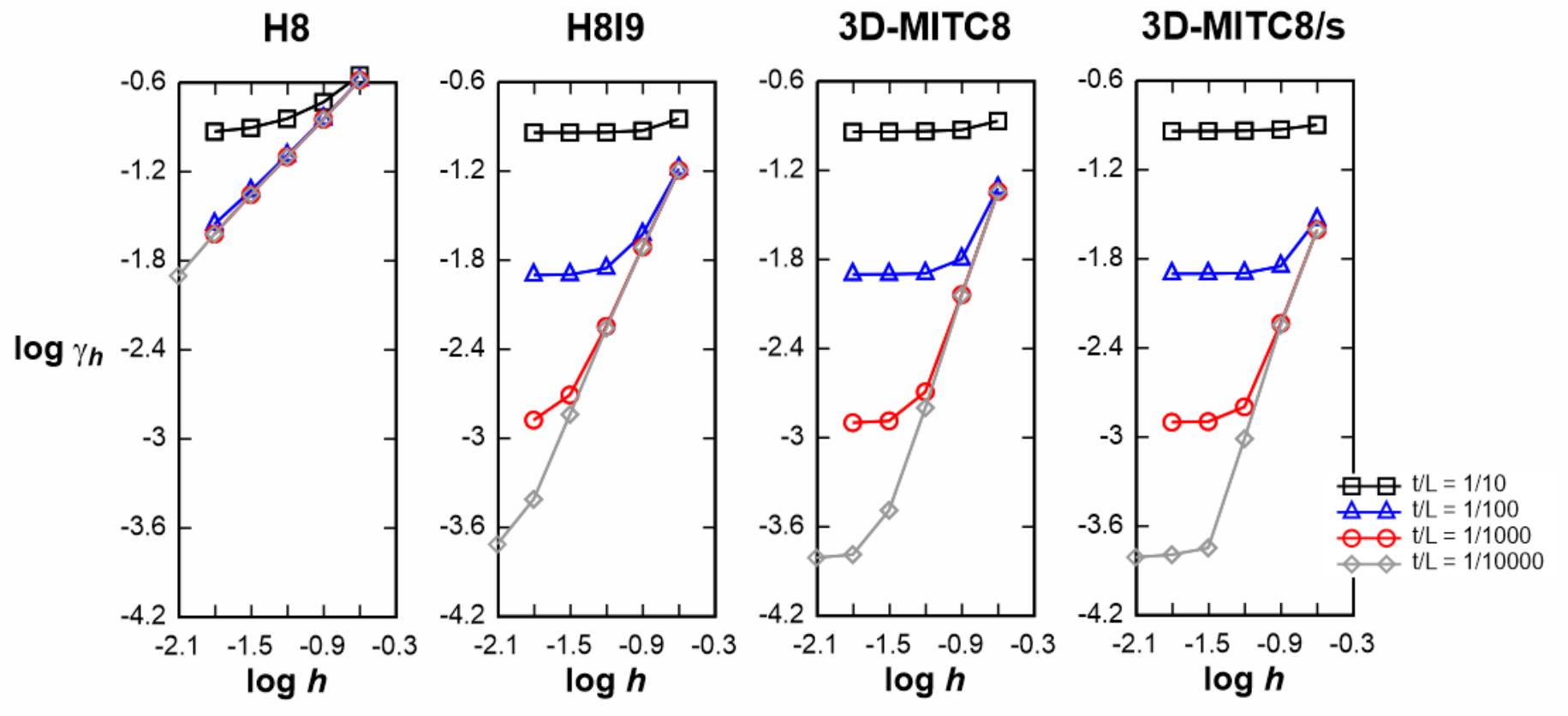
**Inf-sup test,  $\theta = 30$  degrees**



**The hyperbolic shell test problem,  
free ends**



**Hyperbolic shell inf-sup test, regular meshes**



**Hyperbolic shell inf-sup test, distorted meshes**

# The enriched subspace iteration method

We are interested in solving:

$$\mathbf{K}\boldsymbol{\varphi}_i = \lambda_i \mathbf{M}\boldsymbol{\varphi}_i$$

$$0 < \lambda_1 \leq \lambda_2 \leq \dots \leq \lambda_p$$

with  $p = 100, 200, 300, \dots$

$$\mathbf{X}_k = \left[ \Phi_k, \mathbf{X}_k^a, \mathbf{X}_k^b \right]$$

$$\mathbf{K} \overline{\mathbf{X}}_{k+1}^a = \mathbf{M} \mathbf{X}_k^a$$

$$\overline{\mathbf{X}}_{k+1} = \left[ \Phi_k, \overline{\mathbf{X}}_{k+1}^a, \overline{\mathbf{Y}}_{k+1} \right]$$

**Iteration vectors**



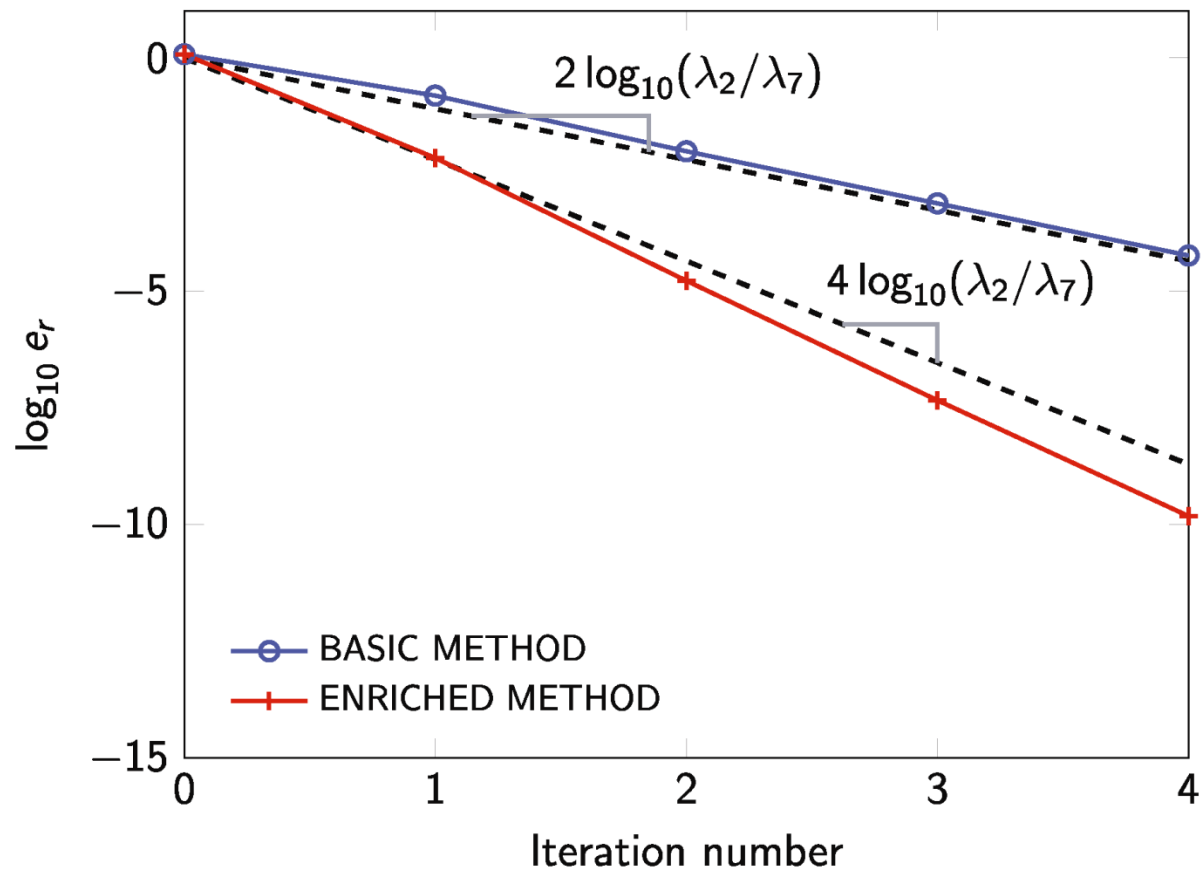
$$\mathbf{K}_{k+1} = \overline{\mathbf{X}}_{k+1}^T \mathbf{K} \overline{\mathbf{X}}_{k+1}$$

$$\mathbf{M}_{k+1} = \overline{\mathbf{X}}_{k+1}^T \mathbf{M} \overline{\mathbf{X}}_{k+1}$$

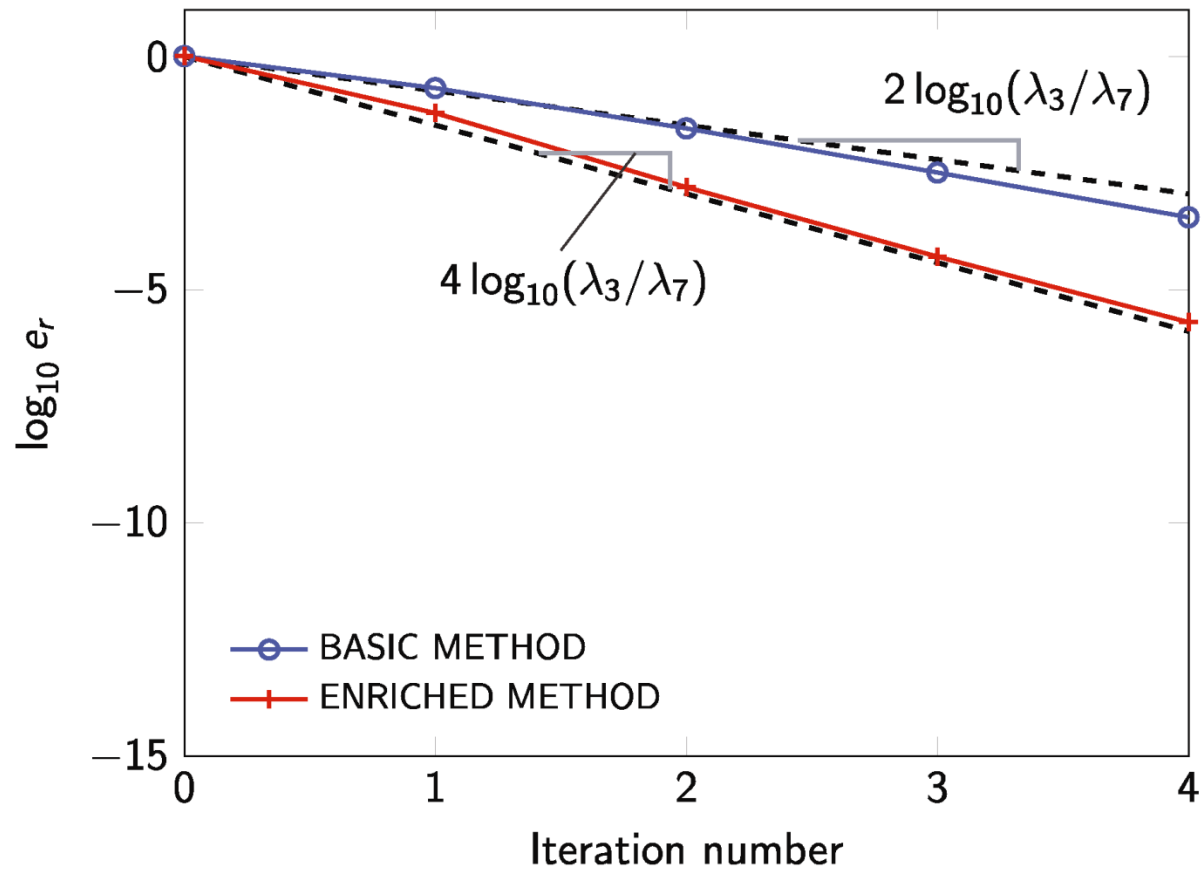
$$\mathbf{K}_{k+1} \mathbf{Q}_{k+1} = \mathbf{M}_{k+1} \mathbf{Q}_{k+1} \mathbf{\Lambda}_{k+1}$$

$$\mathbf{X}_{k+1} = \overline{\mathbf{X}}_{k+1} \mathbf{Q}_{k+1}$$

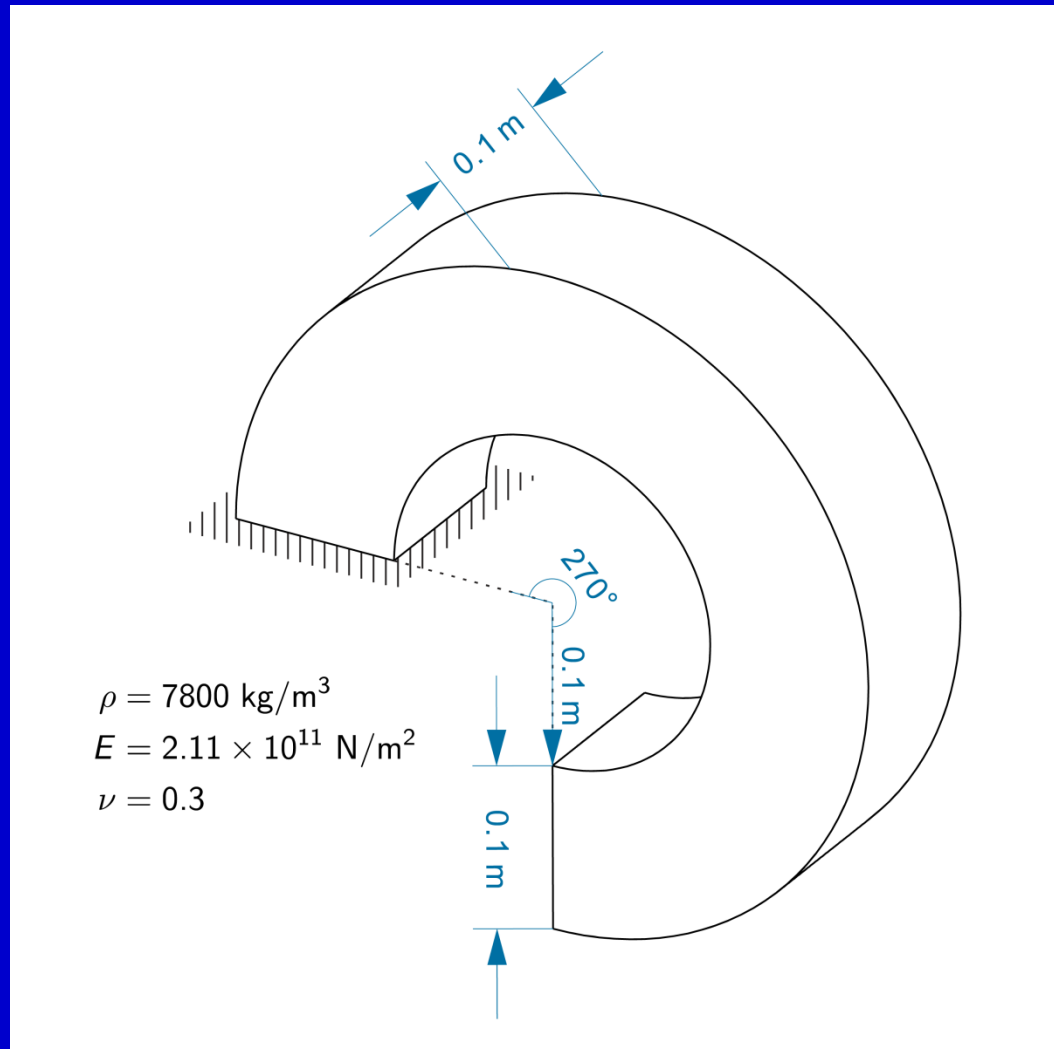
**Update of iteration vectors**



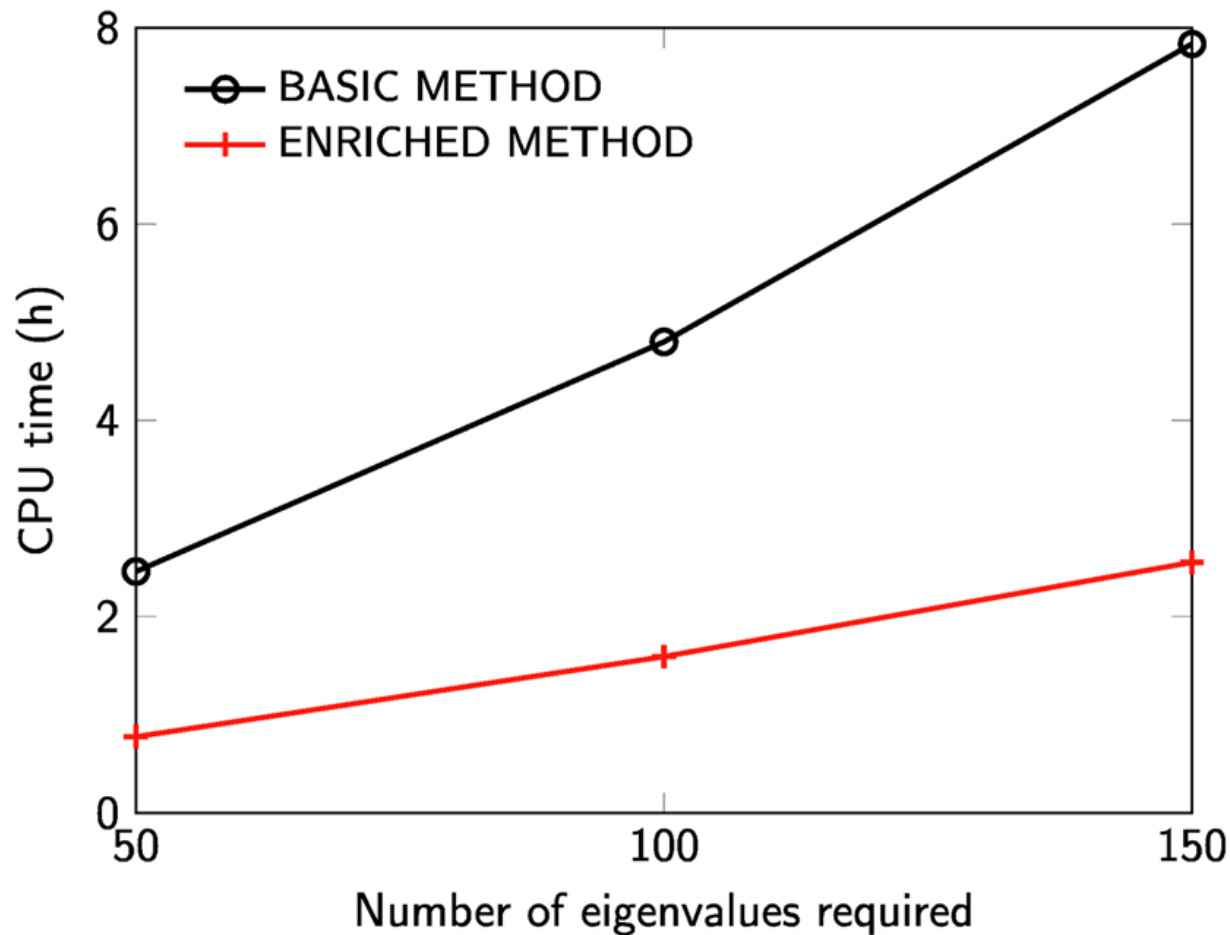
**Convergence rates of eigenvalues**



**Convergence rates of eigenvalues**



**Solution of ring cantilever structure**



**Solution times for 3D ring structure,  
~ 600,000 DOF,  $m \sim 1,600$ ; 1 core m/c, 2.4 GHz**

**Model of exhaust manifold from Volvo,  
NEQ = 2,220,273, w/ contact, 16-core SMP  
( on a DELL 2-proc. computer, each w/ 12 cores, Linux )**

<b>Number of freq./ vectors</b>	<b>Time used (min)</b>
---------------------------------	------------------------

<b>50</b>	<b>7</b>
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<b>100</b>	<b>12</b>
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<b>200</b>	<b>37</b>
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**Almost linear increase in solution times**

**Model of tractor front end from John Deere,  
NEQ = 966,755, 16-core SMP**

**( on a DELL 2-proc. computer, each w/ 12 cores, Linux )**

<b>Number of freq./ vectors</b>	<b>Time used (min)</b>
---------------------------------	------------------------

<b>100</b>	<b>4</b>
------------	----------

<b>150</b>	<b>7</b>
------------	----------

<b>200</b>	<b>11</b>
------------	-----------

<b>300</b>	<b>23</b>
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**Note – no DMP used in solutions, although the  
subspace iteration parallelizes well in DMP**

# Time integration: implicit integration

Use 2 sub-steps per time step, with splitting ratio  $\gamma$

Trapezoidal rule, case  $\gamma = 0.5$

$$\mathbf{M} {}^{t+\Delta t/2}\ddot{\mathbf{U}} + \mathbf{C} {}^{t+\Delta t/2}\dot{\mathbf{U}} = {}^{t+\Delta t/2}\mathbf{R} - {}^{t+\Delta t/2}\mathbf{F}$$

$${}^{t+\Delta t/2}\dot{\mathbf{U}} = {}^t\dot{\mathbf{U}} + \frac{\Delta t}{4} \left( {}^t\ddot{\mathbf{U}} + {}^{t+\Delta t/2}\ddot{\mathbf{U}} \right)$$

$${}^{t+\Delta t/2}\mathbf{U} = {}^t\mathbf{U} + \frac{\Delta t}{4} \left( {}^t\dot{\mathbf{U}} + {}^{t+\Delta t/2}\dot{\mathbf{U}} \right)$$

In general  $\gamma$  can be varied, known as Bathe method



## Euler backward method:

$$\mathbf{M}^{t+\Delta t} \ddot{\mathbf{U}} + \mathbf{C}^{t+\Delta t} \dot{\mathbf{U}} = {}^{t+\Delta t}\mathbf{R} - {}^{t+\Delta t}\mathbf{F}$$

$${}^{t+\Delta t}\dot{\mathbf{U}} = \frac{1}{\Delta t} {}^t\mathbf{U} - \frac{4}{\Delta t} {}^{t+\Delta t/2}\mathbf{U} + \frac{3}{\Delta t} {}^{t+\Delta t}\mathbf{U}$$

$${}^{t+\Delta t}\ddot{\mathbf{U}} = \frac{1}{\Delta t} {}^t\dot{\mathbf{U}} - \frac{4}{\Delta t} {}^{t+\Delta t/2}\dot{\mathbf{U}} + \frac{3}{\Delta t} {}^{t+\Delta t}\dot{\mathbf{U}}$$

Each of these “ingredients” widely known for decades, the valuable point is to use them together in one time step and study the performance of the scheme ...

## **Properties:**

- no parameter to adjust, simply the time step has to be sufficiently small for accuracy**
- solves in nonlinear analysis when the TR fails**
- excellent accuracy characteristics**

**Effective in the analysis of problems in structural dynamics and wave propagations ---  
but some researchers want to adjust the AD, PE**

**Various endeavors have been published**

## Effect of time step splitting --

- In linear analysis the use of  $\gamma = (2 - \sqrt{2})$  means just one coefficient matrix (at max AD, min PE but almost the same as for the case  $\gamma = 0.5$ )
- In nonlinear analysis, different  $\gamma$  values might be used but  $\gamma = 0.5$  is reasonable considering the range of general nonlinear analyses

**For the 1<sup>st</sup> sub-step**

$${}^{t+\gamma\Delta t}\mathbf{U} = {}^t\mathbf{U} + \left(\frac{\gamma\Delta t}{2}\right) \left({}^t\dot{\mathbf{U}} + {}^{t+\gamma\Delta t}\dot{\mathbf{U}}\right)$$

$${}^{t+\gamma\Delta t}\dot{\mathbf{U}} = {}^t\dot{\mathbf{U}} + \left(\frac{\gamma\Delta t}{2}\right) \left({}^t\ddot{\mathbf{U}} + {}^{t+\gamma\Delta t}\ddot{\mathbf{U}}\right)$$

**and accordingly for the 2<sup>nd</sup> sub-step.**

**For different values of  $\gamma$  we obtain different PE and AD,**

**But --- using 2 important natural parameters we can work with the splitting ratio  $\gamma$  and the spectral radius  $\rho_\infty$  for the 2<sup>nd</sup> sub-step:**

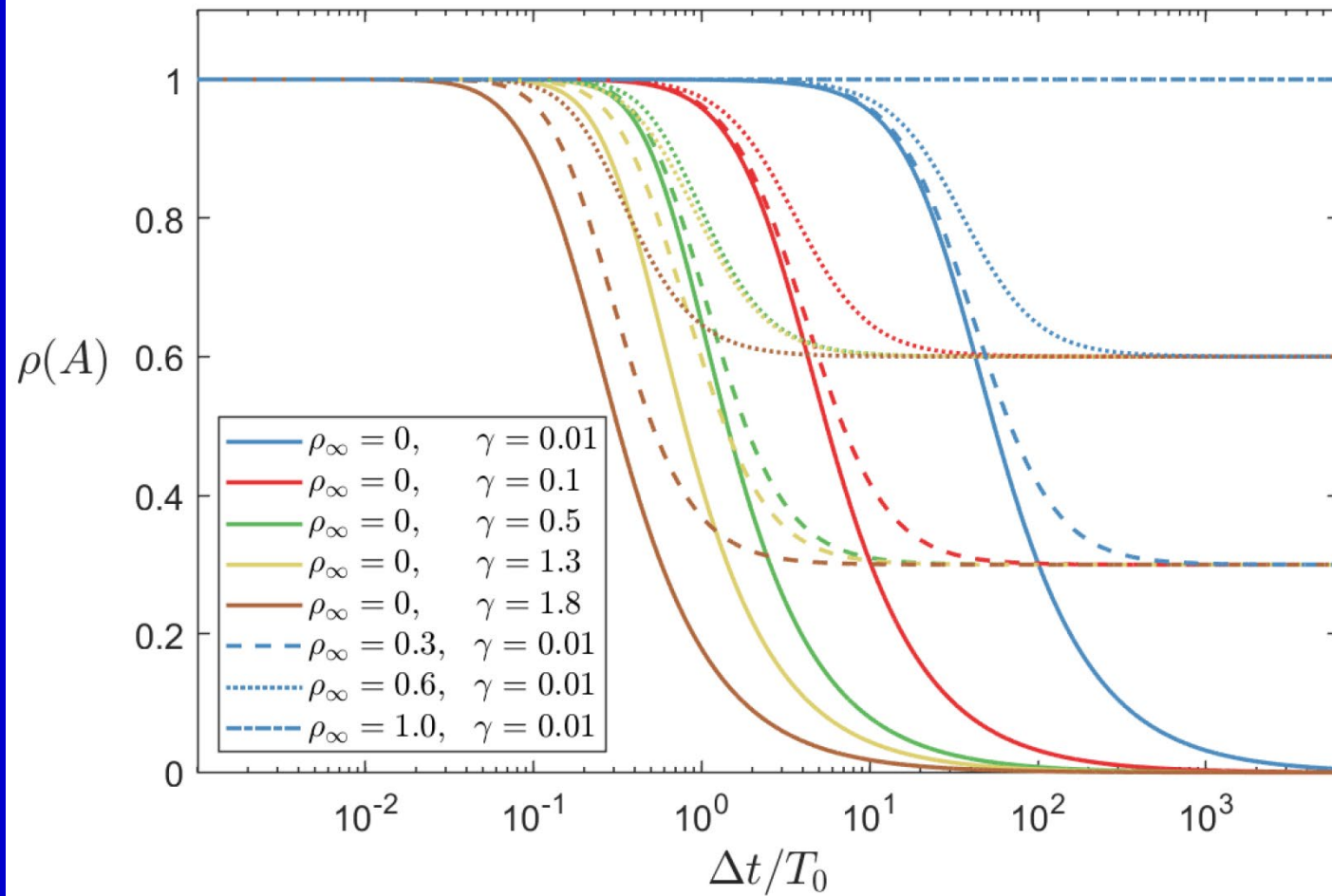
$${}^{t+\Delta t}\mathbf{U} = {}^t\mathbf{U} + \Delta t (q_0 {}^t\dot{\mathbf{U}} + q_1 {}^{t+\gamma\Delta t}\dot{\mathbf{U}} + q_2 {}^{t+\Delta t}\dot{\mathbf{U}})$$

$${}^{t+\Delta t}\dot{\mathbf{U}} = {}^t\dot{\mathbf{U}} + \Delta t (q_0 {}^t\ddot{\mathbf{U}} + q_1 {}^{t+\gamma\Delta t}\ddot{\mathbf{U}} + q_2 {}^{t+\Delta t}\ddot{\mathbf{U}})$$

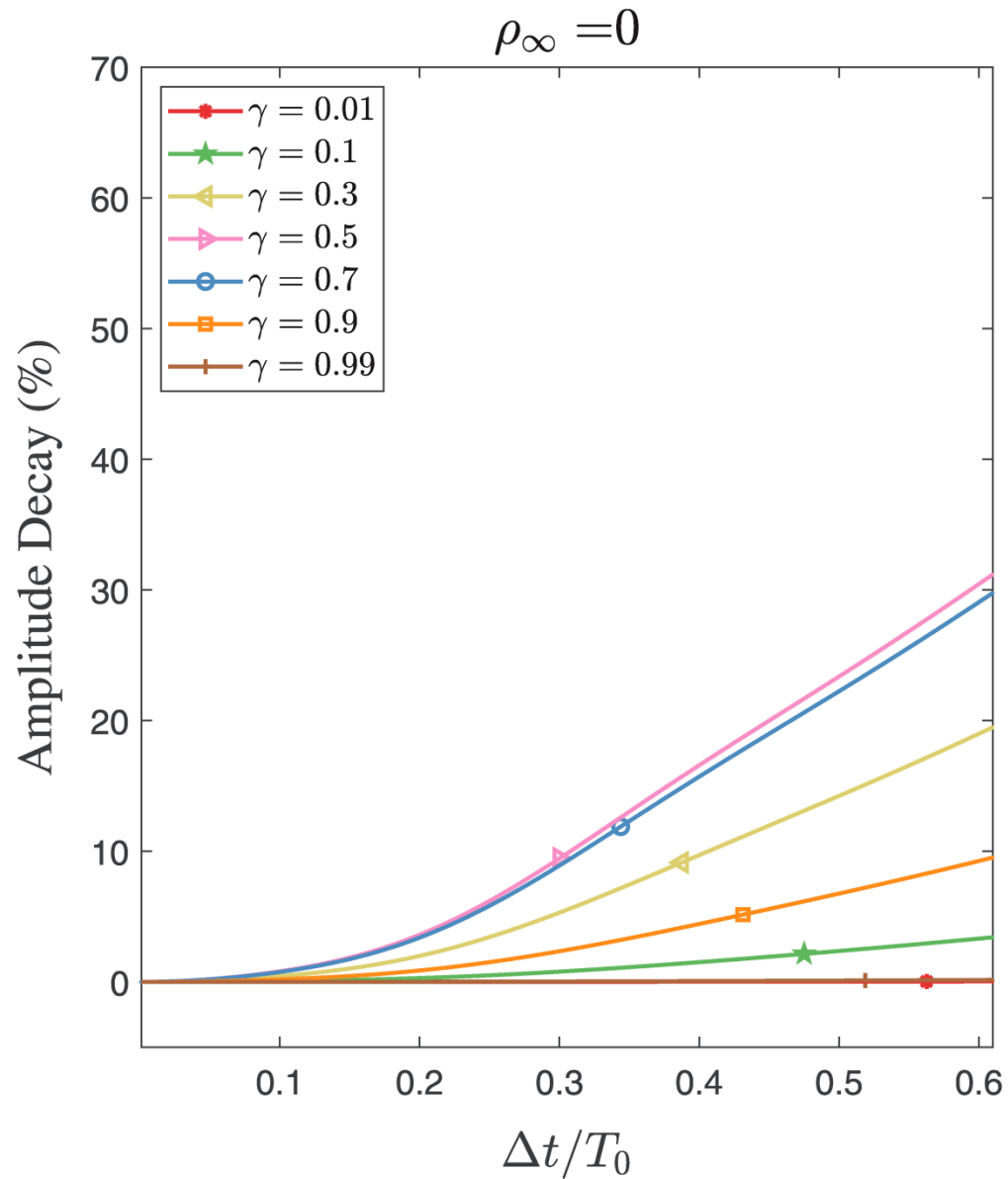
**with**

$$q_0 = (\gamma - 1)q_1 + \frac{1}{2}; \quad q_2 = -\gamma q_1 + \frac{1}{2}$$

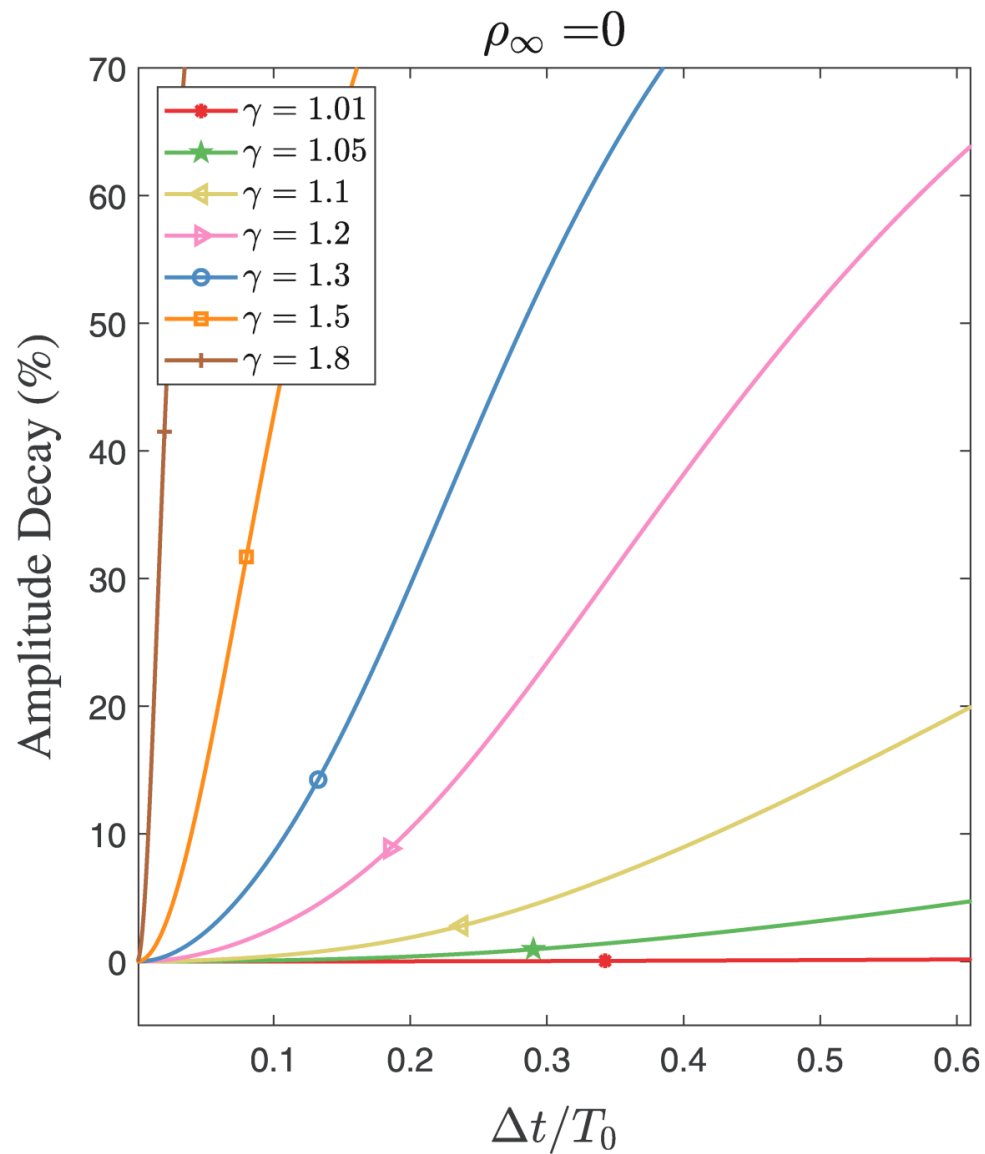
$$q_1 = \frac{\rho_\infty + 1}{2\gamma(\rho_\infty - 1) + 4} \quad \gamma \in \mathbb{R} \mid \gamma \neq 0, \gamma \neq 1$$



**Spectral radii for different values of parameters**



**Amplitude decay**



**Amplitude decay, the case  $\gamma > 1$  can be important**



**The scheme is the standard Bathe method when  $\rho_\infty = 0$  and  $\gamma$  is varying**

**It is the TR for each sub-step when  $\rho_\infty = 1$**

**Further, the scheme is unconditionally stable and second-order accurate**

**In some analyses, the use of the 2 parameters can be valuable -- but we only need to use  $\rho_\infty$  as a parameter and can use the corresponding optimal  $\gamma$ . Hence the method is a 1-parameter scheme using usually  $\rho_\infty = 0$**

**Another way to proceed --**

**use  $\gamma = 0.5$  and for the 2<sup>nd</sup> sub-step:**

$${}^{t+\Delta t}\mathbf{U} =$$

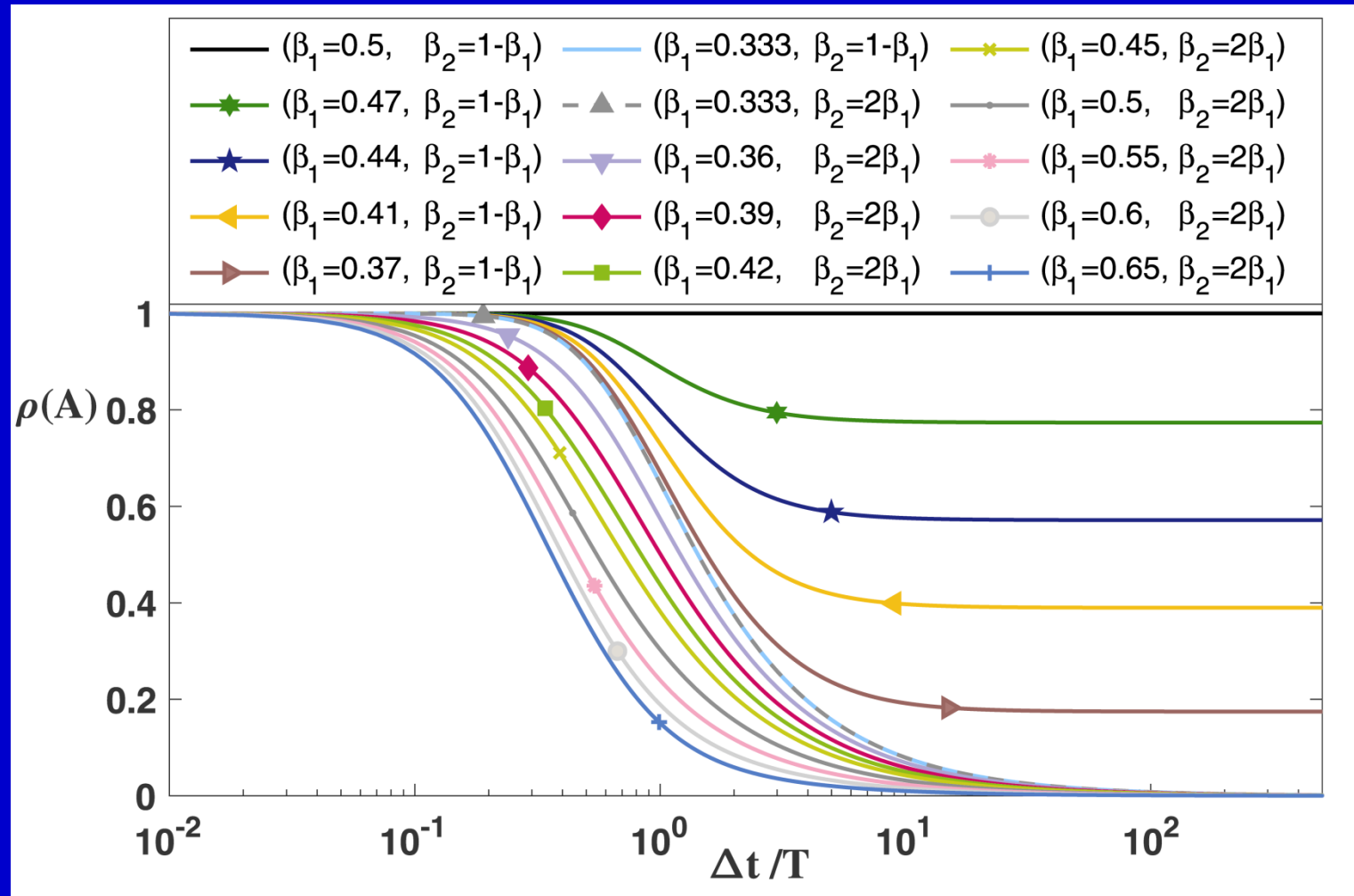
$${}^t\mathbf{U} + (0.5\Delta t) \left( (1 - \beta_1) {}^t\dot{\mathbf{U}} + \beta_1 {}^{t+\gamma\Delta t}\dot{\mathbf{U}} \right) \\ + (0.5\Delta t) \left( (1 - \beta_2) {}^{t+\gamma\Delta t}\dot{\mathbf{U}} + \beta_2 {}^{t+\Delta t}\dot{\mathbf{U}} \right)$$

**Here  $\beta_1$  and  $\beta_2$  are parameters to obtain different AD and PE**

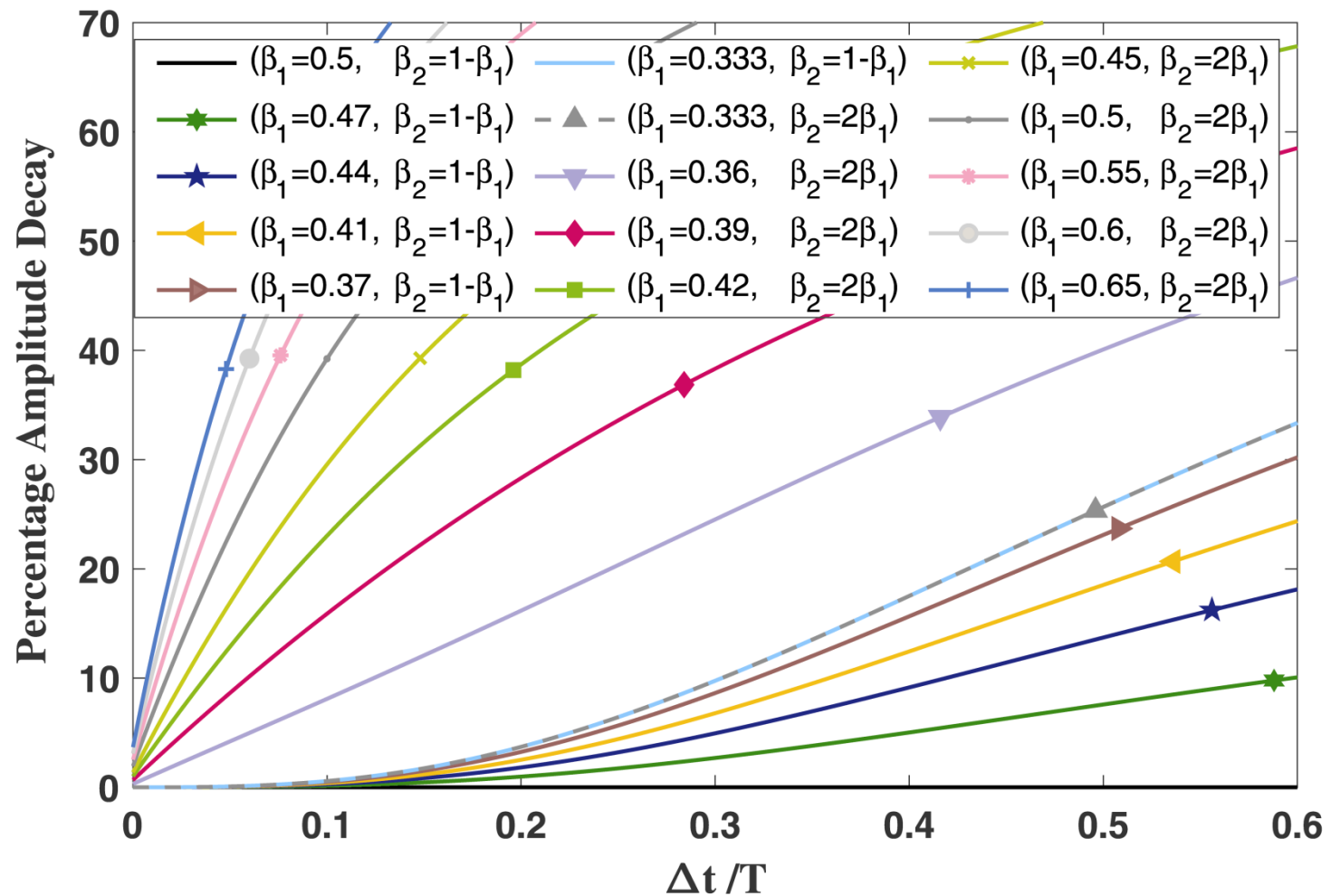
## Assumption for the velocity --

$$\begin{aligned} {}^{t+\Delta t}\dot{\mathbf{U}} = & \\ & {}^t\dot{\mathbf{U}} + (0.5\Delta t) \left( (1-\beta_1) {}^t\ddot{\mathbf{U}} + \beta_1 {}^{t+\gamma\Delta t}\ddot{\mathbf{U}} \right) \\ & + (0.5\Delta t) \left( (1-\beta_2) {}^{t+\gamma\Delta t}\ddot{\mathbf{U}} + \beta_2 {}^{t+\Delta t}\ddot{\mathbf{U}} \right) \end{aligned}$$

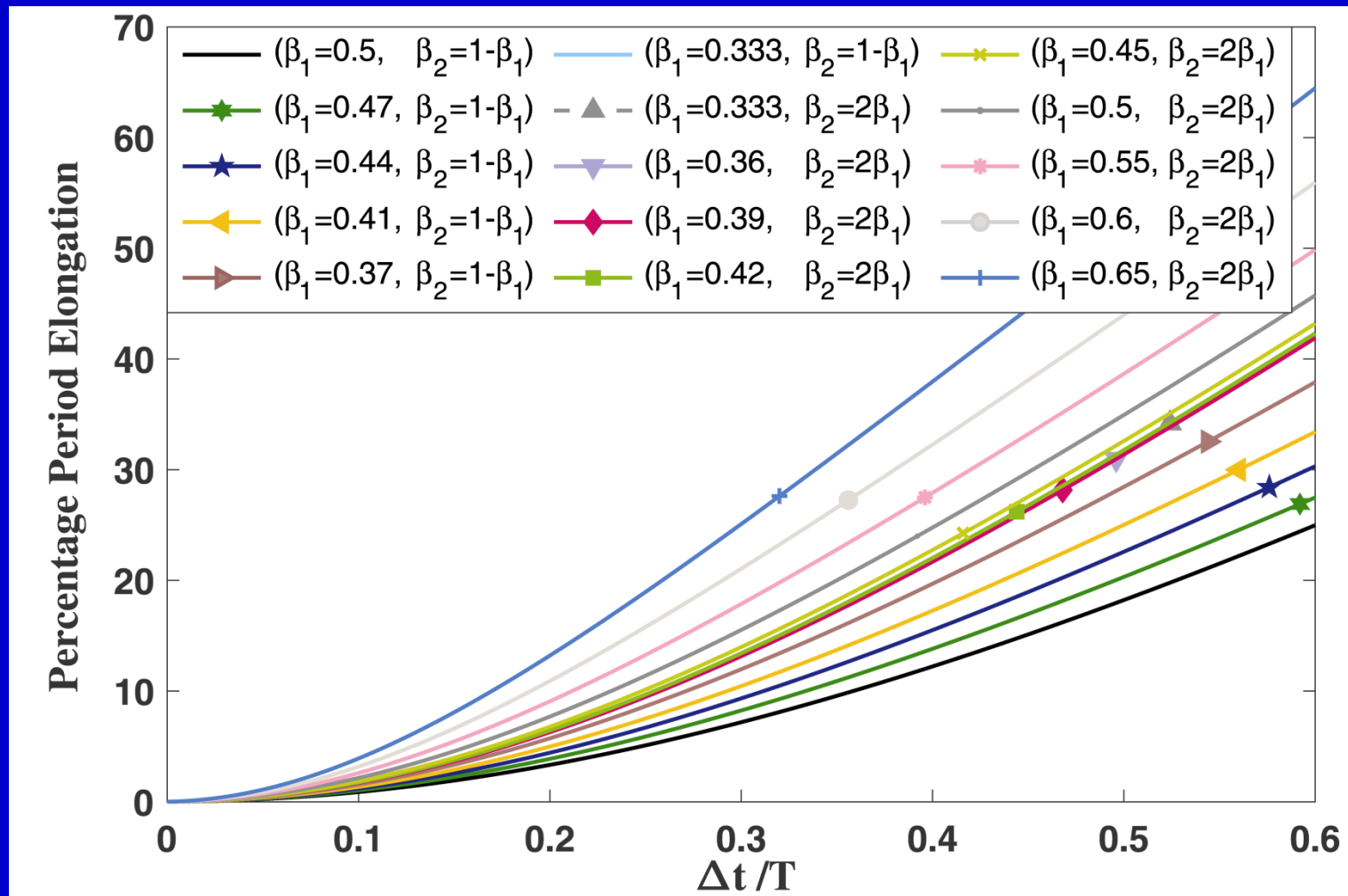
**Again, the  $\beta_1$  and  $\beta_2$  are parameters to obtain different AD and PE . This scheme is a special case of the  $\rho_\infty$  - Bathe scheme**



**Spectral radii, include TR and Bathe std. scheme**

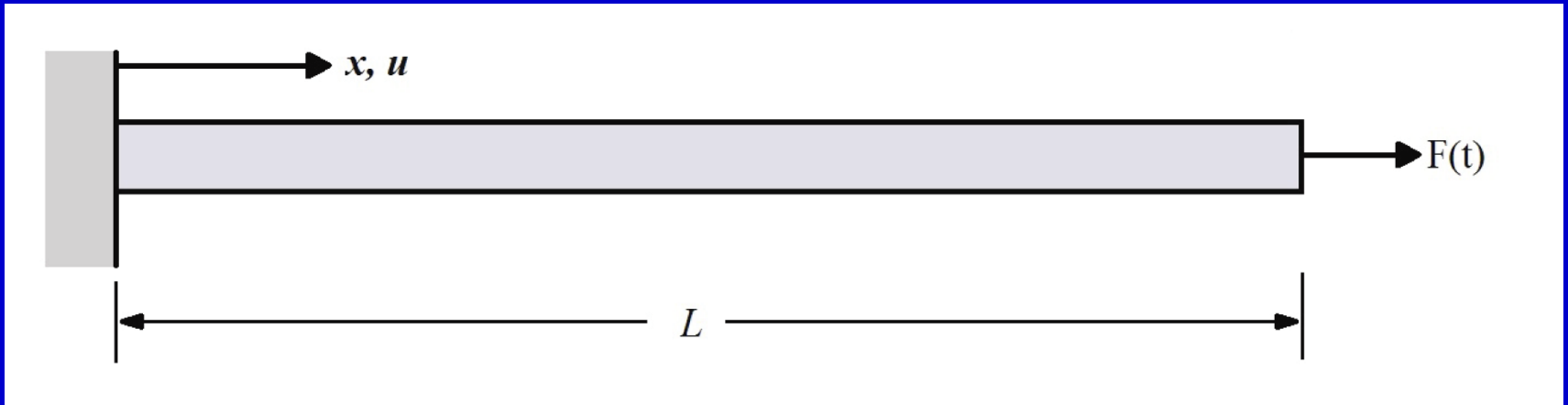


**Amplitude decay (AD)**

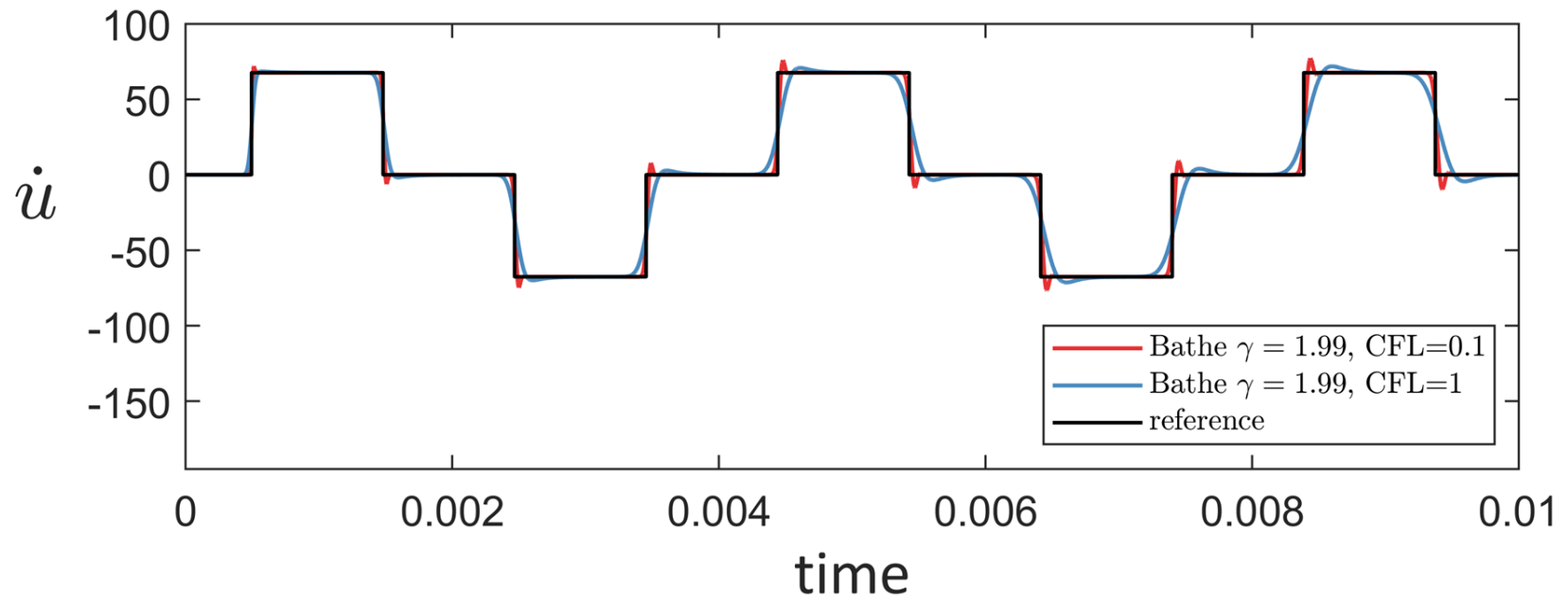


**Period elongation (PE)**

# Illustrative example

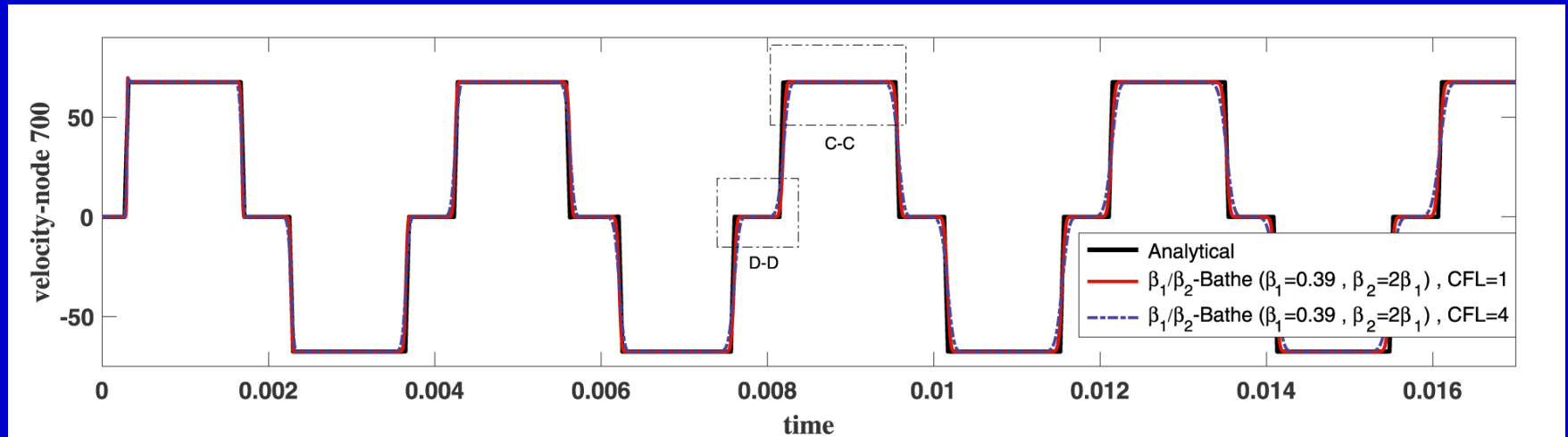


**Analysis of bar subjected to step load;  
consider the velocity at a station  $x$**

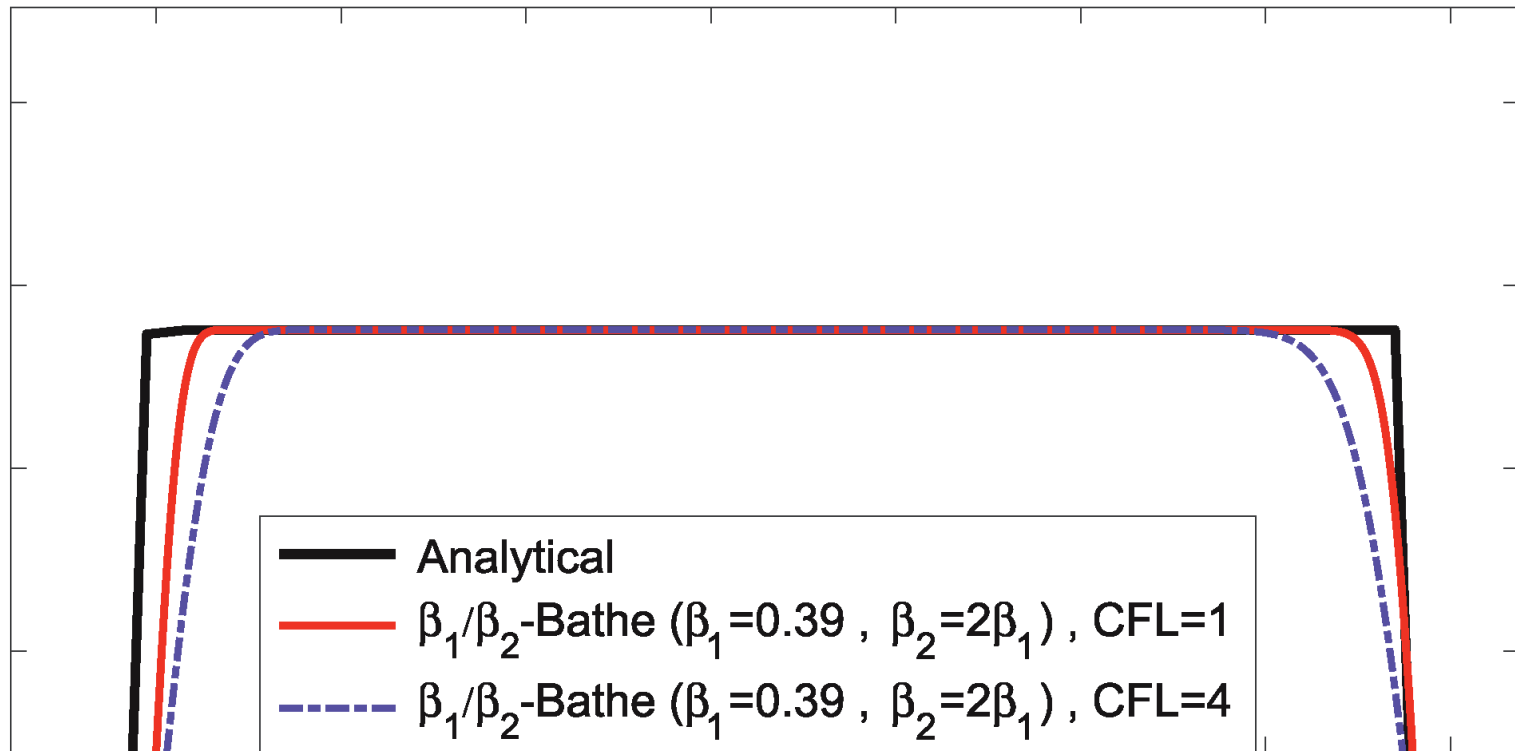


**Using the Bathe method with  
 $\gamma = 1.99$ , CFL = 1, 0.1**





Using the Bathe method with  
**CFL=1, 4;**  $\beta_1 = 0.39, \beta_2 = 2\beta_1$



**Detail of velocity at  $x = 0.7 L$**

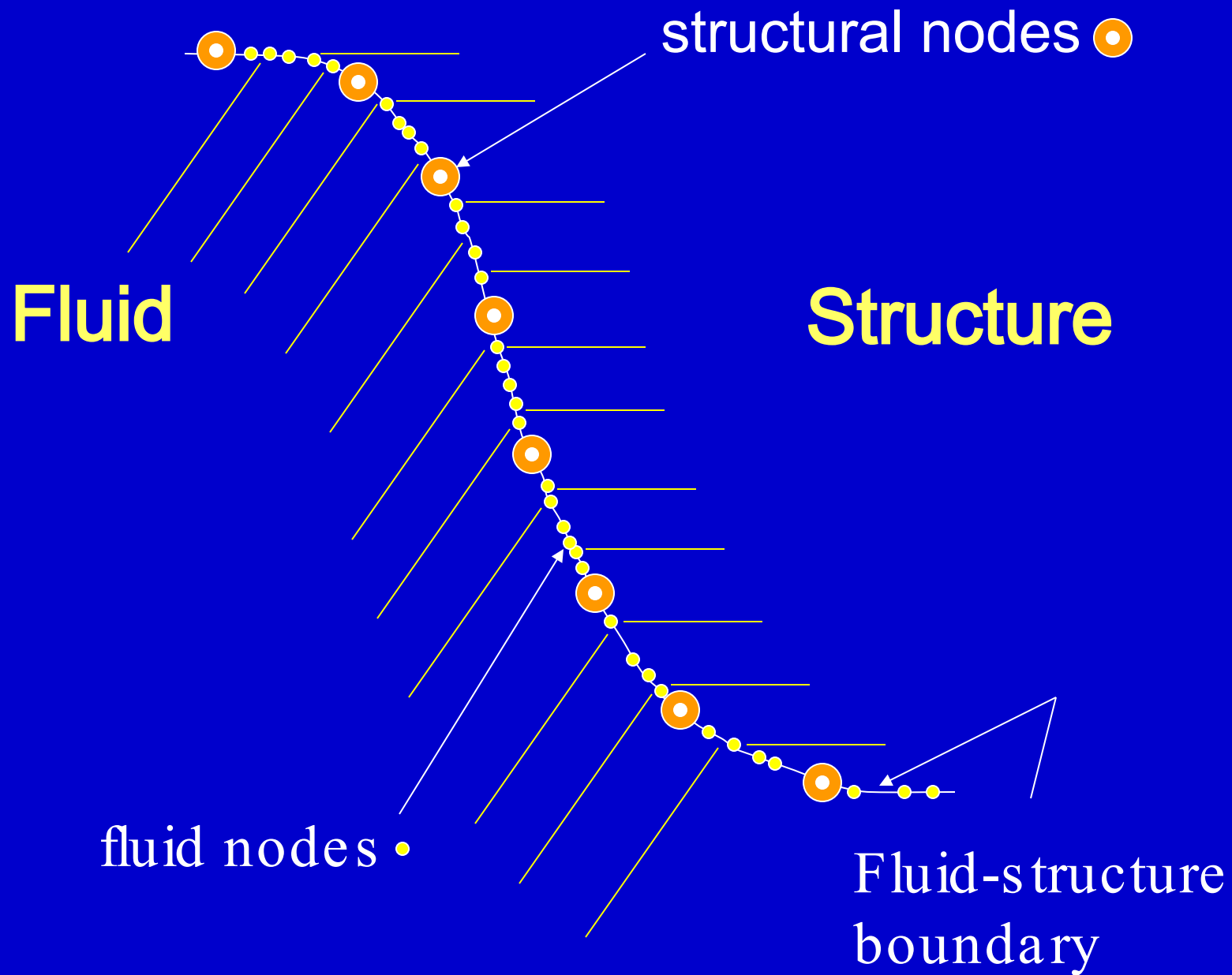
# **Multi-physics simulations**

## **Fluid-structure interactions**

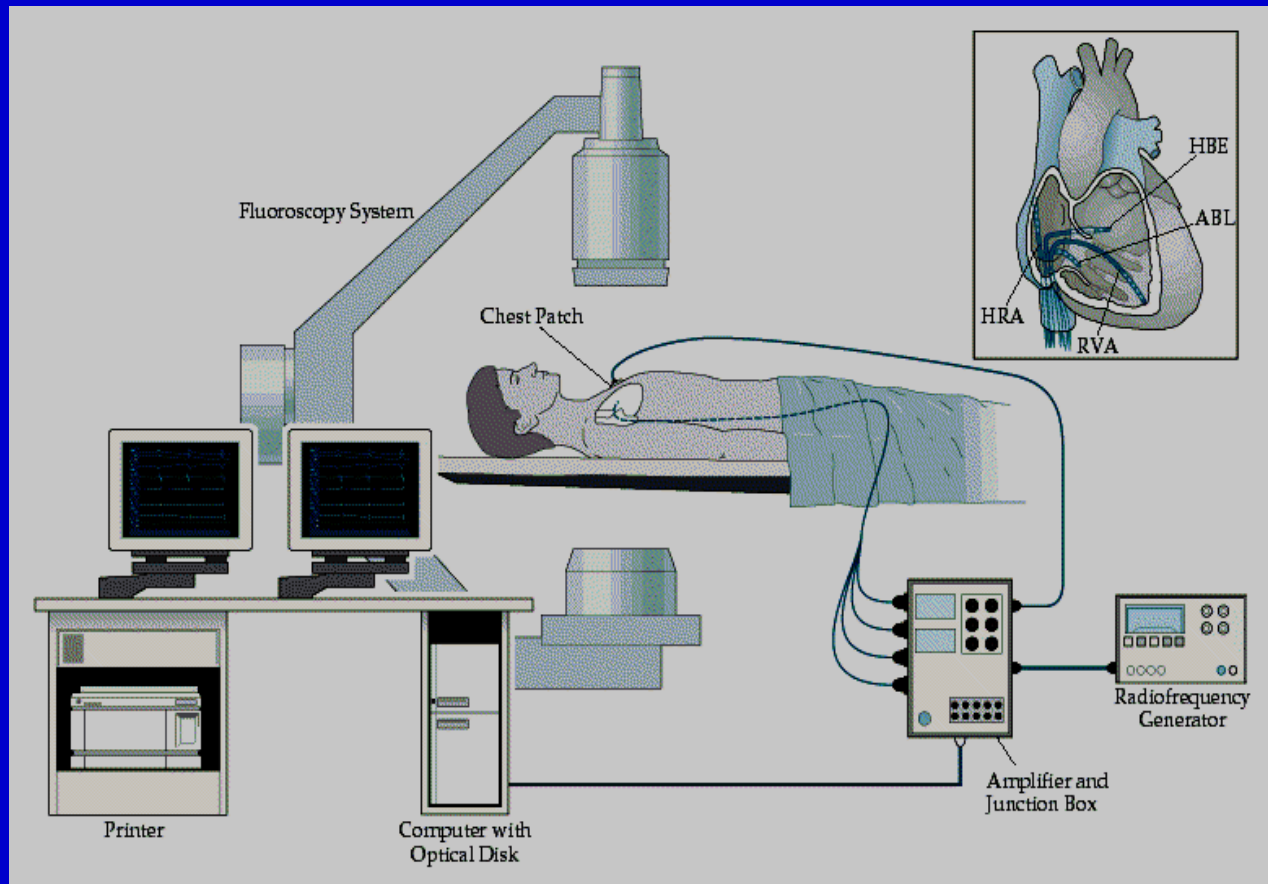
**e.g. blasts, pumps, shock-  
absorbers, hydro-mounts, ..**

## **Electromagnetic effects**

**e.g. pumps, heating, mixing,  
medical ablation, ...**



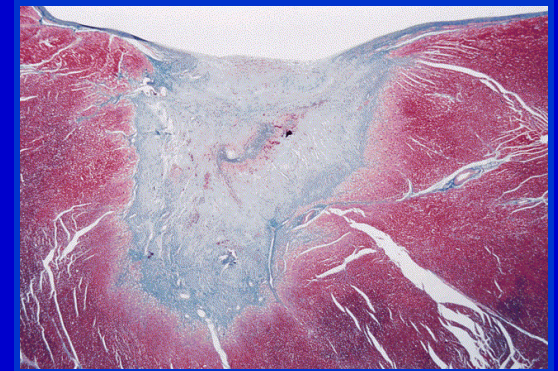
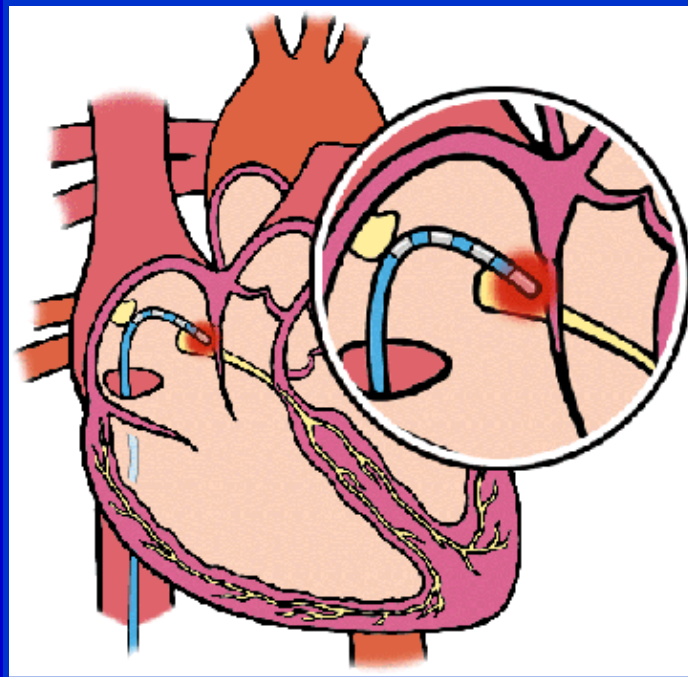
# RF Ablation Lab Set Up



# Radio-frequency tissue ablation



Electrode



Lesion

# Modeling Parameters

## Conventional Catheter

Power: 38W

#Nodes: 111,205

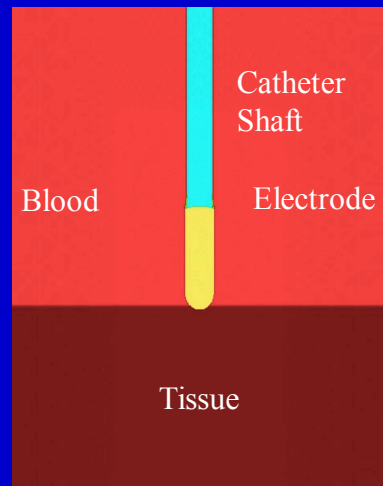
#Elements: 667,880

## 25mm High Density Mesh

Power: 74W, 91W

#Nodes: 1,405,995

#Elements: 8,627,692

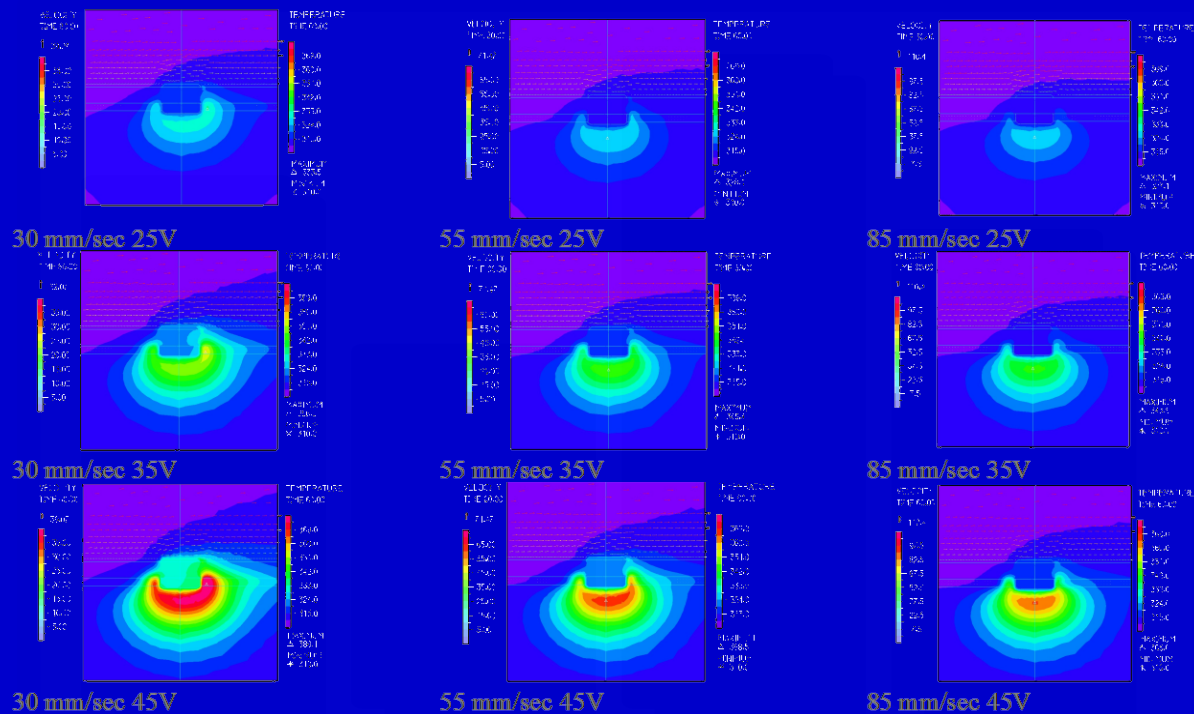


# Temperature/Velocity Profiles for Varying Flows

Increasing Flow



Increasing Power





# **A state of the art industry FSI analysis**

- A scroll compressor, 3D analysis**
- Transient response, many thousands of time steps**
- Bathe implicit time integration used, for fluid and solid**
- Solved for 5 (to 10) revolutions, using adaptive meshing for the fluid**

**Solution obtained by Emerson Inc. using ADINA**

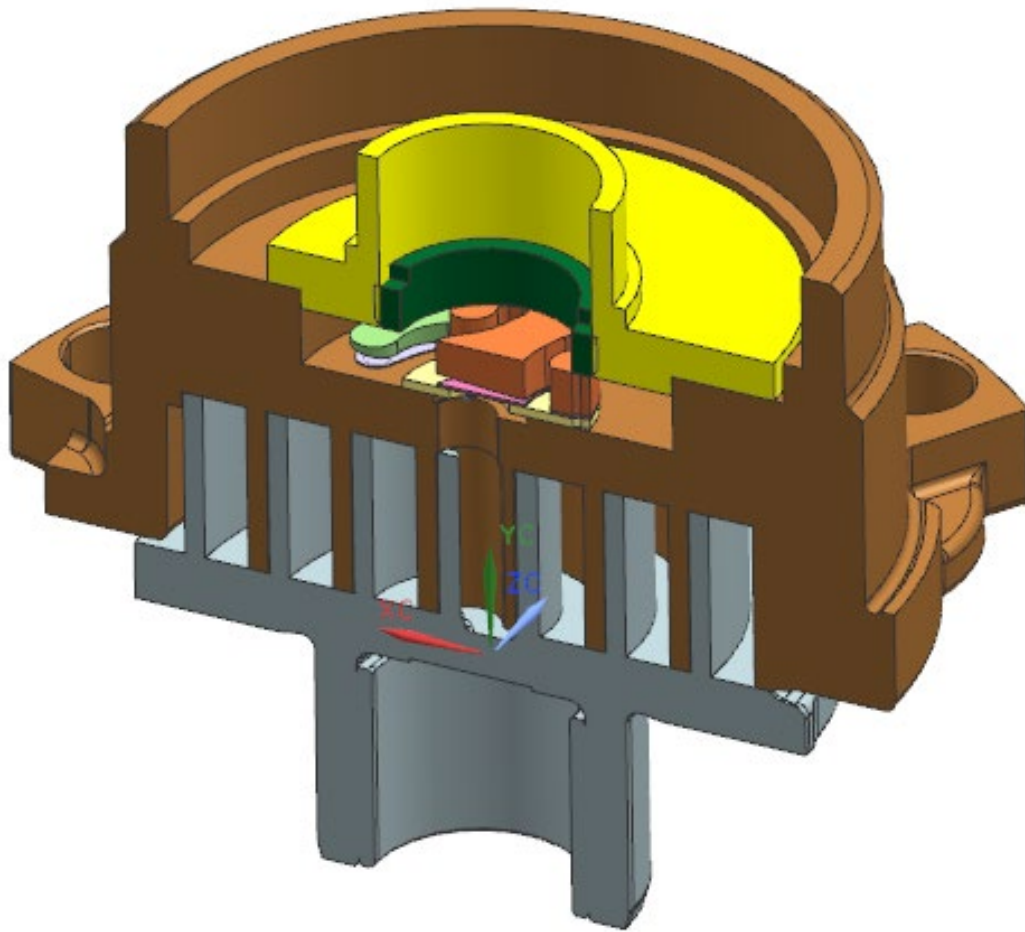
# Emerson Climate Technologies Scroll Compressor Analysis

**Device for compressing air  
or refrigerant**

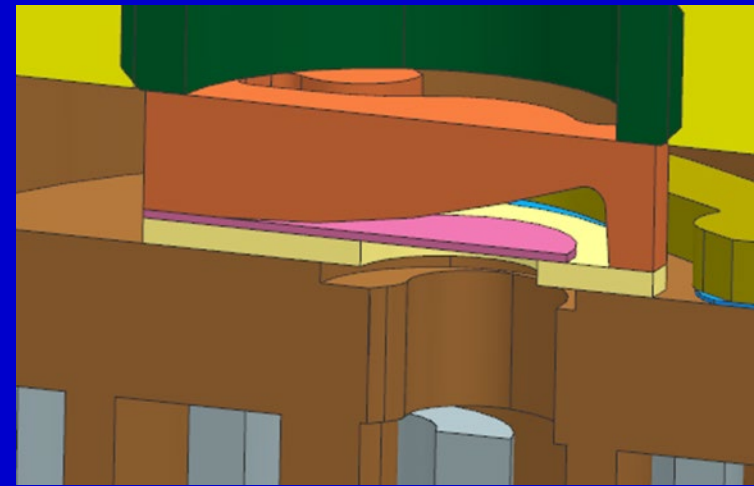
**Used in air conditioners,  
automobile superchargers,  
industrial refrigerators**

**Fewer moving parts,  
quieter, more efficient**

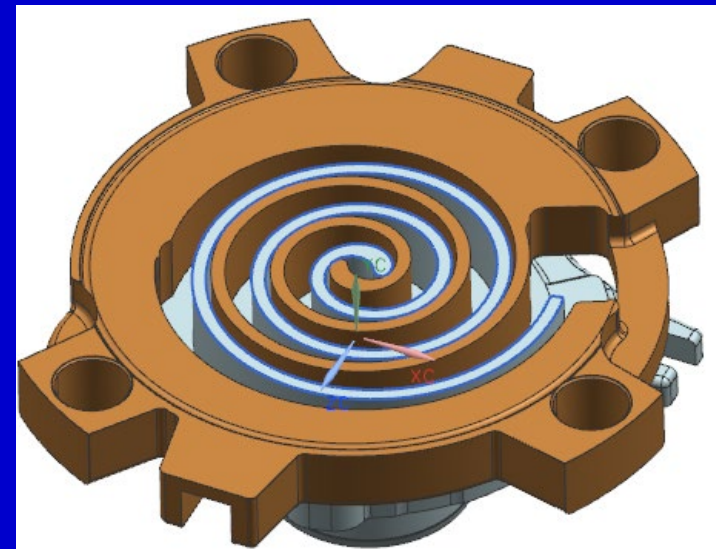




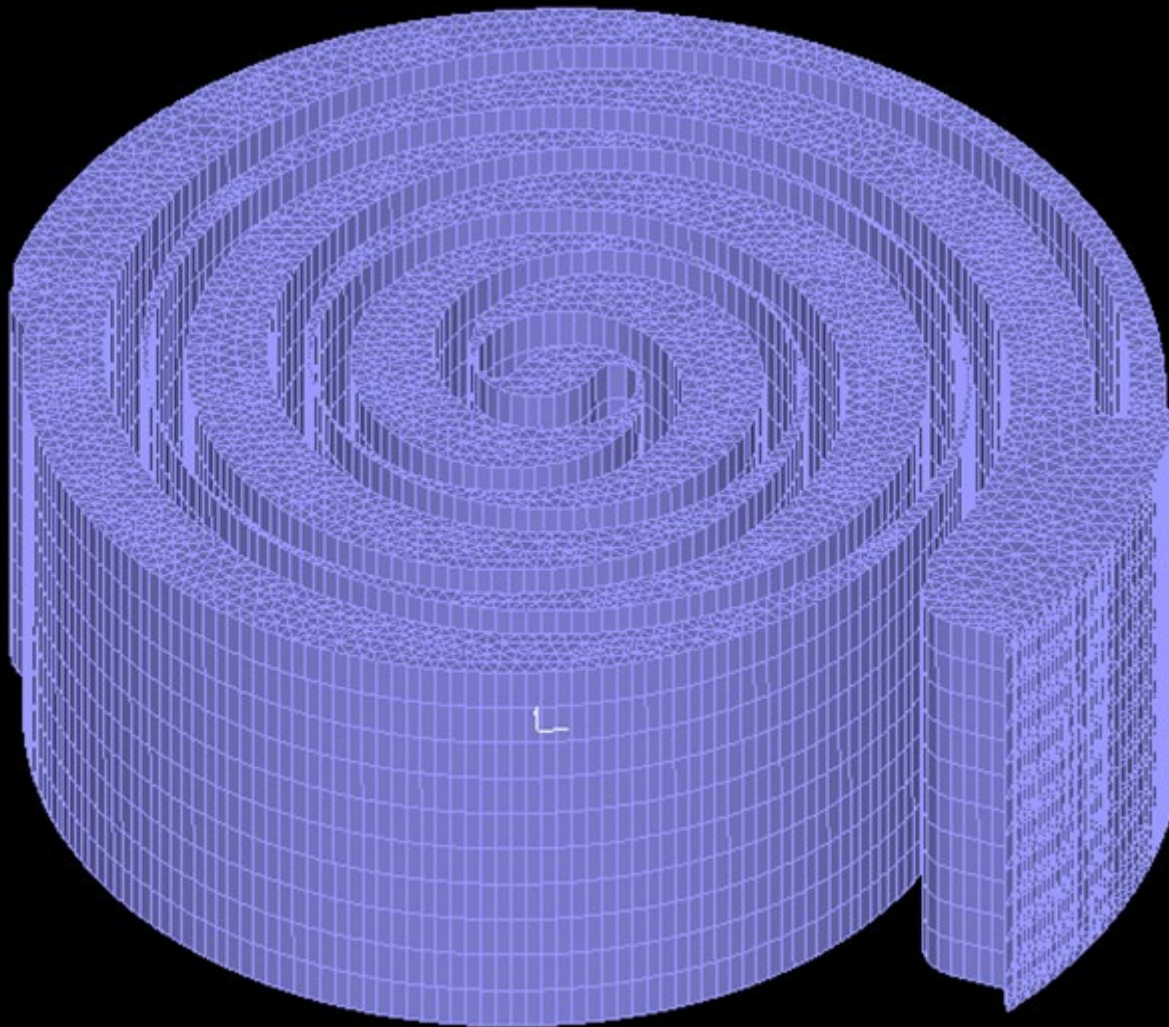
**Orbiting scroll**



**Valve**

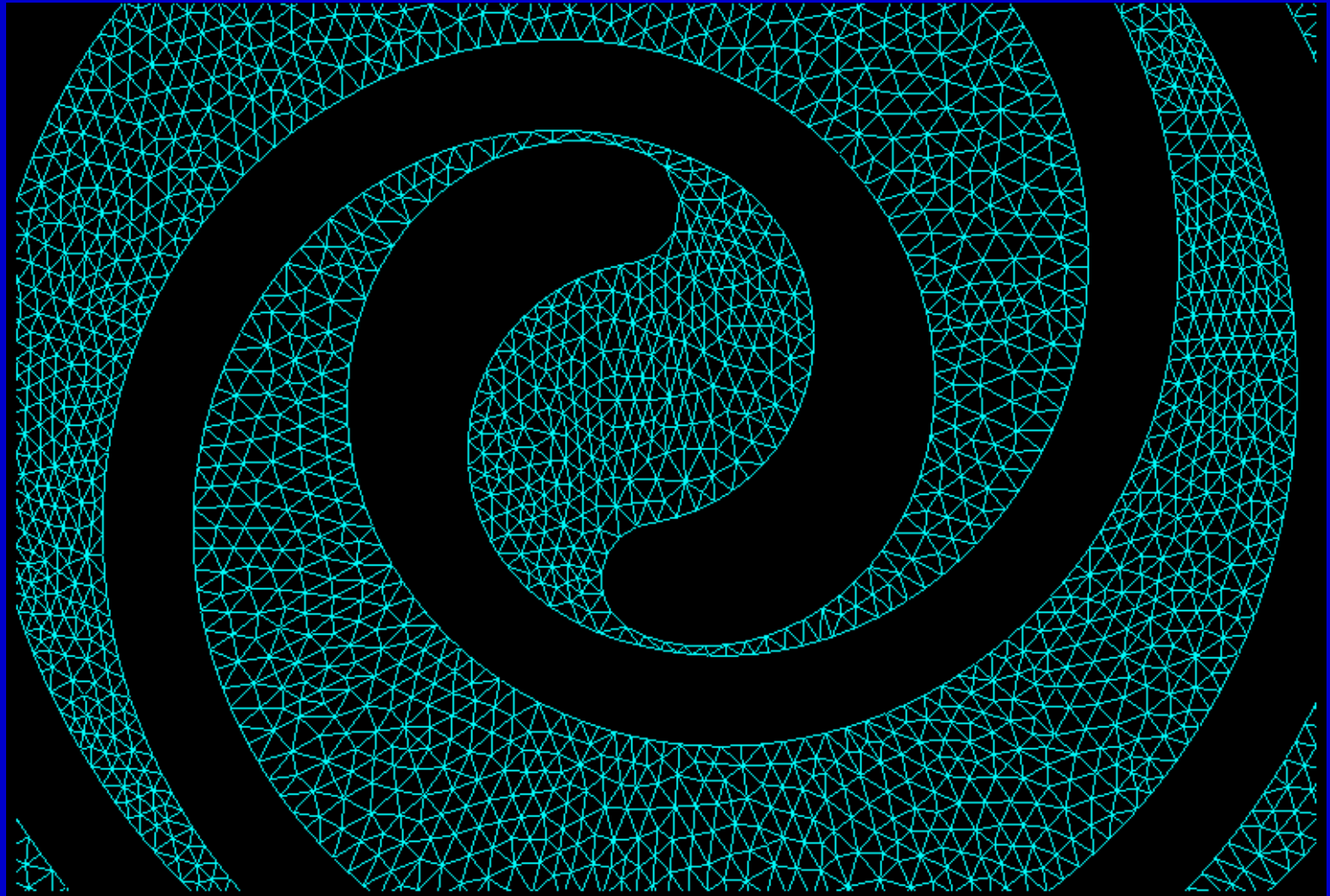


**A  
D  
I  
N  
A**

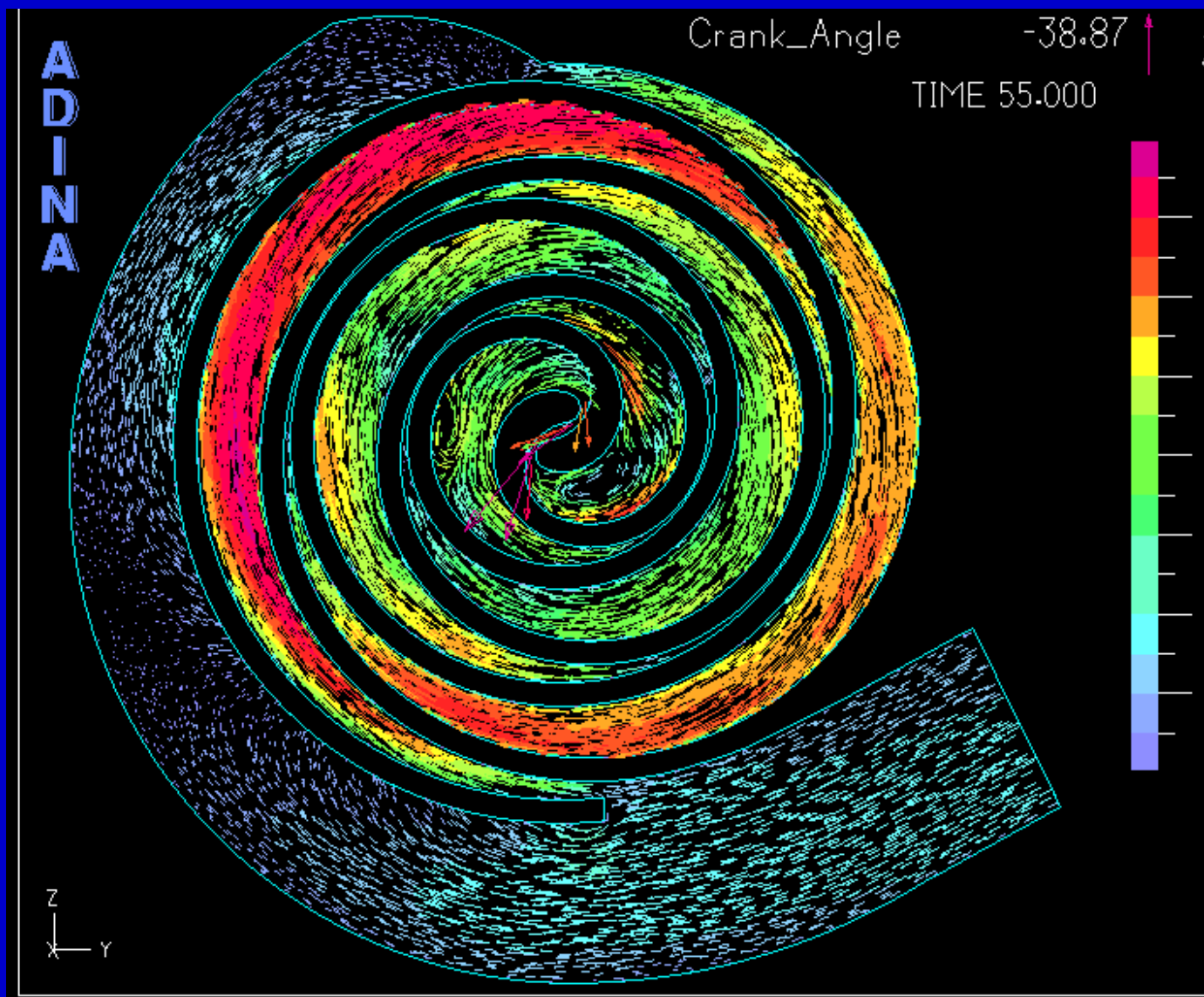


**The fluid mesh used**





**Mesh deformations in horizontal cut**

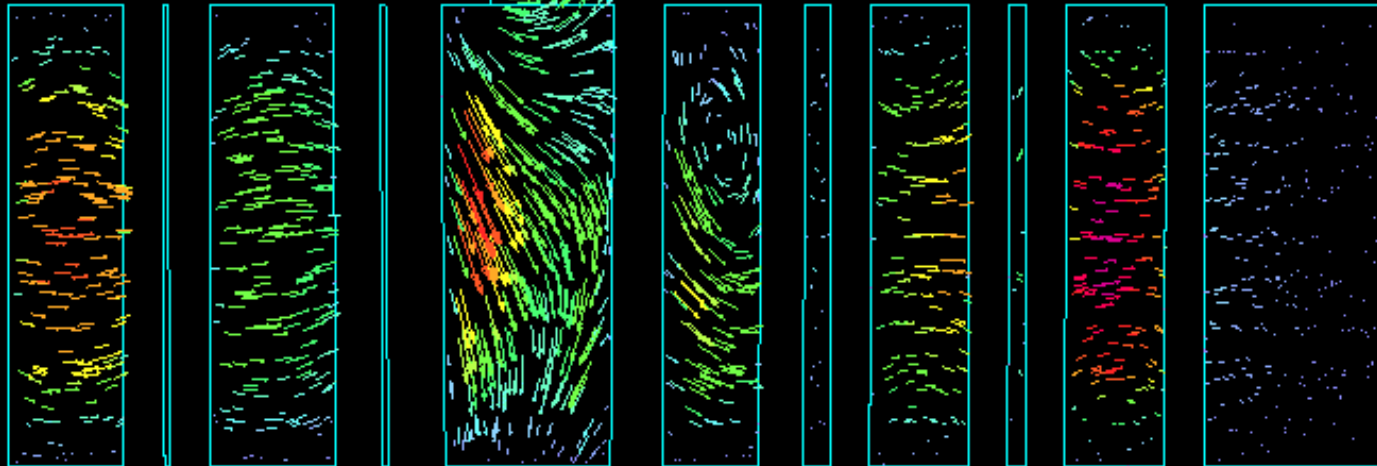


**Fluid velocity in horizontal cut**

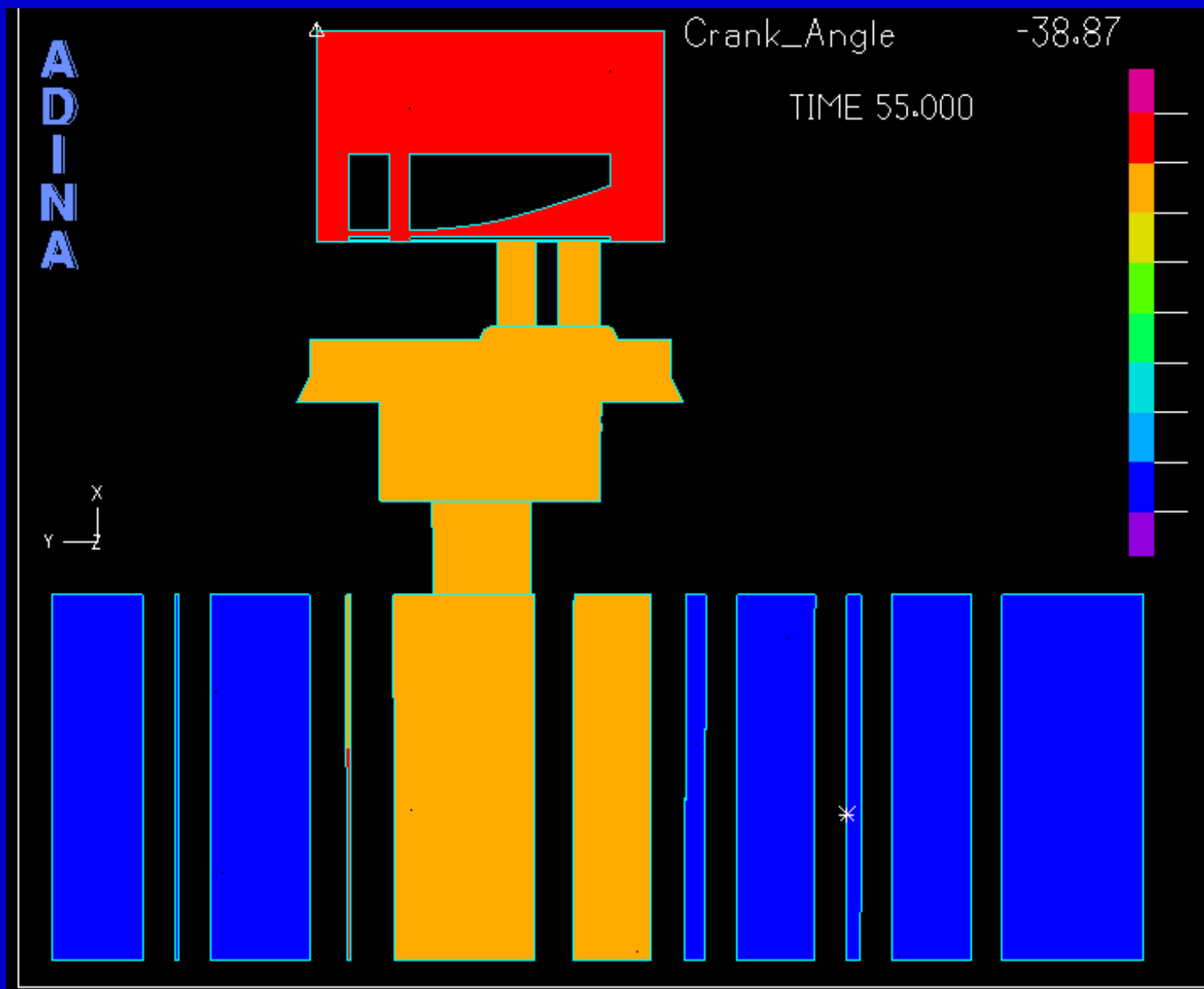
A  
D  
I  
N  
A

Crank\_Angle -38.87

TIME 55.000

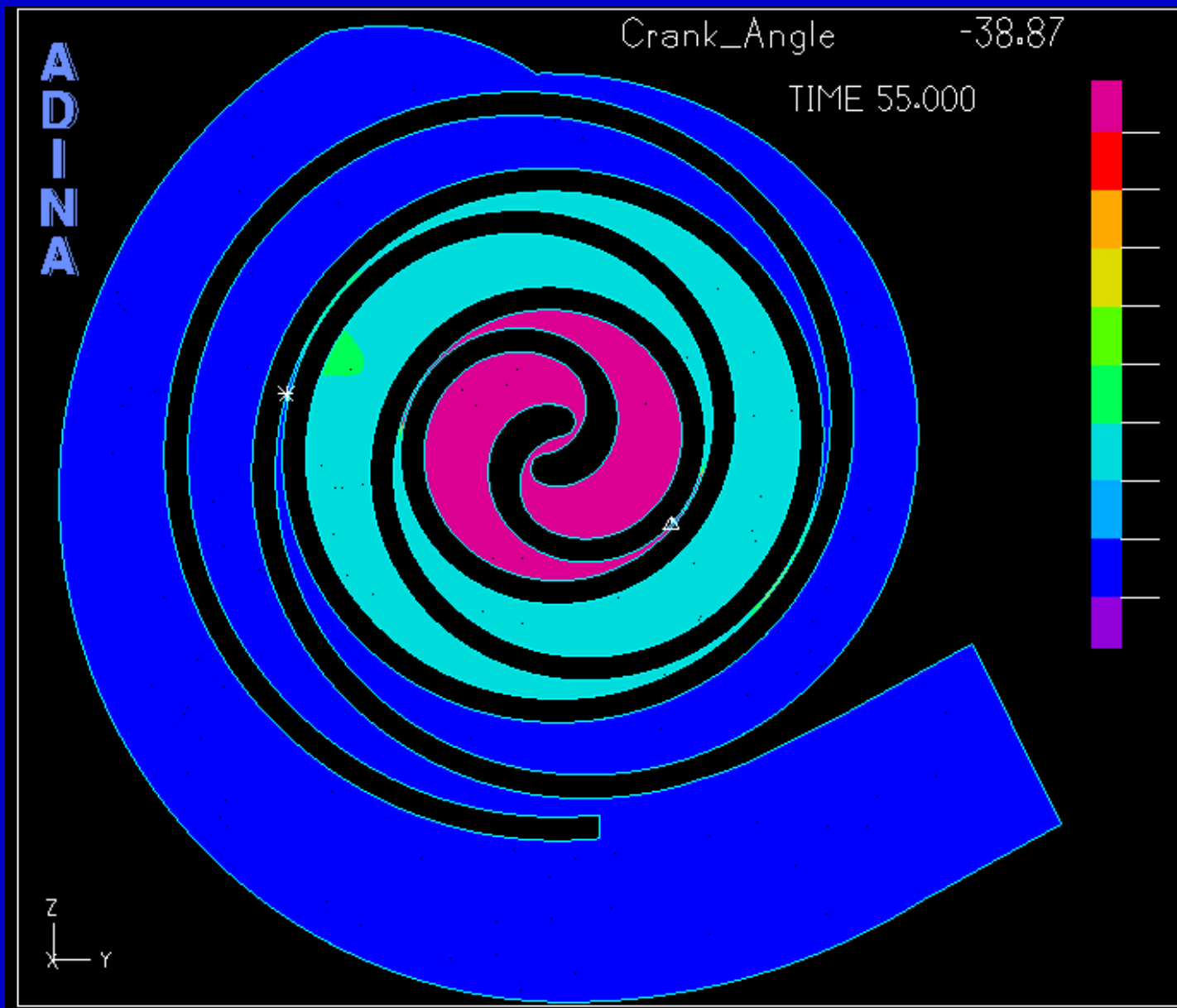


**Fluid velocity in vertical cut**



**Pressure in vertical cut**

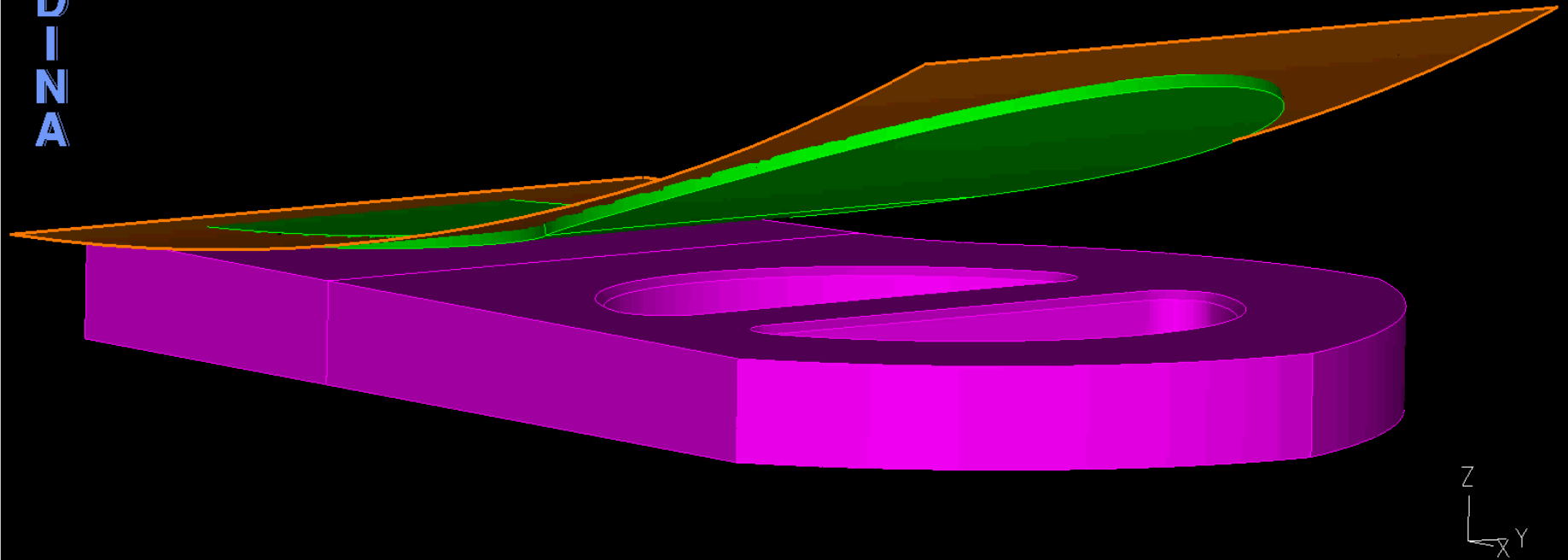




**Gas pressure in horizontal cut**

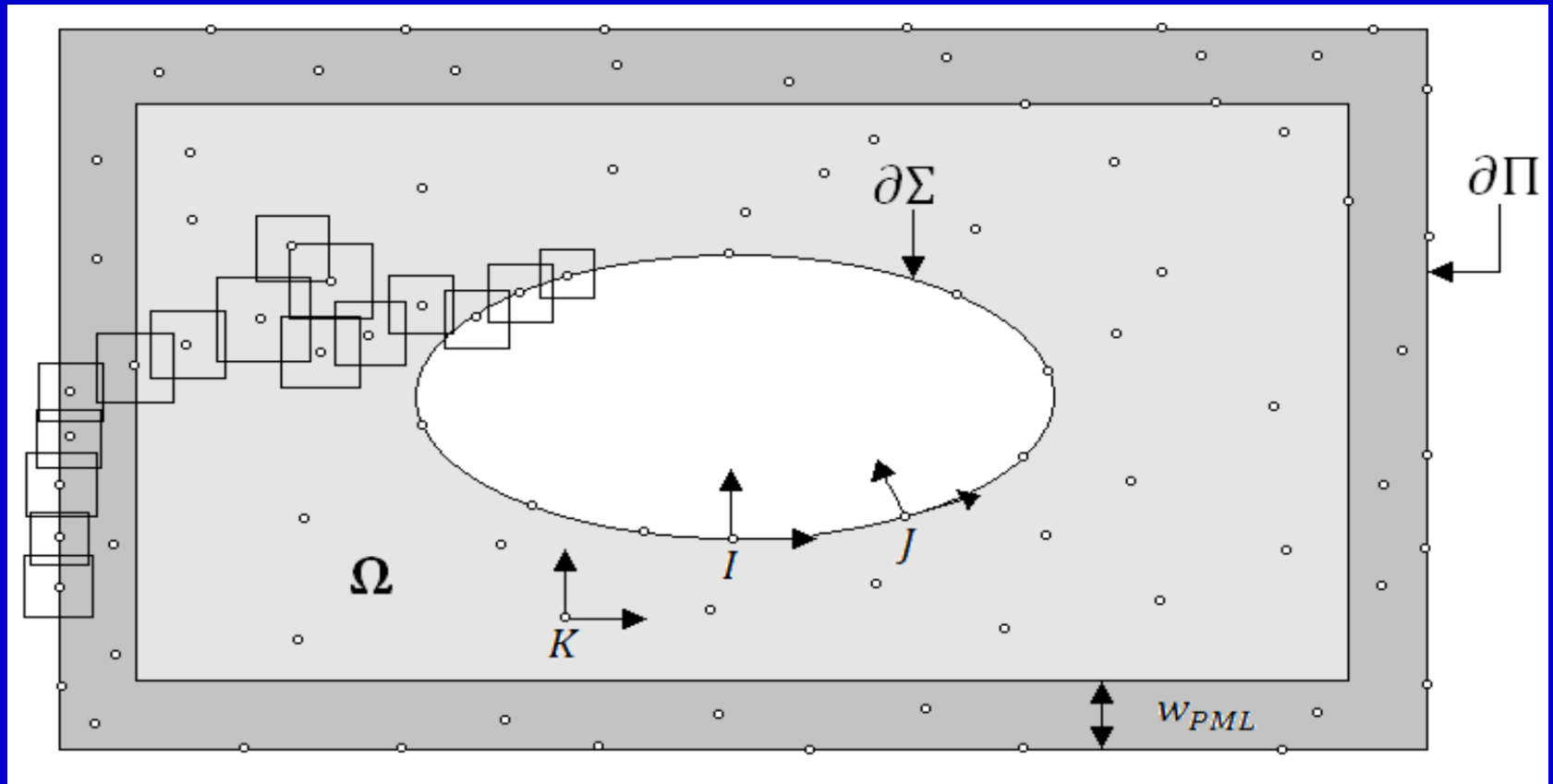
A  
D  
I  
N  
A

TIME 100.000



**Cracking of valve,  
new design reached**

# EM wave scattering by mesh-free method



**Like MFS and ‘overlapping elements’**

W.L. Nicomedes, K.J. Bathe, F.J.S. Moreira and R.C. Mesquita.  
Meshfree analysis of electromagnetic wave ... . C & S 2017.

# **Finite element simulations at the nano-scale**

**Proteins and DNA structures**

**Finite element methods can be very  
effective in analyzing these structures**

**Basic applications in biological engineering,  
energy engineering, medical sciences, ....**

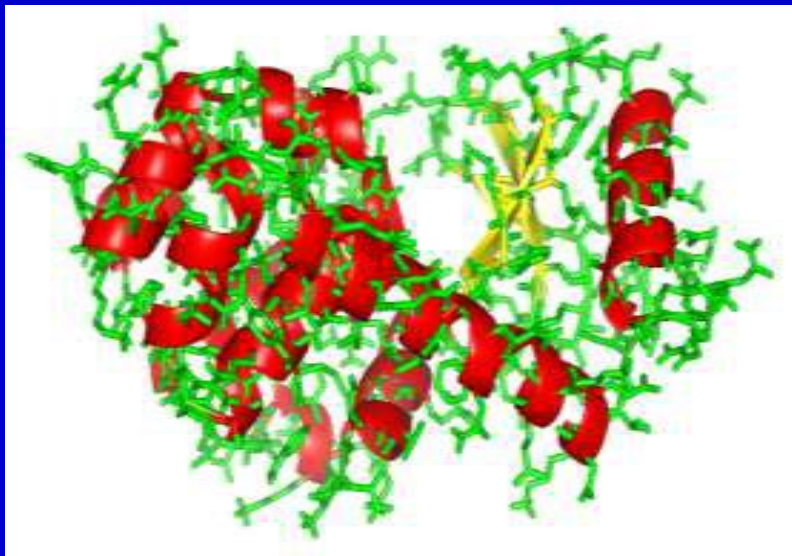
**Huge field for the future ... !**

**The most basic approach is to analyze  
DNA structures and proteins using  
'Molecular Dynamics' ... computationally  
very restrictive**

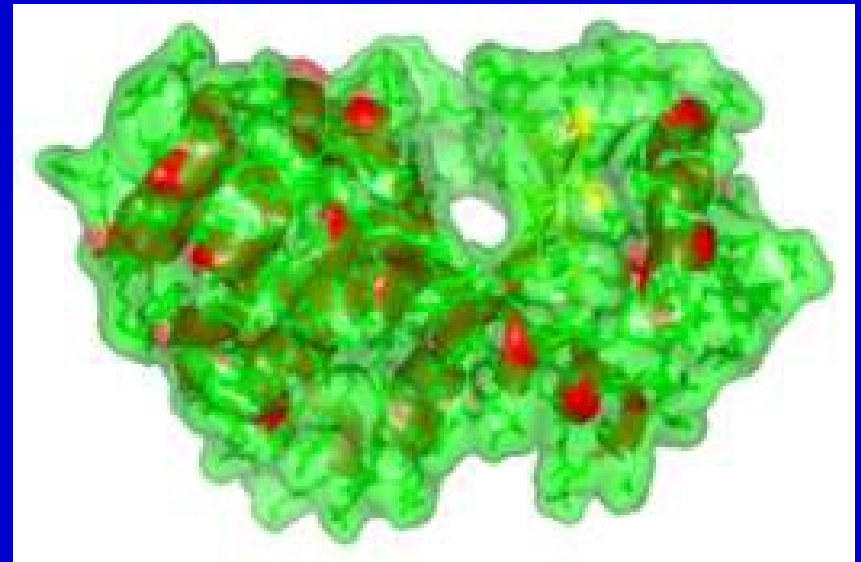
**Langevin or Brownian dynamics is an avenue  
to follow ---**

**We developed finite element procedures for  
effective solutions,**

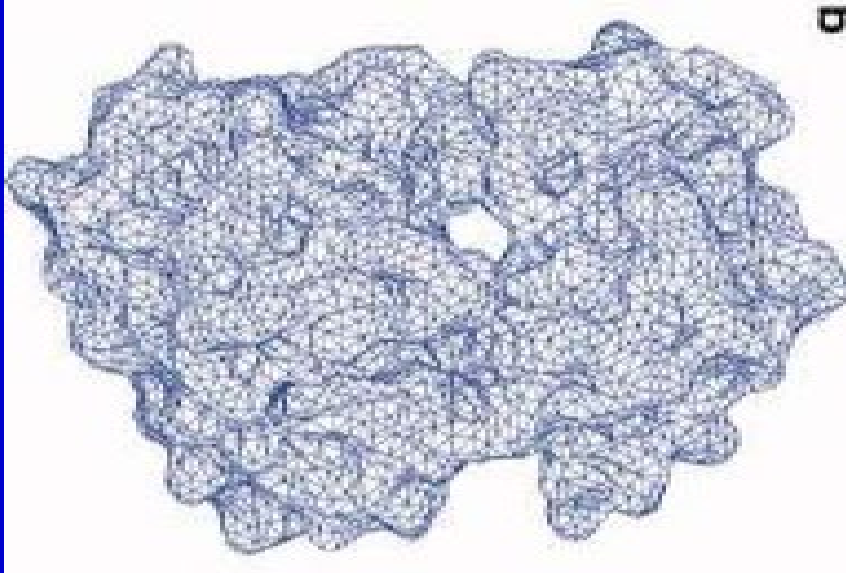
**with these procedures we can efficiently analyze  
even “supramolecular assemblies”**



**Structure**



**Molecular surface**



**Finite element model  
of protein for frequency  
solutions (using the  
subspace iteration  
method) and dynamic  
analyses**

# Equations of motions (Langevin dynamics)

$$\mathbf{M}\ddot{\mathbf{x}} + \mathbf{Z}\dot{\mathbf{x}} + \mathbf{K}\mathbf{x} = \mathbf{f}(t)$$

**K = protein elastic stiffness matrix**

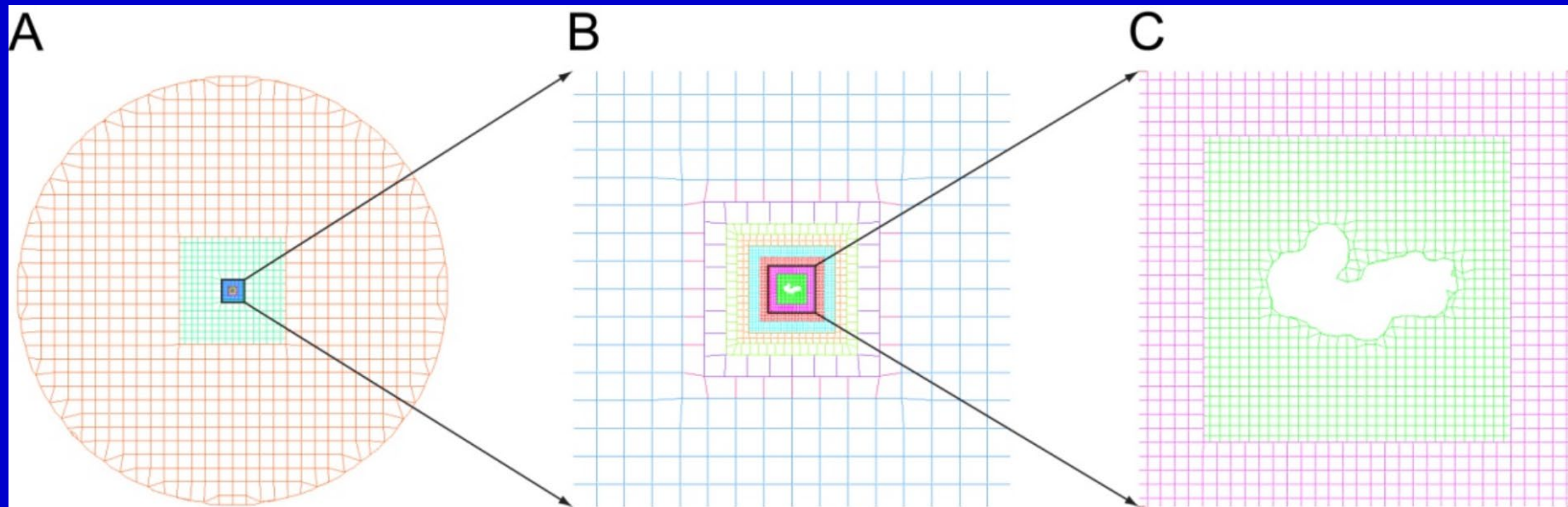
**Z = solvent friction matrix**

**M is assumed = 0 in Brownian dynamics**

**f(t) = solvent-induced**

$$\langle \mathbf{f}_i(t) \rangle = 0$$

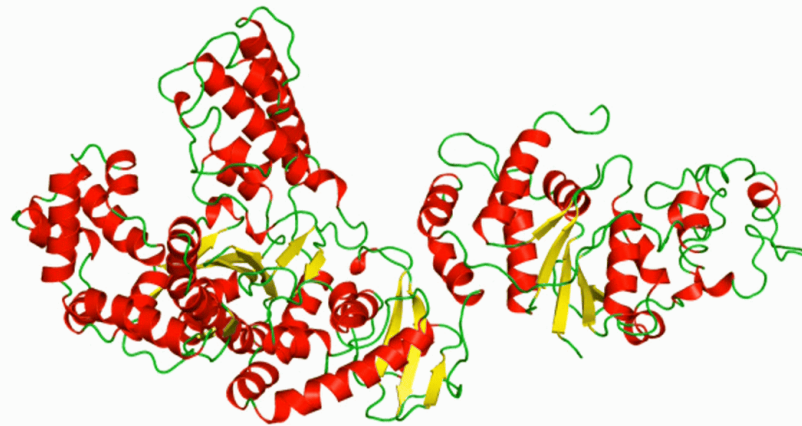
$$\langle \mathbf{f}_i(t) \times \mathbf{f}_j(t') \rangle = 2k_B T \mathbf{Z}_{ij} \delta(t - t')$$



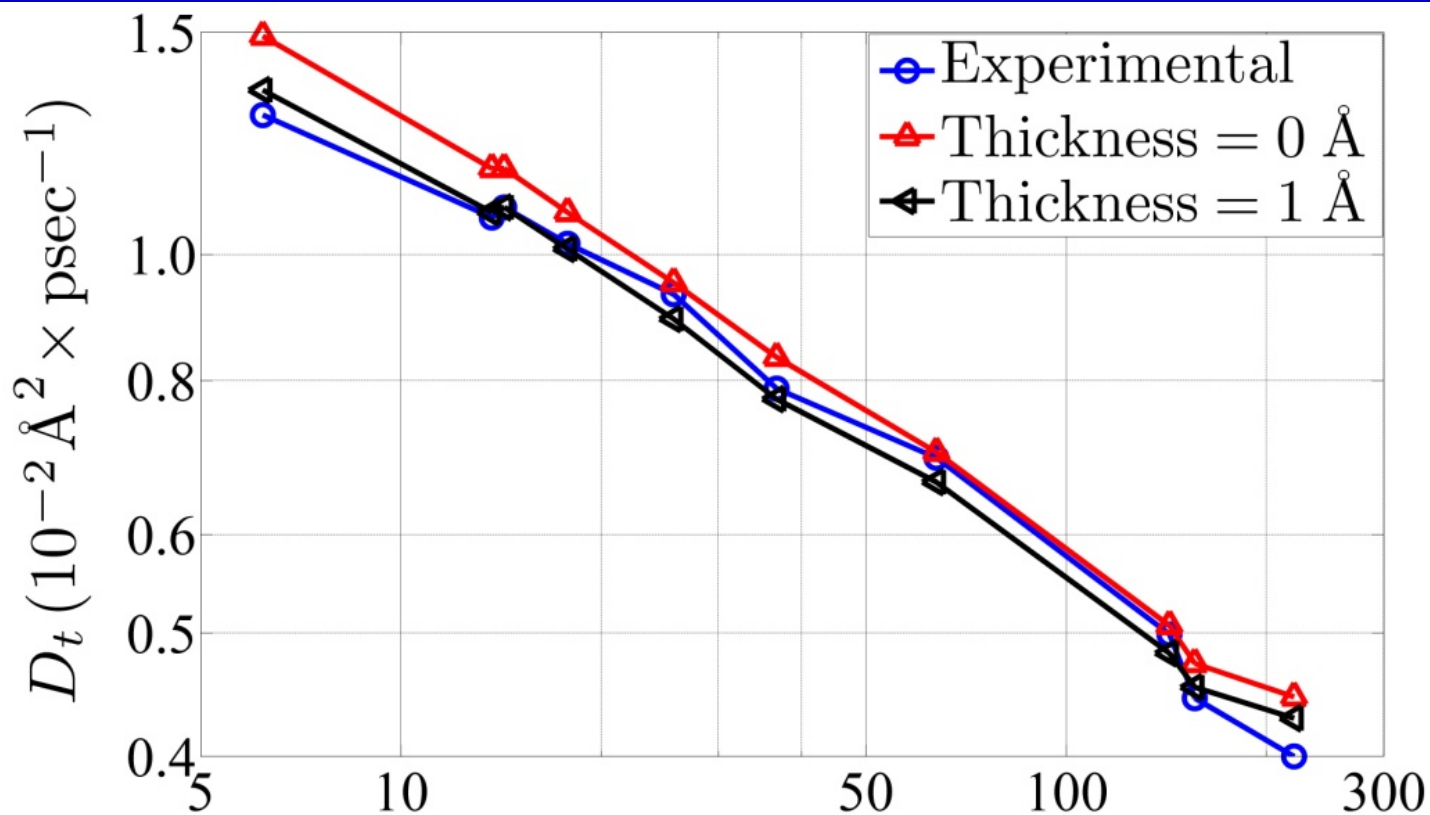
**Finite element modeling, using  
multiple meshes glued together**

**The mesh between the Taq polymerase surface  
and the sphere surface (in cross-section)**





**Brownian dynamics simulation of protein**



Molecular weight (kDa)

**Translational diffusion coefficients of 10 proteins, including Hemoglobin, Lysozyme & Adolase**

# The AMORE paradigm of analysis

## Automatic Meshing with Overlapping and Regular Elements

For any geometry, CAD defined or otherwise,  
that is, not restricted to CAD functions

- Analysis part is immersed in a Cartesian grid
- Boundary of given geometry is discretized

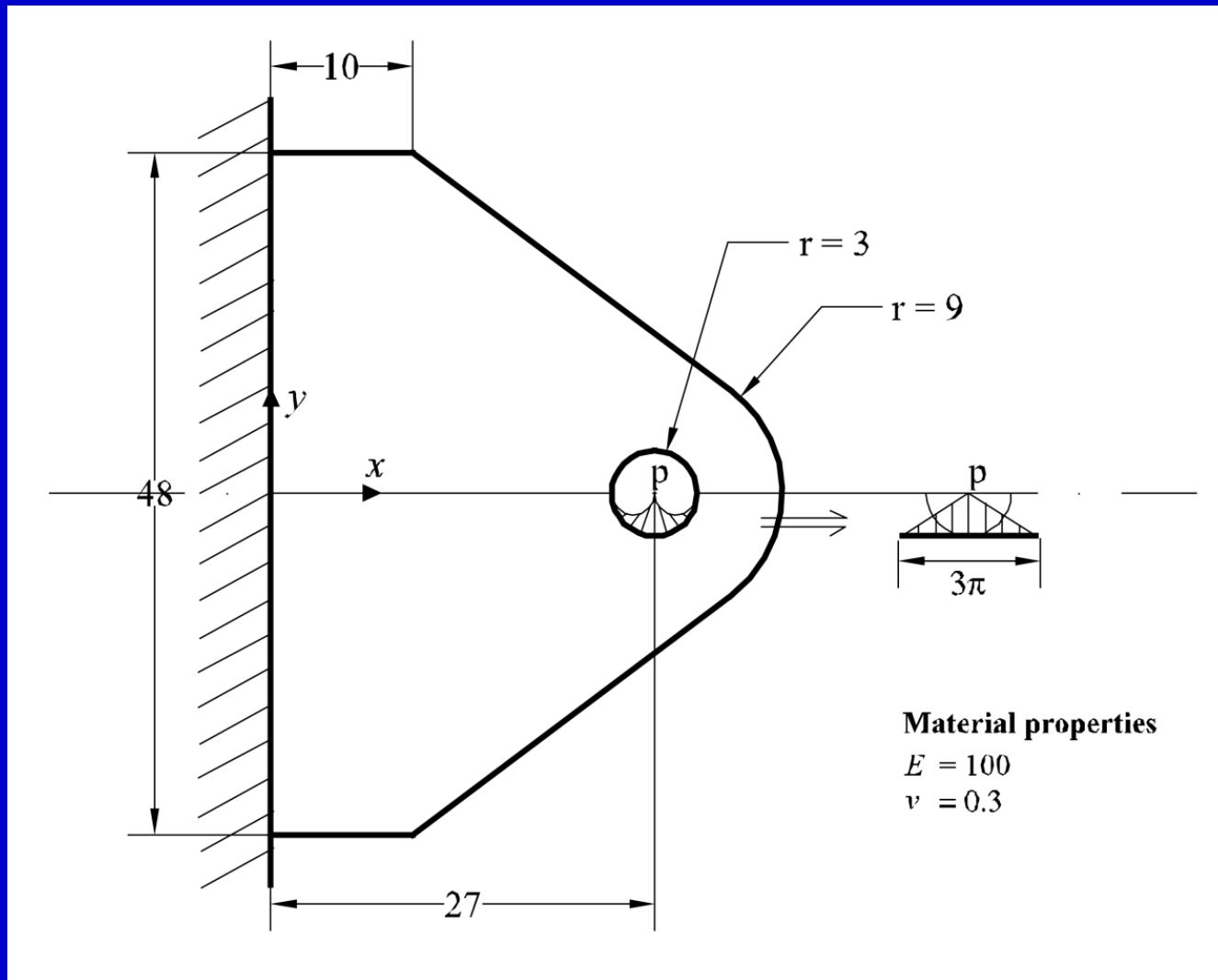
- Cells inside the geometry are turned into finite elements, the other cells are removed
- Overlapping elements are used to fill in the empty space

**Important point: The OFE are distortion insensitive**

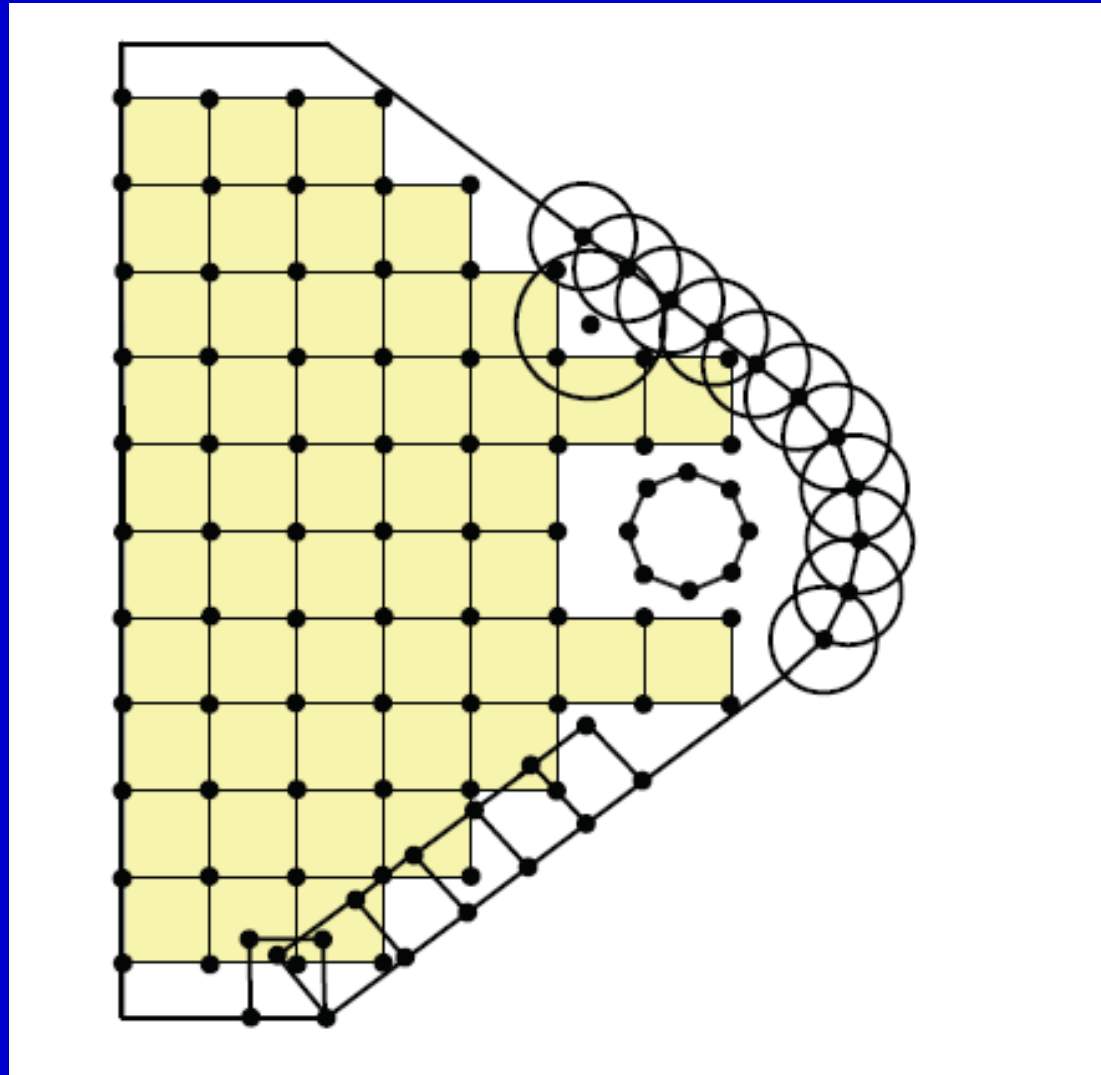
KJ Bathe. Overlapping .... SEMC 2016

KJ Bathe, L Zhang. ... A New Paradigm  
... C & S 2017

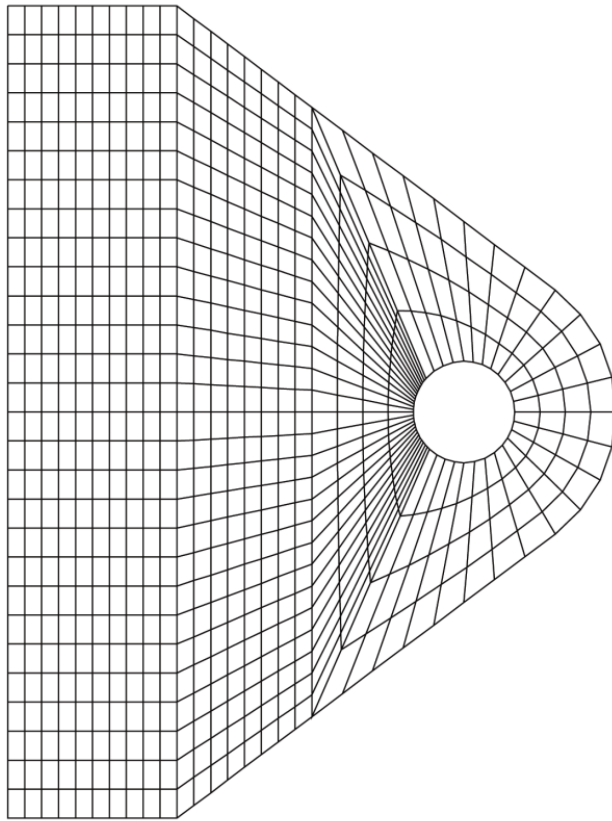
KJ Bathe. The AMORE ... Adv. in Eng.  
Software 2019



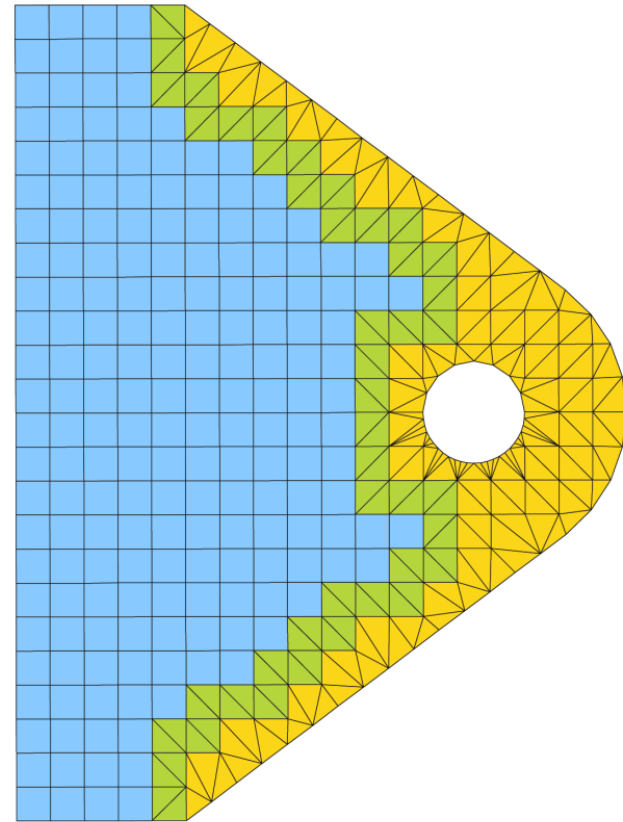
## Example analysis of bracket



**Schematic -- example analysis of bracket**

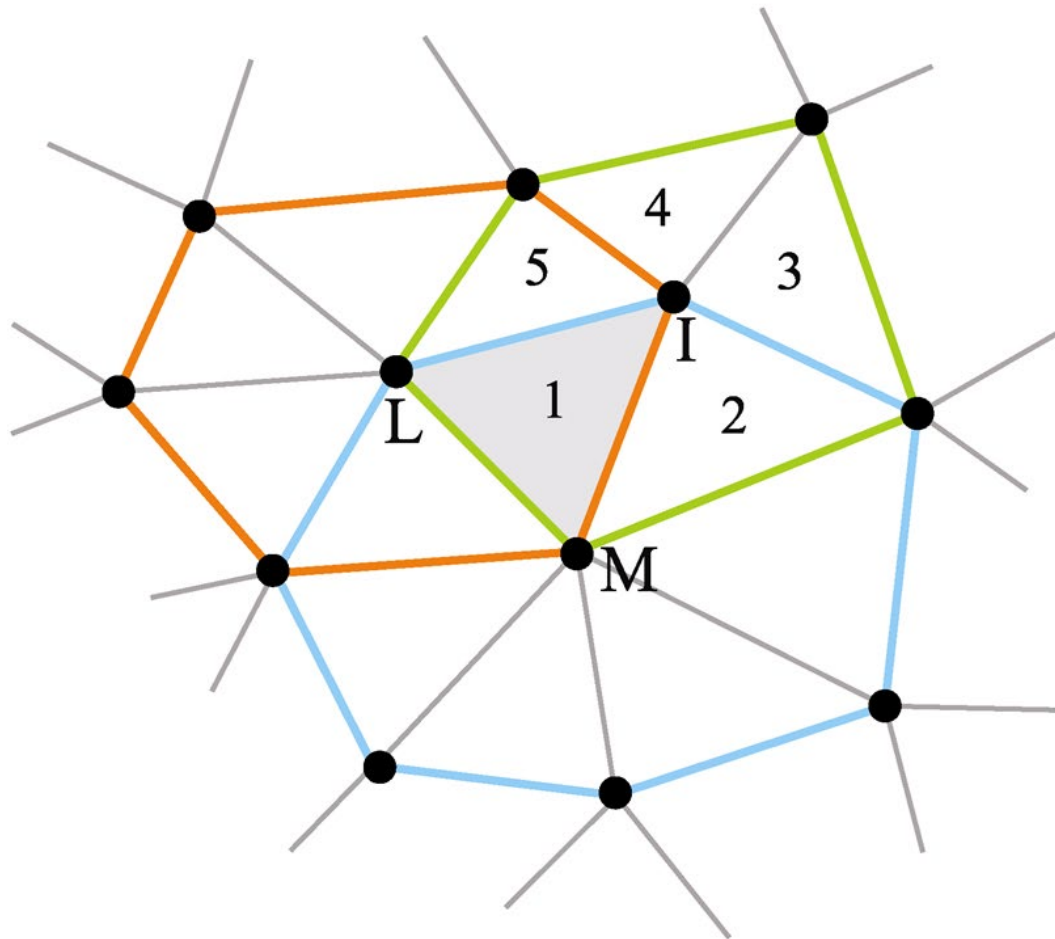


(a) Traditional mesh



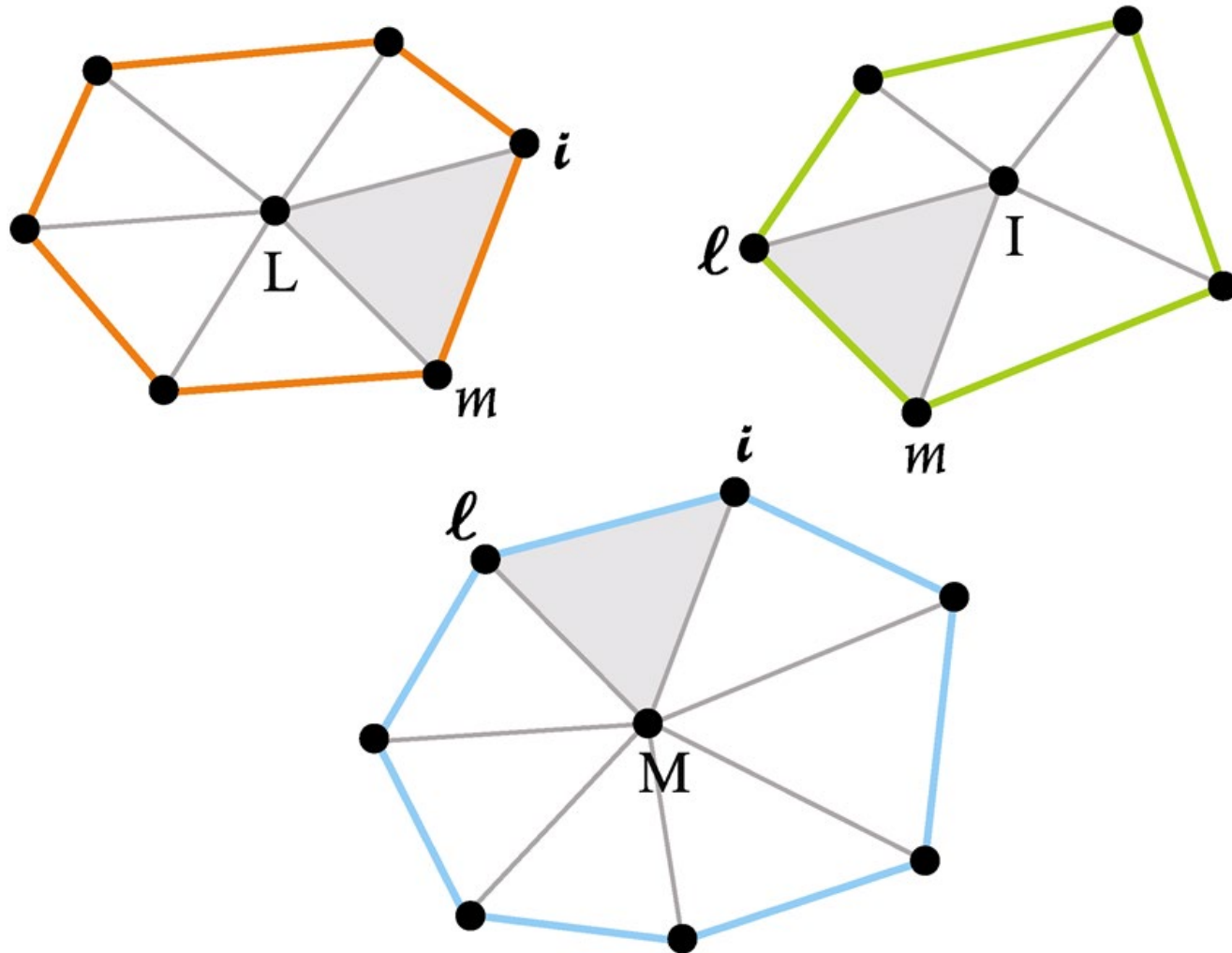
(b) New scheme mesh

**Example analysis of bracket**



**Overlapping elements, with overlapped region shown in grey**





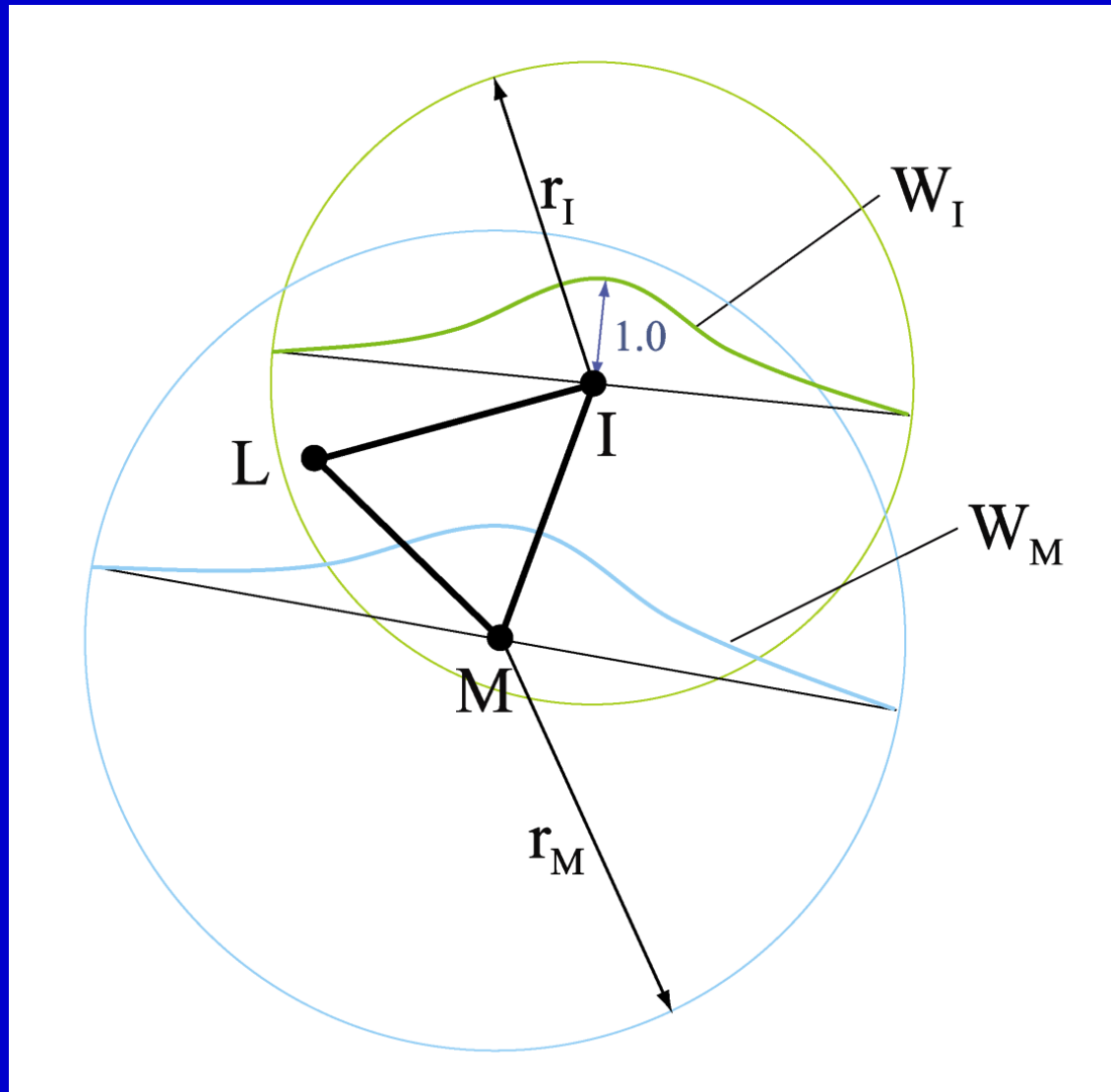
**The 3 overlapping polygonal elements**

**We use the concept of the Method of Finite Spheres (MFS)**

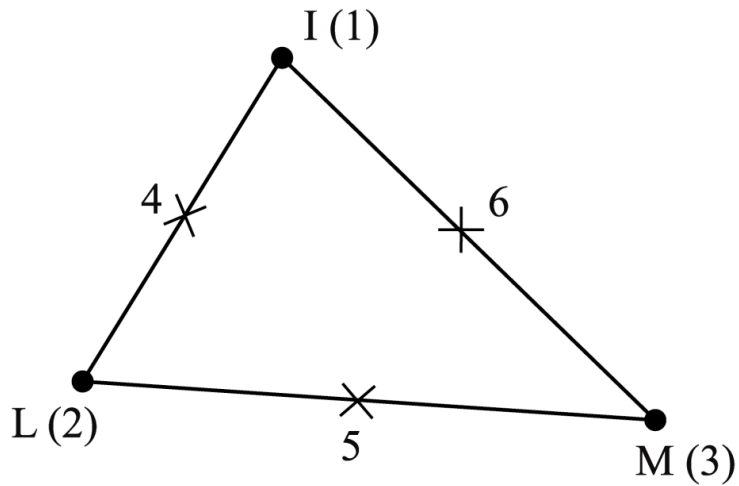
**We place a sphere with its interpolation functions at each node  $I, L, M$  ; hence at each node the DOF of the MFS are used (polynomials, or ... )**

**The displacements of the spheres on the overlapped region  $I$ - $L$ - $M$  are “weighted” by the linear interpolations of the triangular element  $I$ - $L$ - $M$**

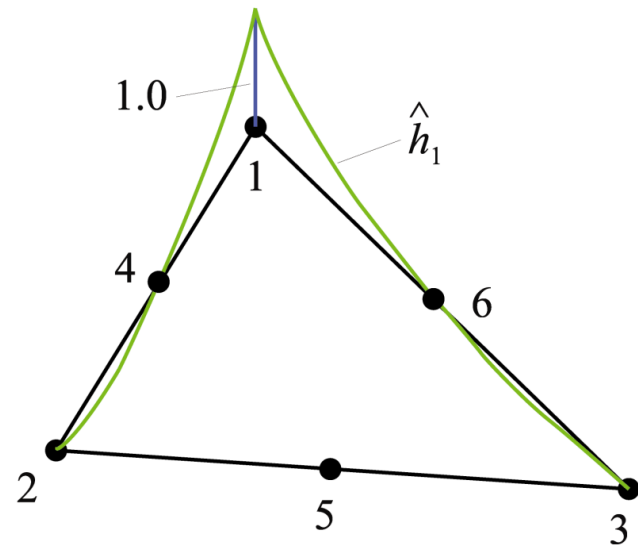
**We interpolate the Shepard functions used in the MFS, to only have polynomials in the interpolations and solution efficiency**



**Shepard weight functions used**



(a)



(b)

## Interpolation of Shepard functions

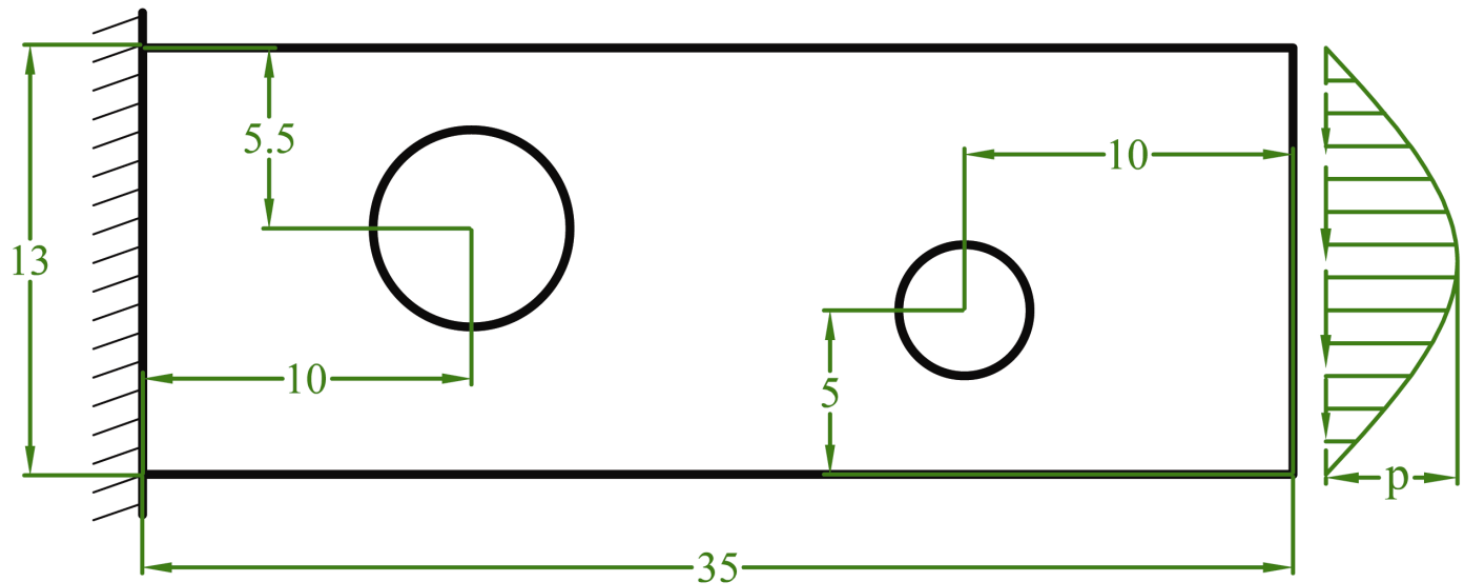
$$\mathbf{u}(\mathbf{x}) = \rho_I p_n \mathbf{a}_{In} + \rho_L p_n \mathbf{a}_{Ln} + \rho_M p_n \mathbf{a}_{Mn}$$

$$\rho_I = h_I \hat{h}_i \hat{\phi}_{Ii}^I + h_L \hat{h}_i \hat{\phi}_{Ii}^L + h_M \hat{h}_i \hat{\phi}_{Ii}^M$$

$$\rho_L = h_I \hat{h}_i \hat{\phi}_{Li}^I + h_L \hat{h}_i \hat{\phi}_{Li}^L + h_M \hat{h}_i \hat{\phi}_{Li}^M$$

$$\rho_M = h_I \hat{h}_i \hat{\phi}_{Mi}^I + h_L \hat{h}_i \hat{\phi}_{Mi}^L + h_M \hat{h}_i \hat{\phi}_{Mi}^M$$

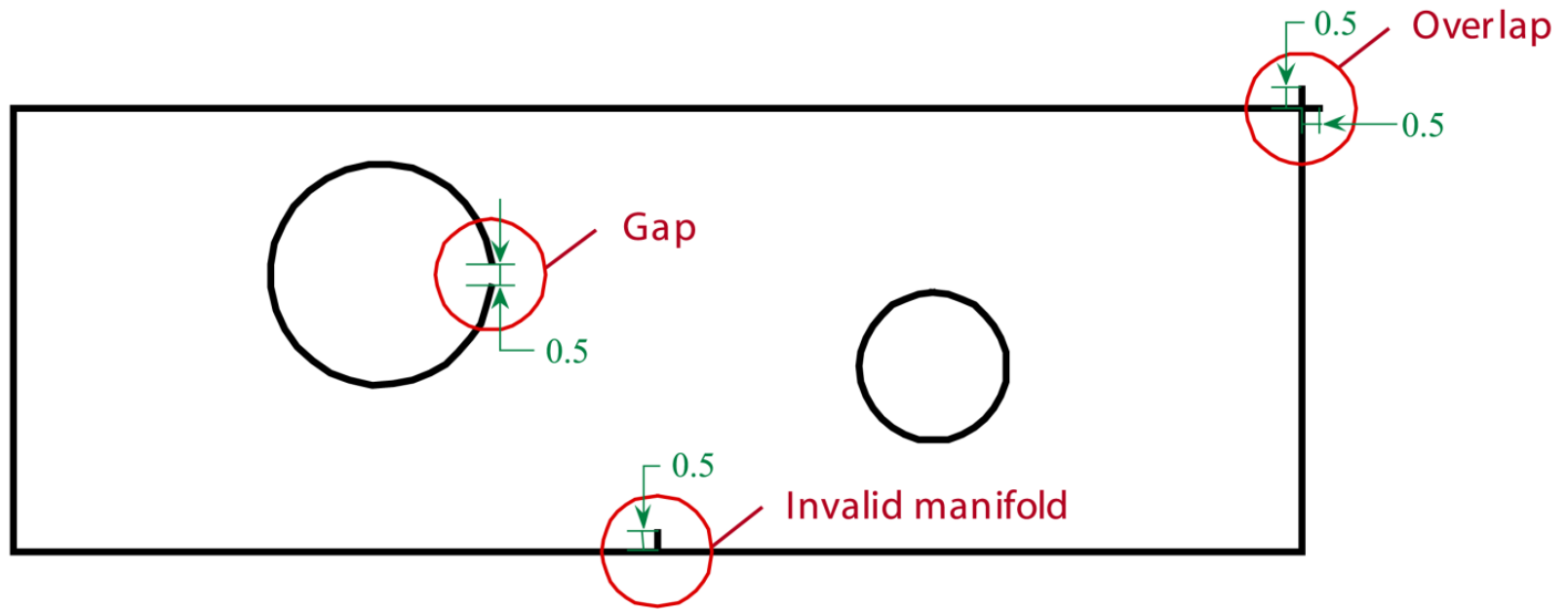
**Interpolation functions for triangle I-L-M**



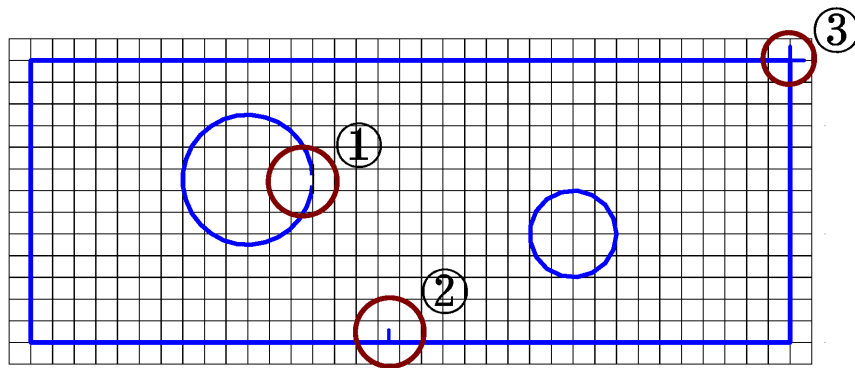
## Example: Analysis of a cantilever plate

L Zhang, KJ Bathe. Overlapping finite elements ... . C & S 2017

L Zhang, KT Kim, KJ Bathe. The new paradigm ... . C & S 2018

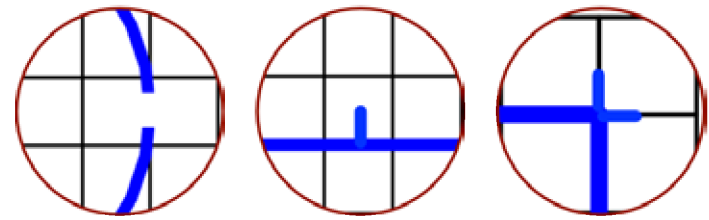


## CAD geometry – Analysis of a cantilever plate



(a) The first step

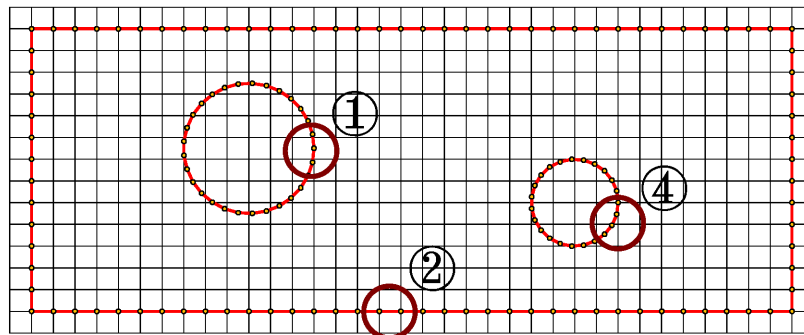
### Geometry defects



①

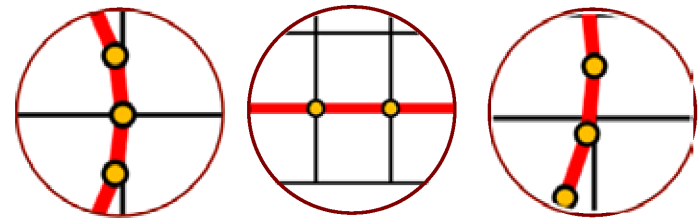
②

③



(b) The second step

### Boundary discretization



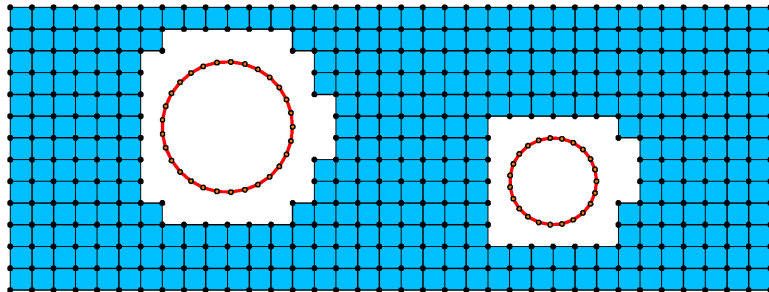
①

②

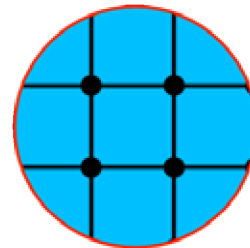
④

## Analysis of a cantilever plate

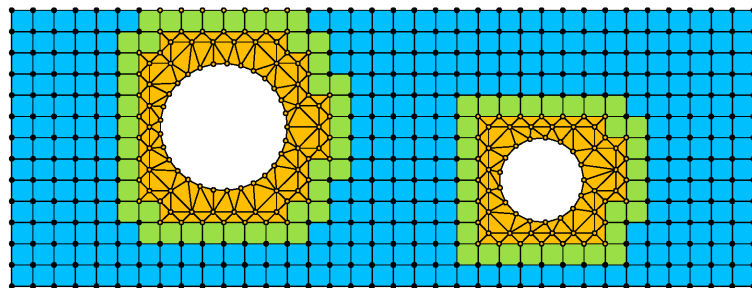




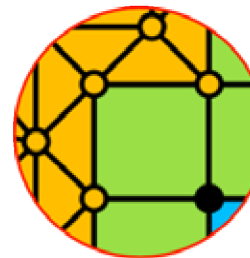
(c) The third step



**Finite elements**



(d) The fourth step



**Overlapping and coupling elements**



4-node finite elements

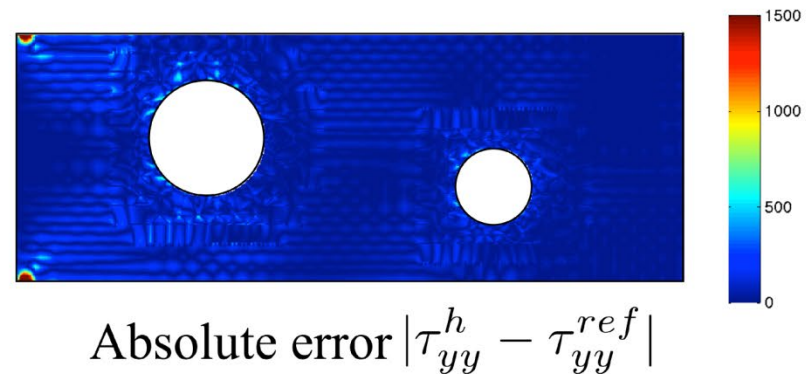
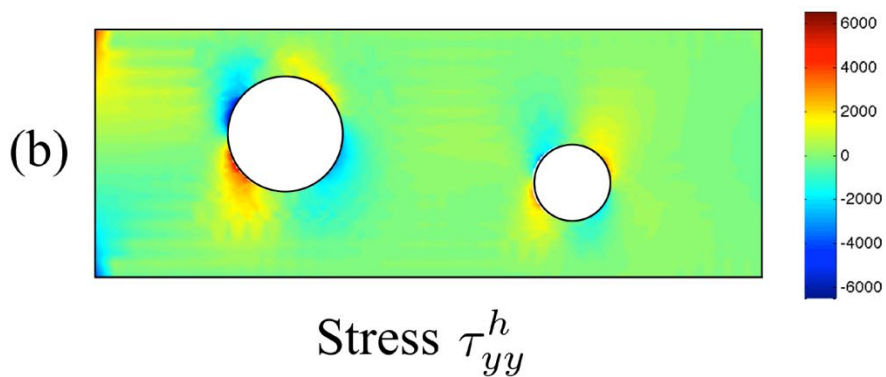
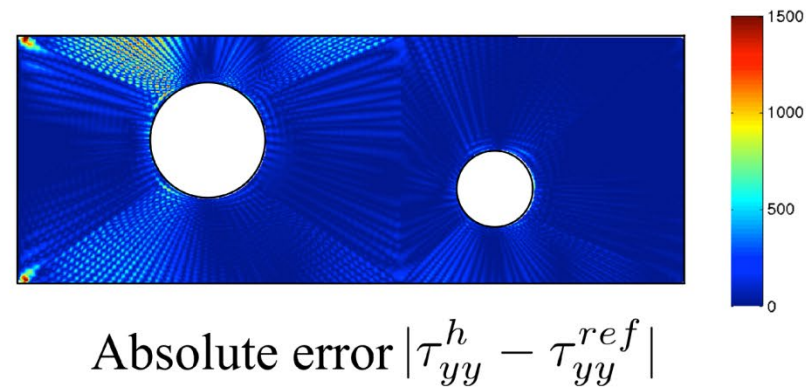
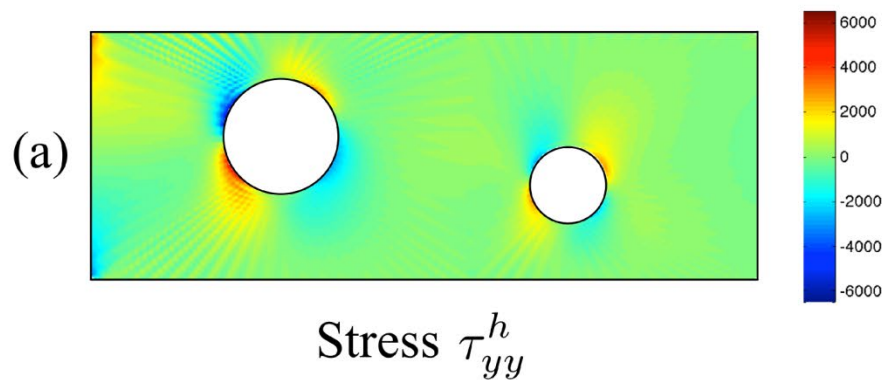


Coupling regions

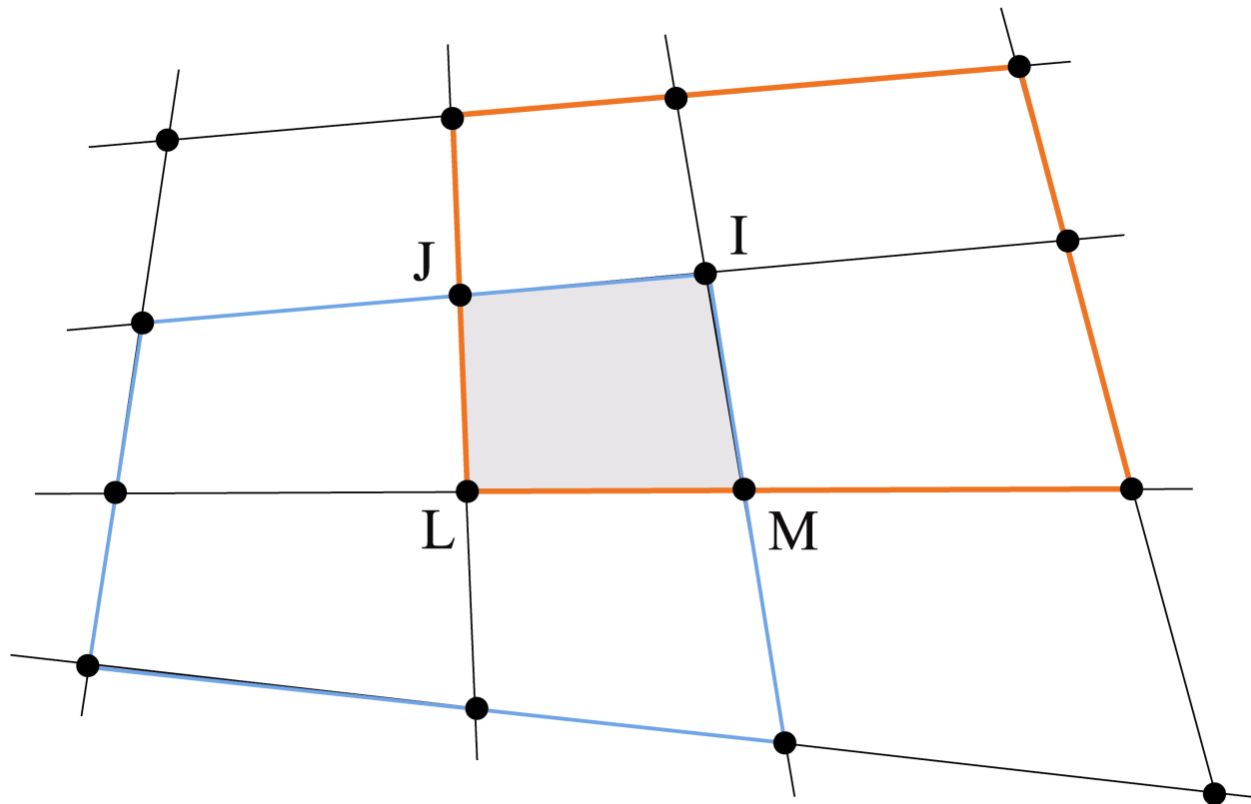


Overlap regions

# Analysis of a cantilever plate

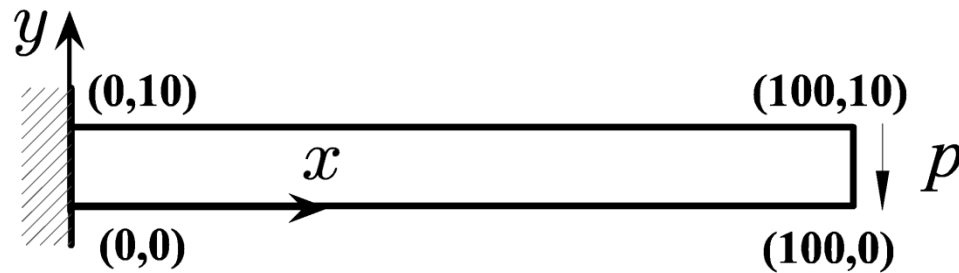


**Results: Analysis of a cantilever plate**  
**(a) traditional analysis, (b) AMORE scheme**



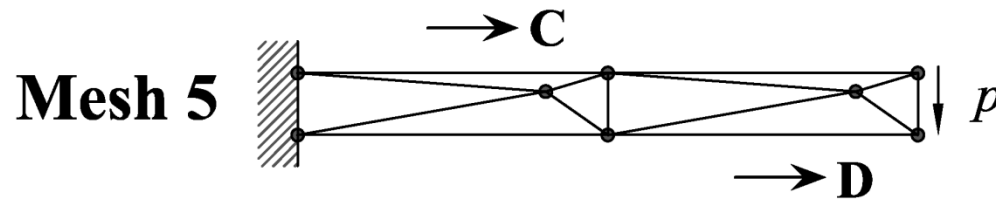
## Quadrilateral overlapping elements

J. Huang, KJ Bathe. Quadrilateral ... . In prep.

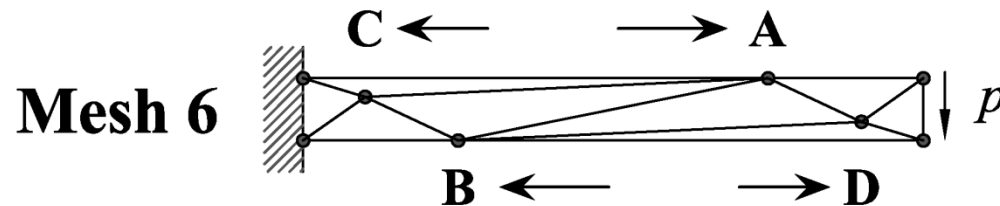


		Point Coordinates
<b>Mesh 1</b>		A: (50,10) B: (50,0) C: (25,5) D: (75,5)
<b>Mesh 2</b>		A: (75,10) B: (25,0)
<b>Mesh 3</b>		A: (70,10) B: (70,0)
<b>Mesh 4</b>		C: (10,7) D: (90,3)

## Analysis of cantilever

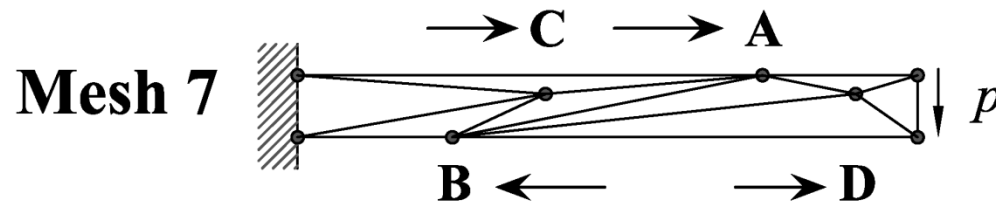


C: (40,7)    D: (90,7)



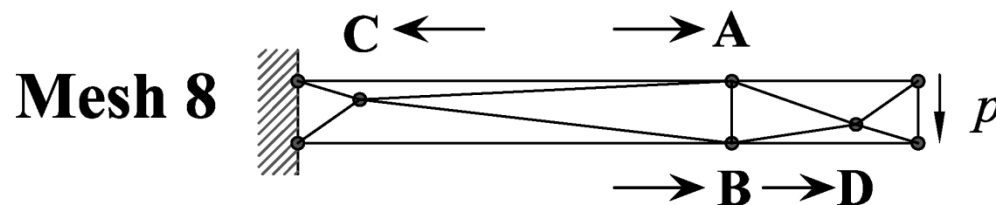
A: (75,10)    B: (25,0)

C: (10,7)    D: (90,3)



A: (75,10)    B: (25,0)

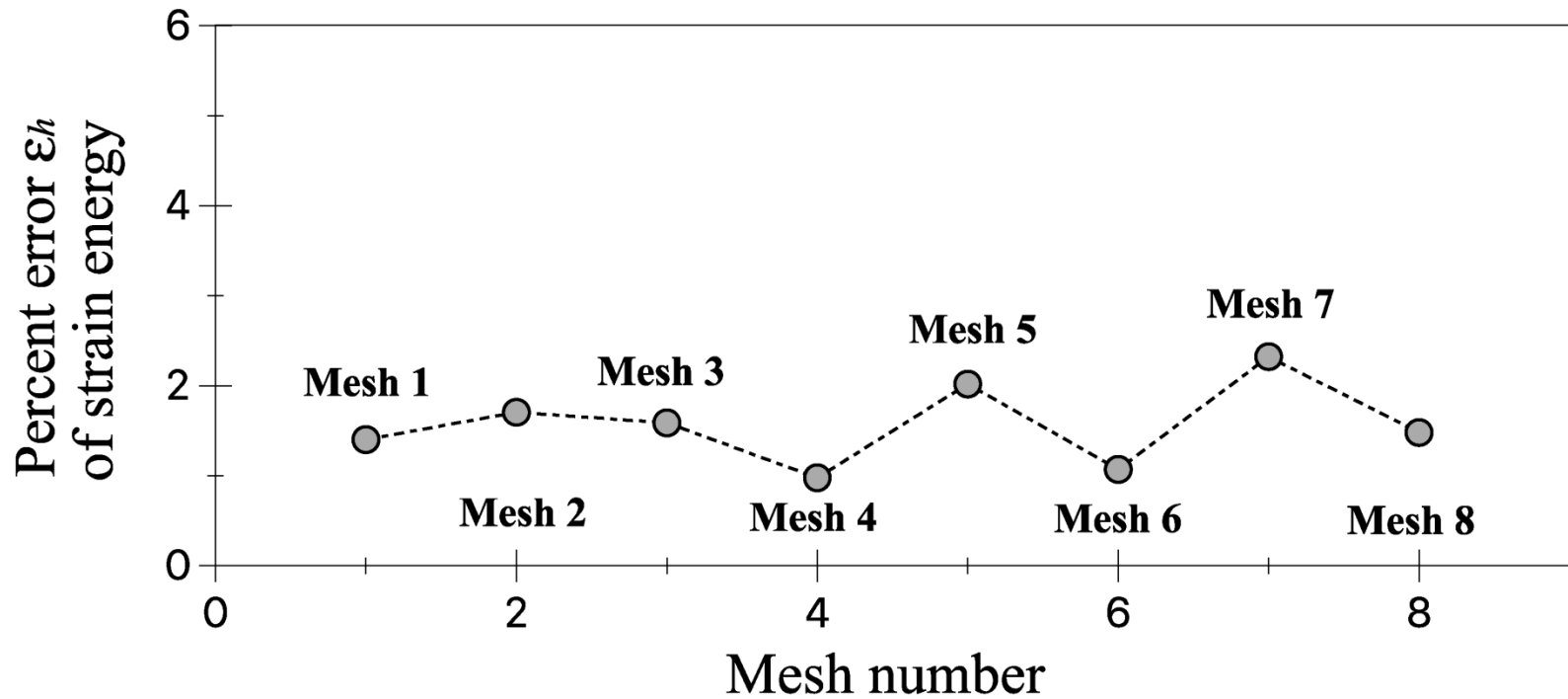
C: (40,7)    D: (90,7)



A: (70,10)    B: (70,0)

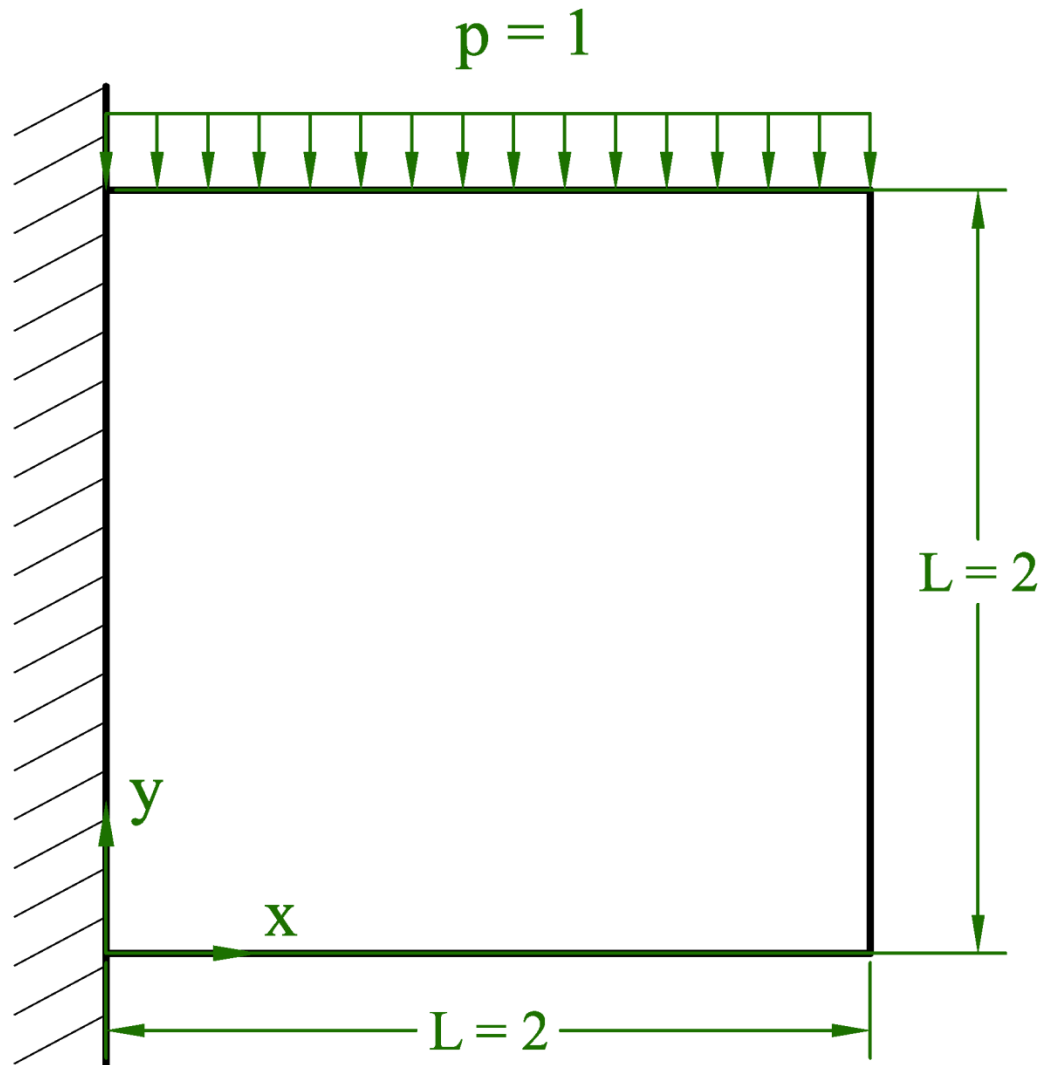
C: (10,7)    D: (90,3)

**Analysis of cantilever, continued**

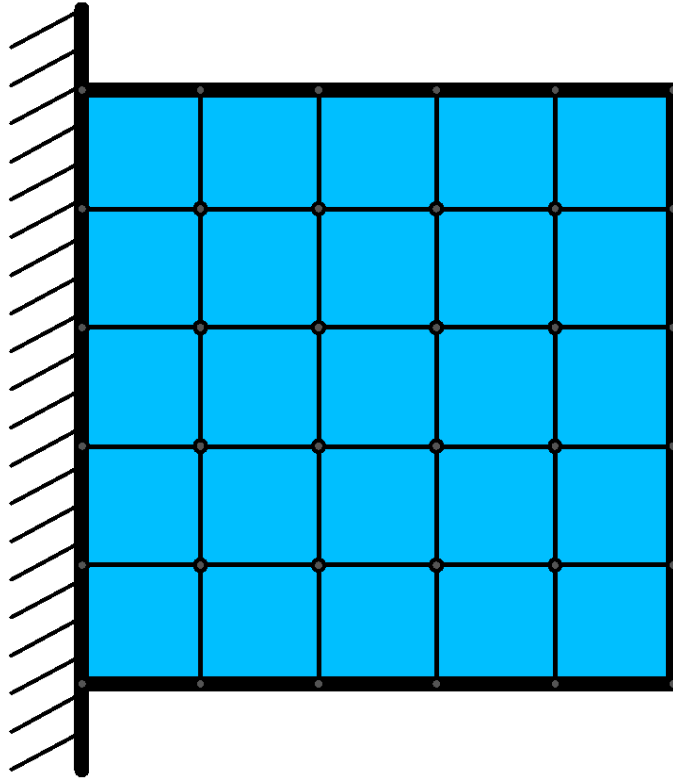


**% strain energy error**

**L Zhang, KT Kim, KJ Bathe. The new paradigm ... . C & S 2018**

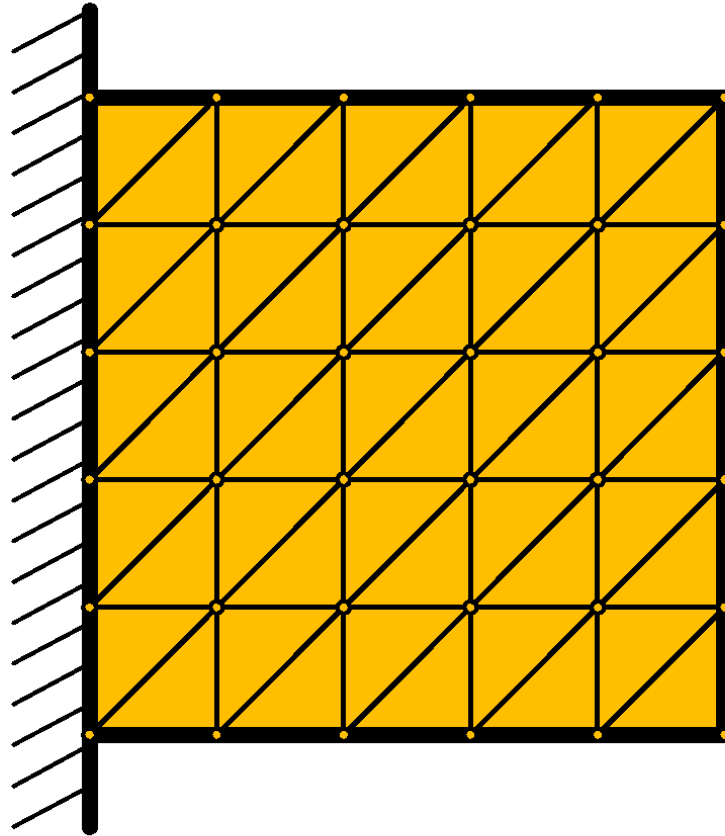


**Analysis of clamped plate**

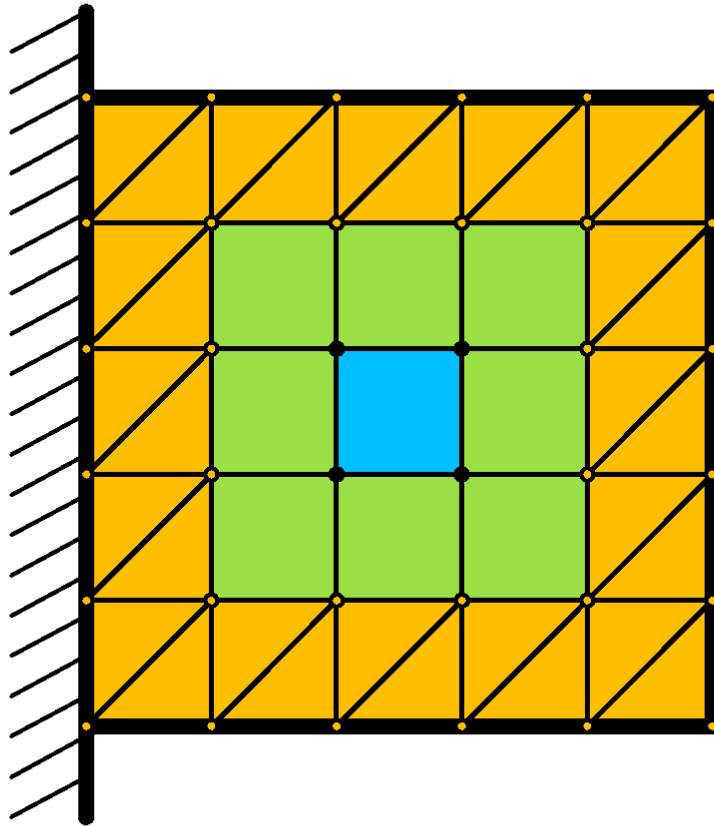


**Meshes used for clamped plate, traditional mesh**

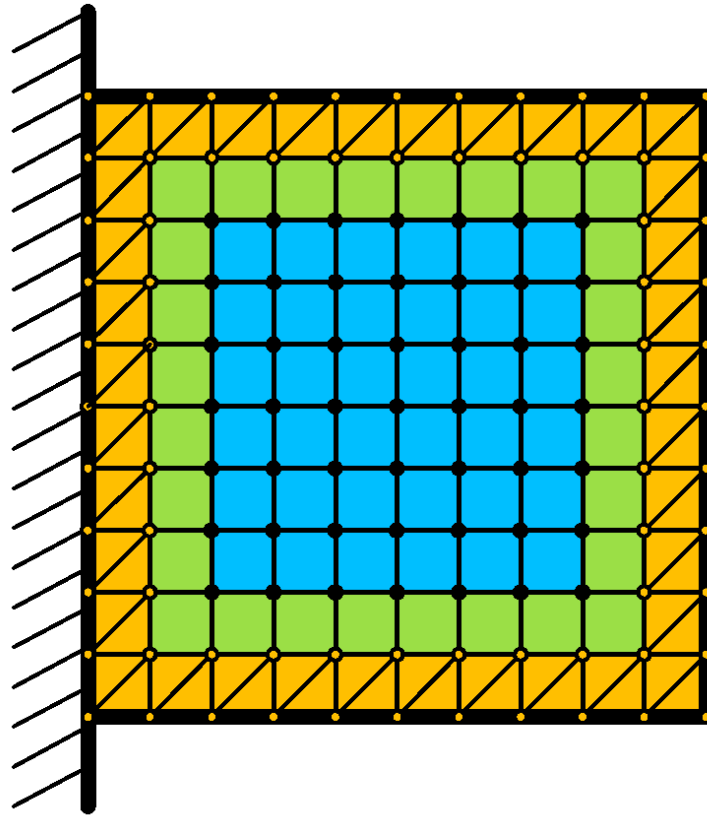




**OFE only used**



**OFE, coupling and 1 traditional element**



**OFE, coupling and traditional elements**

# Solution times for the cantilever plate problem

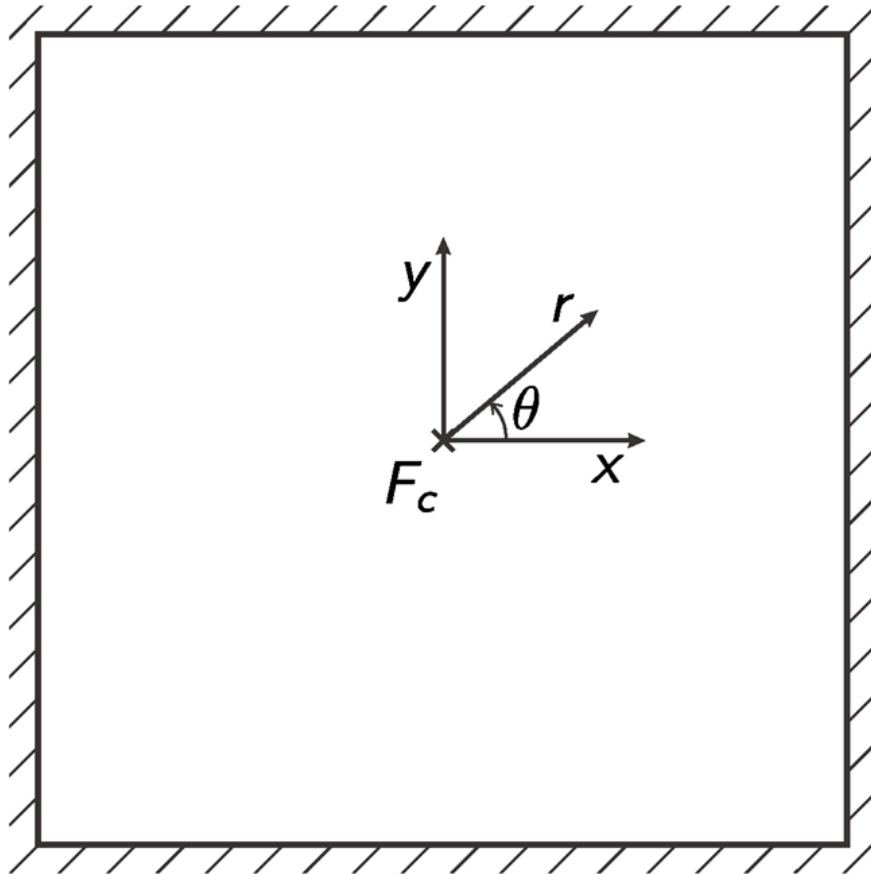
Scheme		$\log( E_{ref} - E_h  / E_{ref})$	Actual CPU time (s)	Predicted CPU time $So/E$ (s)
4-node finite elements		-3.09	1.3	—
Overlapping finite elements	Bilinear basis	-3.25	0.16	0.16
	Quadratic basis	-3.04	0.007	0.004
AMORE	Bilinear basis	-3.35	0.4	0.4
	Quadratic basis	-3.44	1.0	1.0

# **Analysis of wave propagations**

**Need to have ‘effective time integration’  
and ‘effective spatial interpolations’**

**We explored the use of the OFE with  
the Bathe time integration scheme and  
obtained good results**

**In the overlapping finite elements we use  
the bilinear and harmonic functions ...**



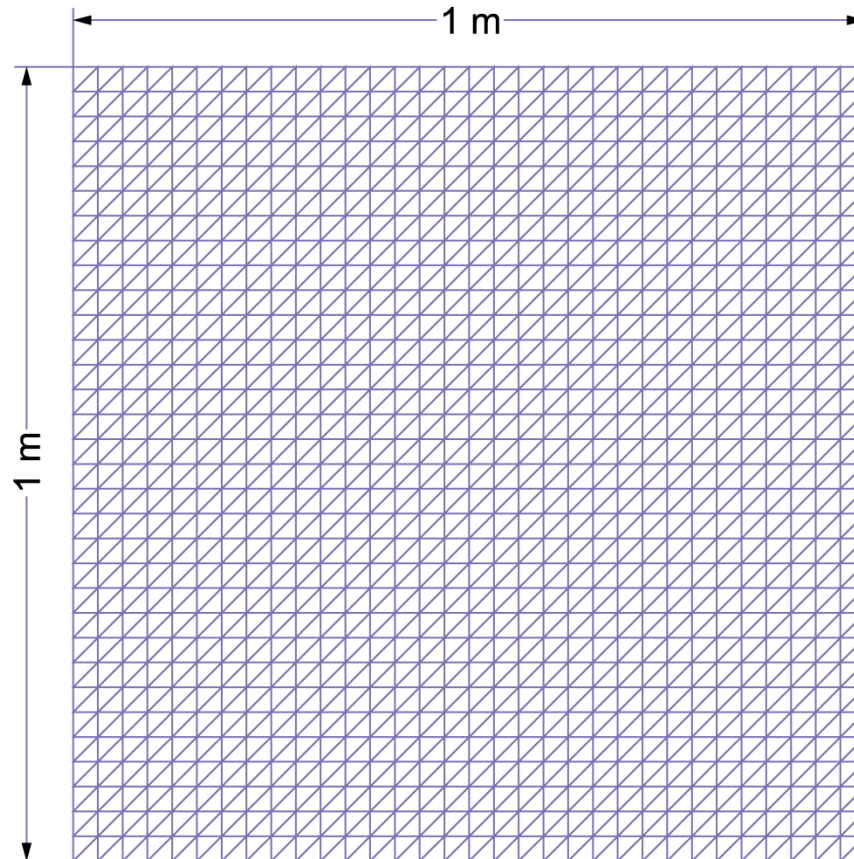
$$c = 1 \text{ m/s}$$

$$u(\mathbf{x}, t = 0) = 0 \text{ m}$$

$$\dot{u}(\mathbf{x}, t = 0) = 0 \text{ m/s}$$

**Analysis of wave propagation problem**

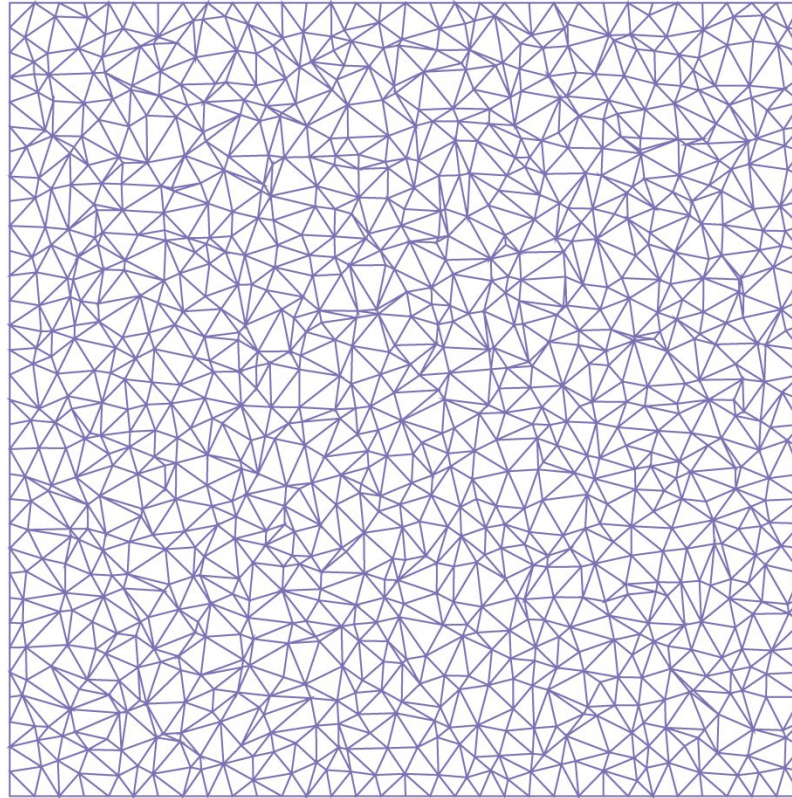
$$N = 32$$
$$h = 1/N = 0.03125 \text{ m}$$



Structured mesh

**Structured mesh used**

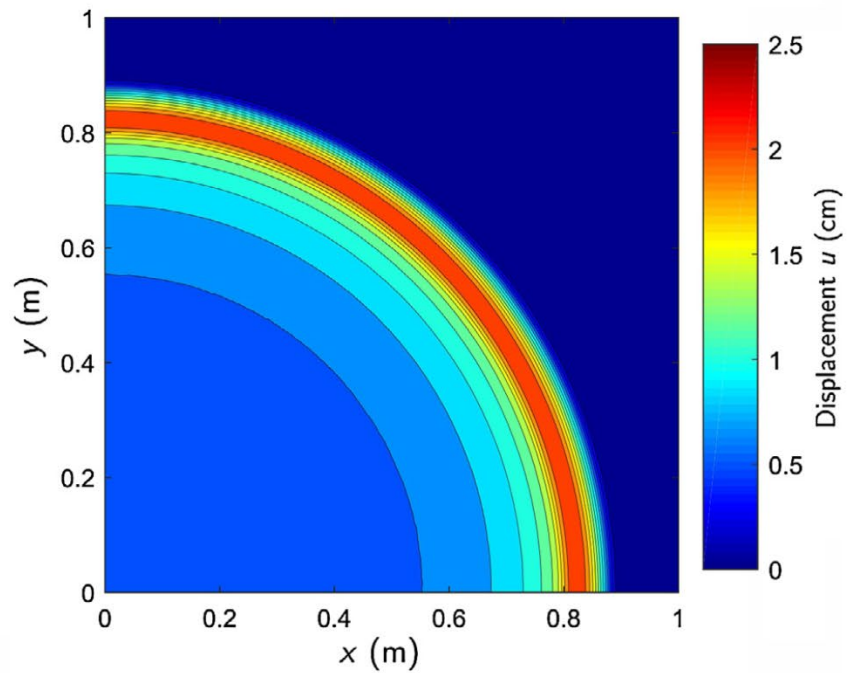
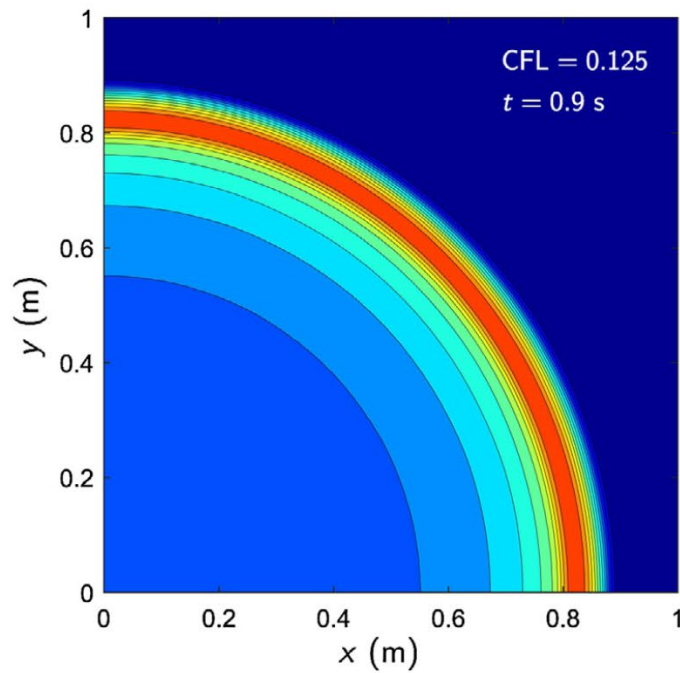
$$N = 32$$
$$h = 1/N = 0.03125 \text{ m}$$



Unstructured mesh

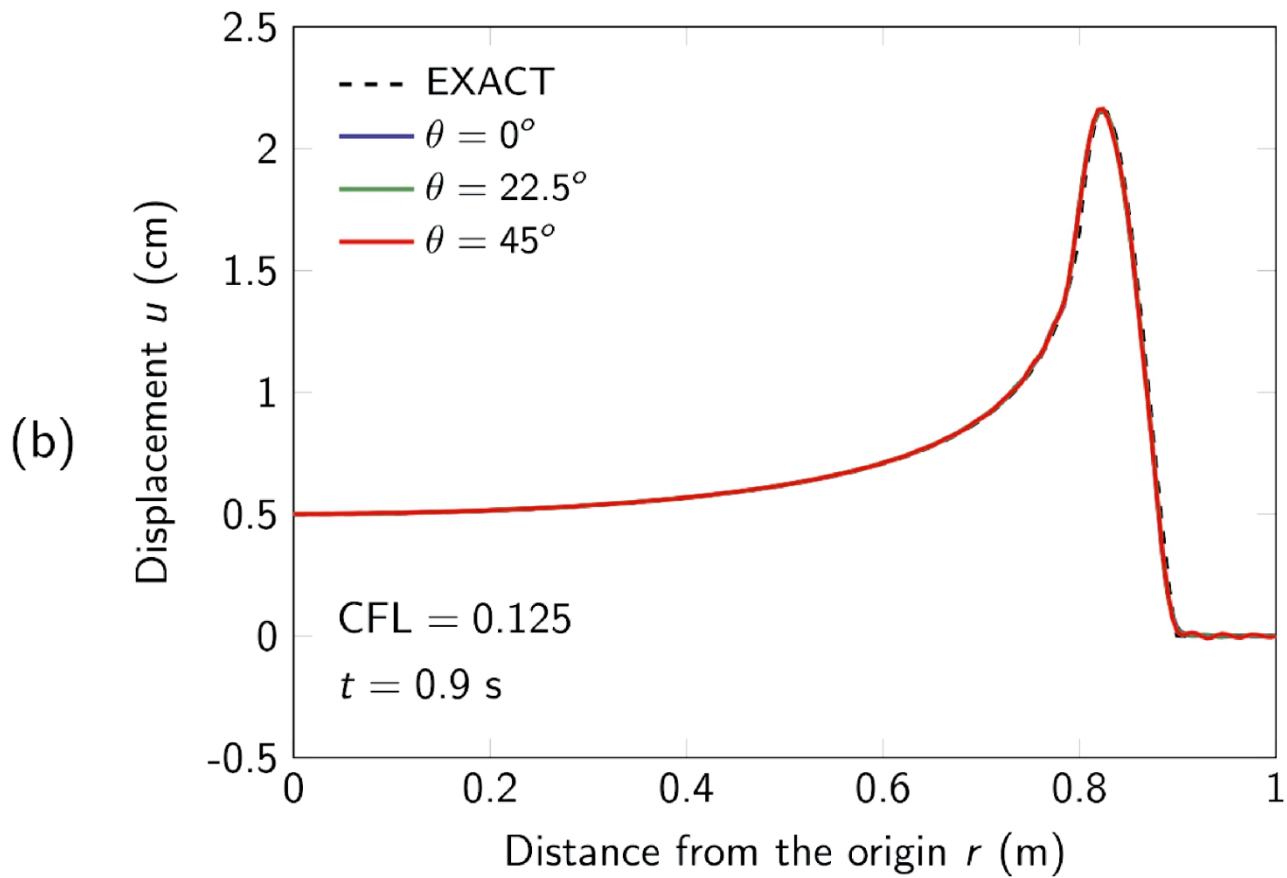
**Unstructured mesh used**





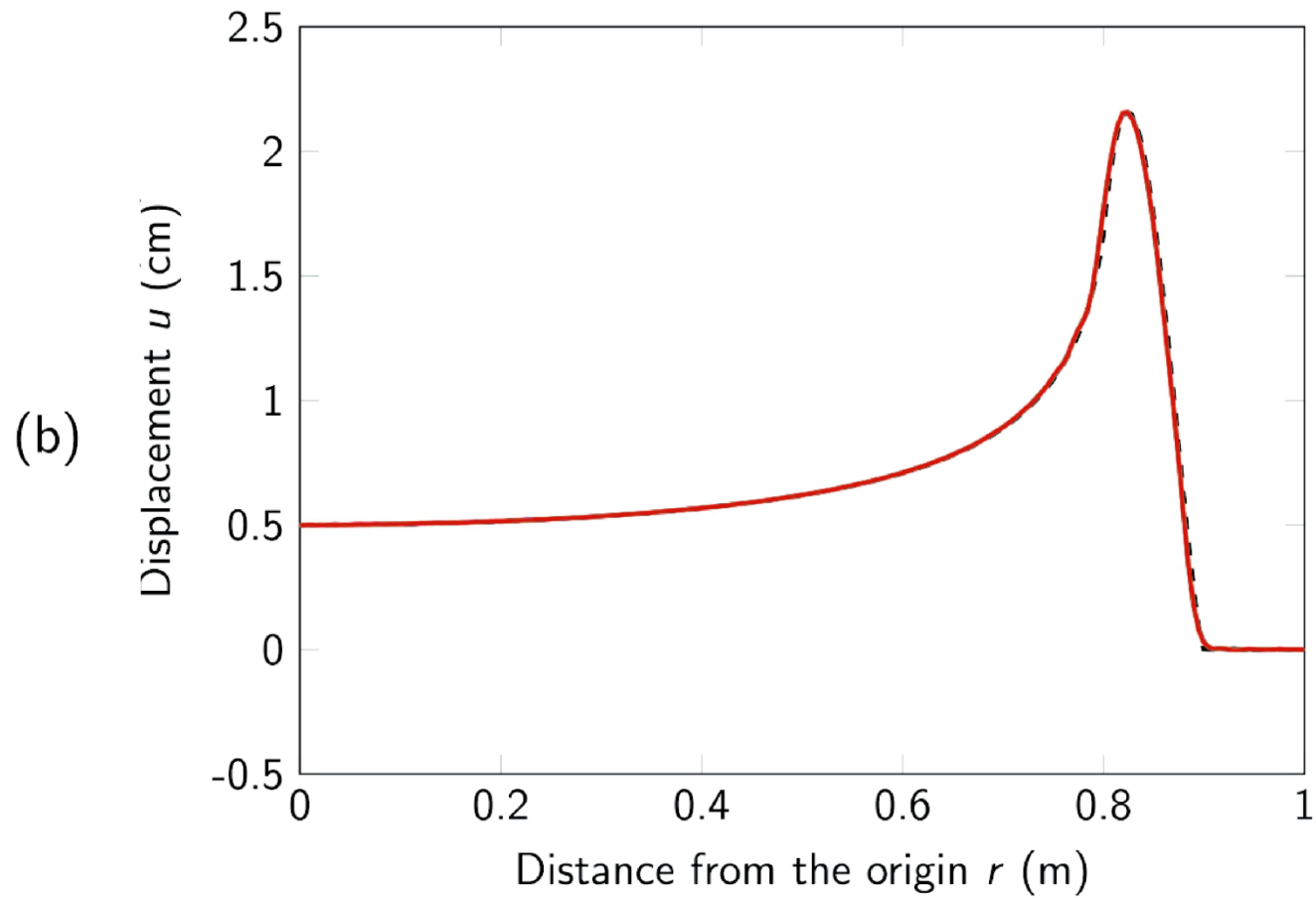
**Results, contours of wave,  
structured mesh left, unstructured mesh right**

## Structured mesh



**Details of results**

## Unstructured mesh



**Details of results**

# Conclusions

## and a look into the future

- Very powerful capabilities are now available, but there are also still many exciting research challenges – a message to our young ( in age and heart ! ) researchers

**Considering analyses and capabilities:**

**“ We are really only at the beginning of the use of simulations on the computer and the extent to which these will greatly**

**ENRICH OUR LIFE ! ”**